

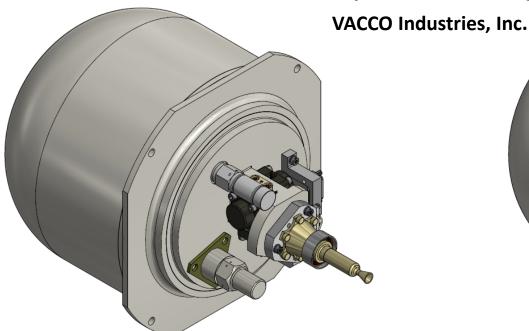


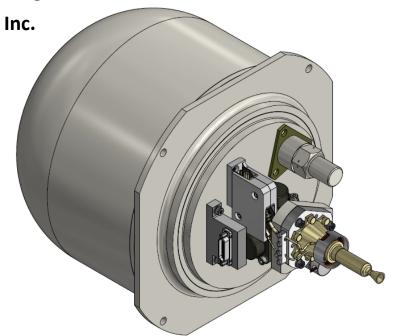
A High-Performance Delta-V Thrust Module for CubeSats and Small Satellites

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Thrust Modules are Simple

- Blow-down design eliminates the need for pressure control:
 - Maximum Design Pressure (MDP): 400 psia
 - 28V input
 - Standardized Manifold Assembly contains all the functional components
 - All-welded titanium construction
 - Features three normally-closed propellant valves directly operated by the user
- Single 1N Axial thruster:
 - Provides delta-V only
 - Flight proven: Eighty-two LMP-103S thrusters flown on twenty-one spacecraft
 - Initial thrust: 1N, final thrust: 150mN
 - Impulse Limit: >50,000N-sec
 - Minimum Impulse Bit: 50 mN-sec at 1N thrust
 - Primary and redundant eight-watt heaters included
 - Redundant thermocouples sense temperature
 - 30-minute pre-heat required before firing
- Flight-proven VACCO three seal Mini Fill & Drain valve

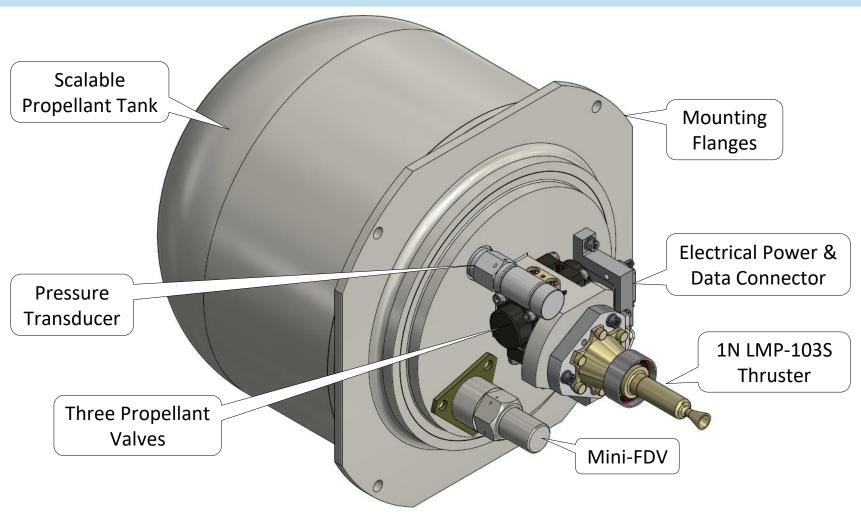


Thrust Modules are Simple

- Flight-proven Pressure Transducer provides fuel state
 - Output 0.1V to 4.9V proportional to pressure
 - Total error band 1.5% of full scale
- Range Safety Features:
 - Green, non-toxic LMP-103S Monopropellant:
 - Can be shipped via commercial aircraft (UN / US 1.4S)
 - Three Seals Against Leakage
 - Benign, flight proven GHe Pressurant
- Optional analog electronics assembly:
 - **Bolt-on subassembly**



