

Flight of *FRNCS-P*, *QuickSAT/Xen* and
LinkStar
on the
RADSat CubeSat Mission

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Cubesat Workshop

sci_Zone, Inc.



Today's Presentation...

- ✦ The *RADSat* Mission
- ✦ *QuickSAT* - *Xen* and *VMS*
- ✦ The *LinkStar* Radio
- ✦ Roadmap and Next Steps



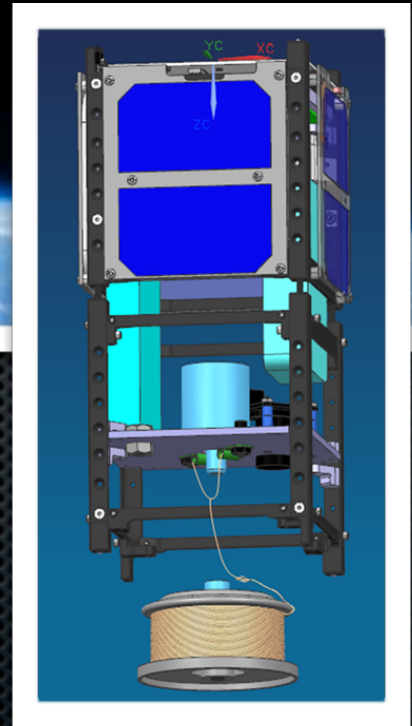
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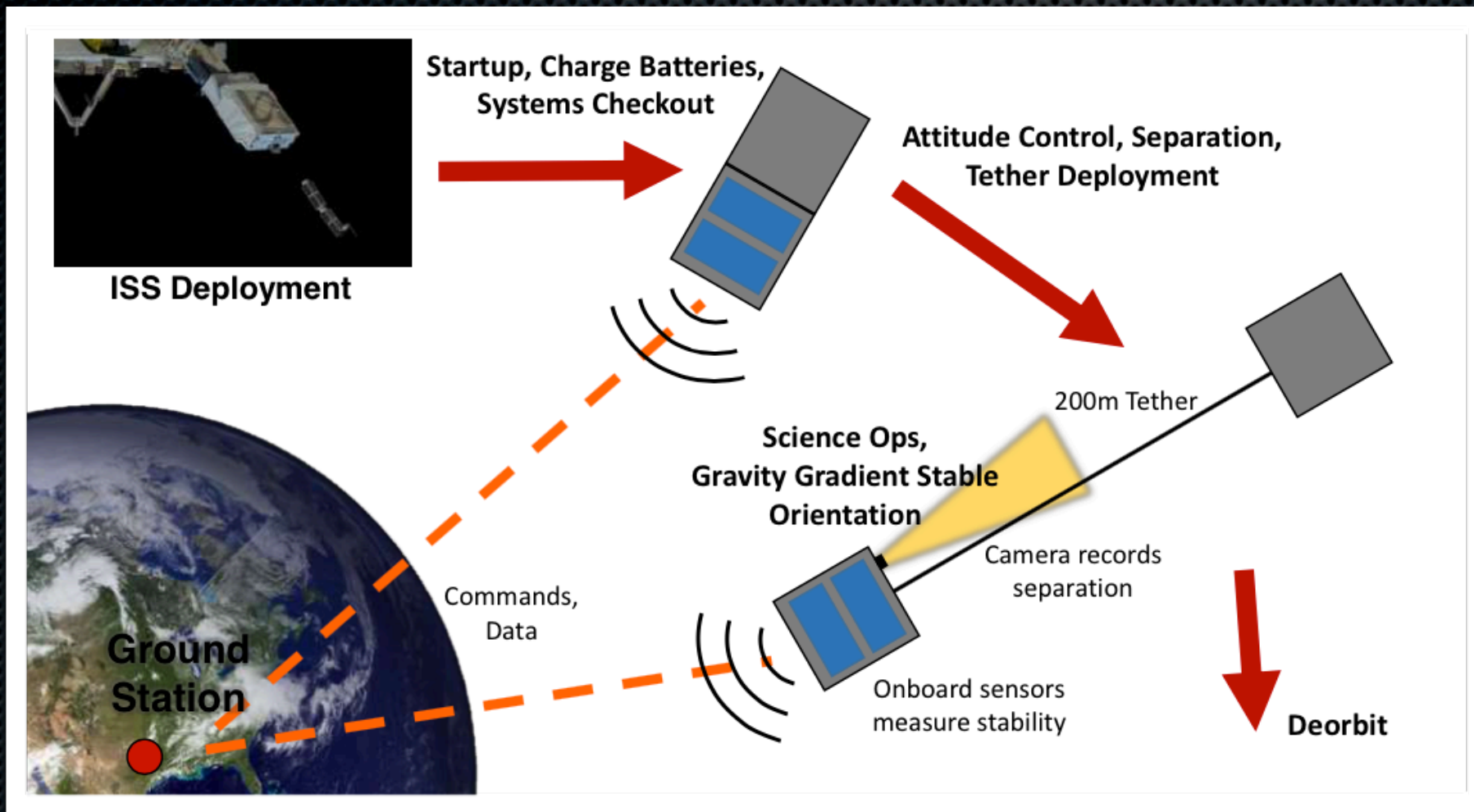
RADSat Mission



- ✦ 2 U CubeSAT
- ✦ Tethered Satellite Mission!
- ✦ Deployed from Nanoracks Platform
- ✦ Launch - Tentatively Late Fall 2015



Mission Concept



Science Objectives

Primary	<ol style="list-style-type: none">1. Characterize the properties of a 2U cubesat in free flight.<ol style="list-style-type: none">1. Power generation / energy demand2. Thermal cycling3. Attitude4. Downlink capability2. Characterize the properties of a 2U cubesat with active attitude control.3. Characterize and demonstrate the deployment of a tether and study the system dynamics.
Secondary	<ol style="list-style-type: none">1. Obtain imagery of separation and tether deployment.2. Activate / operate secondary payload.
Tertiary	<ol style="list-style-type: none">1. Characterize vehicle autonomous operations – nominal operations2. Characterize attitude control and operations – combined vehicle3. Characterize passive attitude control and operations – separated vehicles4. Characterize vehicle autonomous operations – simulated failure operations

Mission Objectives

Primary

1. *Downlink data*
2. *Uplink and execute commands*
3. Separate cubesats
4. Deploy tether to full extent
5. Detumble the combined vehicle
6. Achieve controlled stable attitude (pre separation)
7. Achieve passive stable attitude (post separation)

Secondary

1. Obtain imagery of separation
2. Autonomously identify orbit and propagate vehicle state
3. Activate / operate secondary payload

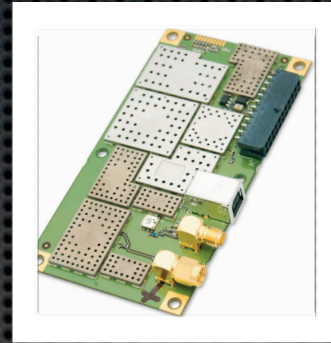
Tertiary

1. Test on-board software autonomous downlink
2. Test attitude control with simulated hardware failure



QuickSAT and LinkStar

- *LinkStar Duplex Radio*
 - *Full Duplex radio*



- *FRNCS-P Flight Computer*
 - *First flight of the BeagleSpace architecture*
 - *Flight Certification of QS/VMS*



