



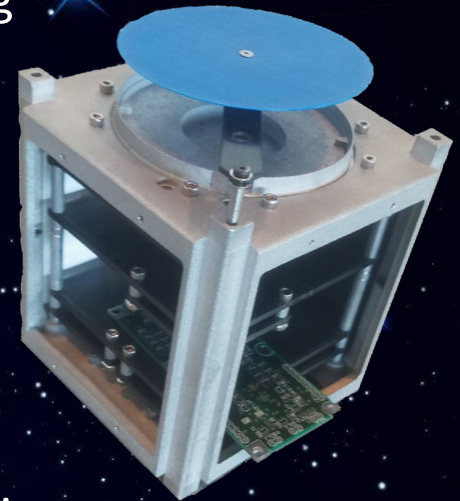
Flight Hardware for the 1U CubeSat EdgeCube

**Student Lead
David House**

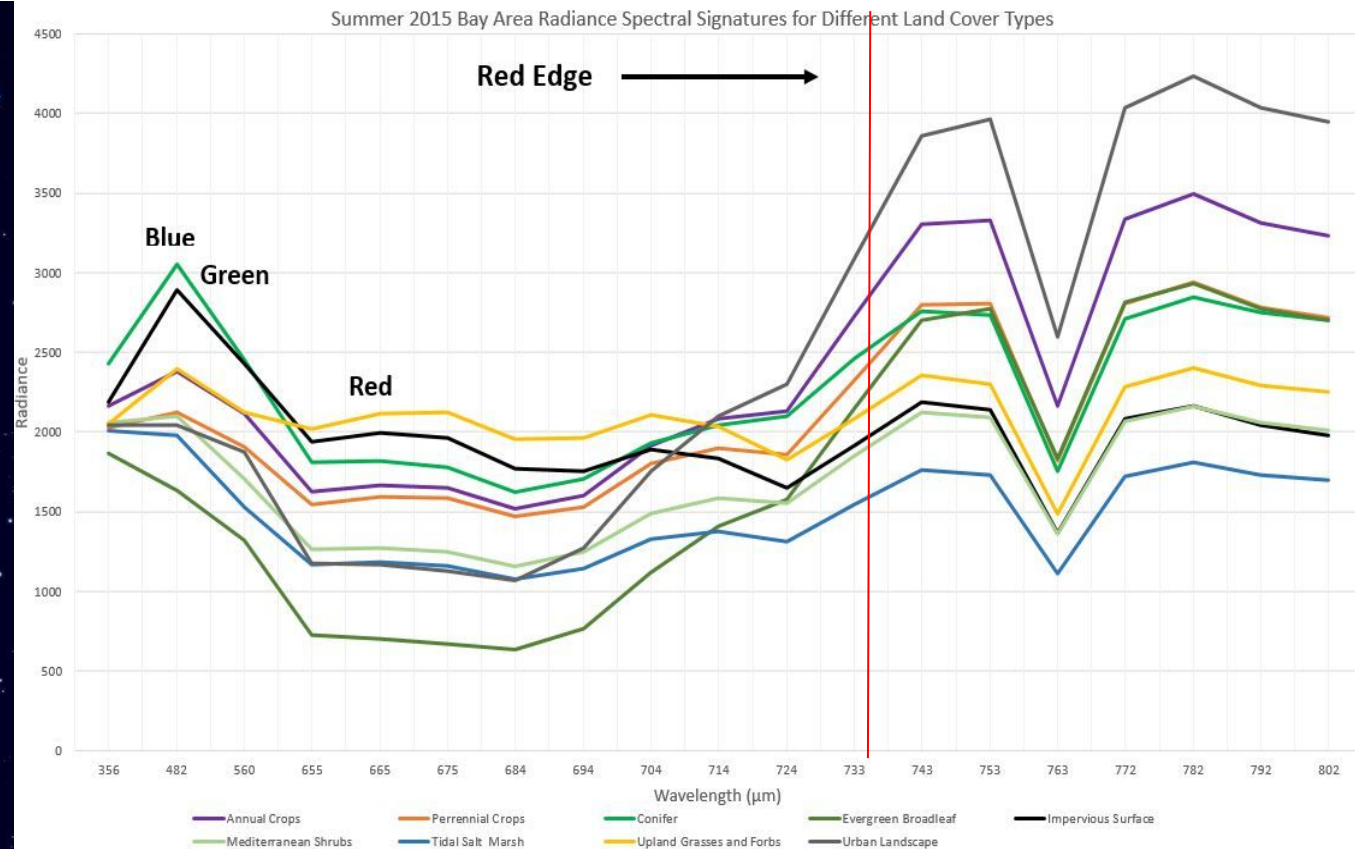
Webpage:
<http://lbym.sonoma.edu/edgecube/>

