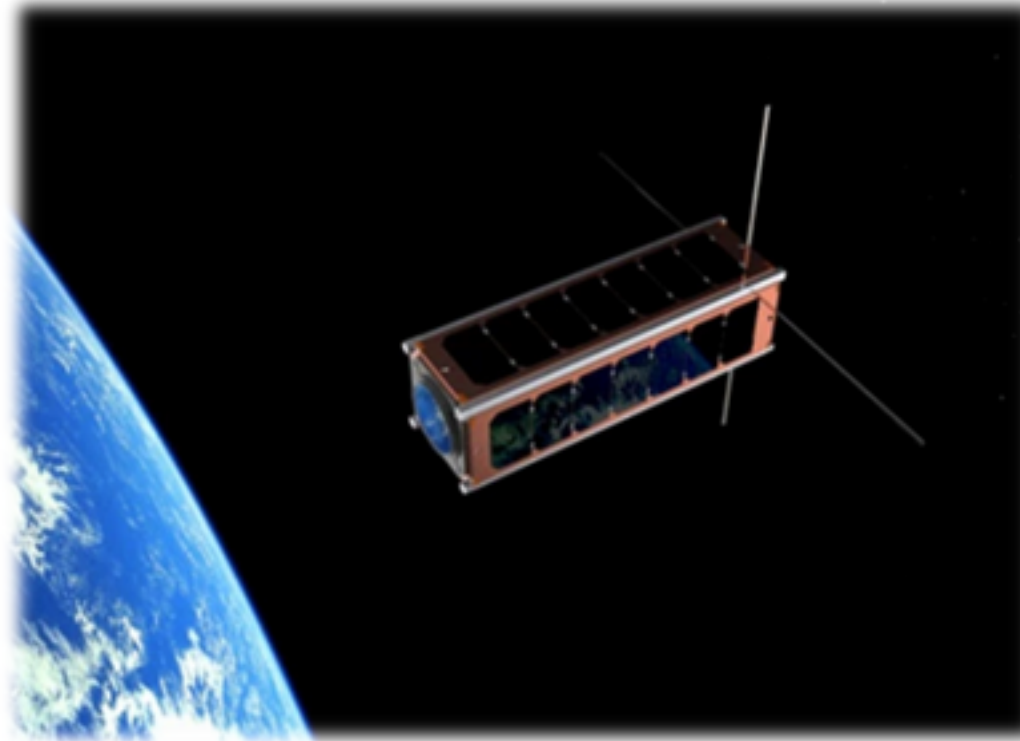


# CubeSat Functionality and Microgravity Testing Platform



*UKube-1 Amateur Radio CubeSat.* Digital image. AMSATUK. N.p., 26 Sept. 2012. Web. 13 Apr. 2016. <<https://amsat-uk.org/2012/09/26/uks-amateur-radio-cubesat-books-a-ride-on-russian-soyuz-2-rocket/>>



# Agenda

- Team Introduction
- Purpose
- Platform Design
- Free Fall Environment
- Benefits
- Scalability
- Additional Applications
- Conclusion



# Team Introduction

- **Alex Crook** – Project Manager
  - B.S., Mechanical Engineering
- **Adam Block** – Team Member
  - B.S., Mechanical Engineering
  - B.S., Energy Systems Engineering
- **Dr. Kevin Kilty** – Faculty PI
  - Adjunct Professor UWYO
- **John Wickman** – Professional Adviser
  - CEO of Wickman Spacecraft & Propulsion Company



