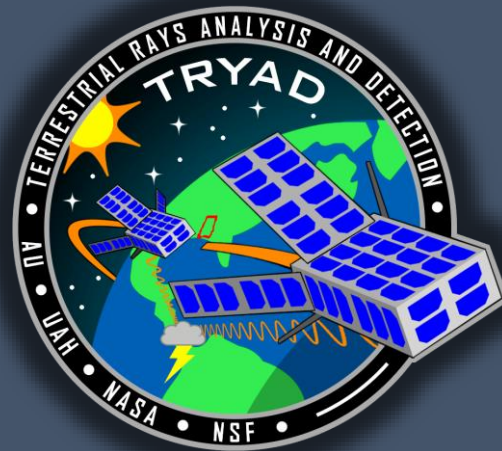


# Terrestrial RaYs Analysis and Detection (TRYAD) Cubesat Mission



Division of Atmospheric and Geospace Sciences (AGS)

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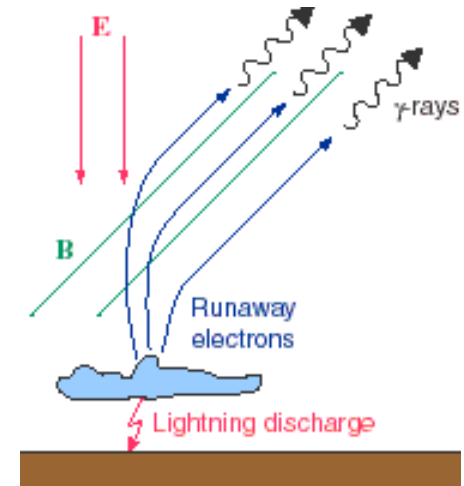


# TRYAD Science Overview

## Primary Science Goal:

*Multi-point Observations of Terrestrial Gamma-ray Flashes (TGFs) to test TGF Beam Models*

- What are TGFs?
- History of detecting TGFs
- What is unique about the TRYAD mission?



- up to 10's MeV Gamma Rays
- $\mu$ s to ms timescale pulses
- Production models unverified

## Short History of TGF Detection

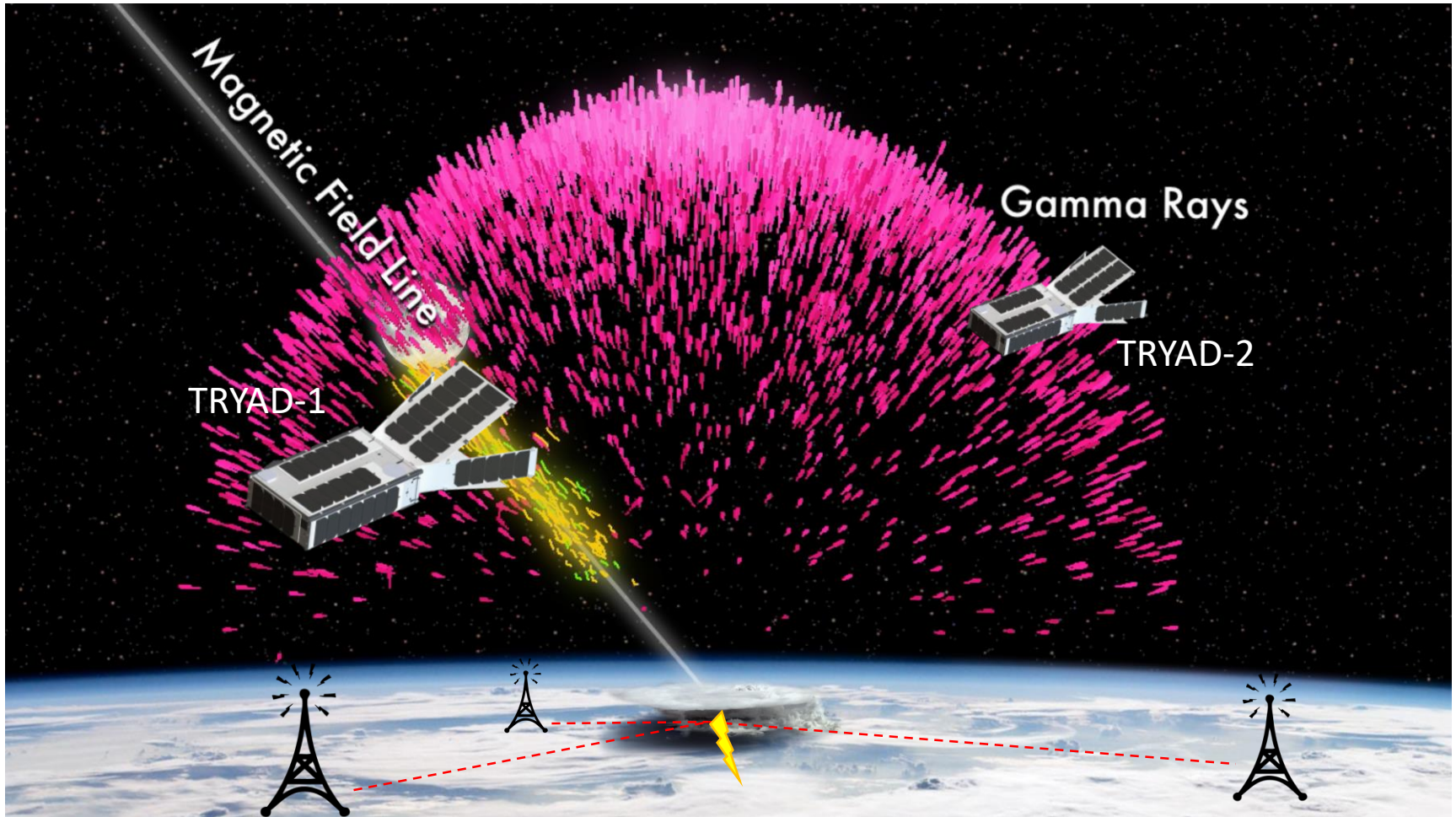
**1994** –Burst and Transient Source Experiment (BATSE) on Compton Gamma-Ray Observatory