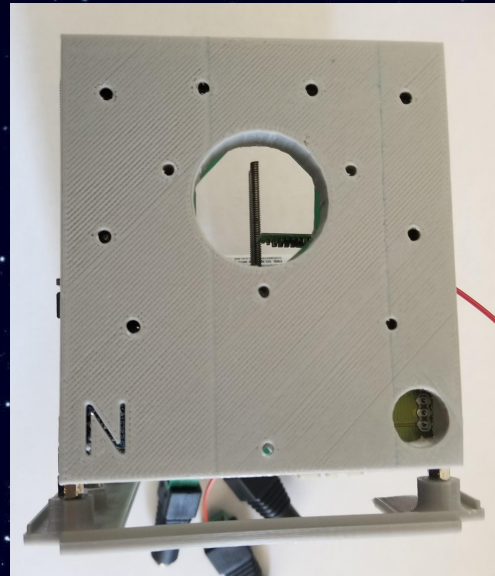
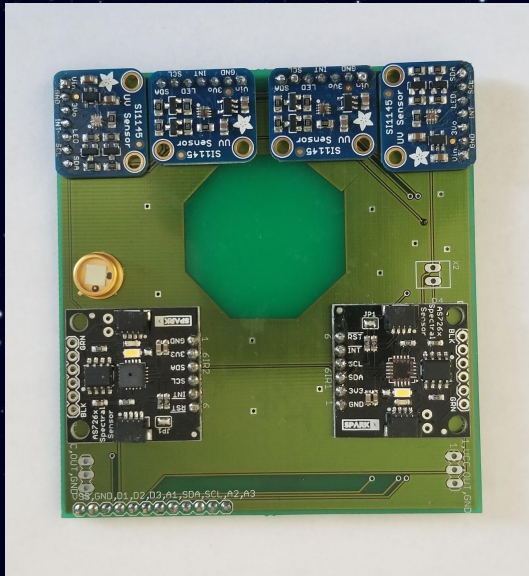
- EdgeCube
- Red Edge
- Science Instruments
- Attitude Control
- Solar Panel Circuit
- One-Wire
- Mechanical Structure
- Acknowledgments

- EdgeCube is a small satellite being built by students from SSU, Morehead State University, Santa Clara University and Moog
- Moog is working on the aluminum body and antenna deployment system
- Morehead is working on a particle detector
- Santa Clara is working on ground stations
- We (SSU) are working on the rest of the electronics
- The goal of EdgeCube is to track the Red Edge
- Launching on ELaNa XXIV, integration by SpaceFlight Industries
- Launch date ~October 1st, 2018



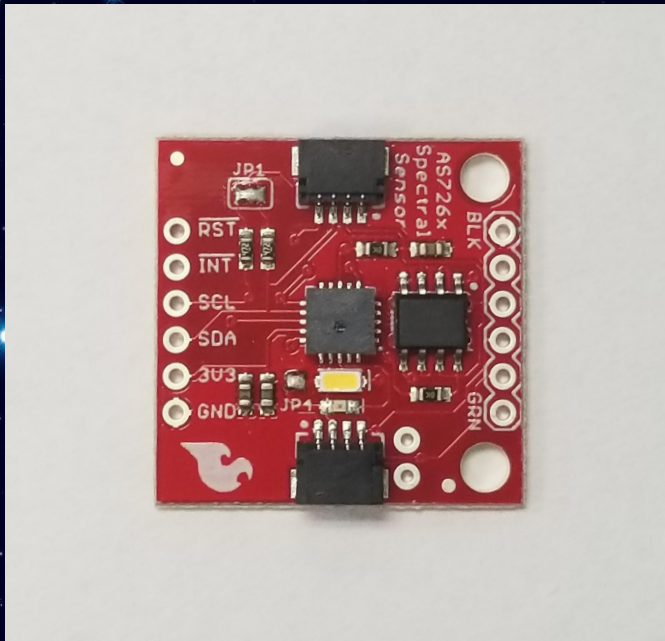
- Natural Difference Vegetation Index (NDVI)
- Calculate vegetation using visible and infrared light
- NDVI bands in figure to the right.



