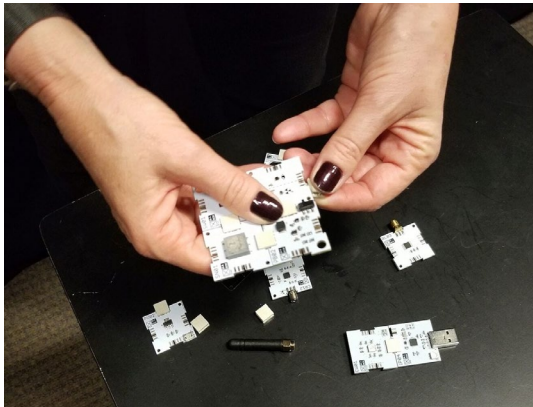
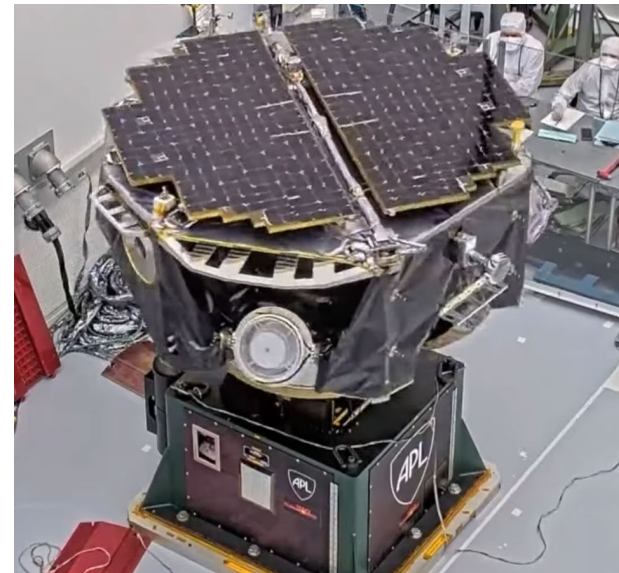
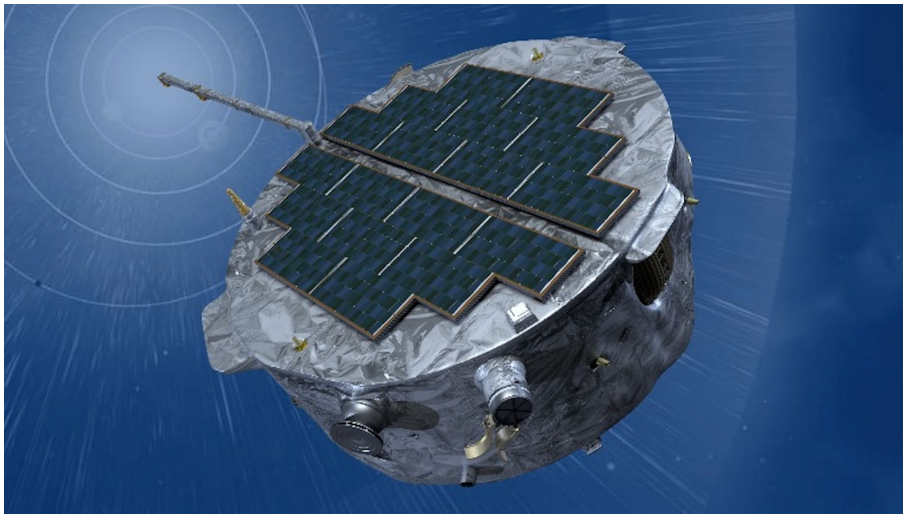
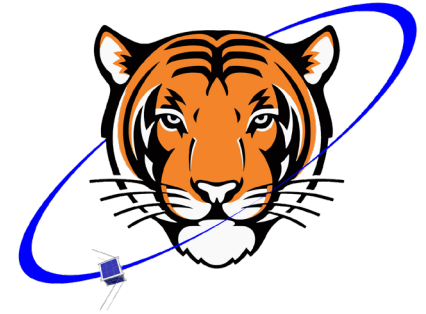


Princeton's Experience in MaxIQ's STEM Launch Program to the ISS

Mike Galvin
mgalvin@princeton.edu



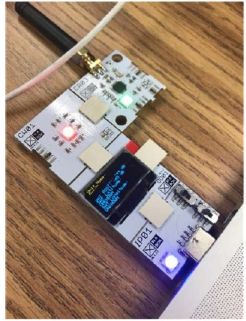
- Lockheed Martin Space Systems
- Lead Mechanical Engineer for Princeton Space Physics Group
- Instructional Staff for MAE342: Space Systems Design
- Principal Investigator for TigerSats student nanosatellite group