1N Thrust Module

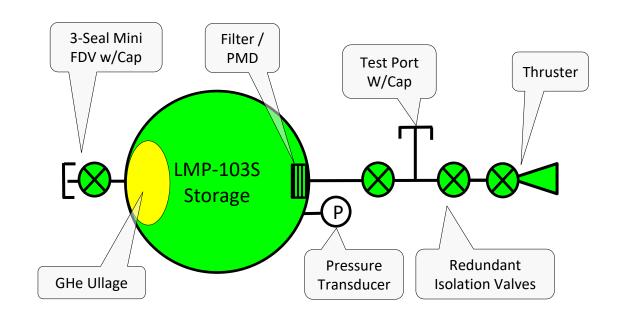


1N Thruster and Manifold Assembly are Standardized, Tank is Sized as Required





Thrust Module "Blow Down" Schematic & Performance

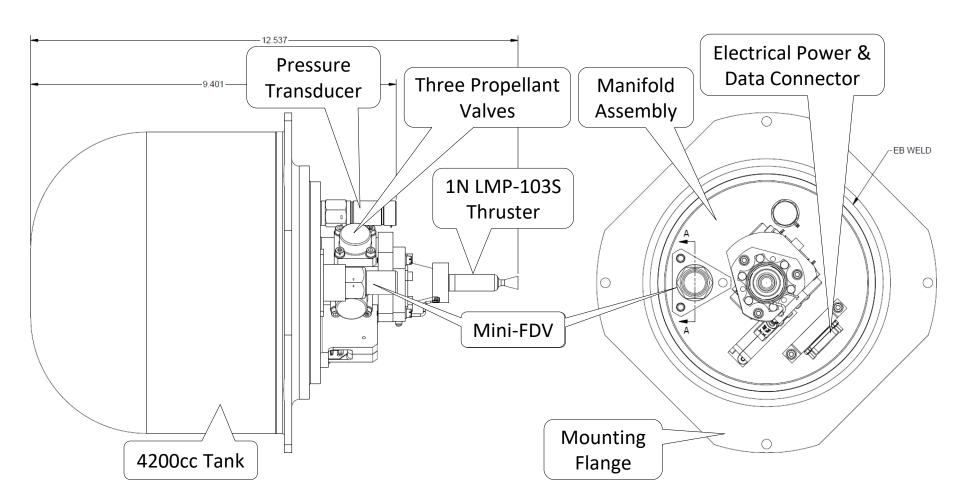


			Usable			MiPS	Initial			
		Prop	ADN		Total	Dry	Satellite		Ghe	Initial
	Tank Vol.	Vol.	Mass	Avg. Isp	Impulse	Mass	Mass	Delta-V	Vol.	Pressure
	(cc)	(cc)	(g)	(sec)	(N-Sec)	(Kg)	(Kg)	(M/s)	(cc)	(psia)
1N LMP Thruster	2000	1560	1774	218	3785	3.2	100	38	440	320
1N LMP Thruster	3300	2575	2973	218	6341	3.5	100	64	725	320
1N LMP Thruster	4200	3300	3872	218	8259	4.0	100	84	900	320





1Nx4200cc Thrust Module: Envelope



Key Component: 1N LMP-103S Thrusters









LMP-103S Monopropellant:

ADN (Ammonium DiNitrimide) Based Propellant

Non-Toxic, Non-Carcinogenic

UN / US 1.4S (Commercial Aircraft Classification)

Safe Fueling without SCAPE suits

>6% Higher Specific Impulse vs. Hydrazine

24% Higher Propellant Density than Hydrazine

Can be Stored for >20 Years

Insensitive to Space Radiation

Insensitive to Air and Humidity

Broad Storage Temperature Range (-7° to +120°C)

1N High Performance Green Propellant Thrusters:

Eighty-Two Thrusters Flown in Space

Flight Proven for Ten Years

Qualified for 60,000 Pulses

Qualified for 1.5-hour Continuous Firing

Broad Operating Temperature Range (-5°C to 60°C)

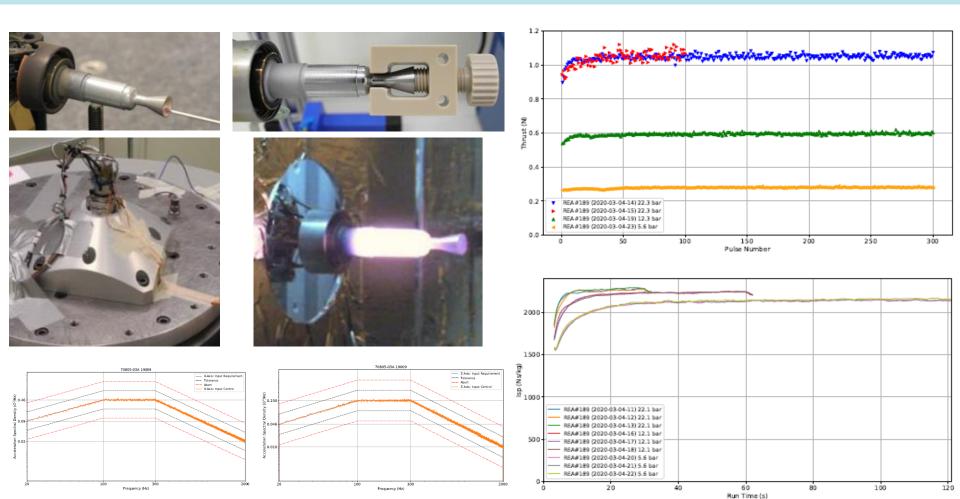
Non-corrosive Exhaust Gases (H2O, N2, H2, CO, CO2)

82 Thruster Flown with Hundreds of Units in Production or Waiting for Launch





1N Thrusters in Continuous Production



100% of 1N Thrusters are Hot-Fire Tested During ATP



Optional Analog Electronics Assembly (AEA)

- Provides independent control of valve driver
- Valves are controlled by digital logic signals
- Valve voltage stepdown is accomplished in the AEA
- Provides a single unified connector for electrical interface
- Simple & Robust system does not require a controller, processor or software
- Current will automatically drop to holding current after 50mS:
 - Reduces power consumption
 - Reduces valve self heating
- Back EMF is automatically blocked



VACCO Flight Controller



Conclusions

- VACCO has been designing and building small satellite propulsion systems since 2003:
 - Forty (40) propulsion systems of sixteen (16) different designs built to-date
 - Experience gained and lessons learned have improved the performance and reliability
- VACCO has designed a simple, high-performance green monopropellant propulsion module designed for shorter lead times and lower cost:
 - Self-contained propulsion assemblies that provide delta-V thrust
 - Green thruster, feed system and propellant storage in a bolt-on, plug-in assembly
 - Host CubeSat controller directly controls of the Thrust Module
 - System provides three seals against overboard loss of propellant
 - Propellant tank geometry can be customized for a total impulse >50,000N-sec
 - A flight-qualified pressure transducer accurately measures remaining propellant
 - A flight-qualified Fill & Drain Valve provides a propellant and Pressurant fluid interface
 - An optional analog electronics assembly provides valve driver circuits that suppress back-EMF and regulate voltage to minimize power consumption
- The initial Thrust Modules use flight-proven 1N LMP-103S green thrusters
- Thrust Module is compatible with other green propellants and larger thrust levels