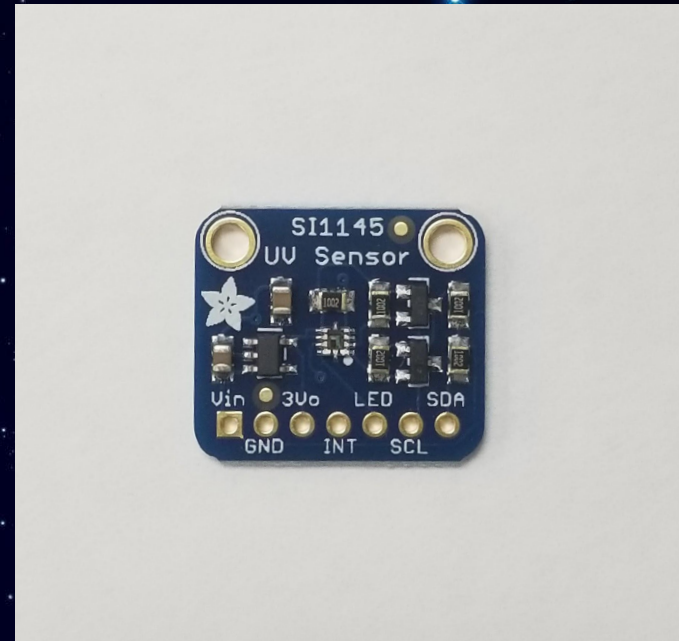


- Plug in sensor array to allow for ease of debugging
- AS7263: 610nm, 680nm, 730nm, 760nm, 810nm and 860nm each with 20nm FWHM
- SI1145: Visible and additional NIR filters

AS7263 Near Infrared Spectral Sensor



SI1145 Visible and Near IR sensor



- 2-axis magnetorquer provides spin rate control and detumbling capabilities.
- Onboard camera provides star tracking aid to the CubeSat.
- Connections to sun sensor on science panel provides sun altitude information to board MCU.

