

TC2M

TIME CAPSULE TO MARS

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Agenda

Mission

Technology

Inspiration

Mission Overview
Why? How? What?

1
Mission Overview



Technology Walkthrough
A deep dive into our systems

2
Technology



Inspiration in Action
Active STEM engagement

3
Inspiration



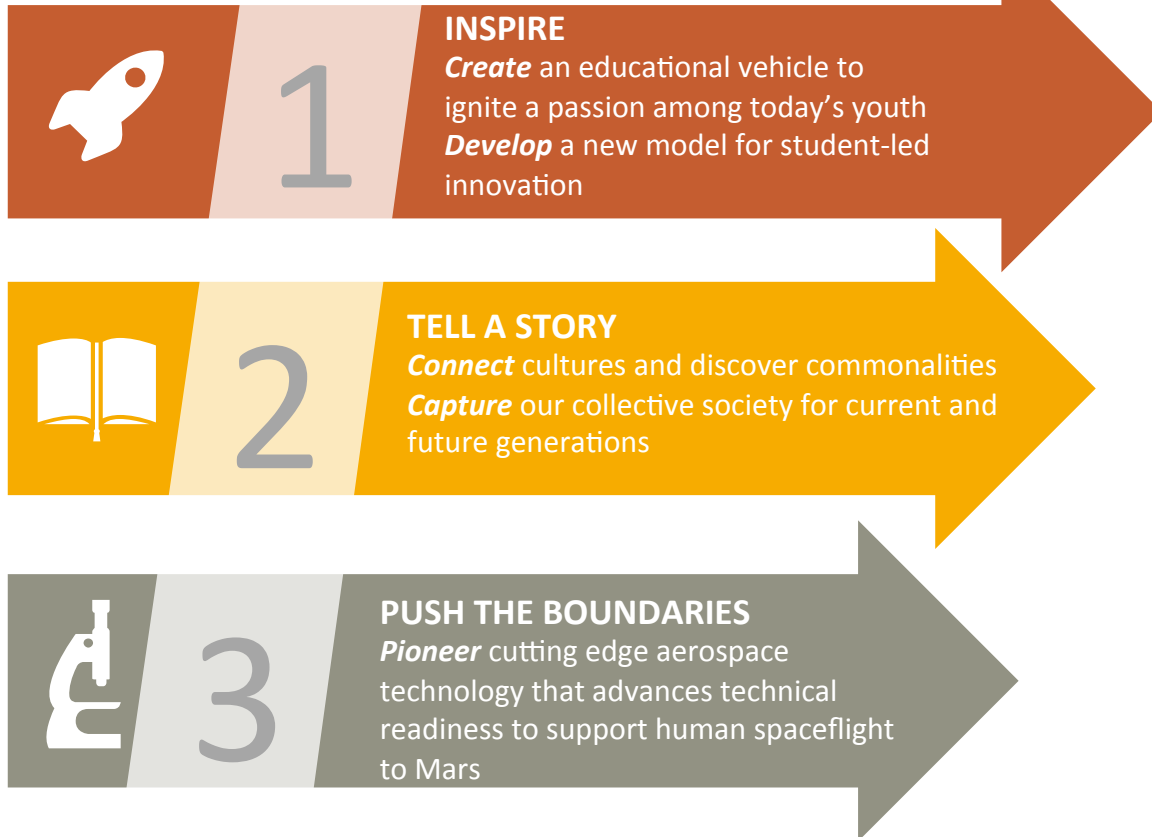
Mission

Technology

Inspiration

MISSION OBJECTIVE Build, launch, and land on the surface of Mars three **student** inspired, **student**-designed, **student**-built, and **student**-funded CubeSat time capsules of humanity

THREE CORE IDEALS



DRIVEN BY THE "WHY"



TC2M in a Nutshell

Mission

Technology

Inspiration

Collect

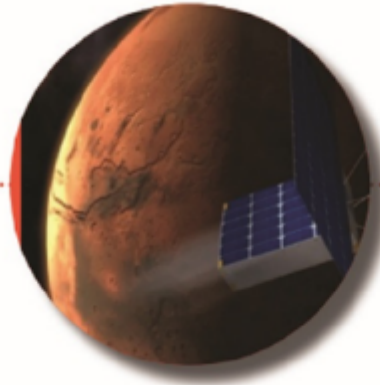
Launch

Arrive

Remain



- Collect **digital content** (text, photo, and video) from across the globe
- Utilize **social media** and STEM partners to reach vast audiences



- **3 CubeSat spacecraft** will thrust to Mars using **ion-electrospray propulsion**
- We want to fly **new experiments** crowdsourced through students



- Enter Martian atmosphere and **deploy time capsule payloads**
- **Signal home** that the landing was a success!