- ▣ Student Lead
- ▣ TigerSats
- ▣ Rocketry Club Treasurer
- ▣ Internships



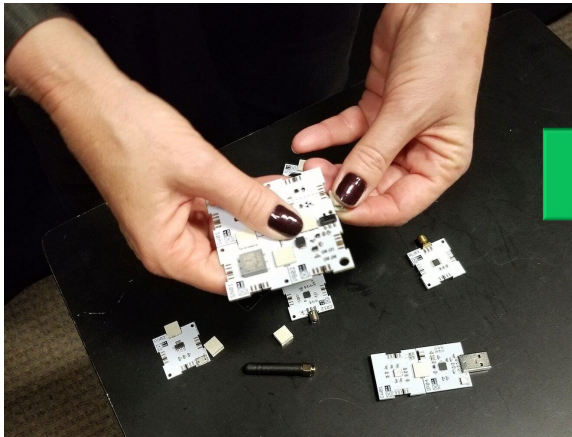
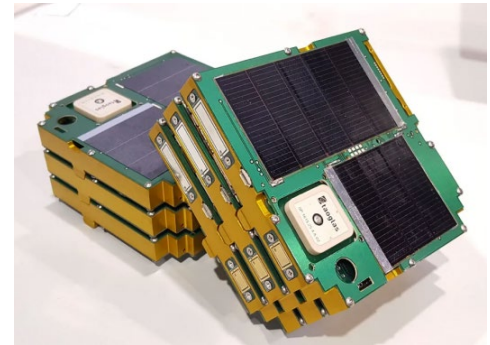
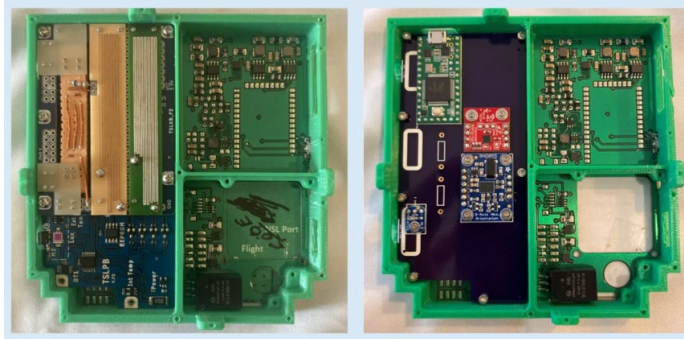
ThinSat (CDW 2018)



Ground Station

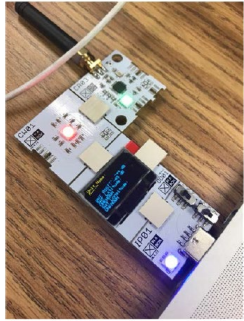


Weather Sat



NG-15 (2021)

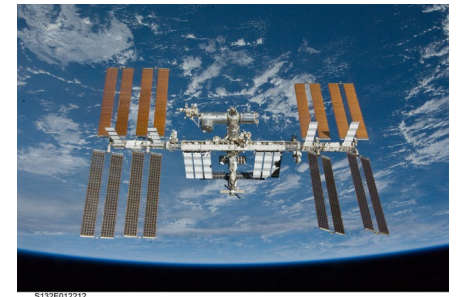
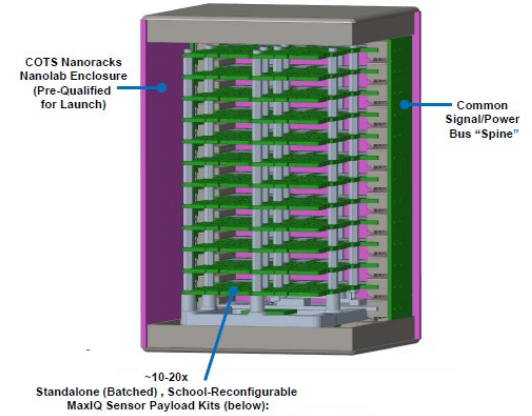
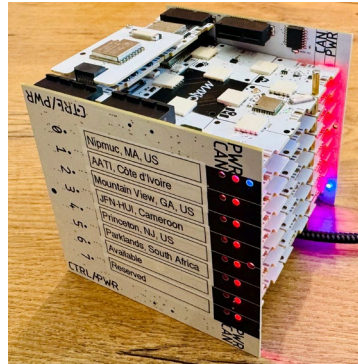
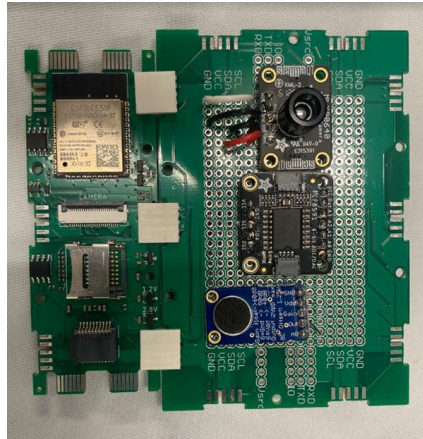
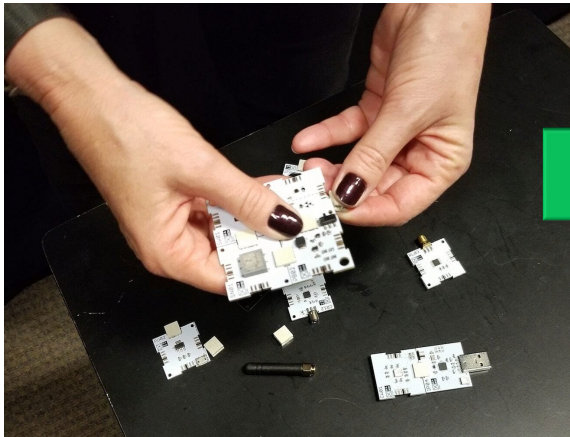