**2005** – RHESSI satellite detected higher energy TGFs

**2009** – Gamma-Ray Burst Monitor on Fermi Gamma-Ray Space Telescope first detects TGFs and positrons

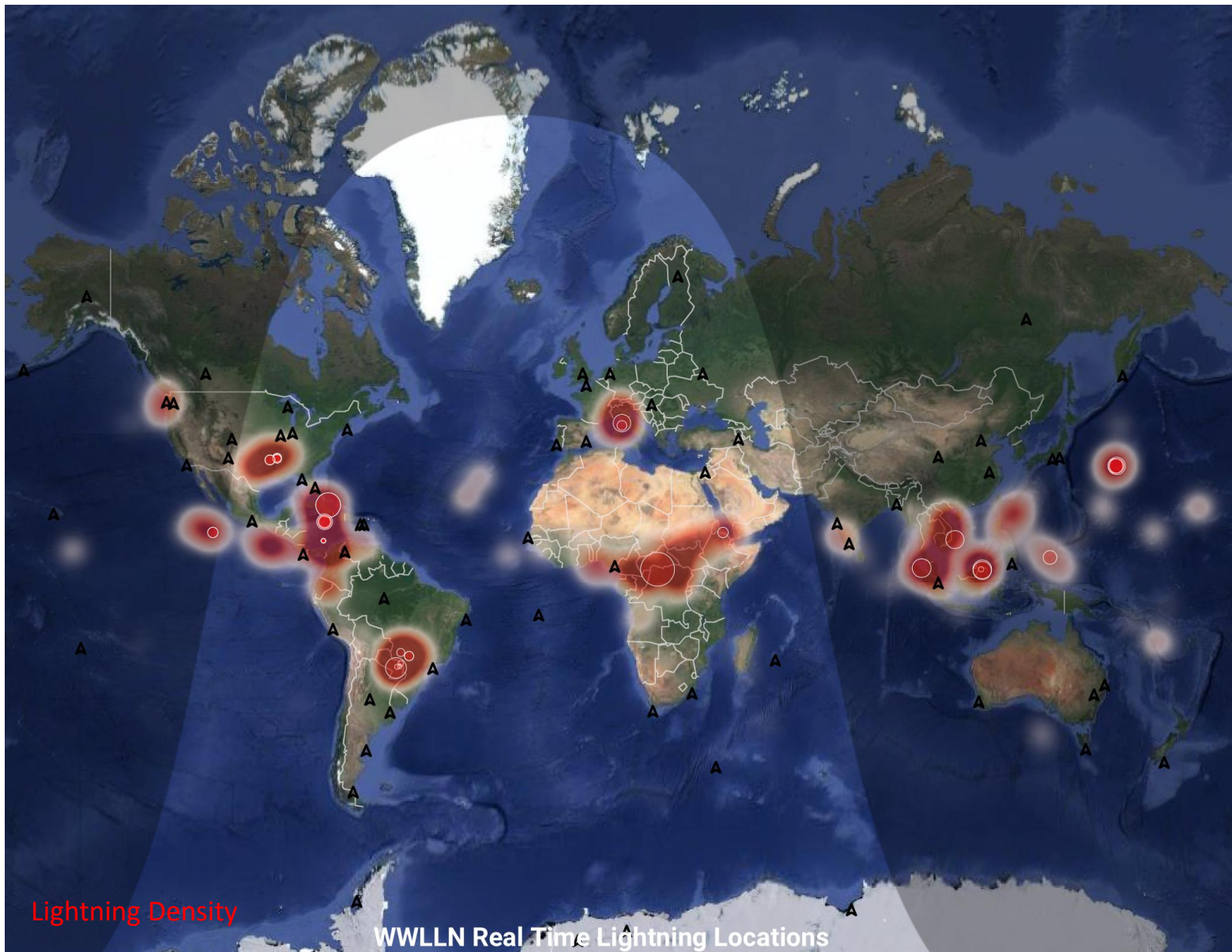
**Present** – thousands of TGFs are detected routinely

## TRYAD Science Overview



TRYAD uses two 6U CubeSats to make coincident measurements of TGFs and correlates to ground-based lightning detection data





[http://wwlln.net/new/map/lightning\\_map.html](http://wwlln.net/new/map/lightning_map.html)