- *QuickSAT/Xen Space Hypervisor*
 - *First Space Hypervisor*



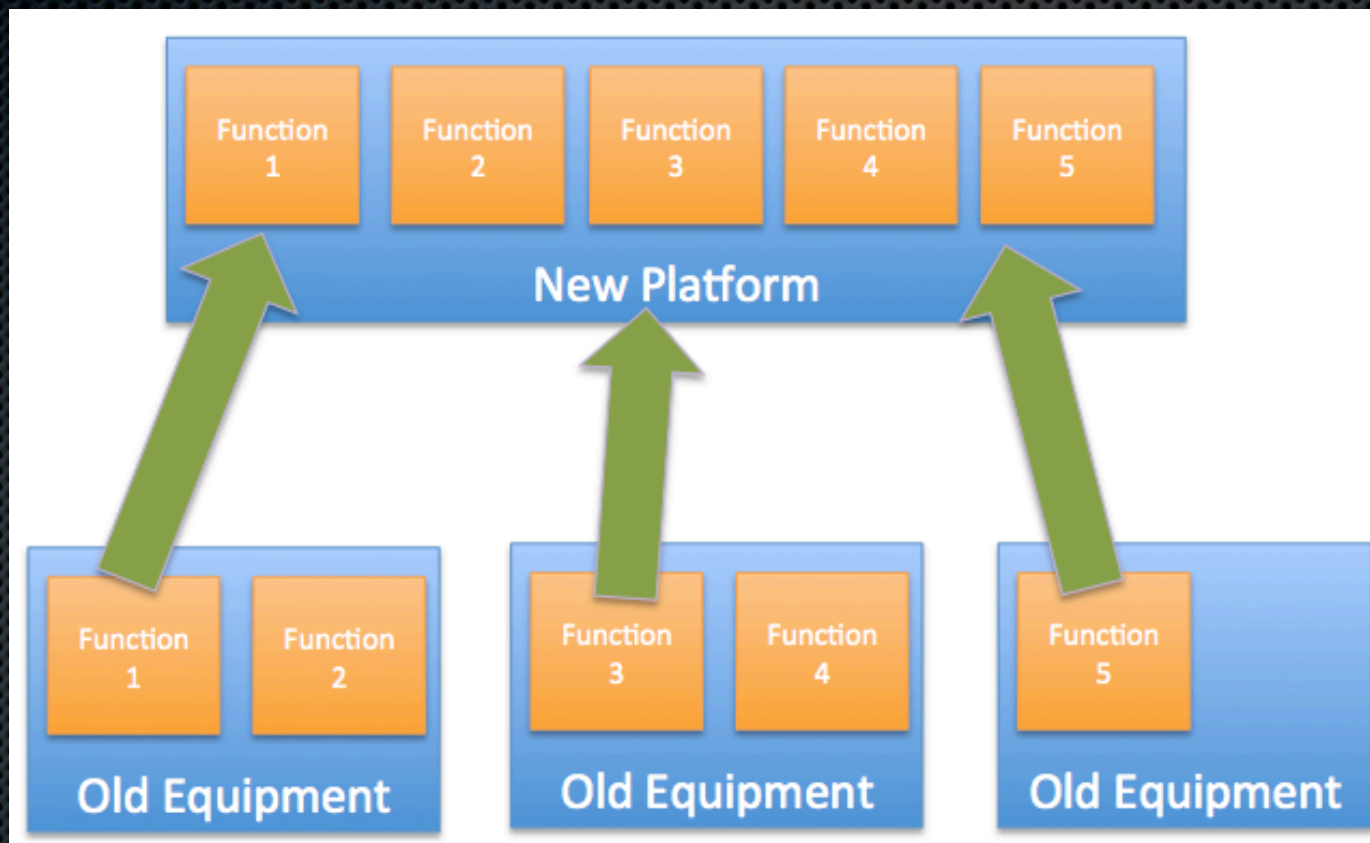
QuickSAT/Xen



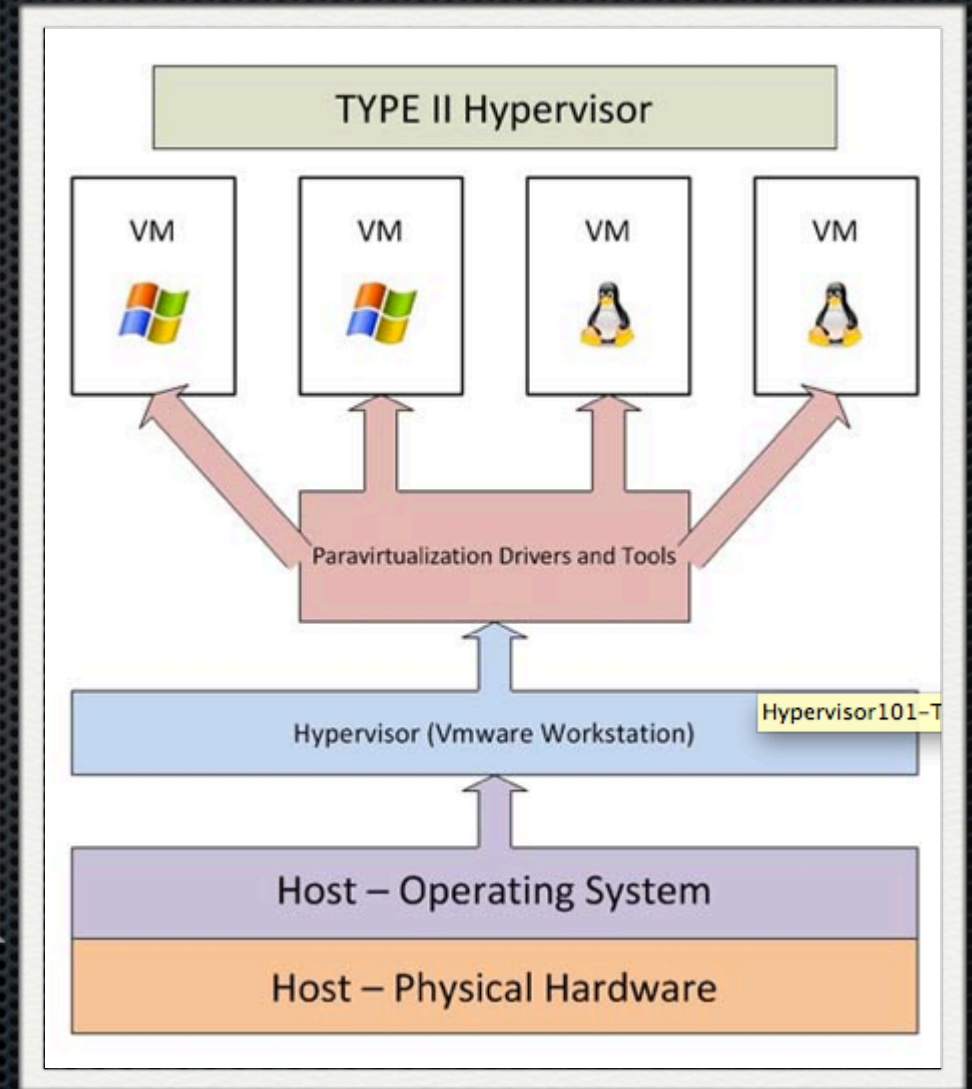
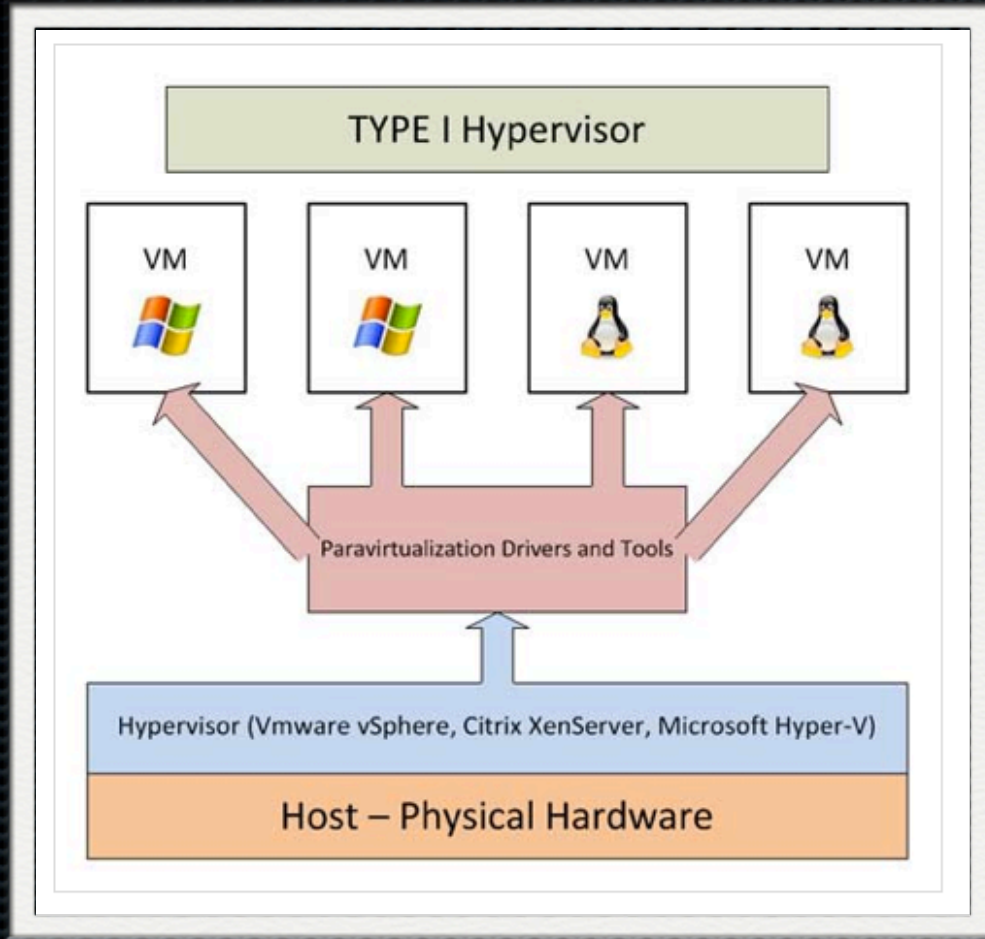
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Sustain Legacy Systems by Migrating Functionality to the Hypervisor



Types of Hypervisors...