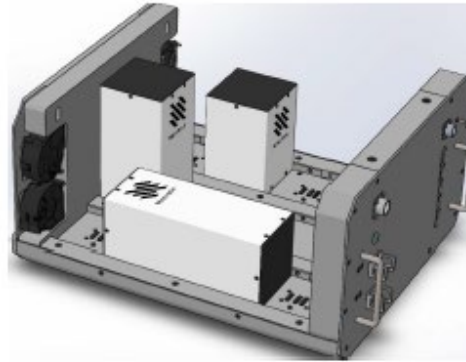
Ground Station



Weather Sat



6132E012212



**Nanoracks "Nanode"
Experiment Locker**



Launch



Delivery



On-Orbit Operations



Return



Landing

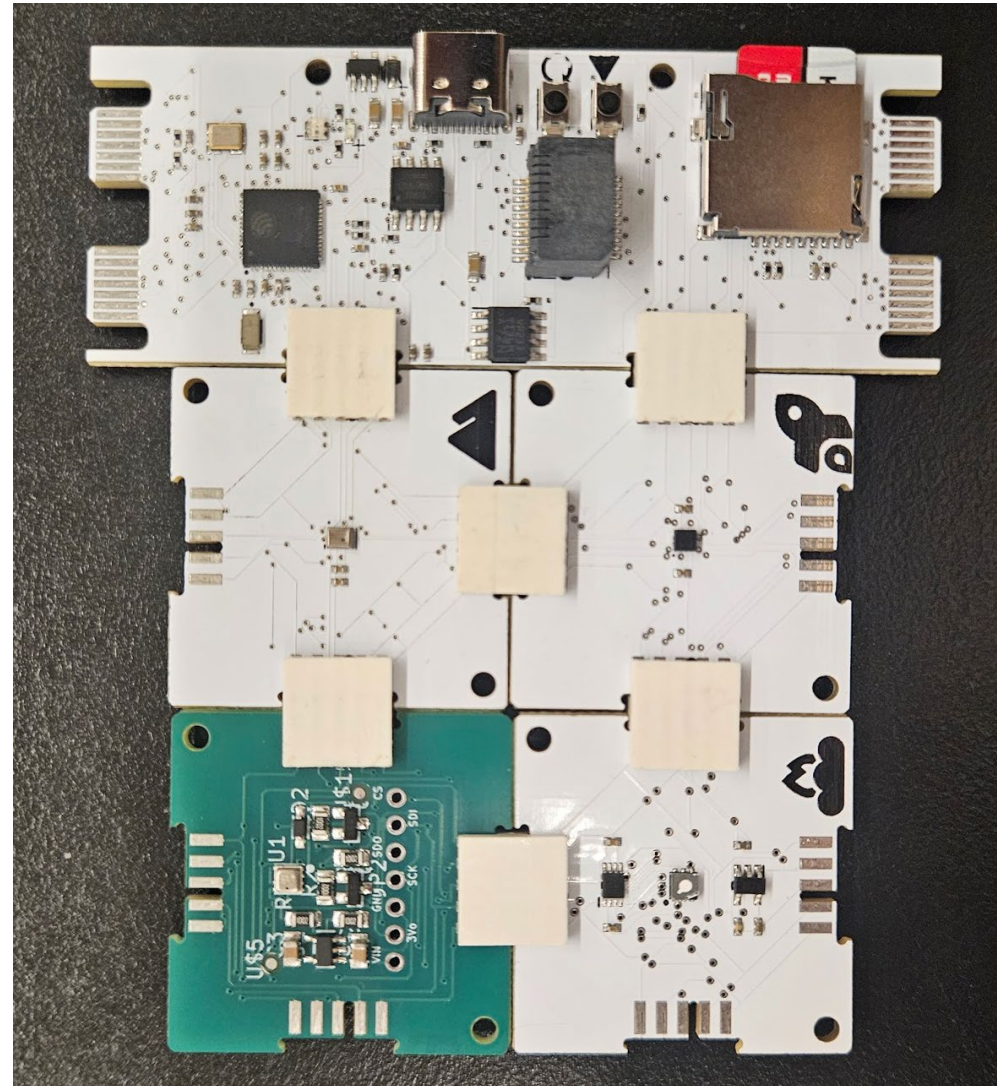
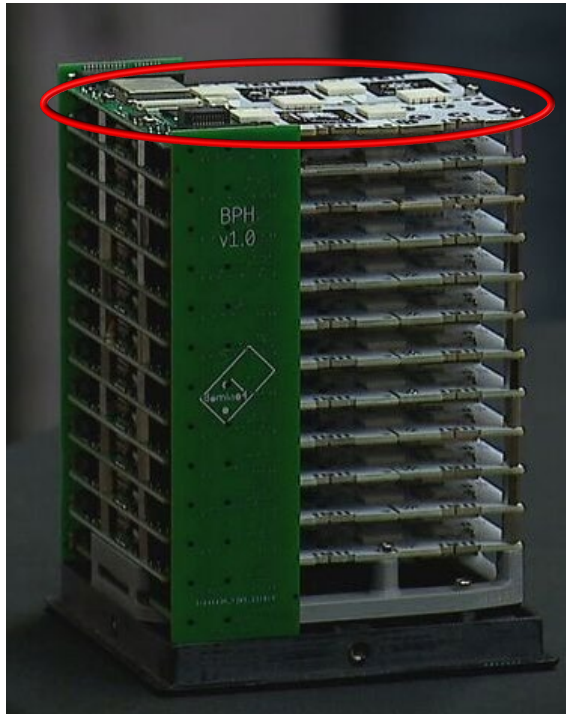