# Purpose

- Risk reduction
- Increase TRL
- Overflow testing
- Microgravity testing



# Platform Design

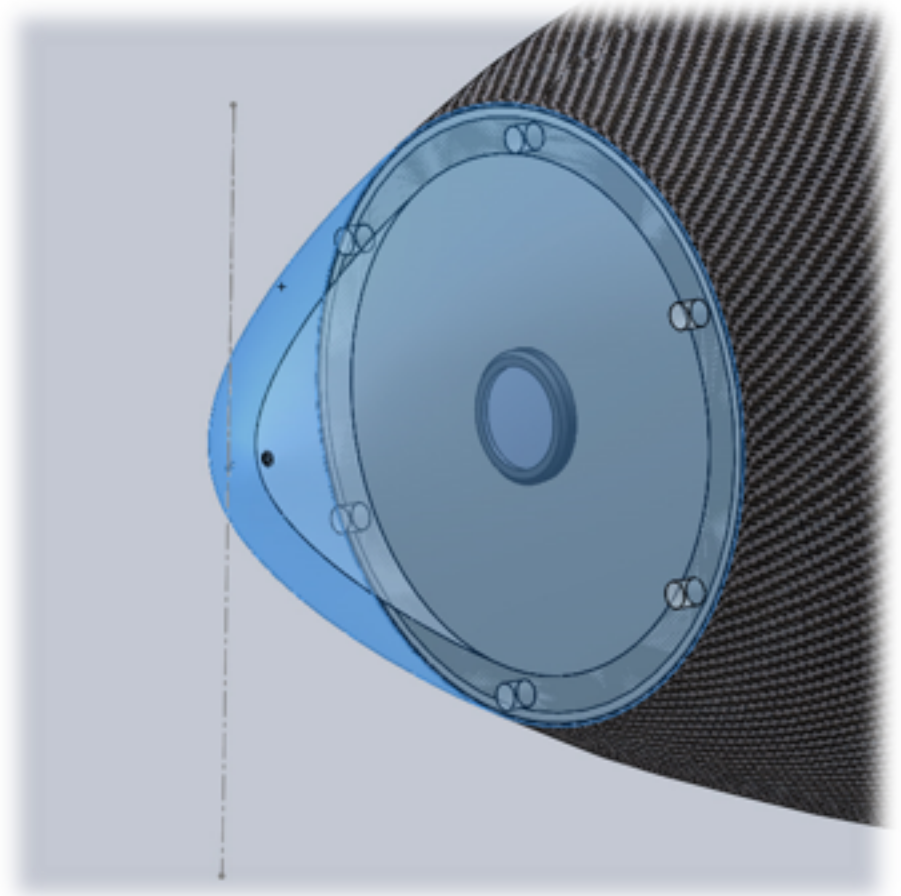


- Drop methodology
  - Weather balloon
  - 30,480 m (100,000 ft.) AGL
- 20 seconds free fall
- Data telemetry during flight
- On board video capability



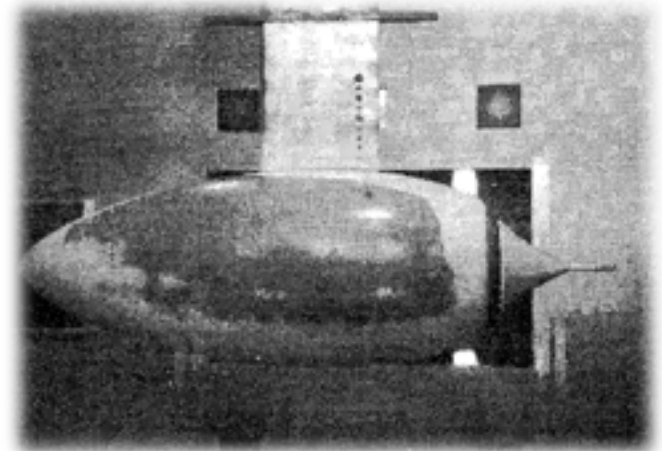
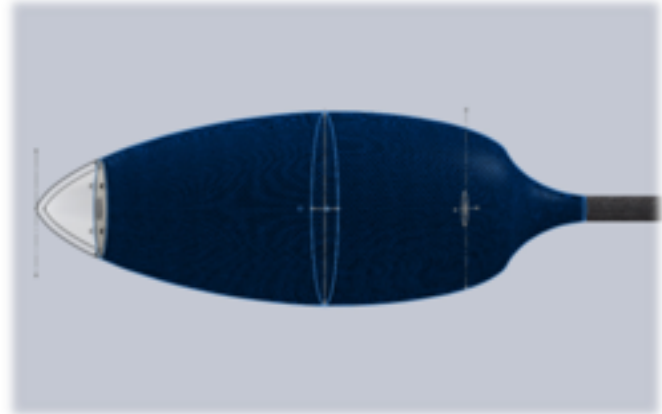
# Nose Cone

- Transparent cast acrylic accommodates video camera
- Transparent to radio frequency