QuickSAT/Xen is an adaptation of Xen, a Type I Hypervisor

Benefits

- Security
 - Multiple Independent Levels of Security (MILS EAL6+) by Formal methods mathematical proof of correctness
 - Isolates software into their own partitions, regulates information transfer
- Safety
 - DO-178C Certification with Design Assurance Level A
 - Certification artifacts
- Performance
 - Low overhead virtualization
 - Low latency via ARINC 653 I/O interrupt
 - Multicore
- Integration
 - Efficiently upload and build virtual machines from the lab or ground station
 - Build VMs from QuickSAT/Designer - automatic parameter definition
 - VMs operation based on modes of operation



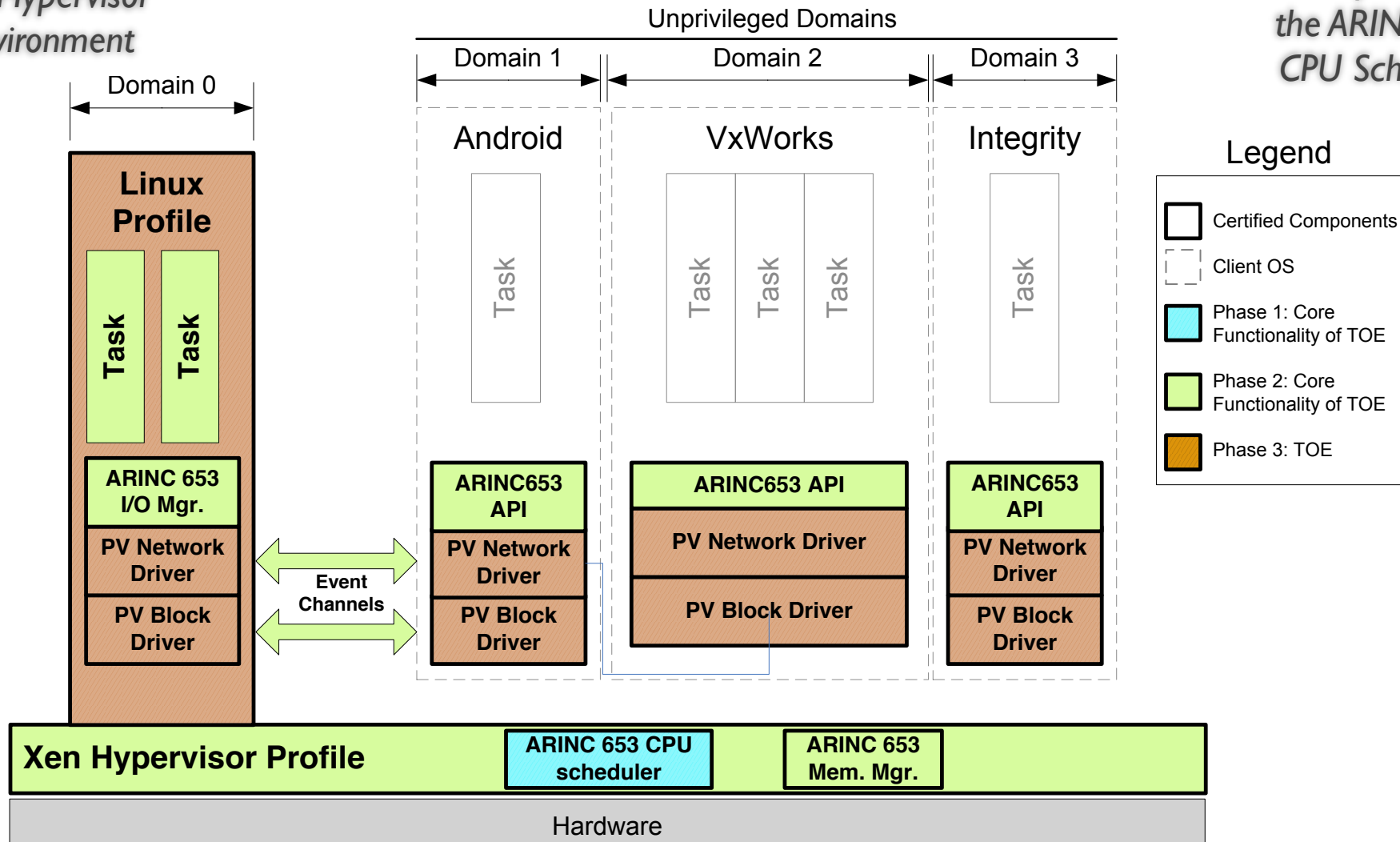
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The Xen Hypervisor Environment

DOM0 Controls
the Hypervisor
Environment

Multiple "Virtual
Machines"
securely linked via
the ARINC 653
CPU Scheduler



QuickSAT/Vehicle Management System (VMS)

Your window into your satellite



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How It Began...

- ❖ sci_Zone commercialized QS code to support the GE Aviation Aircraft Health Management System Project.
- ❖ sci_Zone entered into an agreement with GE to use the open source QuickSAT APIs on their “Operational Ground Program” (OGP).
- ❖ QuickSAT/Vehicle Management System (VMS) created as an expanded version of the OGP program.



QuickSAT/VMS

- Broad Use: *Aviation, Satellites, Cars*
- A complete Flight Management System
- Vehicle Health Management & Monitoring
- Vehicle Commanding Services
- Communications services
- Test/Monitoring interface



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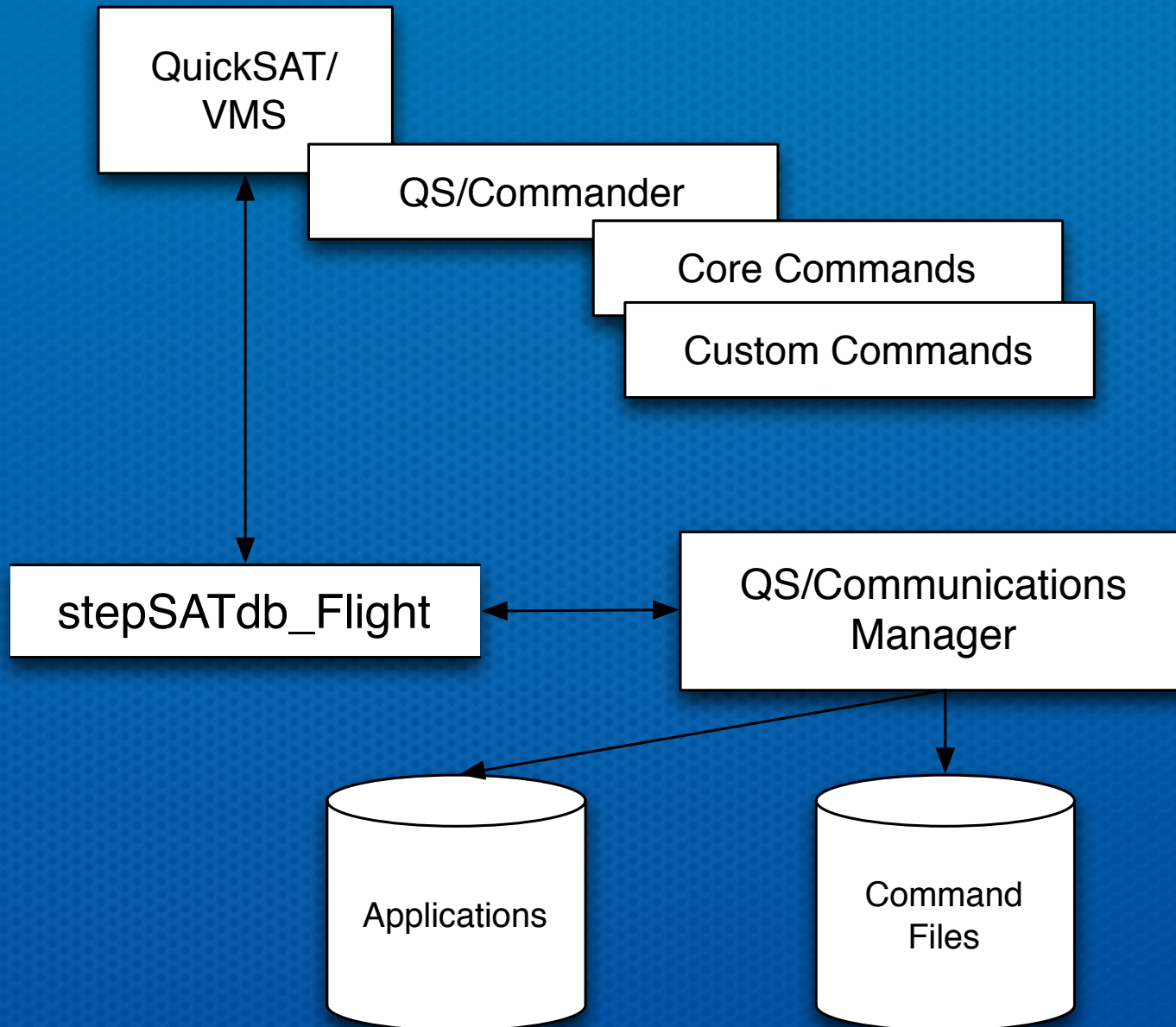
QuickSAT/VMS

- Can serve as a stand alone ground station or part of an expanded environment
- Customizable
- Utilizes open source software where possible
- Works on a range of flight hardware
- Web based - PCs, Tablets, etc.
- Certified DO178B for Aviation



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QuickSAT/Designer

Parts Management

Parameter Defn

C&DH

Mission Flight Plan

Scheduler

Xen Tools

Define Modes

Mission Conops &
Analysis

v6.0

QuickSAT/VMS

Flight Management

File Transfer

Health Monitoring

Communications

Configuration Support

Ground Control

v3.5

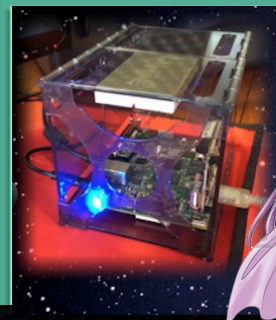
Xen

MCP 1.0

MCT 1.0



FRNCS
Flight computer



ARM-15, -8, -7
Intel Support

LinkStar Radio & QuickSAT

LinkStar: A Paradigm Shift



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Small Satellite Comm Issues

- Difficult to find
- Difficult to acquire
- Limited ground communications

Enter LinkStar...