Retrieval

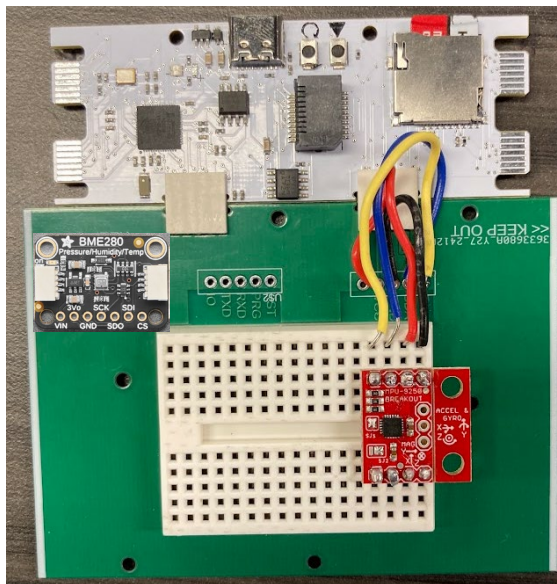
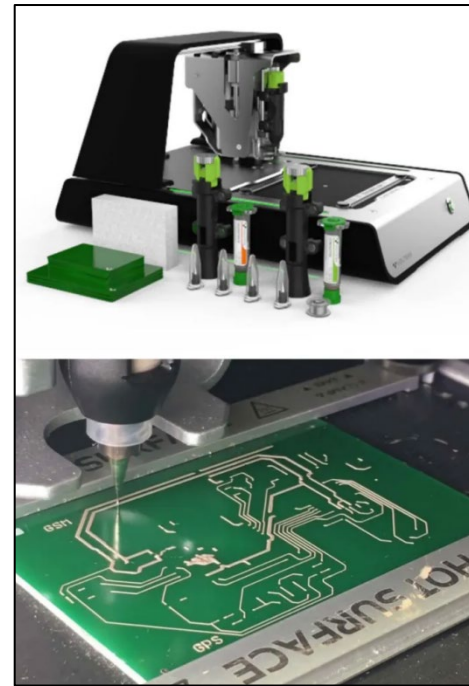
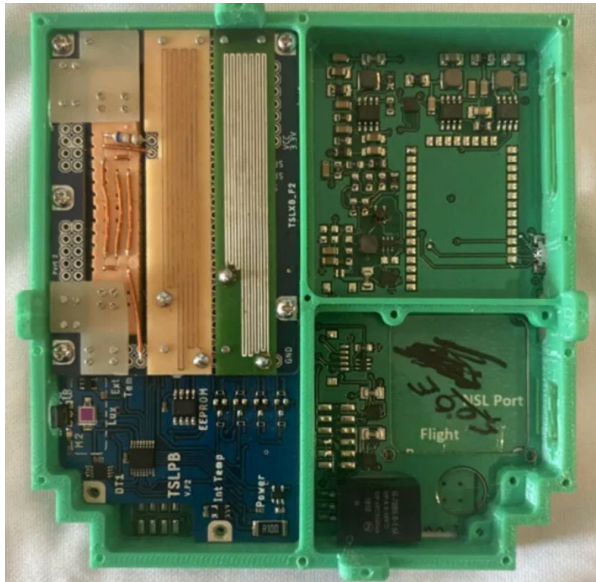


**Receive At
NanoRacks**

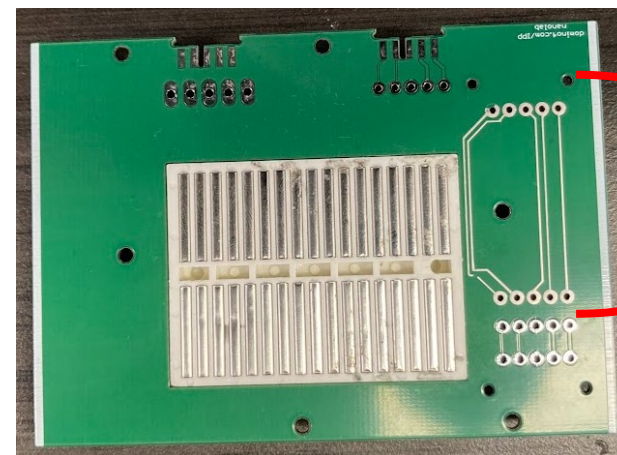
Princeton xChip Experiment (Freshmen)



“ProtoSat”
(ThinSat ‘21)