## Typical TRYAD CubeSat Orbit

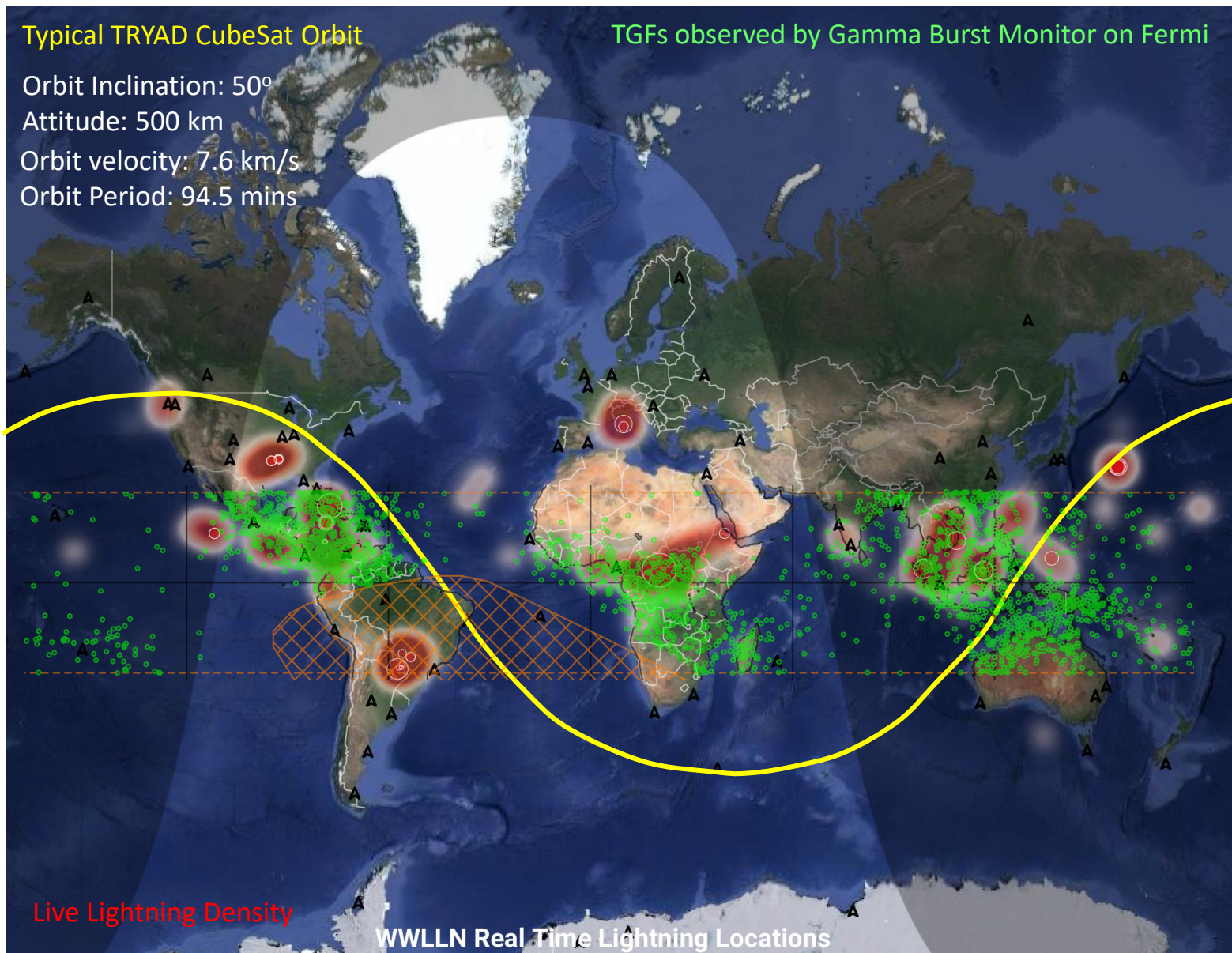
Orbit Inclination:  $50^\circ$

Attitude: 500 km

Orbit velocity: 7.6 km/s

Orbit Period: 94.5 mins

## TGFs observed by Gamma Burst Monitor on Fermi



Live Lightning Density

WWLLN Real Time Lightning Locations

[http://wwlln.net/new/map/lightning\\_map.html](http://wwlln.net/new/map/lightning_map.html)



## Command and Data

### Handling System (C&DHS)

- Embedded Linux
- Beaglebone w/ programmable realtime units

### Attitude Determination and Control System (ADACS)

- Magnetometers, rate gyros, sun angle sensors, orbit propagator
- Novatel GPS
- Magnetorquers & reaction wheels