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LinkStar Quality Satellite Comm Link

“the Edge of Exploration”

- Reliable and seamless data link
- Many communications systems associated with CubeSat failures and require significant flight hardware and tracking time/cost
- Complements high-data rate downlinks
Beacon GPS, Health, Back-up and Sample data sets



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The Foundation: *Globalstar*

- GlobalStar Constellation
 - 32 LEO Satellites (1400 km)
 - Provides global data and voice services for ~ 300,000 customers
- Used primarily for infrastructure/wildlife monitoring
 - Oil Rigs
 - Shipping Containers
 - Gas pipe-lines
 - Endangered animals
- *LinkStar* developed by *sci_Zone* for a range of applications
 - Data links via the GlobalStar network
 - Payload commanding
 - Data downlinks
 - Recovery tracking (with GPS)

Globalstar  Be Heard.



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LinkStar Product Features

- Anytime, anywhere satellite TT&C
- Estimated +70% orbital coverage for Duplex based radios; over 95% for Simplex based radio!
- Common FCC Satellite-to-Satellite License
- No Amateur bands
- Radio astronomy interference was eliminated by limiting the GlobalStar units to operate at 1616.25 MHz with a bandwidth of +/- 1.25 MHz
- No satellite to ground license required



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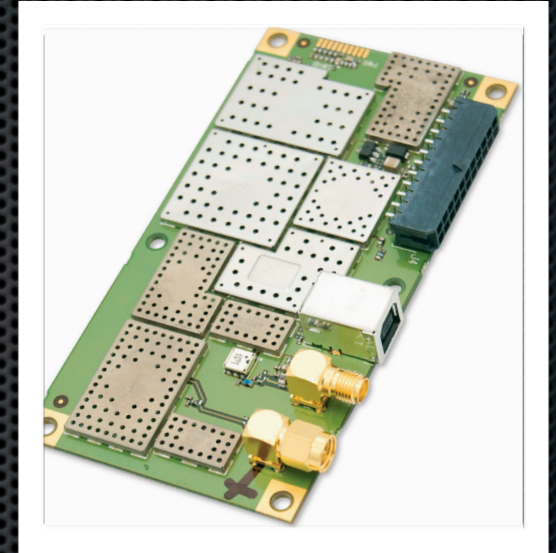
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LinkStar Product Features

- No deployables
 - 2 cm x 2 cm patch for simplex
 - 2.5 cm diameter circular patch for duplex
- Rapid acquisition - simplex beacons within 15 seconds
- Data rates
 - 9600 kbps maximum
 - *LinkStar* can compliment traditional high speed radios
 - *LinkStar* can serve as a primary radio depending mission data requirements.

LinkStar Product Features

- Ground station over Internet Protocol (IP)
 - *Access your spacecraft from anywhere!*
- Piggy-backs on established 2 billion dollar network
- Low Cost
 - No Ground Station
 - No tracking
 - Proven system



Operating Frequencies

Transmit: 1610 MHz – 1626.5 MHz

Maximum transmit power
DC input voltage

Receive: 2483.5 MHz – 2500 MHz

+31dBm EIRP (passive antenna), +34 dBm (active antenna)
+4.7V to 5.1V

Power Consumption
@5VDC input (estimated)

State	Min	Typical	Max	Units
Shutdown	0.0	0.65	1.0	mW
Standby	0.5	0.5	1.1	W
Transmit	2.2	3.65	5.0	W



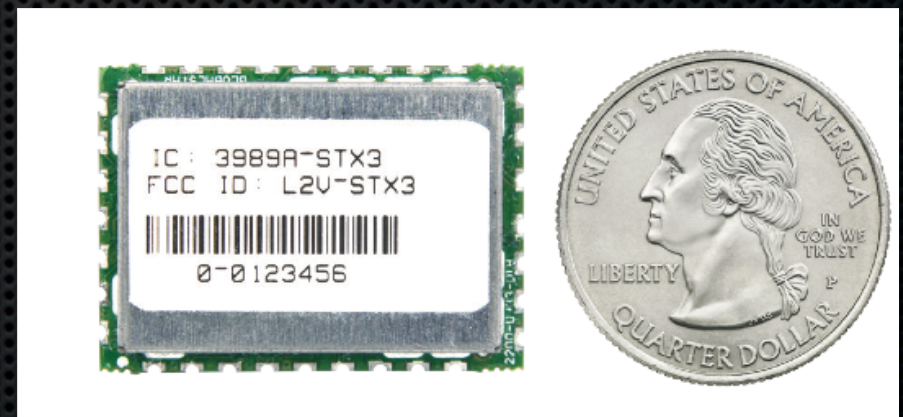
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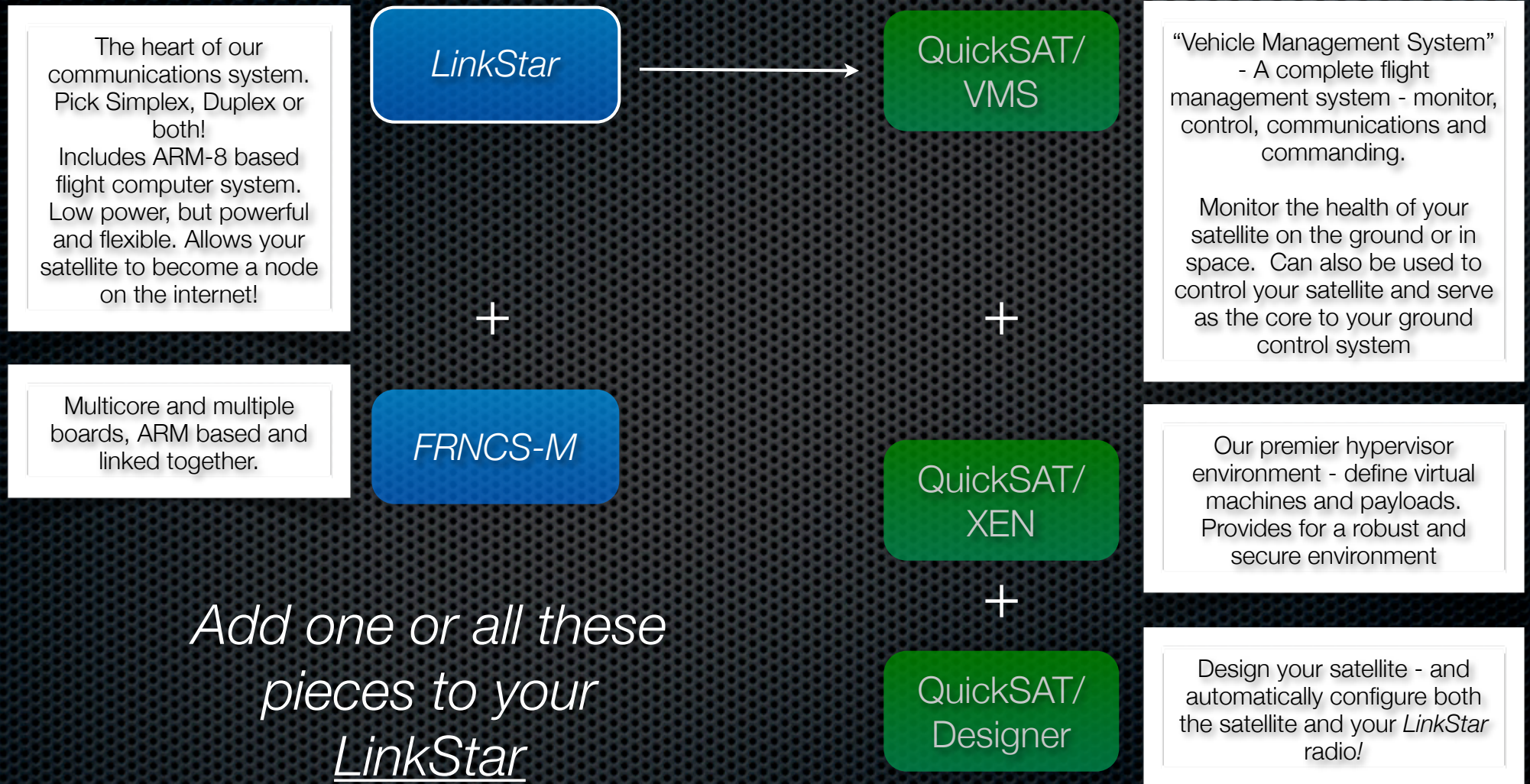
LinkStar-STX3