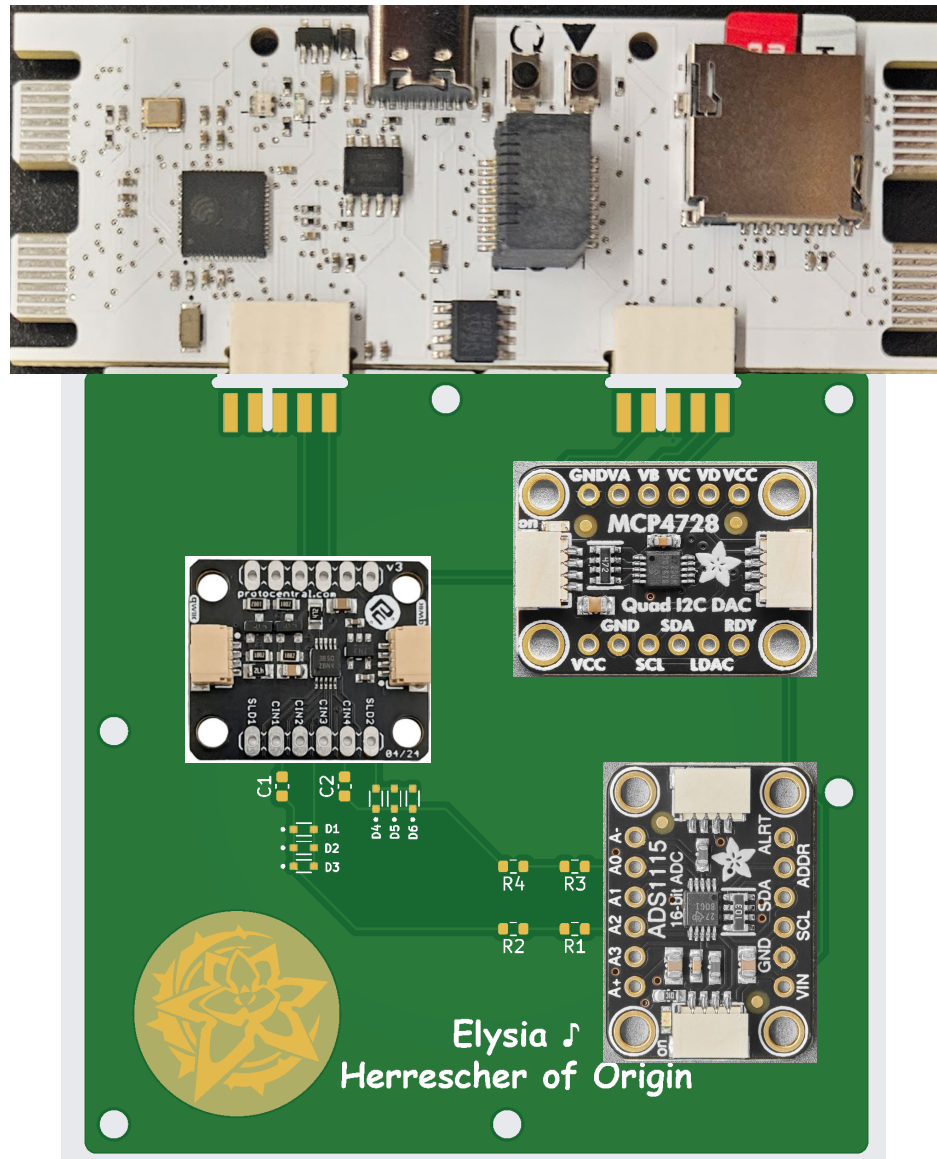


Top

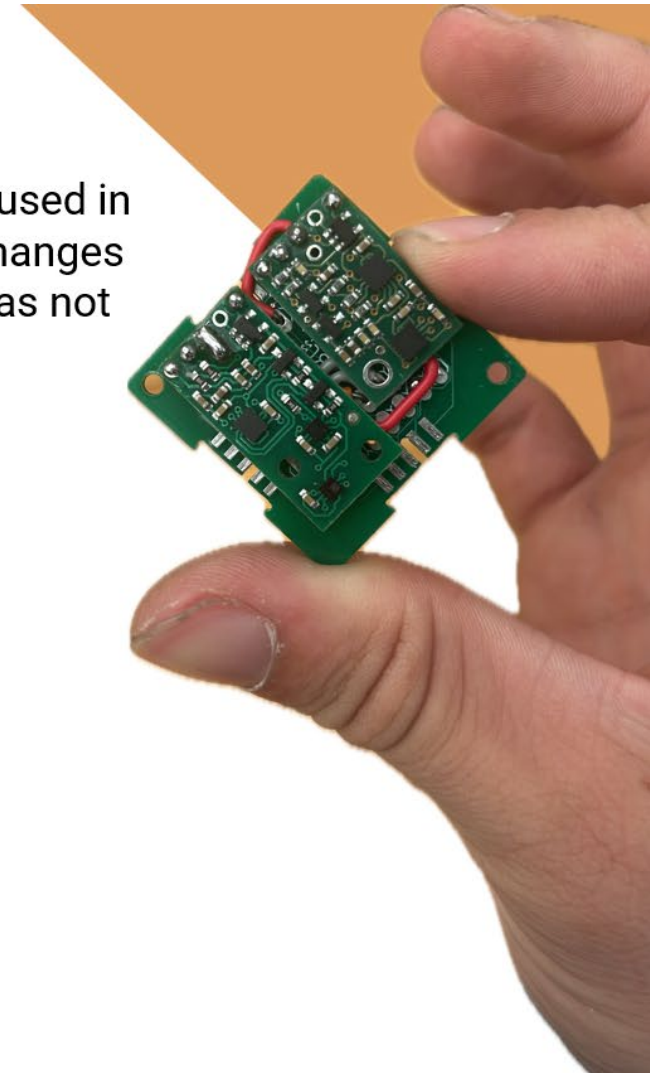
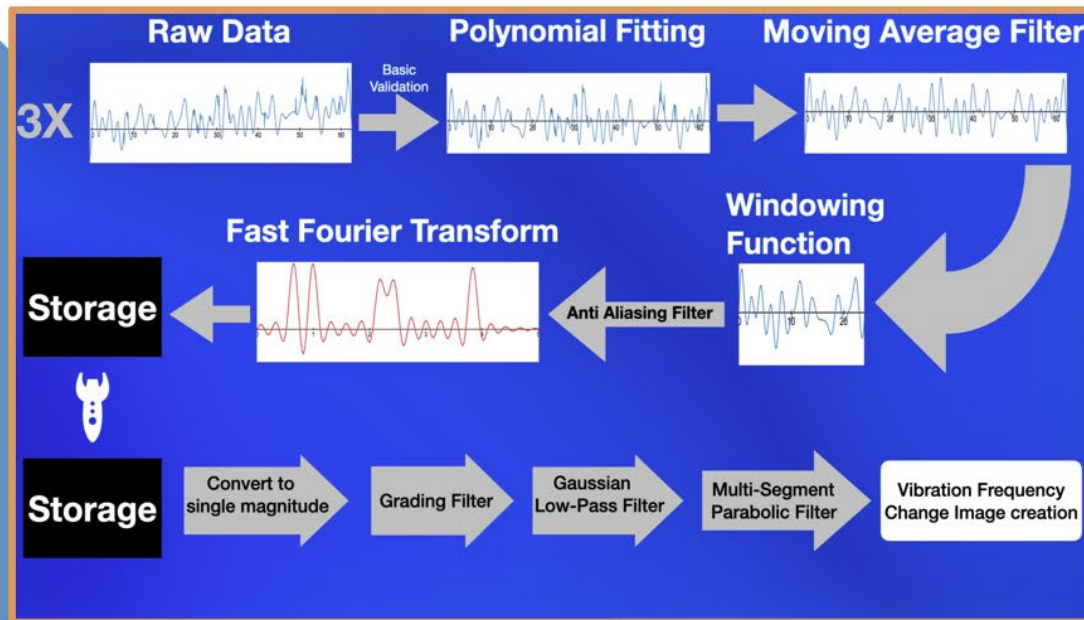