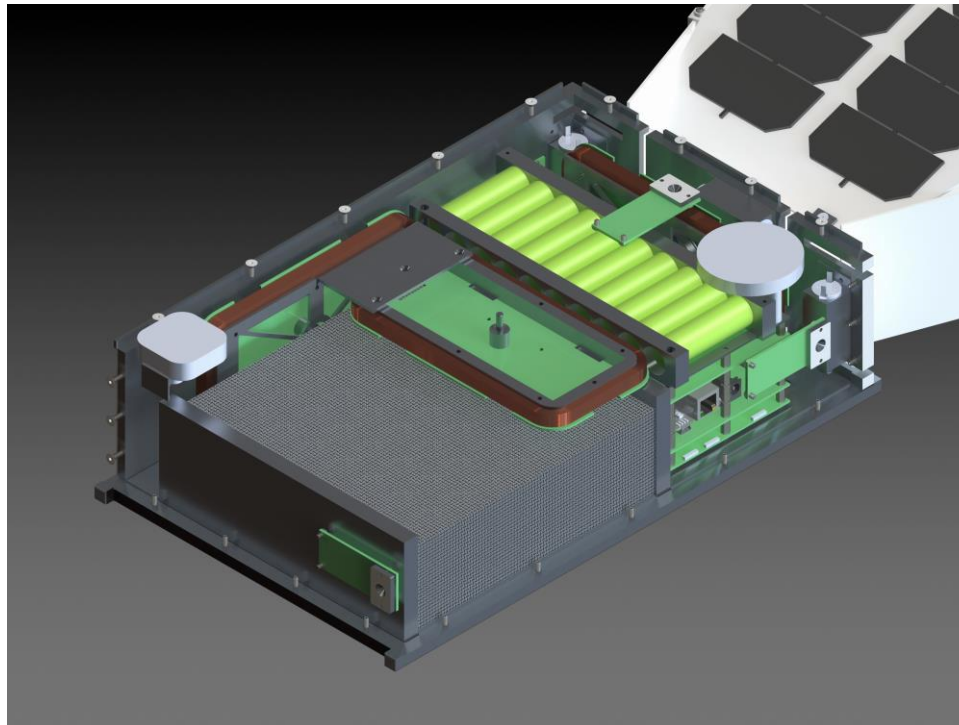
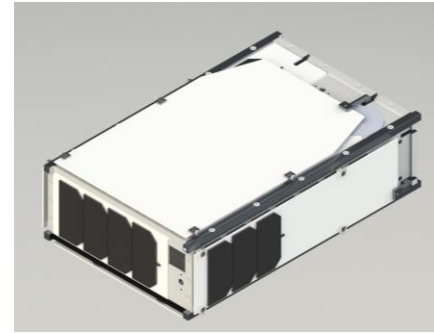
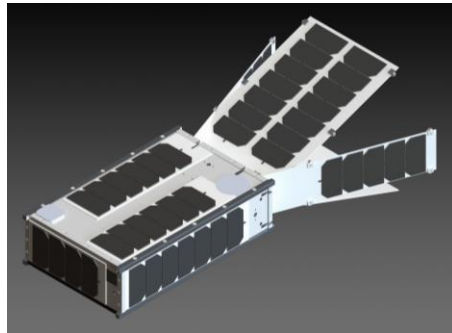
### Electrical Power System (EPS)

- 60 solar cells (29% eff.)
- Max power point trackers
- 10 Li-ion batteries

### Communications

- Globalstar bent pipe COMM
- Full Duplex Command & Control @ 256 kb/s over 45% of orbit
- Simplex telemetry beacons over 90% of orbit

## TRYAD 6U CubeSat



## Mechanical Systems

- Monolithic Al structure panels
- Driven deployable solar panels
- Passive thermal design

## Station Keeping

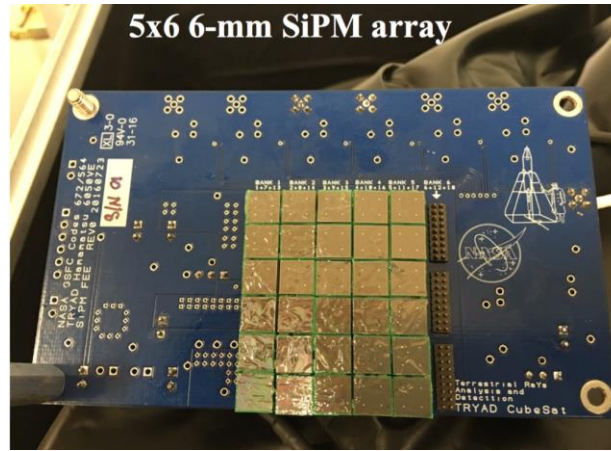
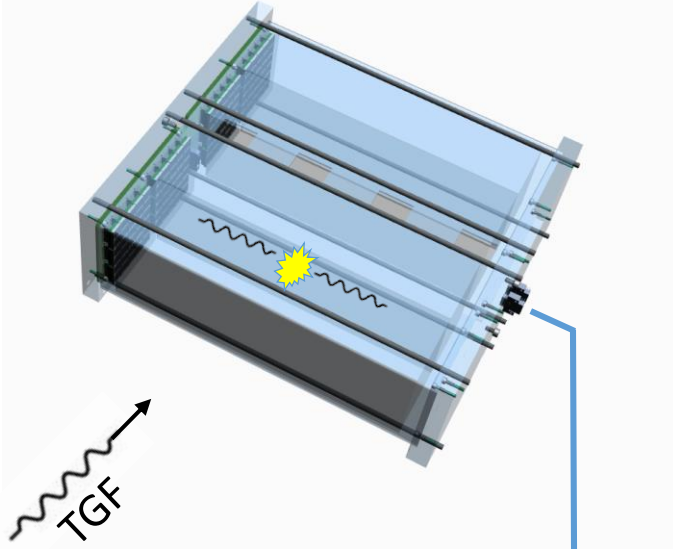
- Deployable "Dart" configuration for passive orientation augmentation
- Station keeping and satellite separation control via aerodynamic differential drag