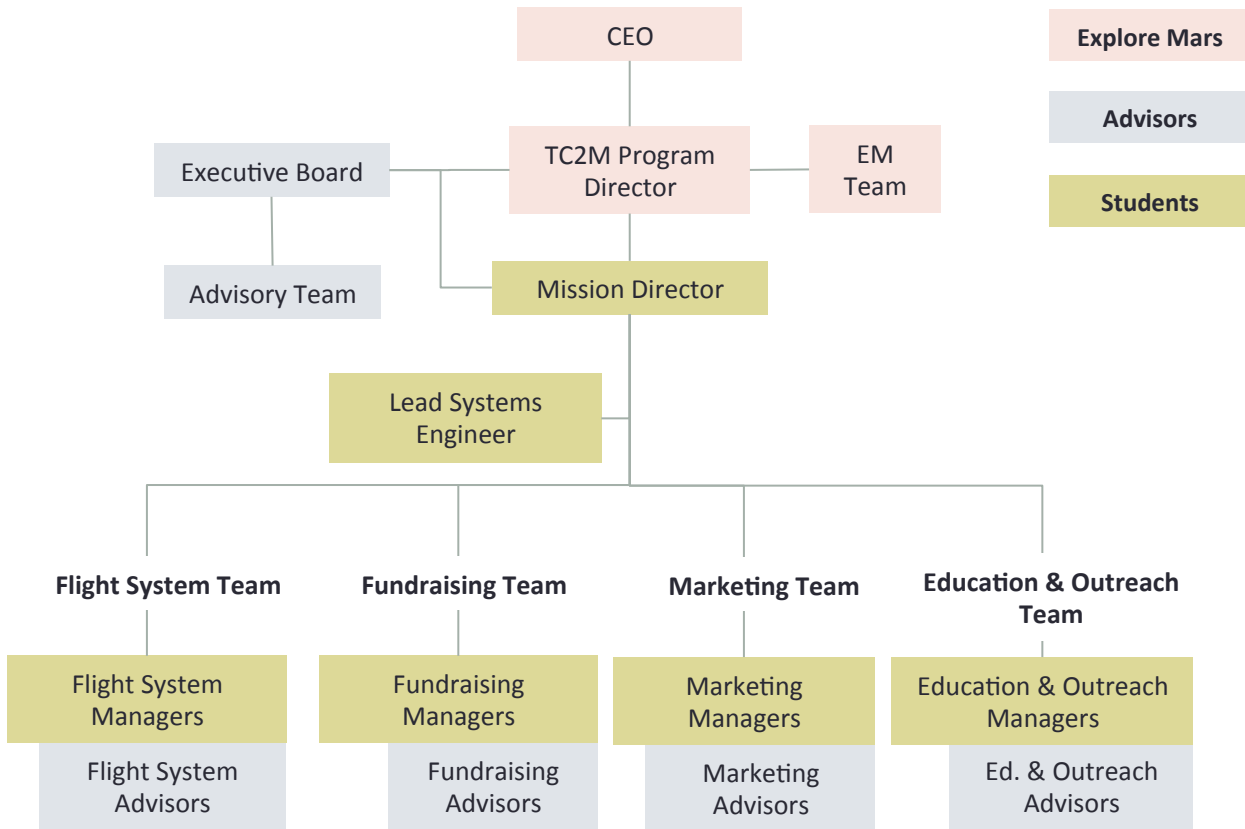
- Payloads land, remain intact for future human **colonists to rediscover**
- Plan to share the same **gallery of digital content** here on Earth to continue to inspire