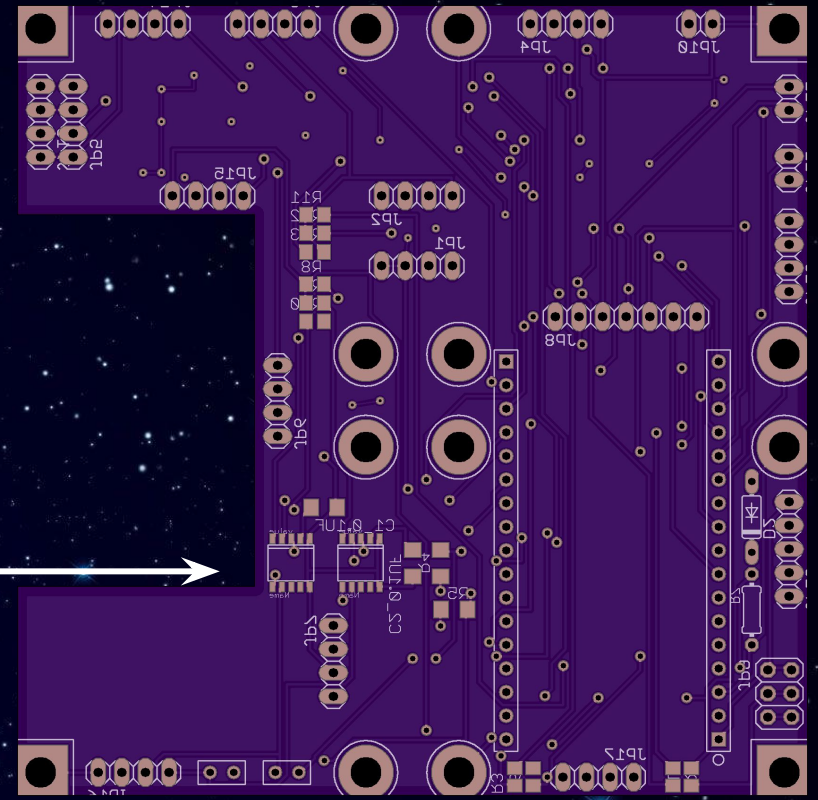
Sun Sensor Connections

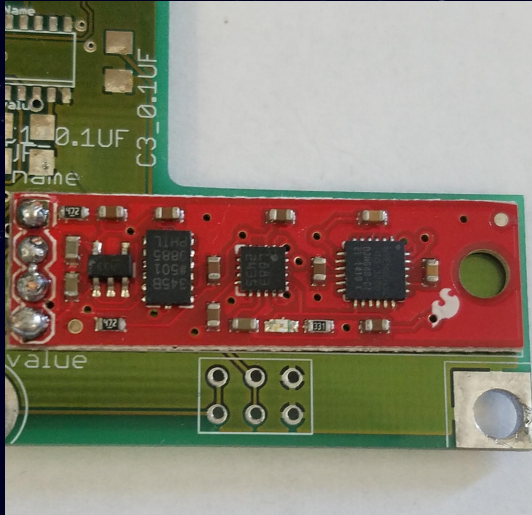


Camera Cavity

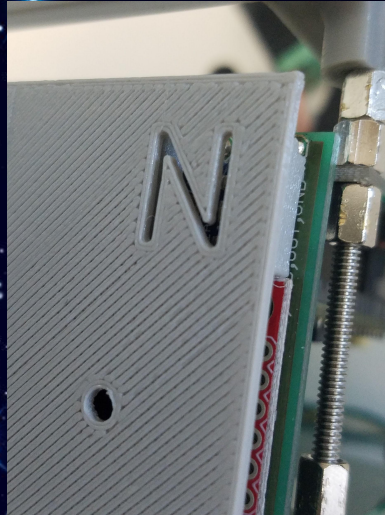


Magnetorquer Control Circuit





3-Axis accelerometer (ADXL345)
3-Axis digital compass (HMC5883L)
3-Axis gyroscope (ITG-3200)

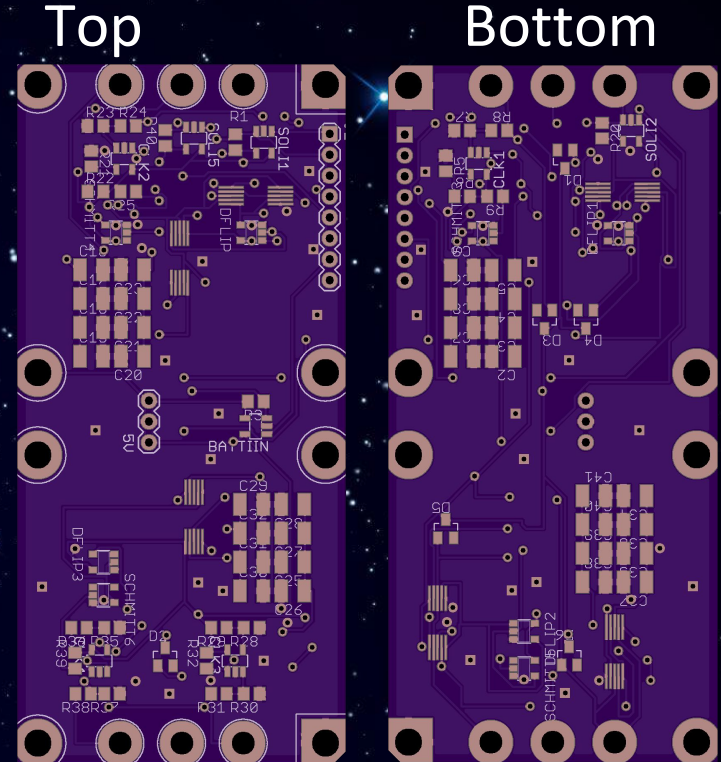


N Diffraction Slit with
GA1A12S202 Light Sensor



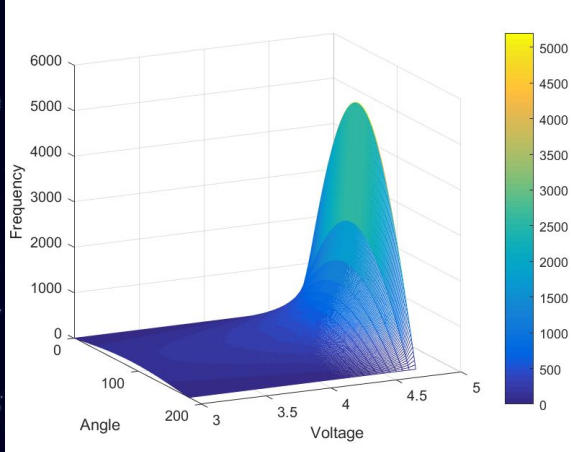
LinkSprite 2MP Camera

- Solar Panels are attached at the ends with lugs
- The circuit can be adjusted to receive voltage from 3.3-12 volt panels by setting a voltage divider
- Frequency range set by capacitor values