# Aerodynamic Body

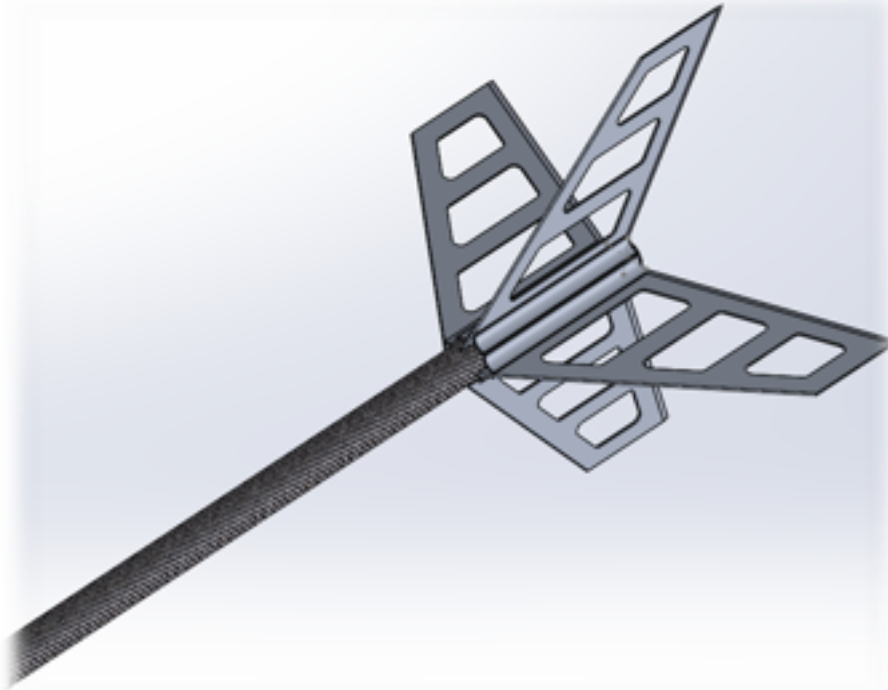
- Light, rigid, durable frame (carbon fiber)
- Proven, aerodynamic shape
- Low coefficient of drag
- Minimal buffeting and vibration
- Constrained to 12 lb. by FAA
- Scalable up to 12U CubeSat with FAA waiver



Goldschmied, F. Aerodynamic Hull Design for HASPA LTA Optimization. Vol 15, No. 9. 1978



# Boom



- Carbon fiber tube
- Low mass
  - Keeps center of gravity forward
- Length of 0.61 m (2 ft.)
  - Keeps center of pressure behind center of gravity
- Stabilizes flight





# Fin Design

- Tapered, swept design to further push back the center of drag
- Addition of slots prevents lateral acceleration and mass reduction
- Moderate airfoil shape



# Parachute

- 1.83 m (6 ft.) panel parachute
- Geometry reduces shock
- 25.31 km/h (15.73 mph) decent rate for 6.80 kg (15 lb.) payload
- Single-stage parachute deployment



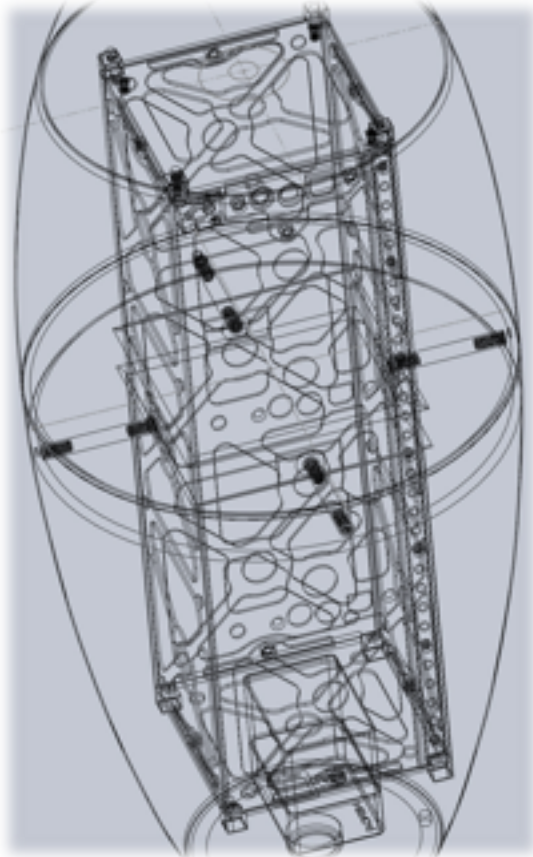
Engelgau, Gene. *Panel Parachute*. Digital image. *Types of Parachutes*. N.p., n.d. Web. <[https://fruitychutes.com/uav\\_rpv\\_drone\\_recovery\\_parachutes/uas-parachute-recovery-tutorial.htm](https://fruitychutes.com/uav_rpv_drone_recovery_parachutes/uas-parachute-recovery-tutorial.htm)>.