Mission

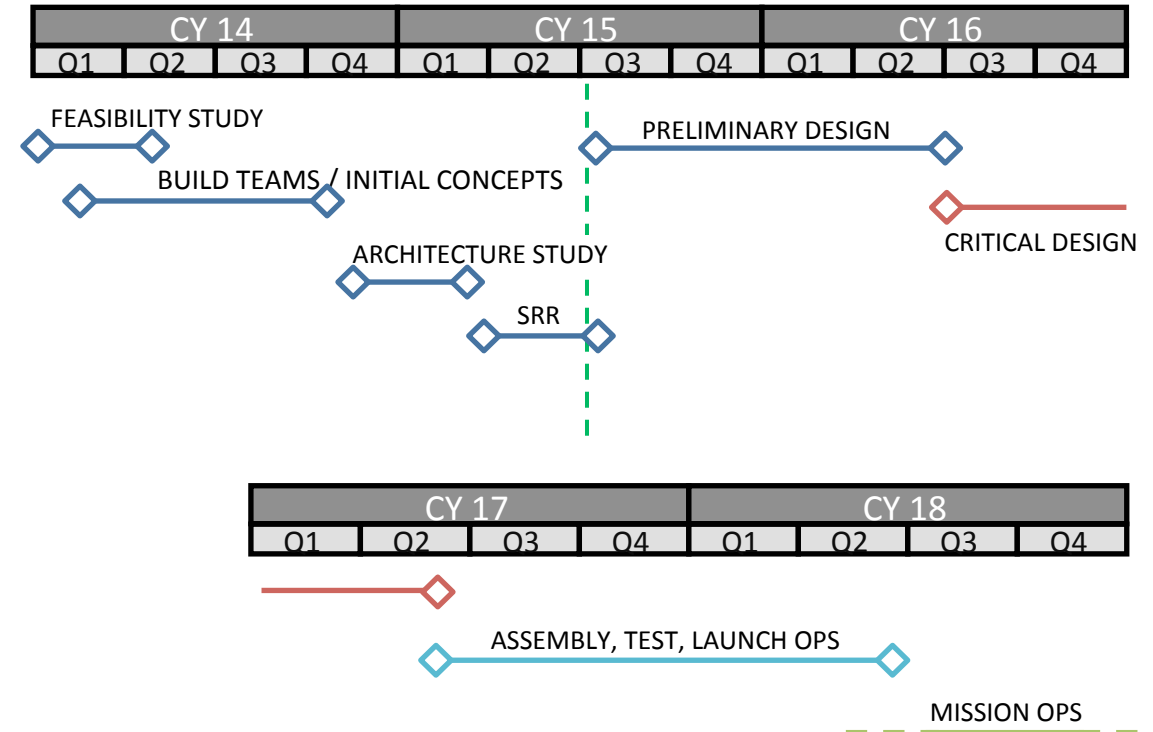
Technology

Inspiration

STRONG ADVISORY STRUCTURE & FEEDBACK LOOP



WHERE WE ARE... WHERE WE'RE GOING



Organizing for Financial Success

Mission

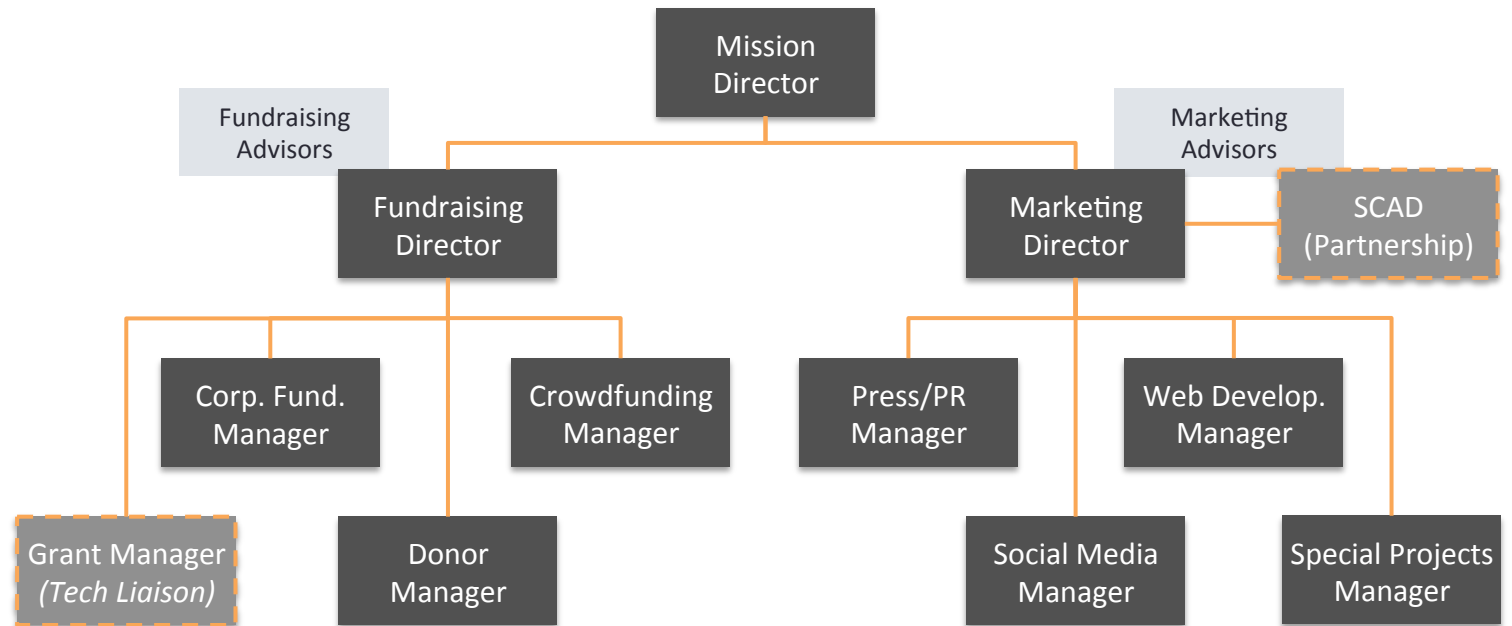
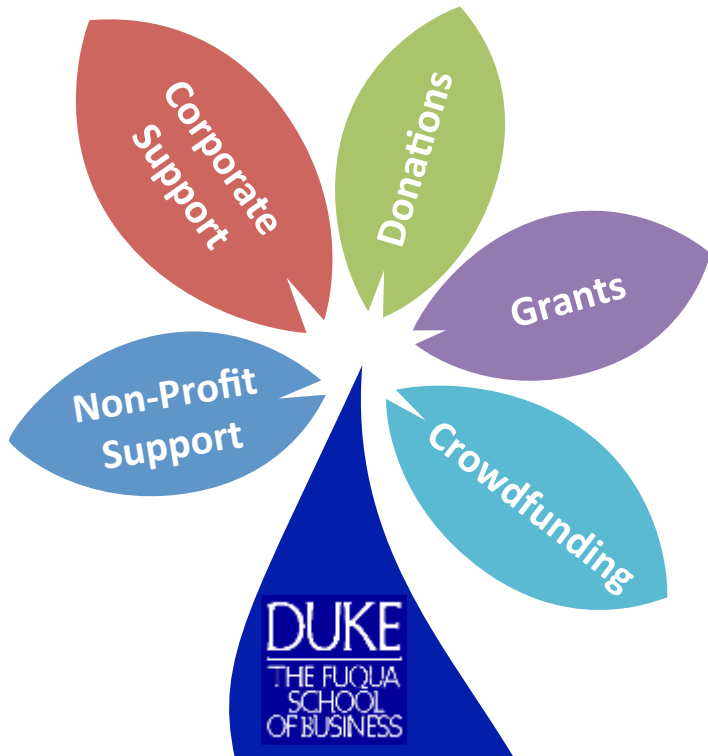
Technology

Inspiration

SOURCES OF SUPPORT



STRONG COMMERCIAL TEAM & PARTNERSHIPS



Strong Partners for Unique System Requirements

Mission

Technology

Inspiration



AVIONICS

Engineer memory that is single event upset and latch up tolerant

Develop method to capture images of objects in and on the spacecraft during flight



GNC PROPULSION

Create propulsion system capable of producing 5.5 km/s ΔV within our limited spacecraft size

Develop semi-autonomous guidance, navigation, and control scheme



TEST ASSEMBLY LAUNCH OPS

Abide by all planetary protection requirements throughout the duration of the mission

Develop a system that is compatible with multiple launch vehicles and trajectories



ENTRY DESCENT LANDING

Separate the lander from the main spacecraft without re-contact

Design 1U CubeSat entry, descent, and landing system



LANDER PAYLOAD

Confirm successful landing on Mars

Protect the lander from dust storms and abrasive environments



COMMS MISSION OPS

Optimize trajectory for efficient propellant use while minimizing the time of flight