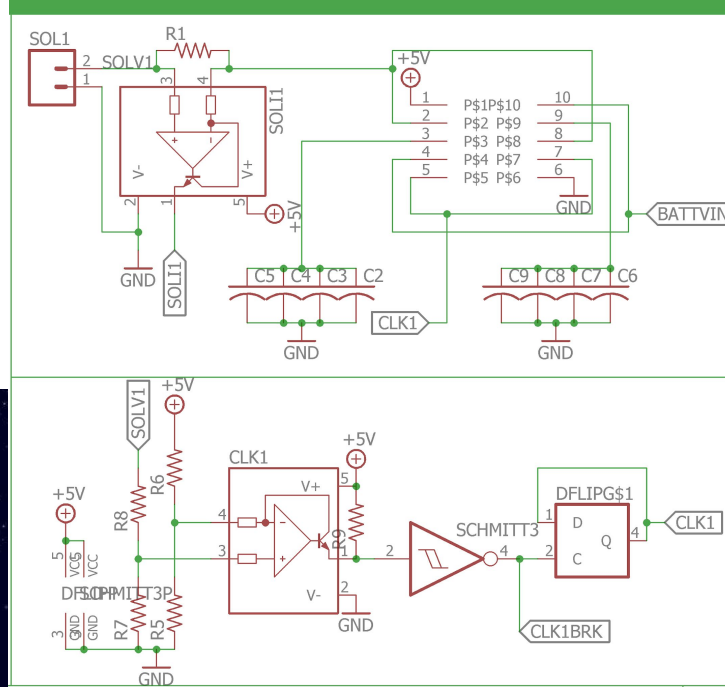
- NiCad Batteries: Higher reliability at the cost of power storage limitation
- Super capacitors to help the battery with impulses of power draw
- When batteries lose capacity SuperCaps will allow the CubeSat to continue running





- The frequency of our circuit shifts based on the amount of power the panel can produce

Sol 1



- Many CubeSat power systems use microprocessors
- A set of capacitors that alternate based on the charge stored in them
- Each panel has its own circuit so the frequencies do not depend on one clock