## Science Payload

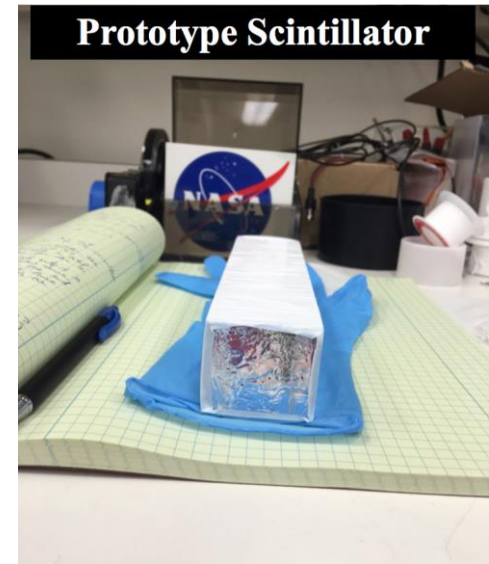
- Plastic Scintillation gamma-ray detector w/ next generation Si photomultipliers (SiPMs)
- >1 M sample/sec event time tagging to 2  $\mu$ s accuracy in real time (slaved to GPS clock)
- ROI's commanded based on weather and lightning data

# Science Payload

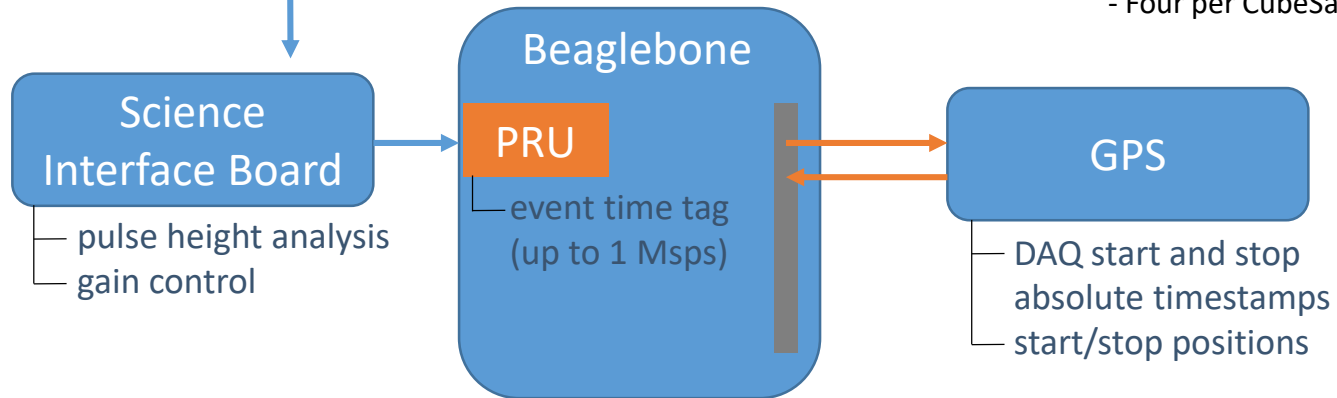
203 x 184 x 71mm



- Hamamatsu Si Photomultiplier (SiPM) Array
- Eight arrays per CubeSat (240 SiPMs)

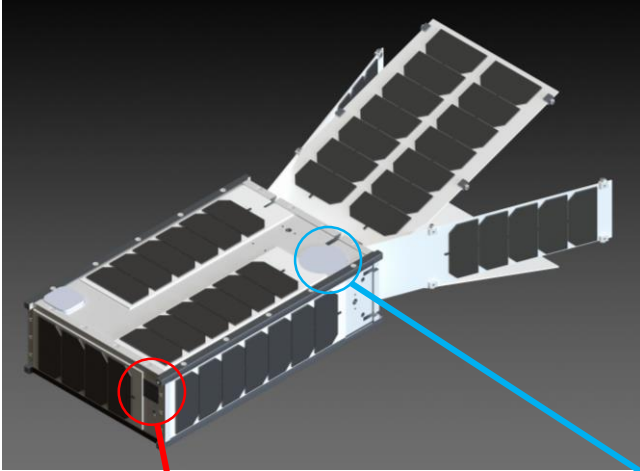


- 5% Lead-doped Plastic (40 x 51 x 166mm)
- Four per CubeSat



*Timing Accuracy Goals: 2  $\mu$ s relative event time tag, 20  $\mu$ s between CubeSats, 200  $\mu$ s w.r.t. ground-based VLF detection*