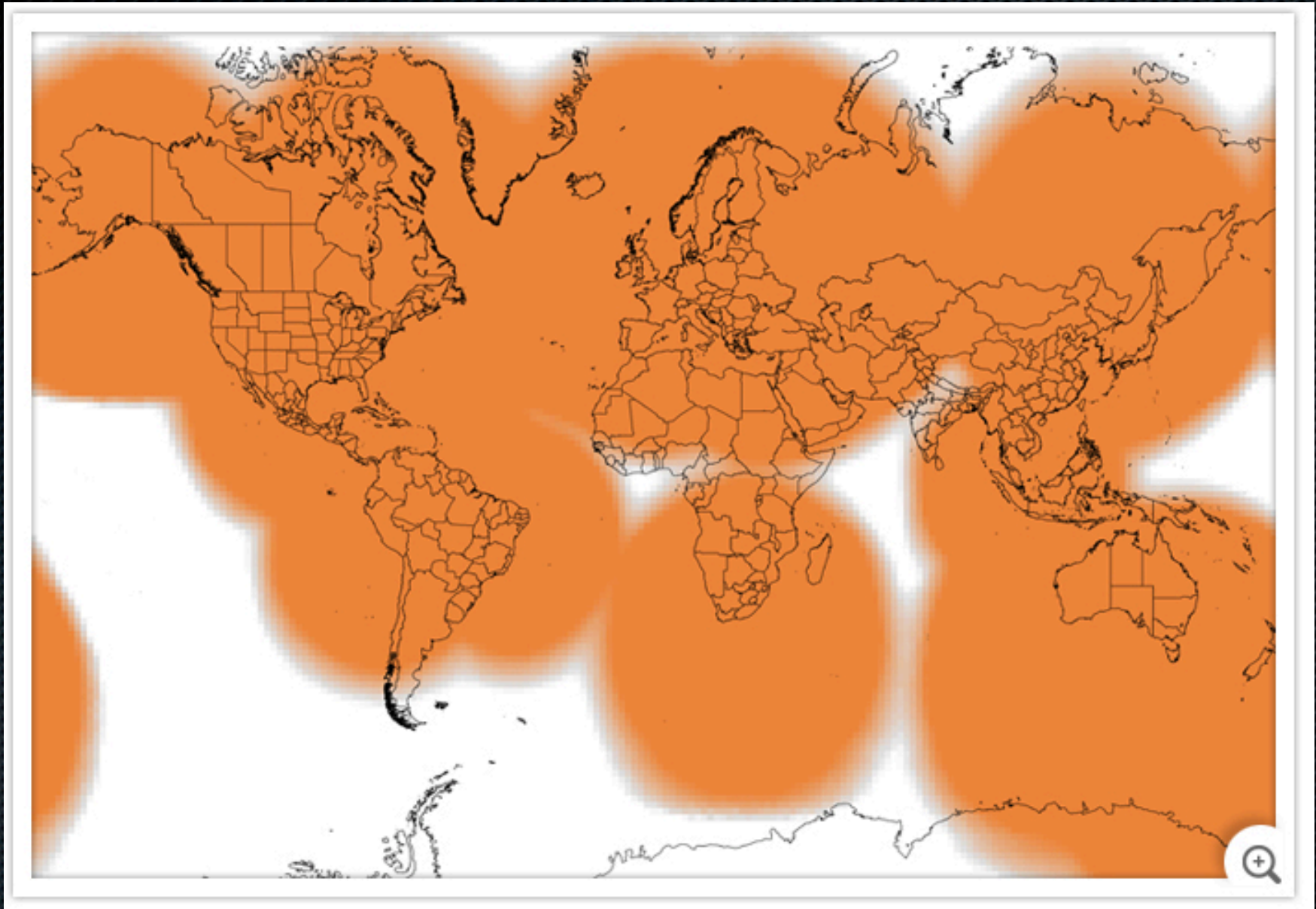
Simplex Gen 3 Features

- Small form factor
 - Power
 - 350 mW Tx power
 - Dimensions
 - 28.7mm x 20.57mm x 4.13mm
 - Electrical
 - Accepts 3.3 V to 12 V
 - TTL Data Protocol
 - Near Global Coverage!



Many ways to configure *LinkStar*



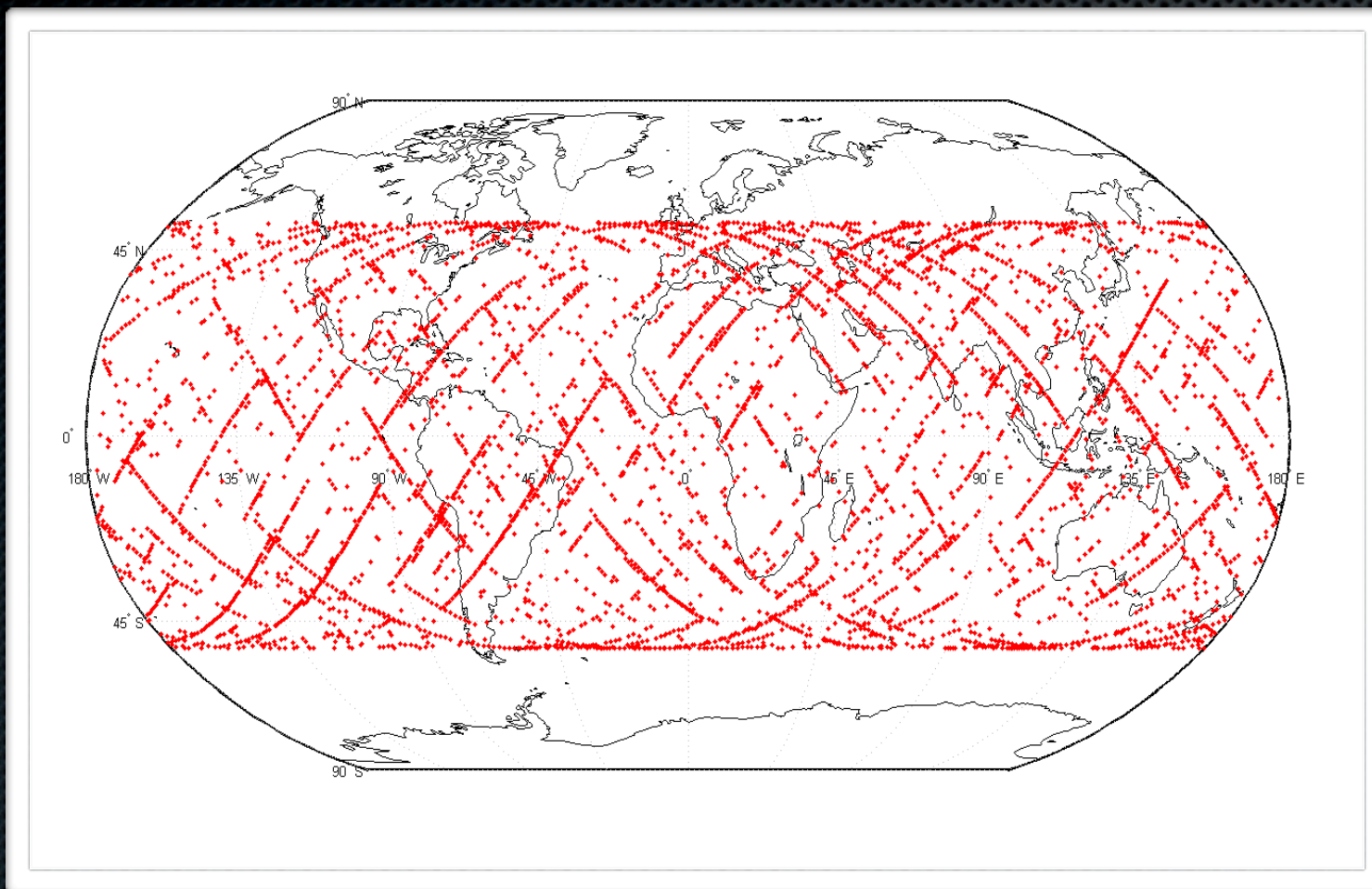


Globalstar  Be Heard.



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Preliminary TSAT Coverage Map Using the Gen2 based simplex radio