- Our circuits efficiency depends on angle and battery voltage

Efficiency

$$\frac{P_{\text{Battery}}}{P_{\text{Solar}}} = P_{\text{in \%Loss}}$$

$$P_{\text{in \%Loss}} = 14.54\%$$

Efficiency = 85.46% at 3.7 Volts

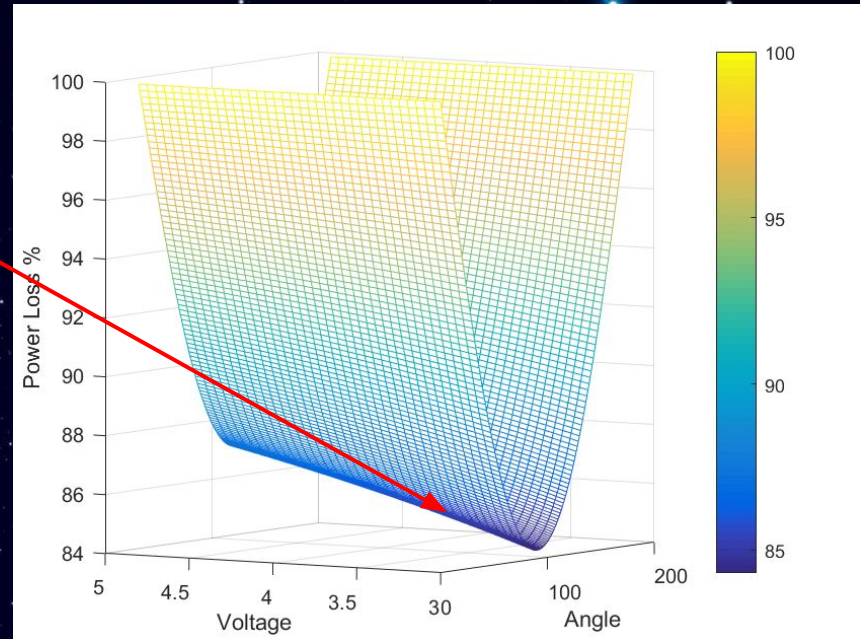
Circuit power

$$P = V * I$$

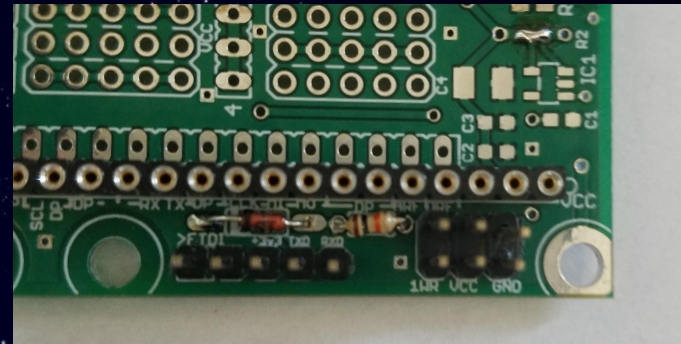
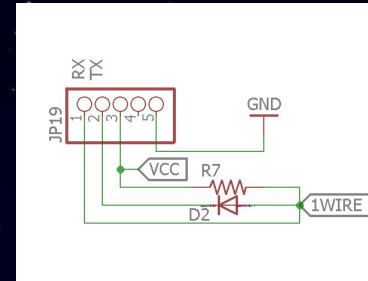
$$P = 3.7 \text{ V} * 1 \text{ mA}$$

$$P = 3.7 \text{ mWatts}$$

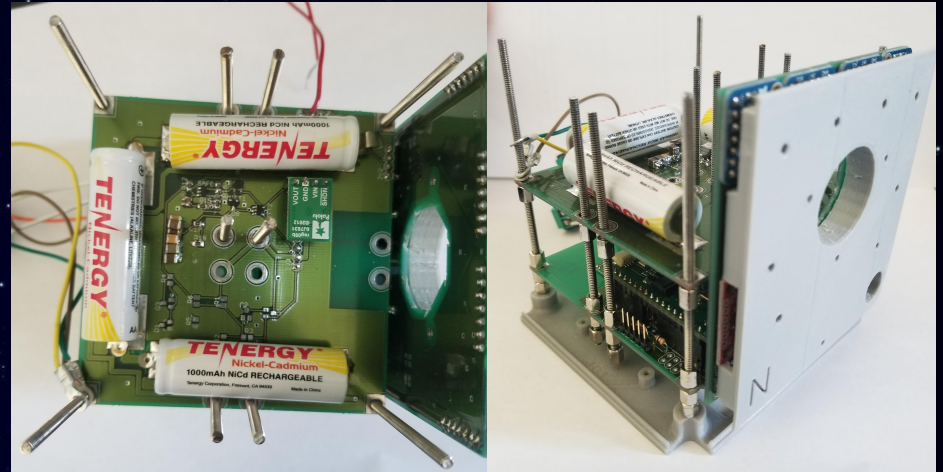
$$P_{\%used} = 3.7/250 = 1.48\%$$



- The UART RX and TX pins for the microcontrollers are tied together to reduce board connections
- A simple circuit was added to allow a USB serial line to be connected to boards for debugging



- 16 rods spaced within the CubeSat hold the boards together
- The rods double as electrical connections between boards





State University
David House (SSU)

SSU Student Team

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Tomas Zavala (Geography)

Acknowledgements:

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EdgeCube Team

<http://lbym.sonoma.edu/edgecube>



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Institutions

HR (H-Bar Research)
 SSU (Sonoma State University)
 MSU (Morehead State Univ. also Montana State Univ.)
 SCU (Santa Clara University)
 NA (Noqsi Aerospace)
 PICO (Playful Invention Company)
 MOOG (Moog CSA)

* formerly Space Sciences Laboratory, UC Berkeley

** formerly Climate Research, NASA GSFC

***Volunteers working with Jernigan at SSU



Questions/Comments