# Communications

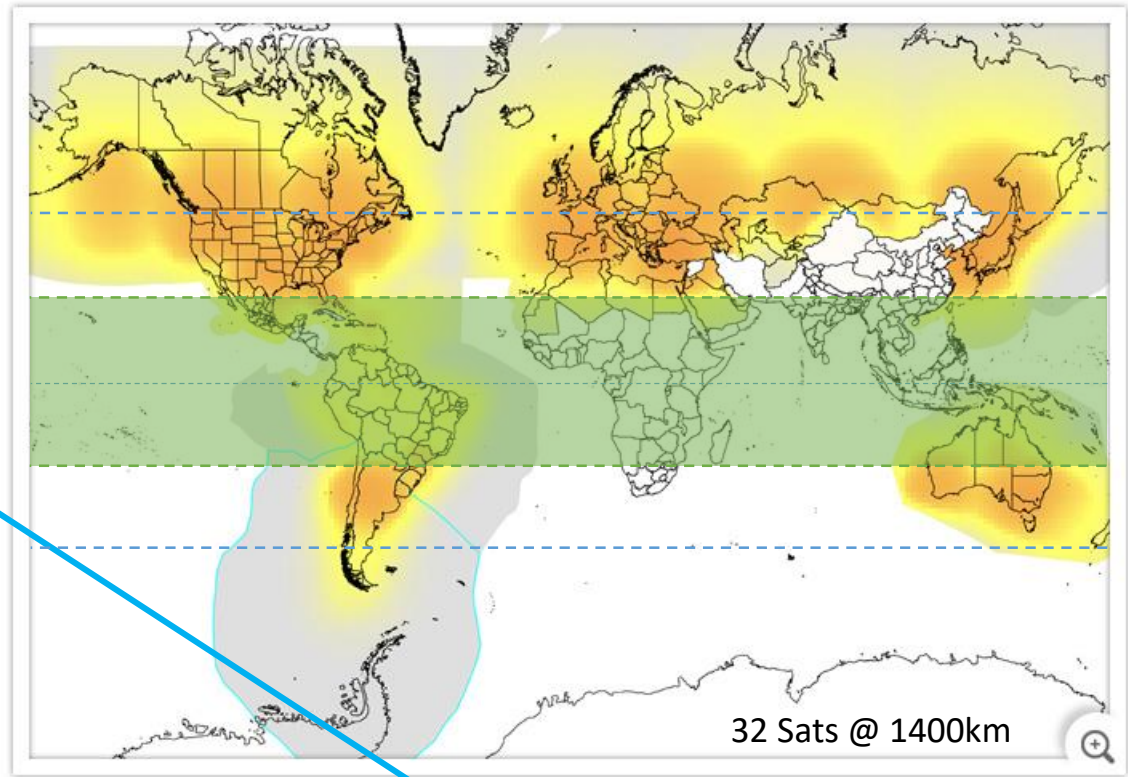


*\*Radios provided by sci\_Zone, Inc*

## Simplex Radio

- Beacon for post-launch phase
- 80-90% orbital coverage
- Abridged telemetry
- Broadcast mode (quasi-roll resistant)

# Globalstar Duplex Coverage

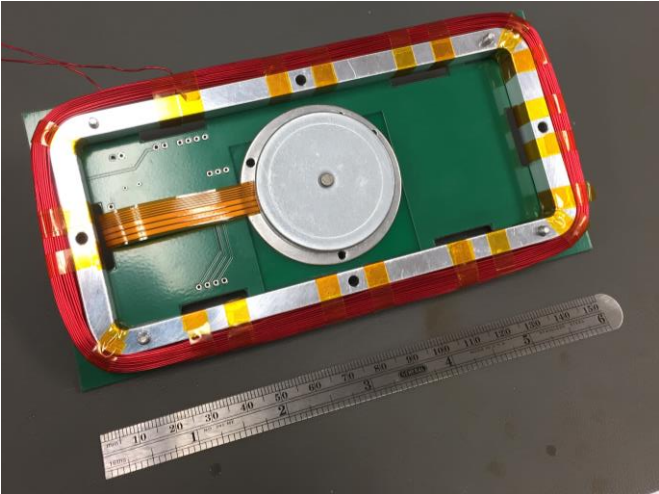
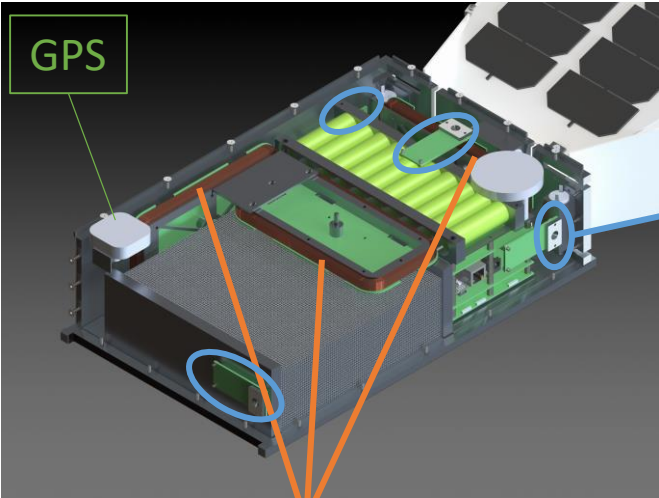


## Duplex Radio

- 256 kb/s, full duplex
- 40-50% orbital coverage
- Science Data / Telemetry
- Command & Control
- Requires +/- 40° zenith antenna pointing