Develop a CubeSat compatible interplanetary communication system



STRUCTURE MECHANICAL

Engineer a passive vibration isolation system

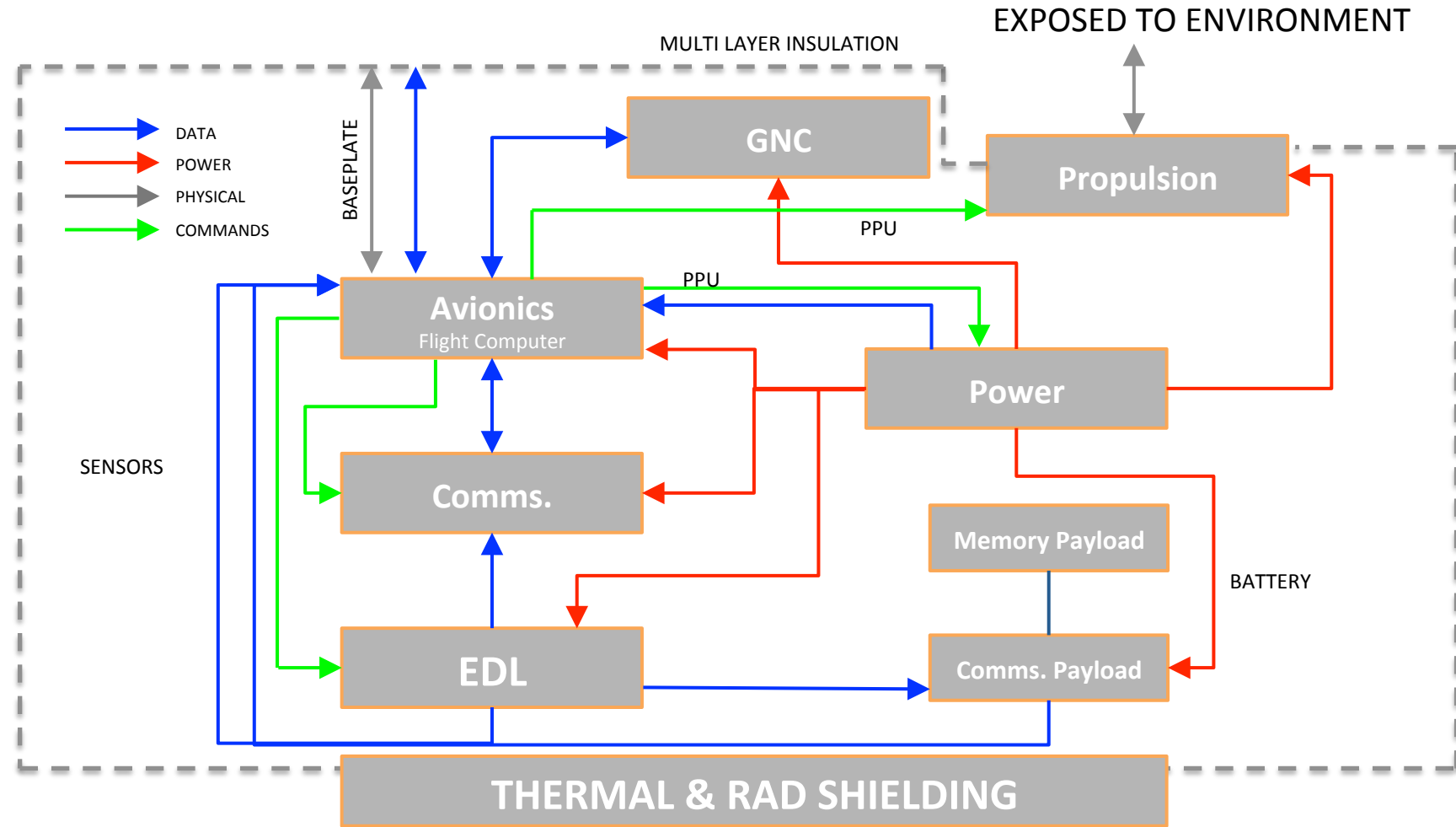
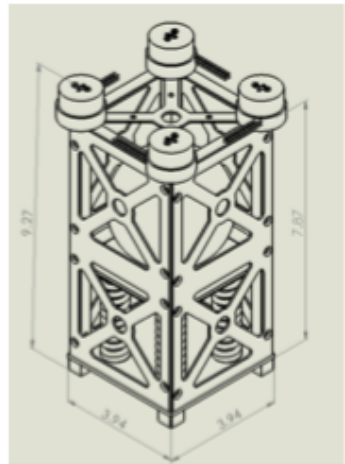
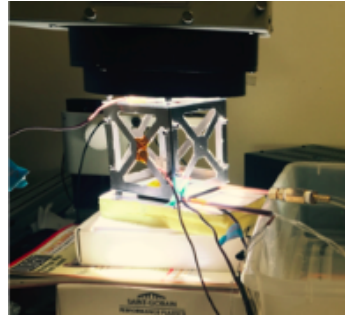
Design thermal control system to maintain temperature of each component within the specified range during all mission phases