**Voltera
circuit**

Bottom

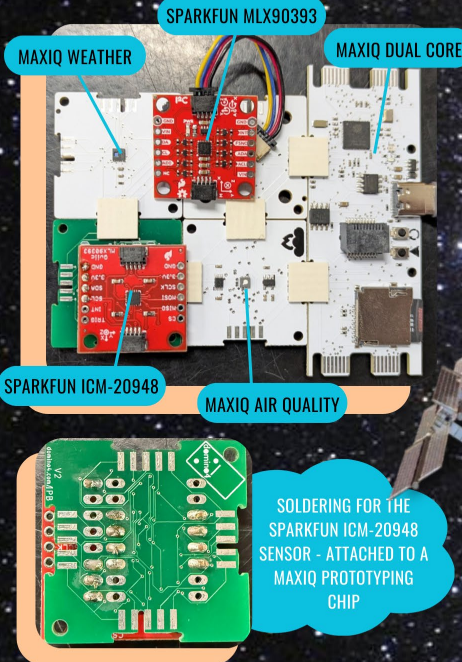


We plan to conduct a vibrational analysis, a technique commonly used in electric pump systems involving measuring vibrations and their changes over time to predict faults before they occur, on the ISS where it has not been attempted before.





PAYLOAD LAYOUT



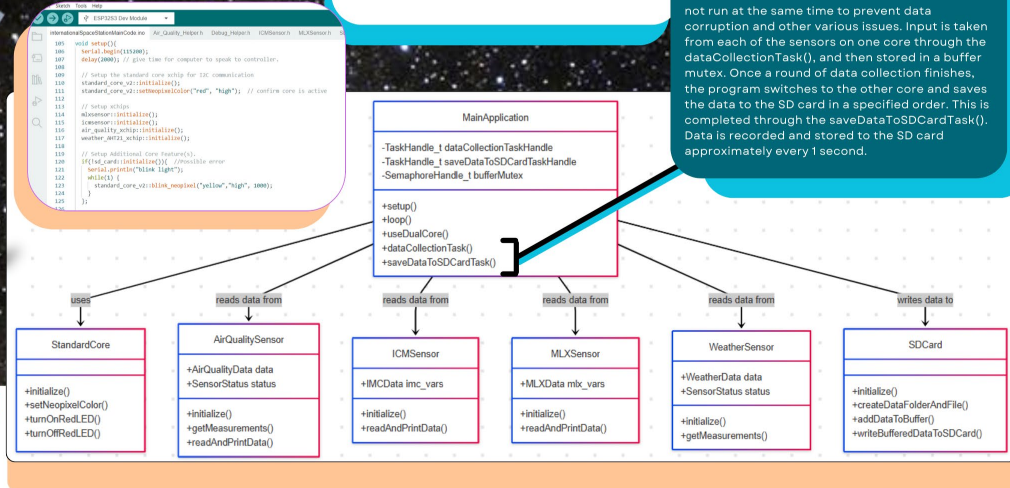
SENSORS

1. **Magnetometer:** SparkFun Triple Axis Magnetometer Breakout - [MLX90393](#)
2. **Gyroscope:** SparkFun 9DoF IMU Breakout - [ICM-20948](#)
3. **Air Quality:** MaxiQ Air Quality sensor (CO₂, Ethanol, Hydrogen, VTOCs)
4. **Weather:** MaxiQ Weather xChip (Temperature and humidity)

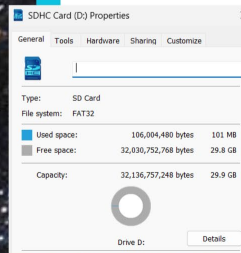
MAXIQ RIDESHARE MISSION TO THE INTERNATIONAL SPACE STATION

PROGRAMMING & TESTING

The main program is composed of two tasks, or functions, that run on separate cores of the MaxiQ Dual Core. The code ensures that the two tasks are not run at the same time to prevent data corruption and other various issues. Input is taken from each of the sensors on one core through the dataCollectionTask(), and then stored in a buffer mutex. Once a round of data collection finishes, the program switches to the other core and saves the data to the SD card in a specified order. This is completed through the saveDataToSDCardTask(). Data is recorded and stored to the SD card approximately every 1 second.



All code was developed in Arduino IDE using the Arduino language, which is based off of C++. Each sensor program is written as a library that can be included in the main program. The libraries all include a namespace for the corresponding sensor, which defines variables and functions to help the program run. Variables are created inside structs for every data column, and each sensor has at least two functions to initialize the sensor and record data.



TEST #1: SD CARD STORAGE TEST

(12:45 pm 3/11/2025 → 12:55 pm 3/17/2025) → 6 days of continuous running and data storage

Used space: 106,004,480 bytes (106 MB)
Total space: 32,136,757,248 bytes (32 GB)
Space per day (used space/6 days): 17,667,413.333 (17.6 MB)

If the payload continuously ran, storage would run out in 1,818 days or just under 5 years (total space/space per day)


The SD Card stored two folders titled "0" and "1", meaning that the payload stopped and restarted at some point. It is helpful to know that this piece of the code works (The data will not be overwritten if power is turned on and off).



PRINCETON UNIVERSITY LIBRARY 2022.09.28 16:27:17 -05'00'


Interface Definition Document (IDD)

Nanoracks Mainframe (Nanode)



Doc No: NR-NANODE-S0001
Revision: A


NANORACKS PROPRIETARY RIGHTS ARE INCLUDED HEREIN. RECIPIENT AGREES THAT NEITHER THIS DOCUMENT NOR THE INFORMATION DISCUSSED HEREIN NOR ANY PART THEREOF SHALL BE REPRODUCED OR DISCLOSED TO OTHERS.