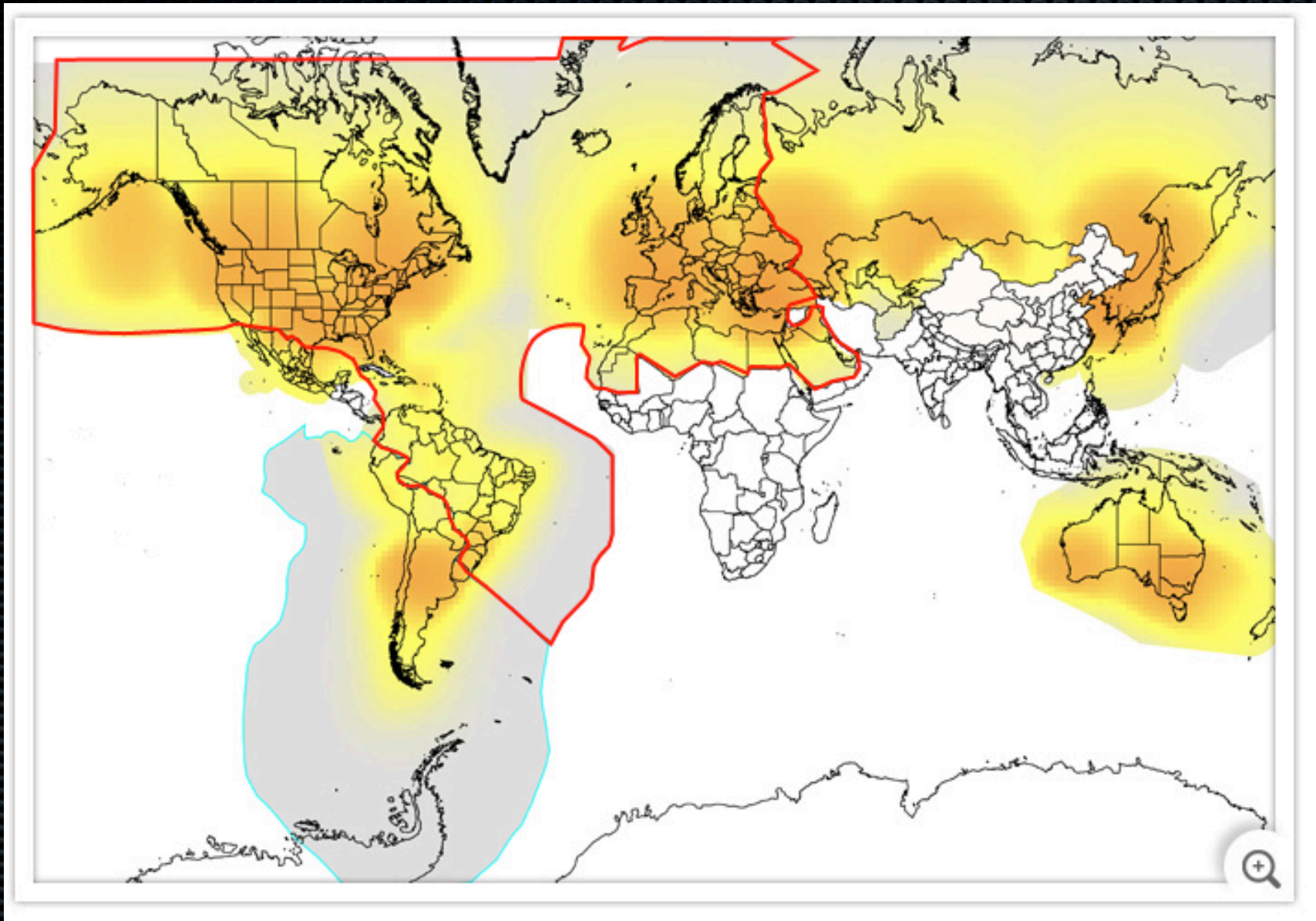


Beacon Cycle

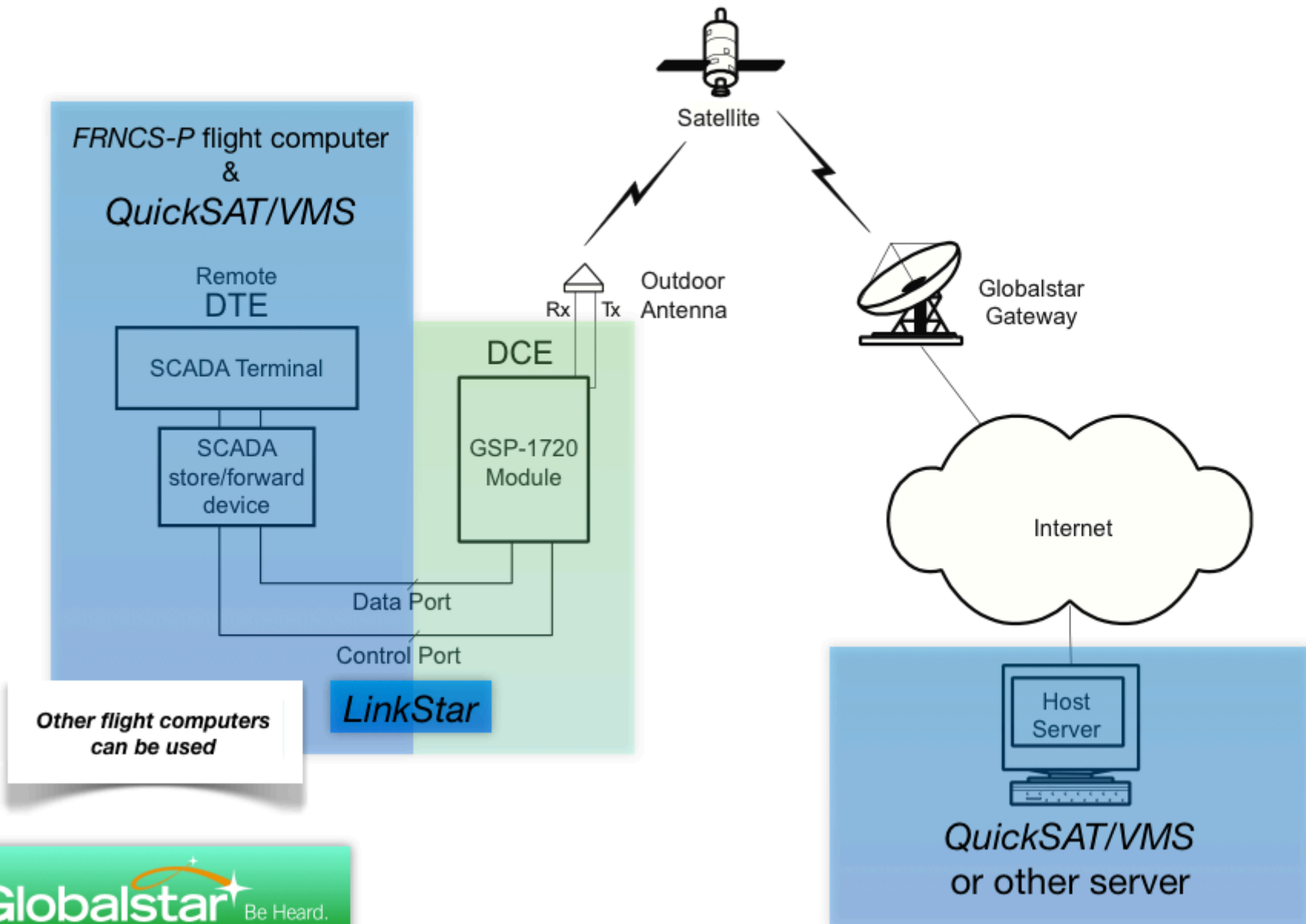
- 32 beacons
- One every 5 seconds
- 15 minute dwell
- over 40,000 packets received

- 45% packet throughput on a tumbling satellite

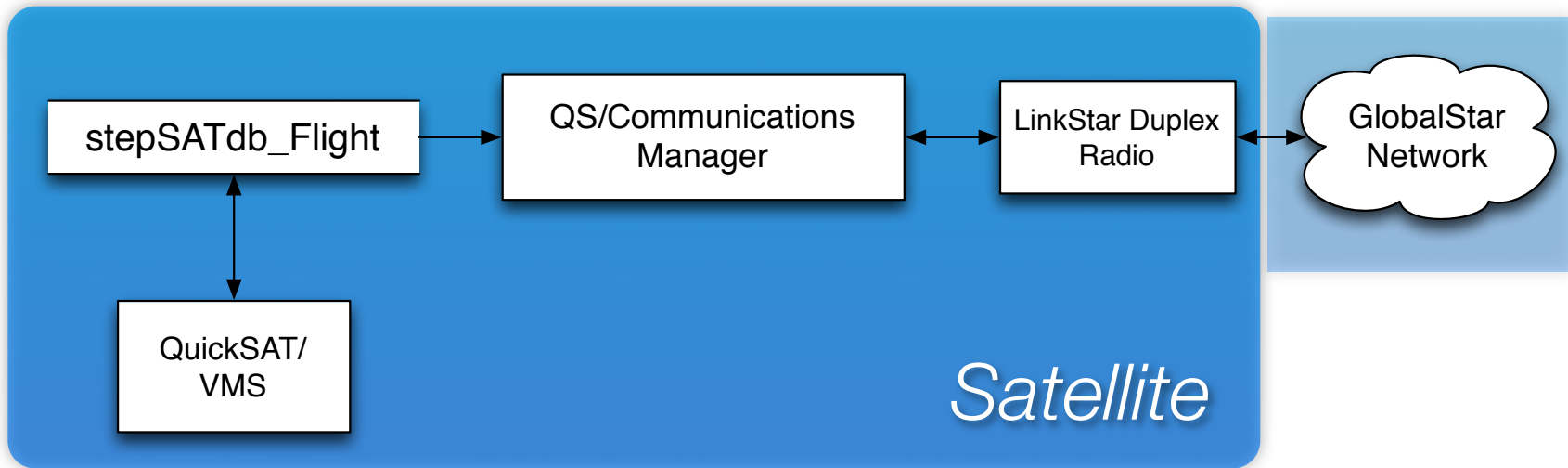
Near Global Data Downlink Coverage



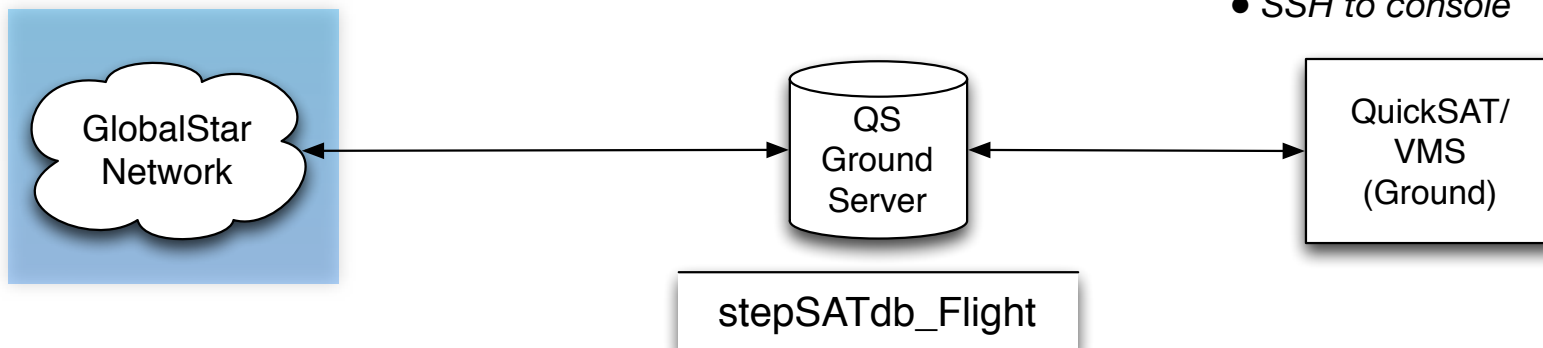
Baseline Communications Scheme with LinkStar