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# Experiment Bay



- 3000 cm<sup>3</sup> (183.1 in<sup>3</sup>) of 3U

CubeSat housing volume

– Scalable

- 1000 cm<sup>3</sup> (61.0 in<sup>3</sup>) of

microgravity testing volume

– Scalable

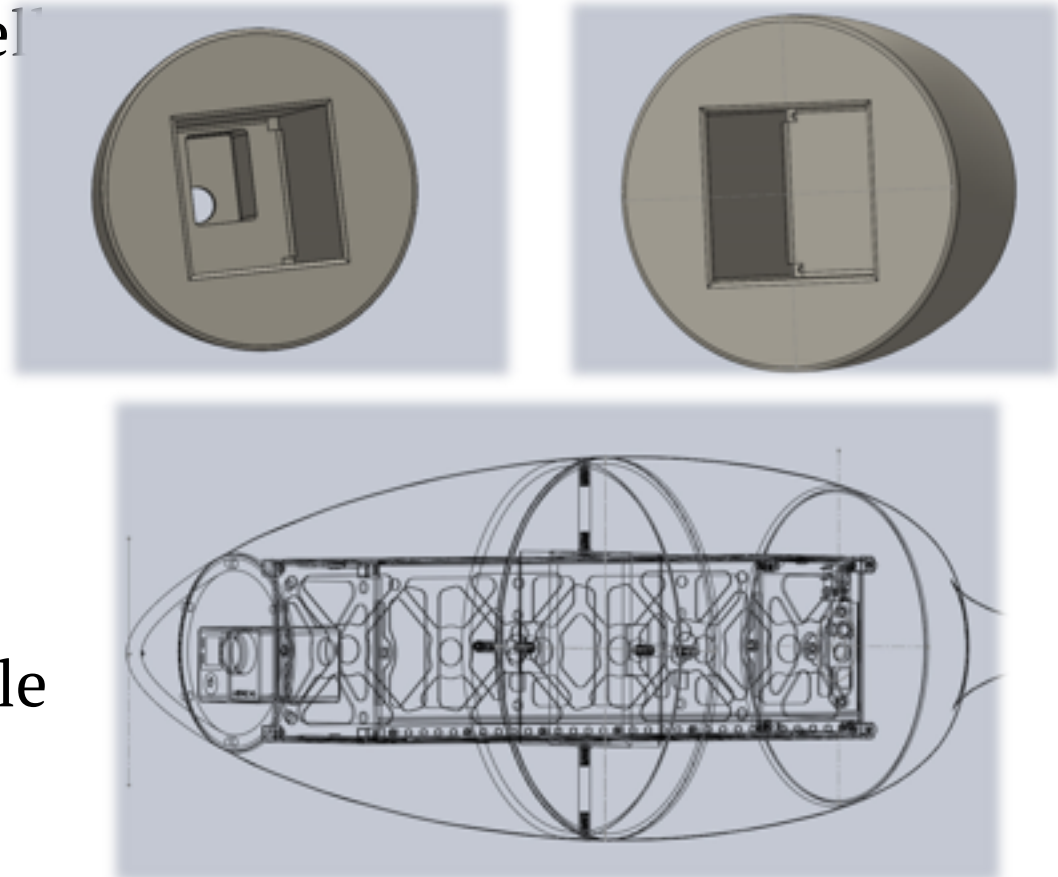


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# Foam Core

- Polyethylene closed-cell foam
- Ideal for shock absorption and vibration
- Durable and machinable
- Low density



Clockwise from top left: Front Form, Center Form, Wireframe Zoom



# Data Acquisition

- Up to 500k samples per second
- 256 KB programmable memory
- Ability to telemeter CubeSat data to ground station
- 16 analog channels
- 85 I/O pins