 NASA TECHNICAL STANDARD National Aeronautics and Space Administration Washington, DC 20546-0001	NASA-STD-6016
	Approved: 07-11-2008 Expiration Date: 07-10-2013 Superseding NASA-STD-(I)-6016

STANDARD MATERIALS AND PROCESSES REQUIREMENTS FOR SPACECRAFT

MEASUREMENT SYSTEM IDENTIFICATION: METRIC/SI (ENGLISH)

APPROVED FOR PUBLIC RELEASE – DISTRIBUTION IS UNLIMITED

 INTERIM NASA TECHNICAL STANDARD National Aeronautics and Space Administration Washington, DC 20546-0001	NASA-STD-(I)-6001B
	Approved: 04-21-2008 Expiration Date: 04-21-2010 Superseding NASA-STD-6001

FLAMMABILITY, OFFGASSING, AND COMPATIBILITY REQUIREMENTS AND TEST PROCEDURES

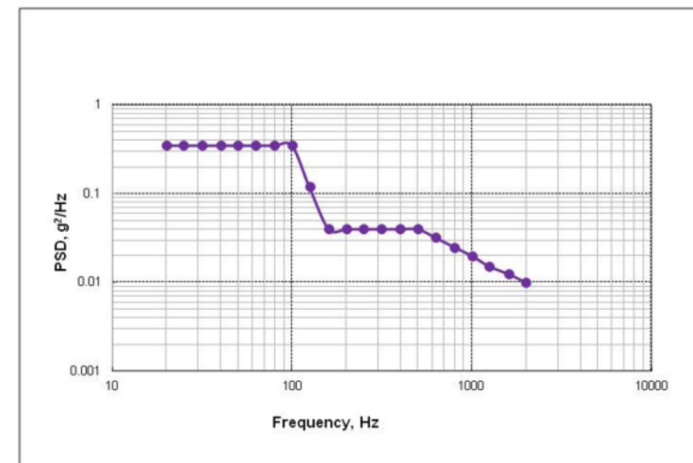
MEASUREMENT SYSTEM IDENTIFICATION: METRIC (INCH-POUND)

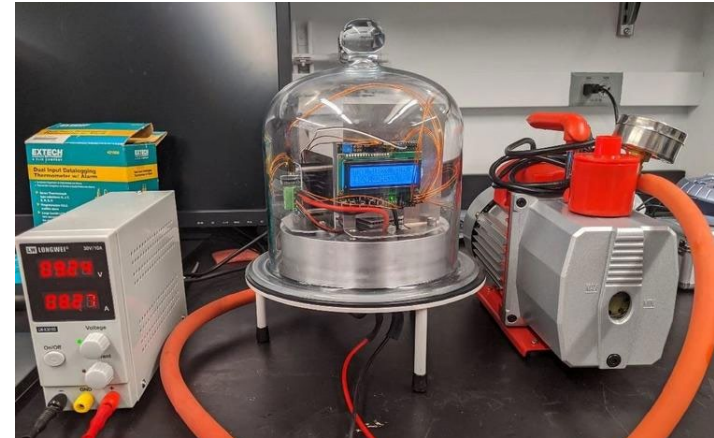
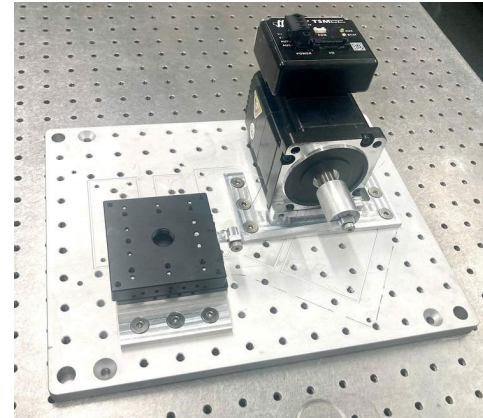
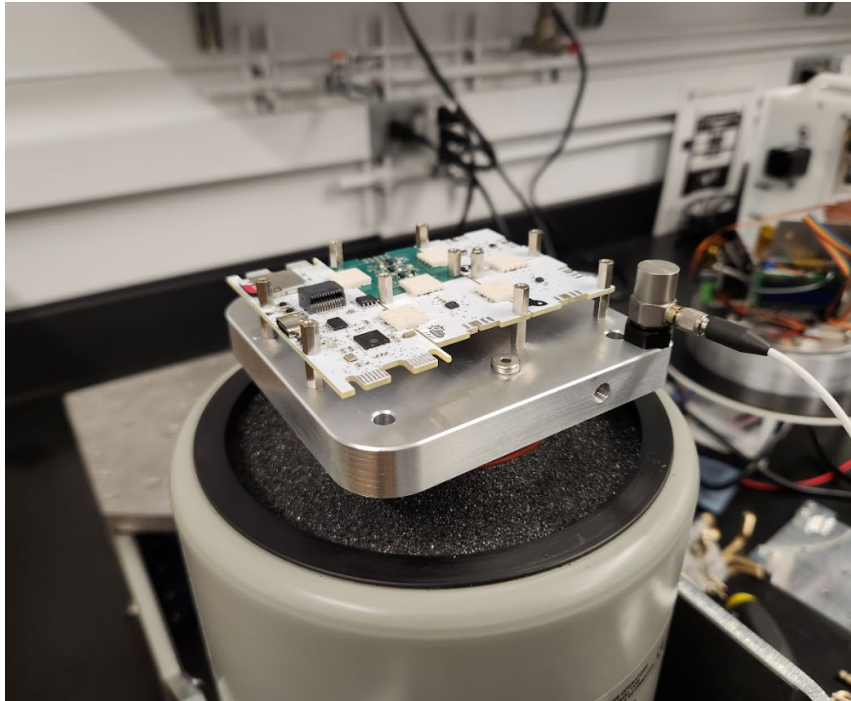
This document represents the technical consensus of the developing group but does not yet have final NASA approval.

