

# A Failure Analysis of the ExoCube CubSat

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# Background

- To characterize Hydrogen, Helium, Nitrogen and Oxygen, ions and neutrals in the Exosphere.
- Also measure total ion density above ground stations, and Incoherent Scatter Radar (ISR) stations.
- Collaboration between Scientific Solutions, University of Wisconsin in Madison, the University of Illinois, Cal Poly and Funded by the NSF.
- Mass Spectrometer built by NASA Goddard
- Launched aboard a Delta II on the NASA SMAP mission on January 31, 2015.







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Establish a set of failure modes

- Develop a fishbone and investigation plan
- Coordinate a plan among personnel to implement root cause analysis
- Establish a battle rhythm and communications protocol (e.g.: tag-ups, telecons)



# **Failure Modes**



Antenna did not deploy

One of gravity gradient booms did not deploy

Radio stopped working 6 months after start of mission







Anomaly Description

Description: RF antenna did not deploy

### Evidence:

Gain insufficient without 150' diameter SRI dish or NASA Wallops facilities to talk to satellite - using larger dish, satellite behaved nominally.

List and timeline of relevant on-orbit events :

Date: 2/1/2014 Observation: Weak signal from ExoCube







Fishbone Diagram



### Anomaly #1: RF Antenna Deploy *Pictures of Hardware*



Matching Board And Burn Wire 0

Antenna Route







Fault Response of the Burn-circuit IC

Fault line - Active Low Fault Occurs When: Overcurrent - Over 928.4mA Over temperature - Over 135°C Reverse Voltage – Over 190mV Prevents false reports with an internal delay deglitch circuit

Telemetry Output - First number is the number of times a burn was stopped, and the second is number of times a burn was initiated.





#### Telemetry Evidence



- Antenna deployment draws approximately 1 A of current
- However, no large current draw events seen in the first month

### Current did not flow when commanded





Fishbone Diagram



Likely that burn wire electrical path broke (burn wire itself or wires leading to it)



*Evidence for Bone Closure* 

### **Operations Failure**

Battery Charge State

Closed by showing that battery voltage was in the nominal range during deployment times.

### **Analog Failure**

Heating Wire Burned Out

Possible contender since no current was flowing through the burn wire.

Wiring to Device is Broken

Possible contender vibrations could have disconnected the wire leading to the burn element.





Evidence for Bone Closure

### Commanding

### Command not Received, or not Executed

Closed by viewing telemetry showing total number of completed burn commands.

### Environmental

Temperature Outside of Operational Ranges

Deemed not probable since command executed and satellites temperature was not out of nominal range

### Radiation Dose

Closed considering EXOCUBE was only in orbit a few days before deployment was commanded, hard to test.

### **Assembly Workmanship**

Device wiring failed due to launch.

Possible that wire to heating element solder joints could have broken.





Anomaly Description

<u>Description</u>: One out of two booms did not deploy

#### <u>Evidence</u>:

Boom magnetometer readings did not align with spacecraft navigation magnetometer readings

#### List and timeline of relevant on-orbit events :

Date: July 20th 2015 Date: July 27th 2015 Observation: Extra Current draw indicating burn Observation: Mag boom data indicates one deployed





Magnetometer readings



Pre: Y, B1, B2 (Dark Blue, Red, Green respectively) Values do not line up. Post: Y, B1, B2 (Violet, Light Blue, Orange respectively) Values of Post Y and B1 line up.

Post Y and B2 do not line up. B1 deployed and B2 did not.

#### On boom did not deploy



### Anomaly #2: Magnetometer Boom Deploy *Pictures of hardware*





### Process for installing burn wire has challenges





Fault Response of the Burn-circuit IC)

Fault line - Active Low Fault Occurs When: Overcurrent – Set by an external resistor - Over 450mA Over temperature - Over 135°C Reverse Voltage - Over 190mV

Prevents false reports with an internal delay deglitch circuit.

No Faults Were detected





- Approximately 1100 mA indicates both burn wires active.
- Data from day of boom deployment July 20th.

Commanded current was realized





Telemetry evidence

fuelGaugeOne volt: 4.040588 V current: -0.297989 A currentAccum: 2.026993 A 4.040588 V volt: -0.528000 A current: currentAccum: 2.026993 A 4.035690 V volt: current: -0.528000 A currentAccum: 2.026993 A volt: 4.035690 V -0.528000 A current: currentAccum: 2.026993 A 4.035690 V volt: -0.728989 A current: currentAccum: 2.025986 A 4.035690 V volt: -0.728989 A current:

- About 430 mA current per boom.
- Differences can be explained with the error associated with the switch error of approximately +/- 40mA.

Commanded current was realized





Fishbone Diagram



Conclude insufficient contact between burn wire and restraint filament



Evidence for Bone Closure

### **Operations Failure**

Battery Charge State

Closed by showing that battery voltage was in the nominal range during deployment times.

### **Analog Failure**

Heating Wire Burned Out

Closed by showing current flow through the burn circuit. *Wiring to Device is Broken* 

Closed by showing current flow through the burn circuit.





Evidence for Bone Closure

### Commanding

### Command not Received, or not Executed

Closed since current telemetry shows commands executed successfully

#### Environmental

Temperature Outside of Operational Ranges

Closed by showing the circuit works and the current flow was nominal. *Radiation Dose* 

Closed by showing the circuit works and the current flow was nominal

#### **Assembly Workmanship**

Device wiring failed due to launch.

Closed by showing current flow was nominal.

Burn Wire did not Make Good Contact with Fishing Line

Deemed the most probable since in lab testing shows that burn wire was hot, but did not cut the fishing line..





Anomaly Description

Description: Radio transmissions ceased

#### Evidence:

Nothing has been heard from satellite since 6 months after mission start Noticeable degradation of performance 1 week prior to silence

### List and timeline of relevant on-orbit events:

Date: 7/30/2014	Observation: RF Temperature Sensor Reset
Date: 7/30/2014	Observation: Poor Radio Performance



### Telemetry evidence

Overview:

- Radio amplifier saw a large amount of reflections.
- Large amount of power dissipated in amplifier.







Fishbone



Suspect RF amplifier failed due to reflected RF energy from undeployed antenna



Evidence for Bone Closure

#### **Operations Failure**

Battery Charge State

Closed by showing that battery voltage was in the nominal range during deployment times.

#### Analog Failure

Reflections From Radio

Possible contender because the antenna did not deploy which causes RF reflections.

Overcurrent

Possible, subset of RF reflections. Reflections could increase the voltage reflected which would drive current back through the amplifier.

#### Environmental

*Temperature Outside of Operational Ranges* 

Possible contender since a mismatch of the antenna causes more current to be used.

Radiation Dose

Closed because it can not be easily tested.





## Lesson's Learned

Design for testability and repeatability NiChrome needed to be reset after every test

Design more reliable systems

- Technically
- Procedurally

"Failure is only the opportunity to begin again, only this time more wisely." - Henry Ford





# Questions?