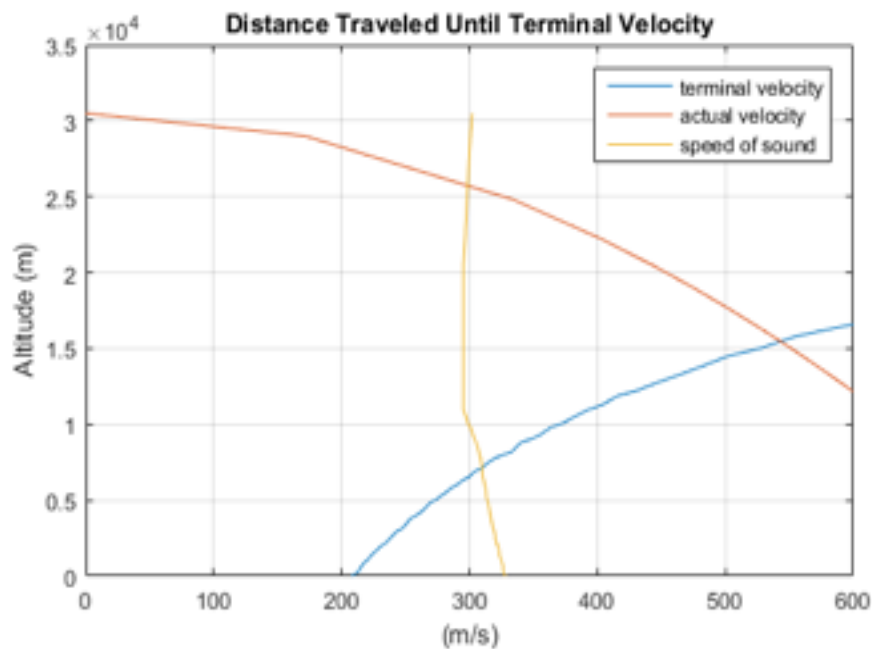
Hart, D. Nodes. Digital image. *CubeSats undergo final inspection at NASA's Ames Research Center in Moffett Field, California.* [www.nasa.gov/press-release](http://www.nasa.gov/press-release). 2015



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# Free Fall Environment

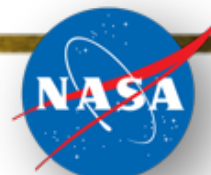


- 30 seconds until sound barrier
- Terminal velocity 540 m/s (1207 mph)
- Absolute pressure at altitude 1116 N/m<sup>2</sup> (0.162 lb/in<sup>2</sup>)
- Density at altitude 0.017 kg/m<sup>3</sup> (0.33 x 10<sup>-4</sup> slugs/ft<sup>3</sup>)
- Temperatures as low as -46.64 °C (-51.10 °F)

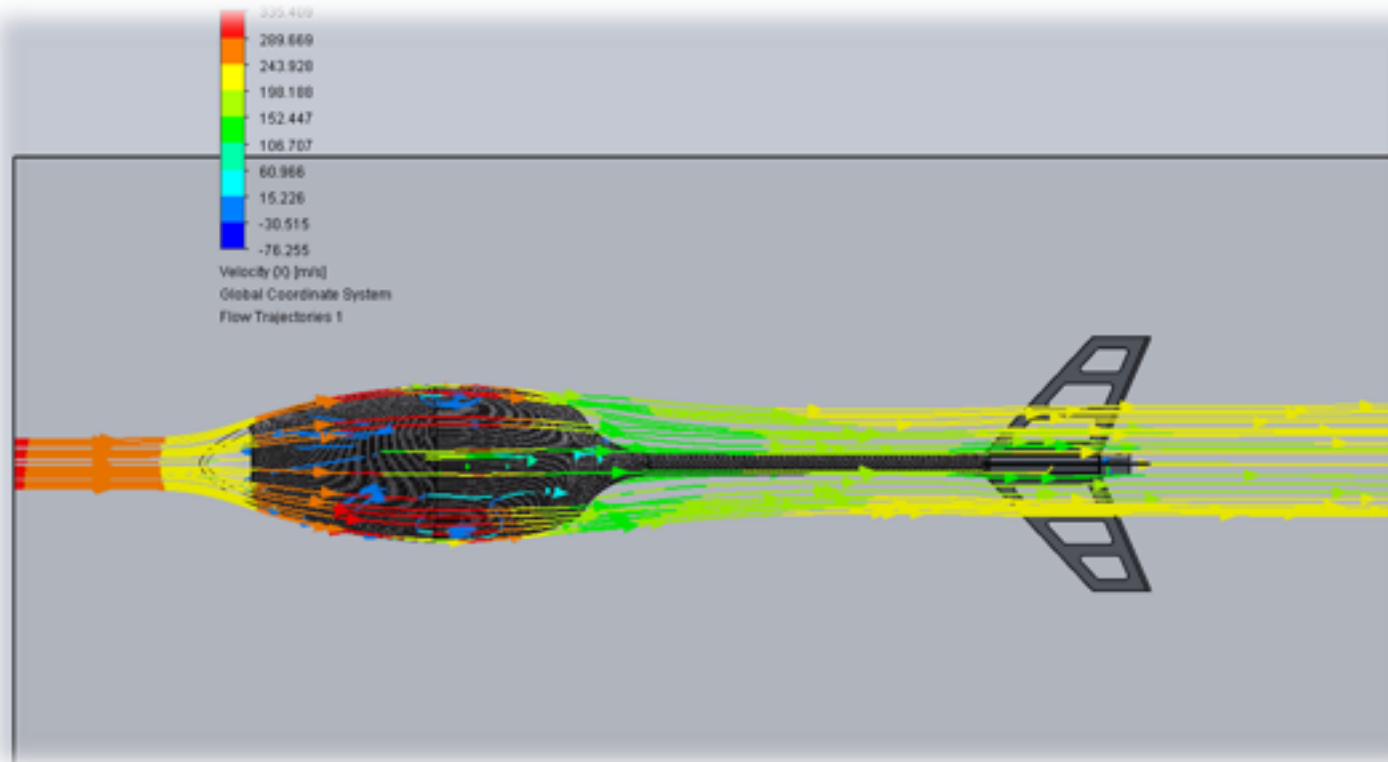
Predicted aerodynamic events during free fall



