Direct Thrust Measurements and Full Performance Mapping of the IFM Nano Thruster at ESA ESTEC Facilities

David Krejci et al.











9

(in)

f

IFM Nano Thruster



MODULARIZATION -> STANDARDIZATION -> MULITPLE CONSTELLATIONS







y

in

t.



INDIUM FEEP ion emitter

0

Building on 30 Years of Development at FOTEC



Debris safe

No Pressure

Non-Toxic

ENPULSION

100% INCOMING INSPECTION OF ALL MECHANICAL PARTS
100% TRACEABILITY OF ALL PARTS, MANUFACTURING AND TESTING
LEAN PRODUCTION (KANBAN) FOR HIGH THROUGHPUT
BATCH TESTING ON PART, COMPONENT AND SYSTEM-LEVEL



2.3



•

• ENPULSION 100% Testing: Every emitter fired 2x

Batch testing to increase throughput



ESA performance mapping



ESA thrust balance

2 thrusters with low impedance emitters were selected

- $\rightarrow\,$ Allow to maximize thrust at low power
- \rightarrow Do not allow to reach Isp > 4000s



2 identical thrusters

www.enpulsion.com

ESA performance mapping



ESA thrust balance

thrusters

thrusters

with low

emitters

selected

were

Verification of thrust telemetry with direct thrust balance measurements



From: Krejci et al: Full Performance Mapping of the IFM Nano Thruster including Direct Thrust Measurements, submitted to JoSS



y

ESA performance mapping



ESA thrust balance

2 identical thrusters





From: Krejci et al: Full Performance Mapping of the IFM Nano Thruster including Direct Thrust Measurements, submitted to JoSS



7

High Δv optimized thrusters



High emitter impedance selection optimized for high delta V missions



Thrusters currently in delivery: high impedance to increase total impulse



3U Cubesat SSO orbit After full commissioning verifying all subsystems

30 mins firing





y



in

www.enpulsion.com

3U Cubesat SSO orbit After full commissioning verifying all subsystems

Thrust steps: uncontrolled steps





3U Cubesat SSO orbit After full commissioning verifying all subsystems

Thrust steps with controlled transients



y



3U Cubesat SSO orbit

After full commissioning verifying all subsystems

Thrust steps with controlled transients





Conclusion



High rate

Independently verified performance envelope

In-orbit

Multiple thrusters per week shipped to customers, enabled by standardization and batch testing

Performance mapping at ESA thrust facility provided additional verification of thrust models to measured thrust and showed capability to operate at any operational point within the performance envelope

verification

Record-setting

Double digit number of thrusters in space now, firing tests presented show good controllability of thrust

-SFR 33

Operation at >5000s impulse on a 3U Cubesat

www.enpulsion.com



ENPULSION SPACECRAFT TECHNOLOGY

MARKET LEADER FOR SMALL SAT PROPULSION