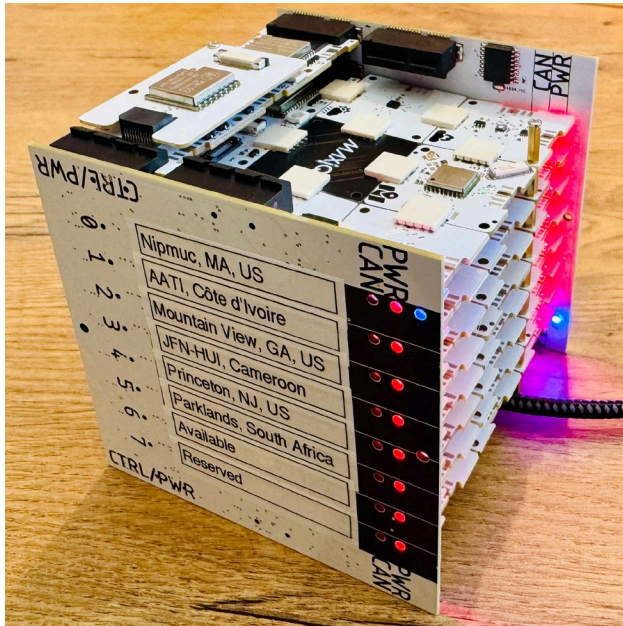
APPROVED FOR PUBLIC RELEASE—DISTRIBUTION IS UNLIMITED

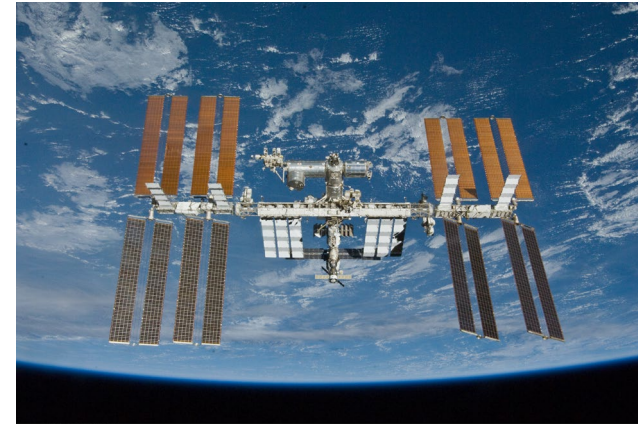
Table 4.2.4.2-1: Random Vibration Environment
Ref. SSP 57000, Rev S, Table D.3.1.2-1

FREQUENCY (Hz)	PROTOFLIGHT TEST LEVEL (g ² /Hz)
20	0.35
100	0.35
100 - 160	-15.4 dB/octave slope
160 - 500	0.04
500-2000	-3 dB/octave slope
2000	0.01
Overall	8.8 g _{rms}
Duration	1 min/axis

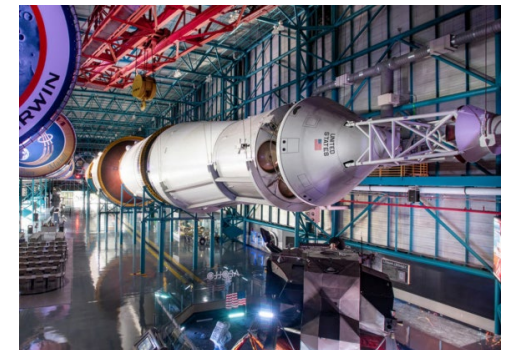








S132E012212





Launch



Delivery



On-Orbit Operations



Return



Landing



Retrieval

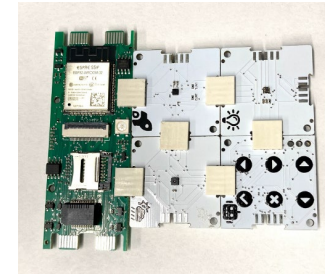


Receive At
NanoRacks

Early-Bird Pricing: \$15K total

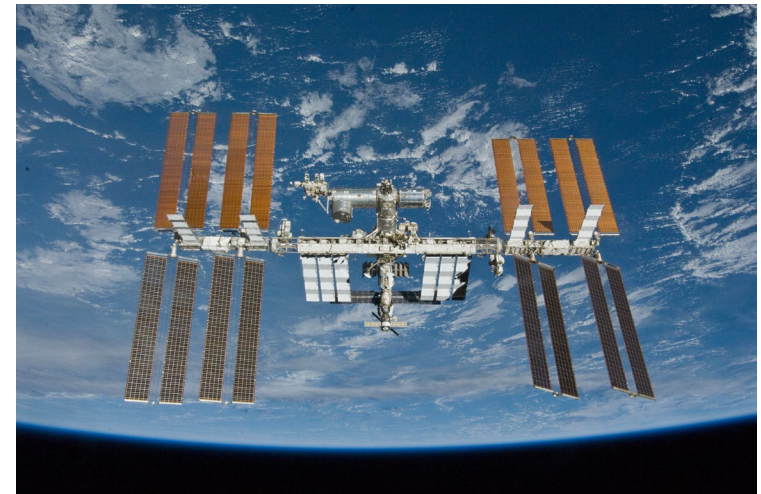
First 40% payment (\$6K) =

- No-solder sensor/coding kits for your whole classroom
- Furnished curriculum, tutorials & live MaxIQ-hosted Zoom workshops
- Includes HAB launch!



Second 60% payment (\$9K, due later) =

- Continue on to the batched launch to the ISS
- Astronauts bring experiments back to Earth after 6-week mission!



S132E012212

(7th-12th grade or adaptable to universities)

Mike Galvin

mgalvin@princeton.edu

tigersats.princeton.edu

<https://themaxiq.com/products/international-space-station>