System Architecture & Interfaces

Mission

Technology

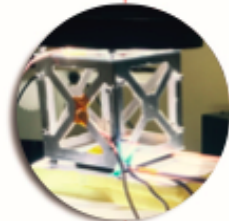
Inspiration





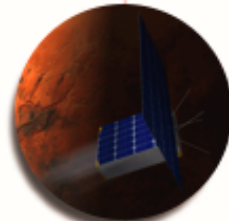
GUIDANCE & NAVIGATION CONTROL

With limited power available for the comms system, and thus limited frequency of communication with Earth, how will we **maintain our intended trajectory?**



PROPULSION

Given the relatively small size of our spacecraft and room for payload how will we meet **the minimum required ΔV to reach mars?**



SURVIVING THE SPACE ENVIRONMENT

How will we **block out high energy solar particles** and background cosmic radiation?
How will we **moderate S/C temp.** to survive a dynamic mission environment?



ENTRY DESCENT & LANDING

Given that Mars has a thin and variable atmosphere with idiosyncratic densities how do we **survive entry and land** in a region of our choosing with relative accuracy?

Mission

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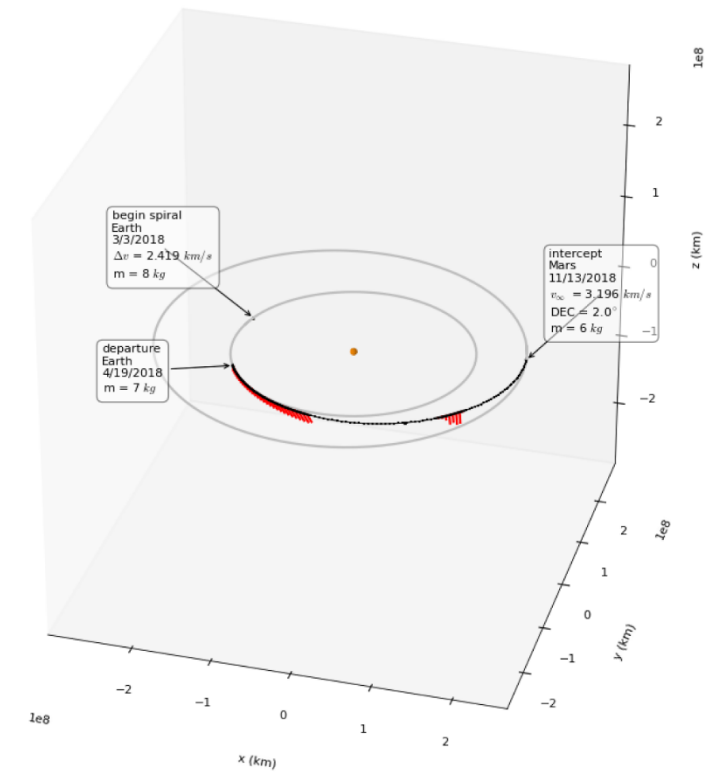
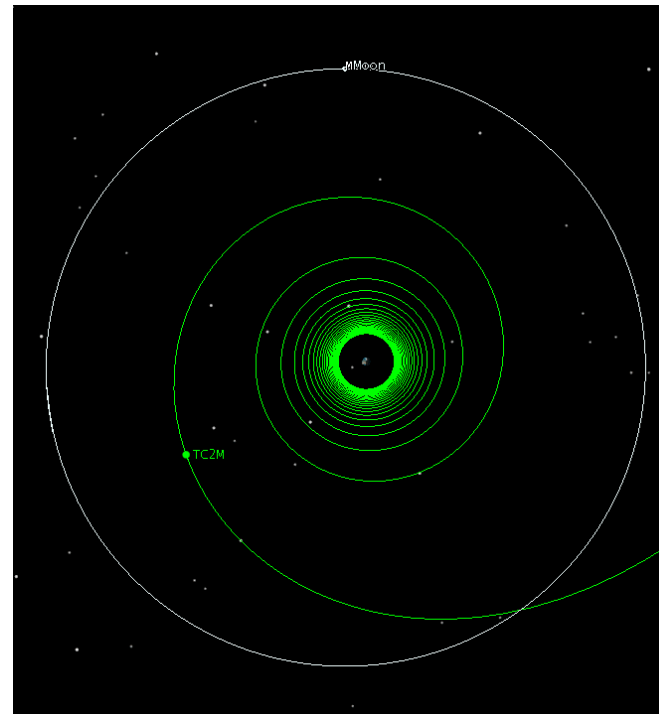
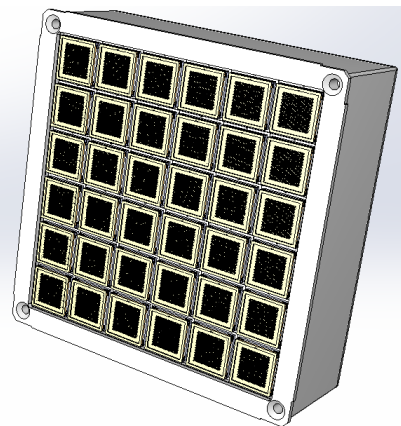
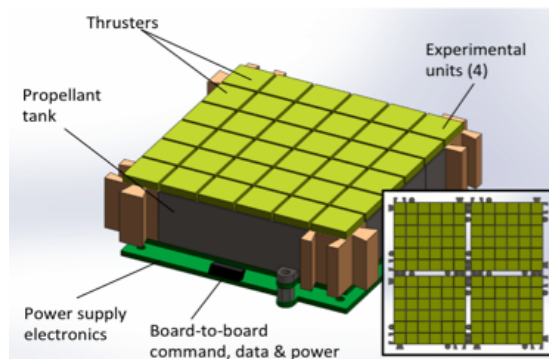
PROPULSION