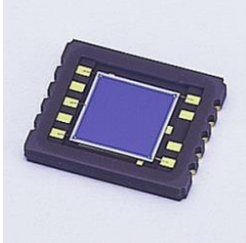
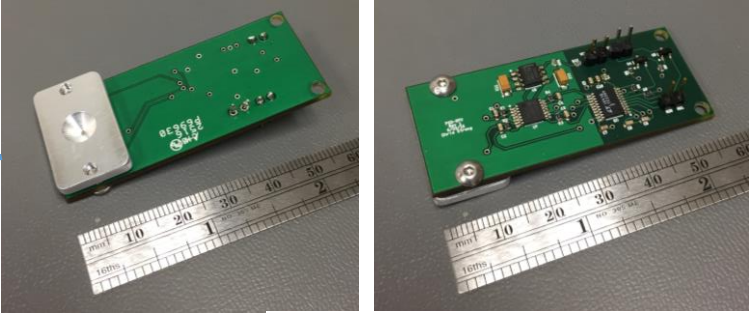


# Attitude Determination and Control System (ADACS)

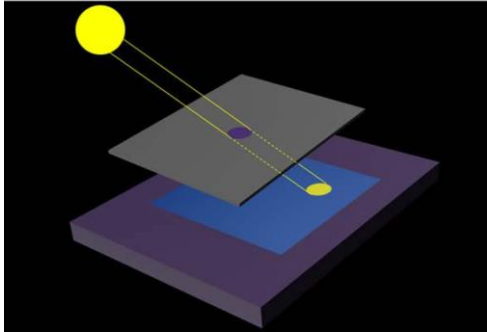


Reaction Unit: Magnetorquer + Reaction Wheel

Sun Angle Sensor



Hamamatsu PSD



**Magnetorquer**

180 x 60 x 5 mm  
 $\mu = 0.91 \text{ Am}^2$

**Reaction Wheel**

45mm brushless DC  
 8,000 rpm



Magnetometer