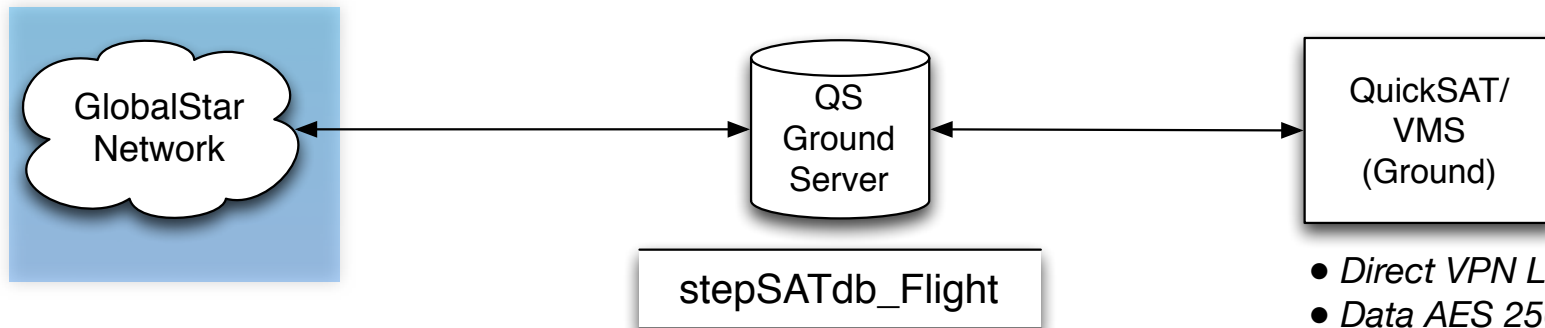
Satellite to Ground



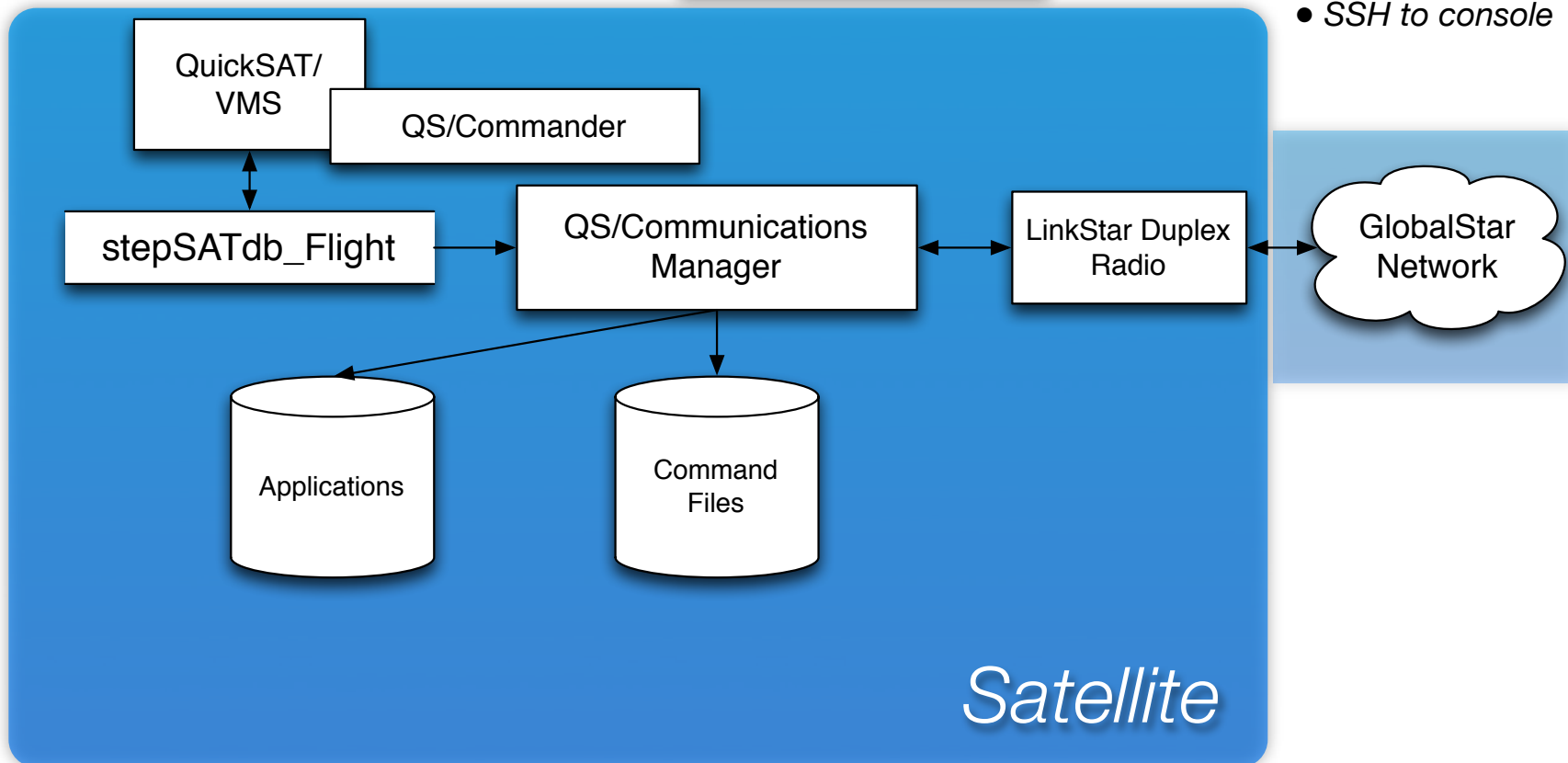
- *Direct VPN Link*
- *Data AES 256 Encryption*
- *SSH to console*



Ground to Satellite



- Direct VPN Link
- Data AES 256 Encryption
- SSH to console



FRNCS-P



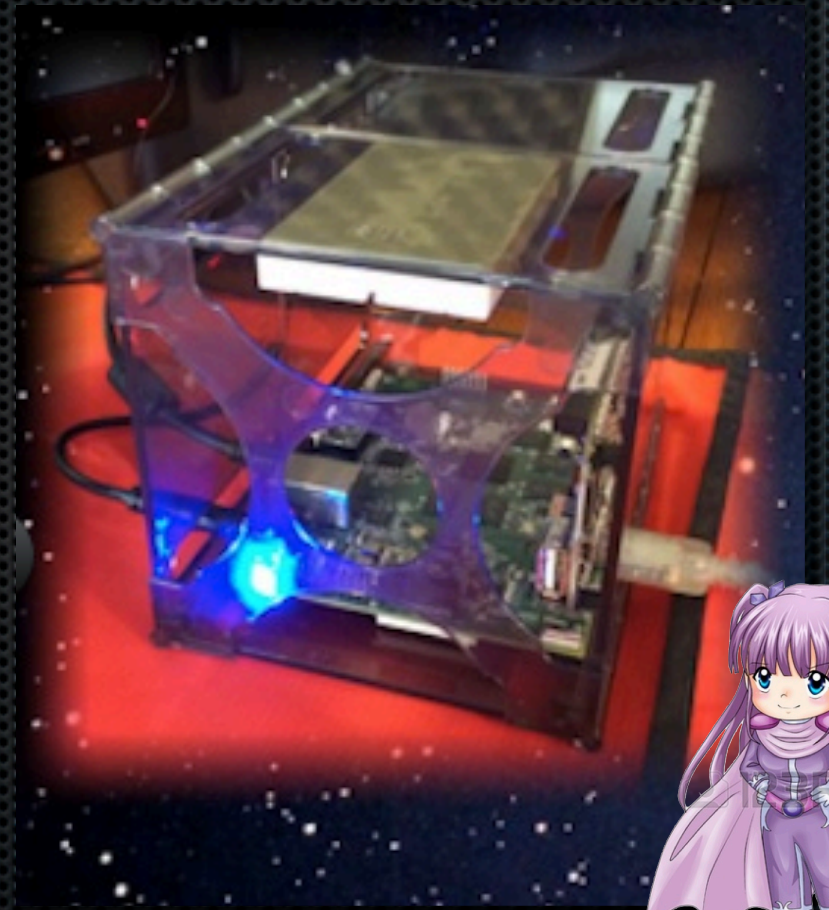
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High Speed Computing in Space

- Supports Multiple ARM Processors & Boards
- Internal/External Networking
- Single and Multi-Core
- Configure for flight using QuickSAT/Designer
- Comes with QuickSAT/VMS for ground station front end, systems monitoring and payload health management and for command, control support, and functions as communications gateway

RADSat will fly a BeagleBone Black (Rev C) with the BeagleSpace architecture.