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# Free Fall Environment



Flow trajectory of flow at maximum velocity, max  $C_d=0.02$



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# Benefits

- Increase Technical Readiness Level (TRL) to 6-7
- On demand testing capabilities
- Opening doors to CubeSat microgravity testing
- Overflow functionality testing
  - Further develop requirements

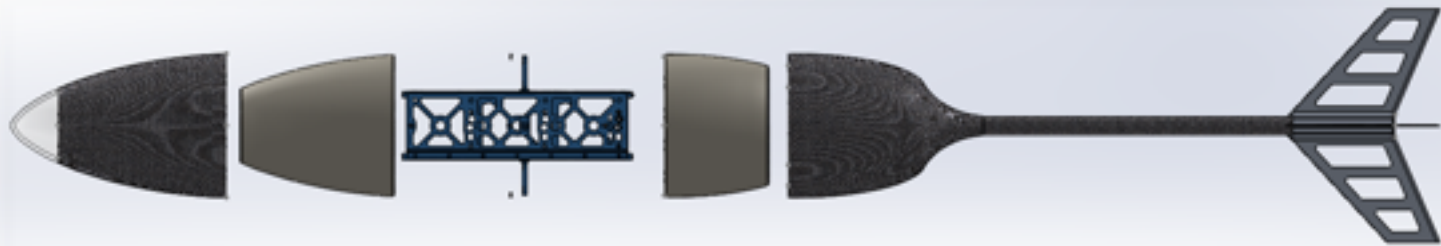
<b>TRL 1</b>	Basic principles observed and reported
<b>TRL 2</b>	Technology concept and/or application invented
<b>TRL 3</b>	Active research and development is initiated
<b>TRL 4</b>	Component or breadboard validation in a laboratory environment
<b>TRL 5</b>	Component or breadboard validation in technological relevant environment
<b>TRL 6</b>	System model or prototype demonstrated in a relevant environment
<b>TRL 7</b>	System prototype demonstration in an operational environment
<b>TRL 8</b>	System completed and "flight qualified" through test and demonstration
<b>TRL 9</b>	Actual system "flight proven" through successful mission operations





# Scalability

- Fly up to a 12U CubeSat
- 10,000 cm<sup>3</sup> (610.2 in<sup>3</sup>) microgravity testing volume
- Increase microgravity duration
- Increase velocity
- Launch Service Requirements Document (LSRD)



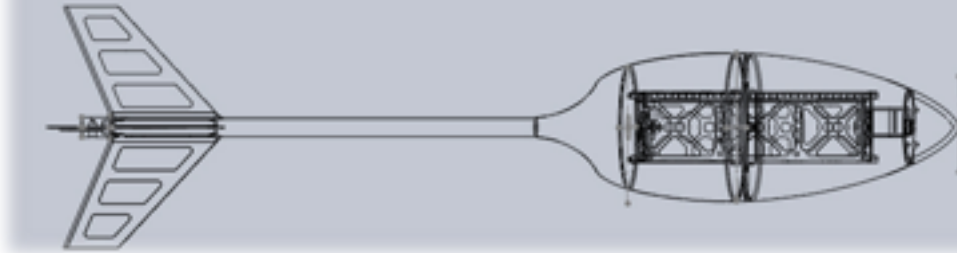
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# Additional Applications



- Fly multiple balloons
  - CubeSats communicate during flight
- Flying in formation
  - Situational awareness
- Fly to a laser designator
  - Movable fins





Q & A

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