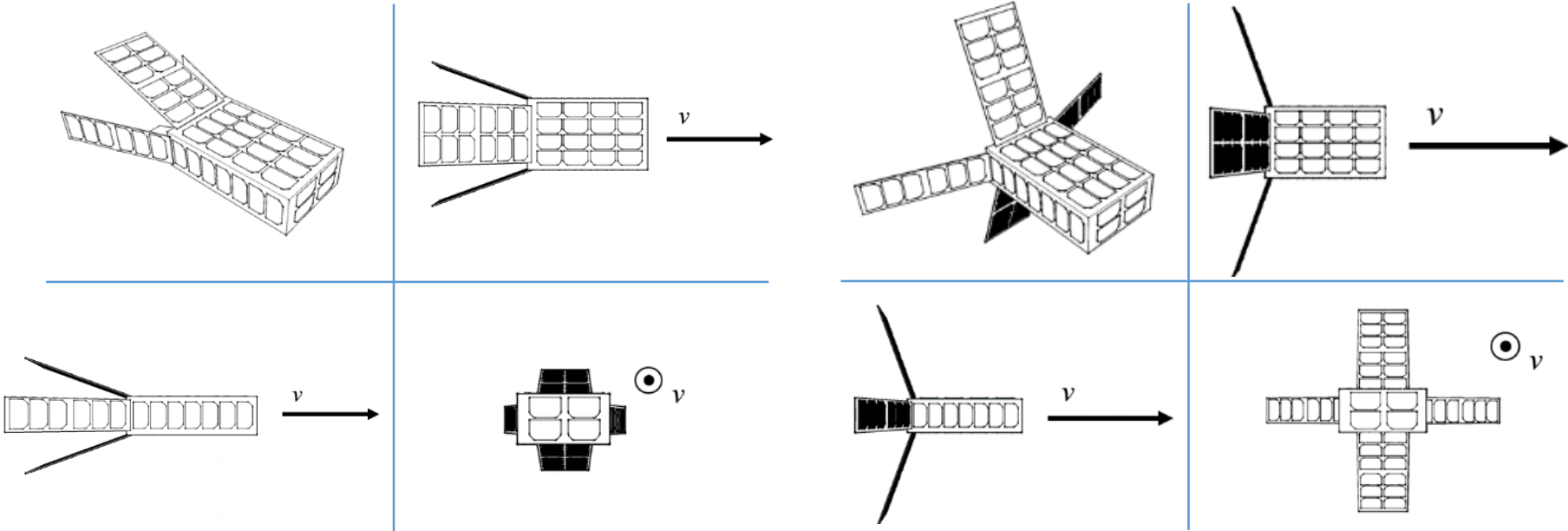


Rate gyro

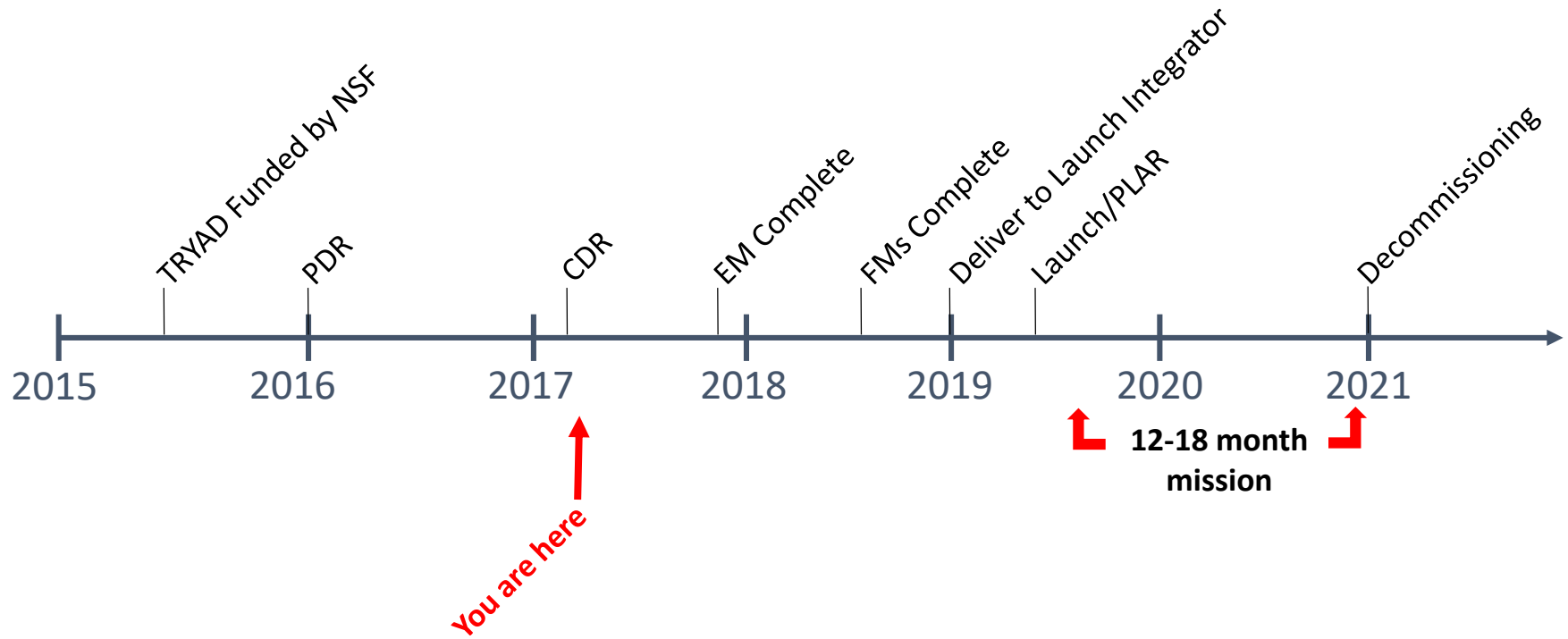
# Station Keeping via Differential Drag

## Min Drag

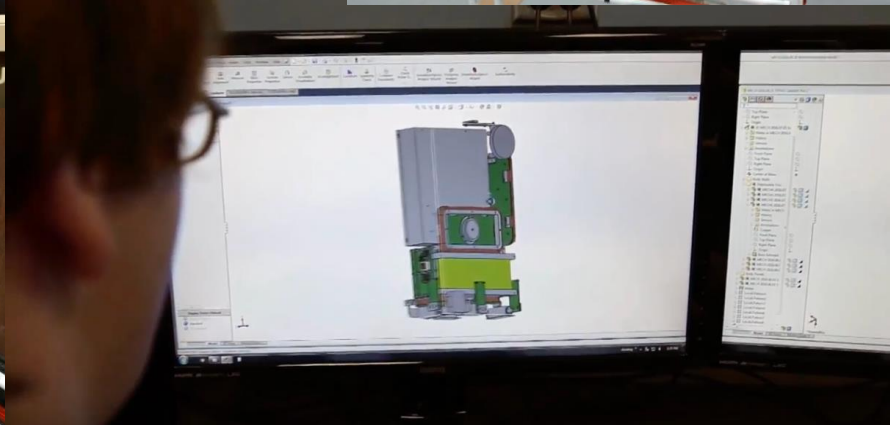
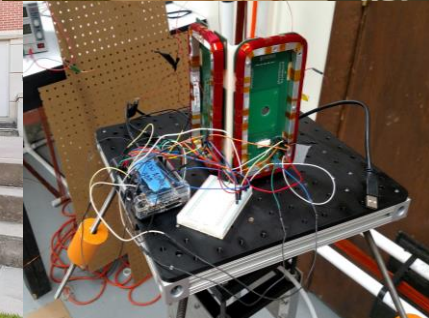
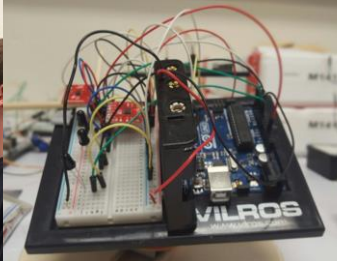
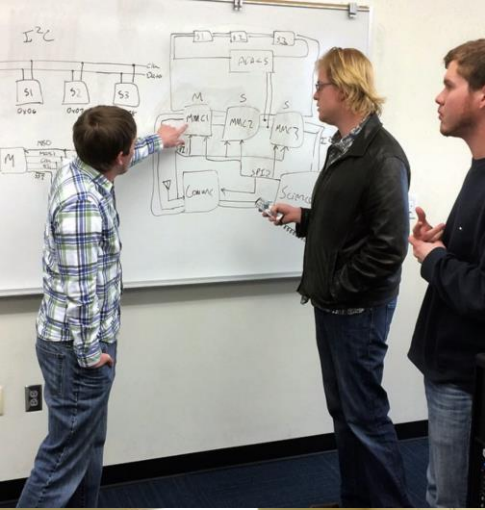
## Max Drag



# TRYAD Timeline







# Thank You!



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