EARTH SPIRAL OUT



MARS ENTRY

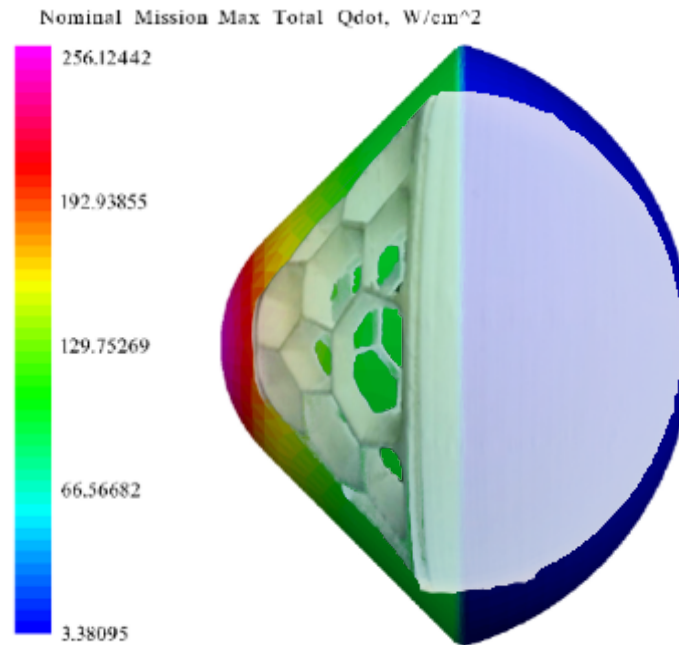


LOW MASS, LOW VOLUME PASSIVE EDL SYSTEM

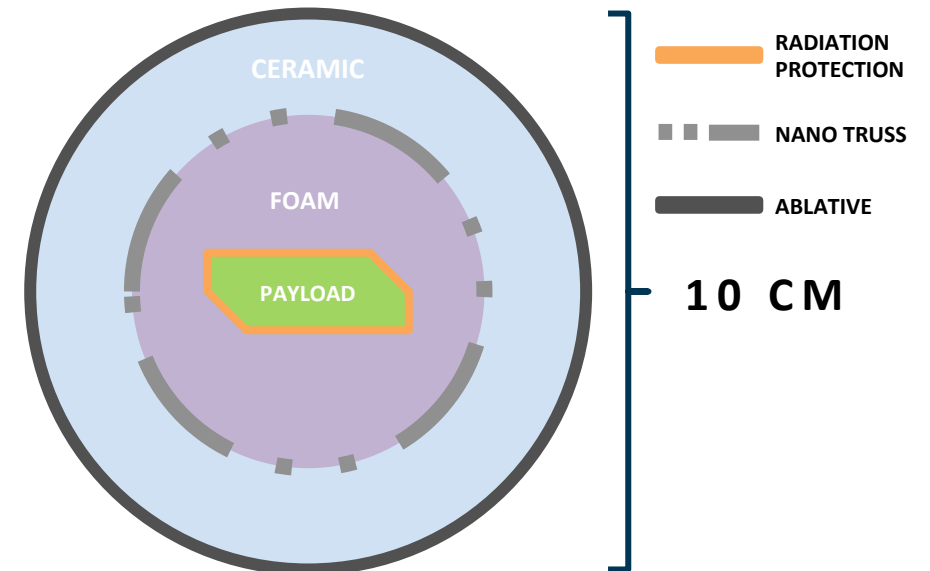
CHALLENGES

1. **Incorporate** EDL system into small volume and mass
2. **Analyze** potential atmospheric entry conditions
3. **Design** long term storage of large memory capacity in a small volume
4. **Ensure** data payload survives the passive descent

EXTERIOR DESIGN



INTERIOR DESIGN



TC2M and Explore Mars are interested in science investigations that advance knowledge needed for future human missions to Mars

POSSIBILITIES

RADIATION MEASUREMENTS

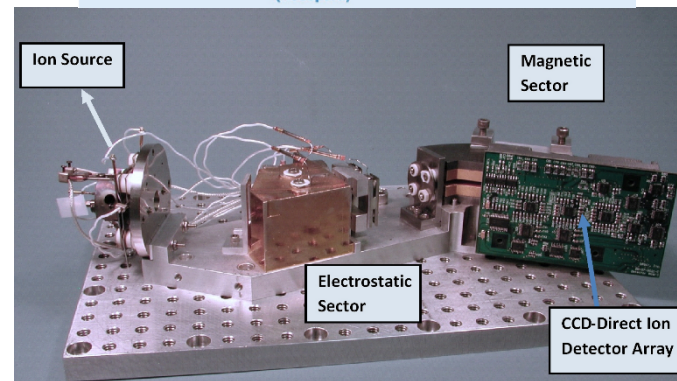
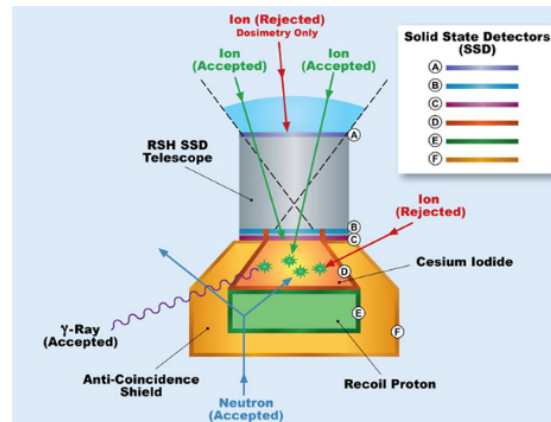
In route between Earth and Mars

On surface measurements timeline, total dose variations, uplink to Mars orbital assets (NASA, ESA)

MARS RESOURCE DETECTION

Spectral imager for precipitation of water and CO₂ in form of snow or frost with daily relay frequency

Electrodynamic dust properties experiments



ASK

CANDIDATE

University or undergraduate students



PROPOSAL

Submit a planned experiment that could fit within the spacecraft's limited resources and return viable data

It's More Than Just Getting to Mars

Mission

Technology

Inspiration

We've been connecting and inspiring schools, students, mentors, and professionals around the country and soon the world



TCEM

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+



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