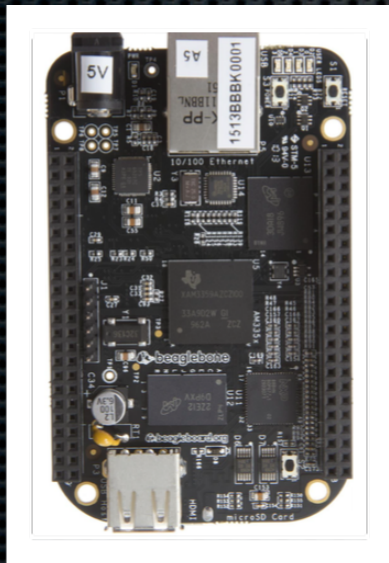
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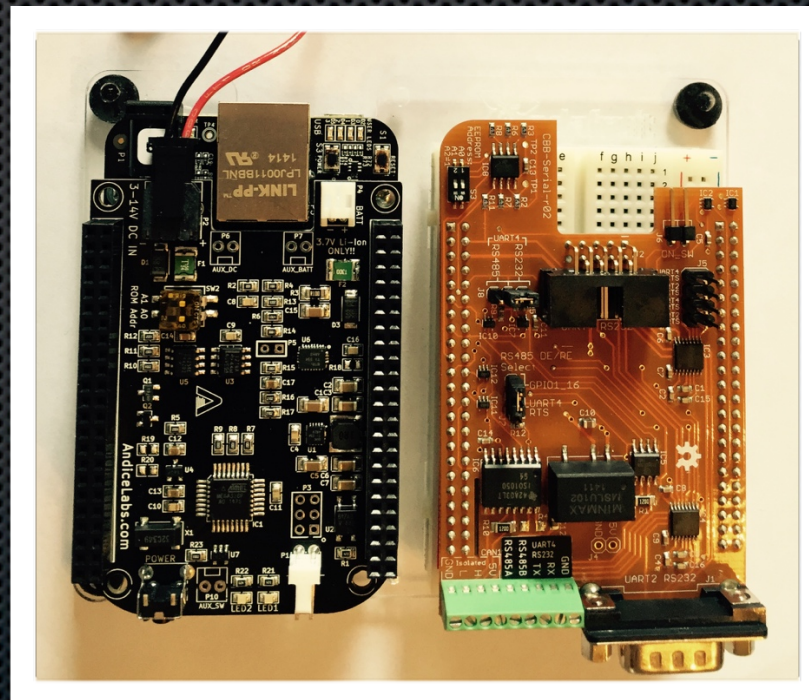
The Details



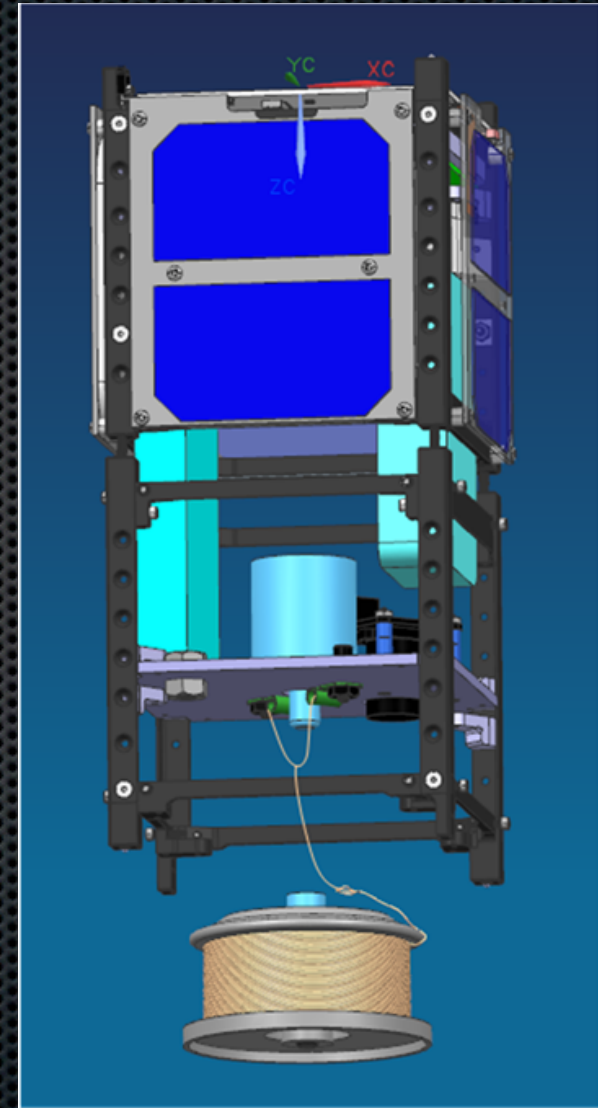
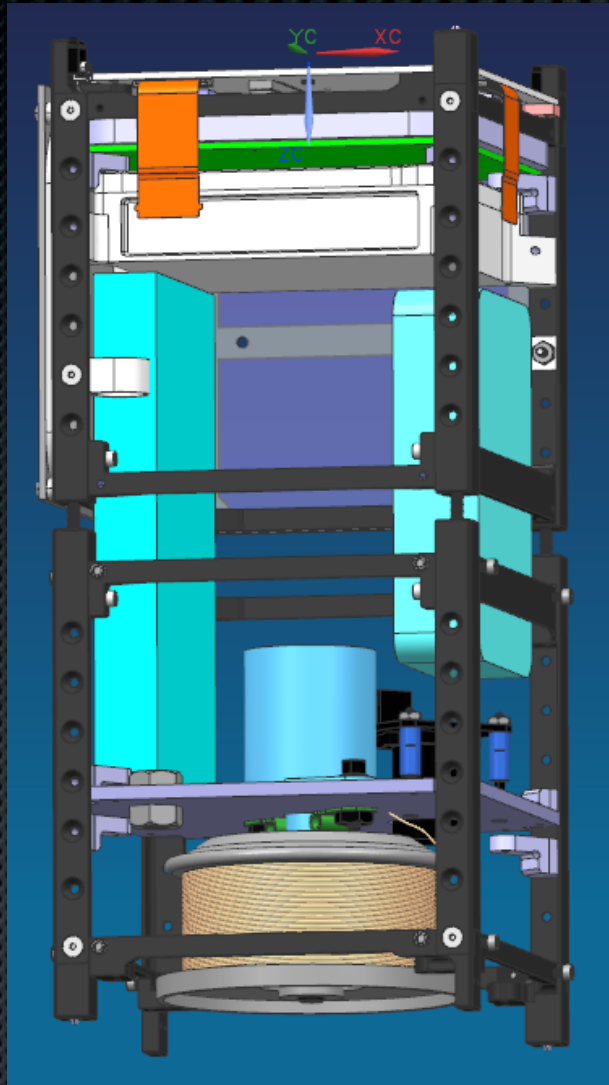
Processor	Sitara AM3359AZCZ100 1GHz, 2000 MIPS (ARM 8)
SDRAM Memory	512 MB
On Board Flash	4 GB
Software	Debian (7.6), PHP, C, MySQL, QuickSAT/VMS



BeagleBone Black
(rev C)



- Power Cape
- Serial Cape
- **Coming soon:** Beagle Space Cape with STX3, rate gyros, magnetometer, and accelerometer



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Schedule

- Work started with the Boeing team in March 2015
- June, 2015 - Software PDR
- Late August, 2015 - Software CDR
- September - Delivery for Test & Integration
- Fall - Ship to NanoRacks
- Late Fall - Launch
- January/February 2016 - Deploy
- Jan/Feb 2016 through March - Mission Operations



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Next STEP - Join the Fun!

- ✦ email: andrew_santangelo@sci-zone.com
- ✦ web: www.quick-sat.com

QuickSAT Satellite Designer
Mission: AP7-0 SAT
Configuration: v6
Configuration Date: 2012-02-13 13:15:45
Created By: Andrew Santangelo

ADCS
Group: Primary Satellite Bus
Total Mass: 20.15
Average TRL: 8

Selected Part Information
Part Name: Coarse Sun Sensor
Part Type: Sun Sensor
Part Number: 29450
Vendor ID: Adcole/SpaceWorks/Da
Entered By: null
Description:
Comment:
Flight History:
TRL: 7
Mass (kg): 0.524
Average Power (W): 1.3
Length (m): 0.11979
Width (m): 0.0762
Height (m): 0.11052
Min Operating Temperature (C): -24
Max Operating Temperature (C): 61
Min Survival Temperature (C): -50
Max Survival Temperature (C): 100
Radiation Tolerance (kRad): 0

Part Name	Part Number	Multiplier	Vendor	Mass	Avg Power (Watts)	TRL
Aerogel MR-103G (GEO)	Aerogel MR-103G (GEO)00001	1		0.33	8.25	8
Coarse Sun Sensor	29450	2	Adcole/SpaceWorks/A FRL	0.524	1.3	7
Coarse Sun Sensor	29450	1	Adcole/SpaceWorks/A FRL	0.524	1.3	7
IMU	imu	1	Analog Devices/Morcoom/S spaceWorks	0.36	1.5	6
PHP GPS Receiver with Antenna	prpgps	2	UTData Design/AnsAntenna/SpaceWorks	0.896	2.3	6
PHP GPS Receiver with Antenna	prpgps	1	UTData Design/AnsAntenna/SpaceWorks	0.896	2.3	6
PHP Magnetometer	MAG 101	1	Southwest Research Institute	0.48	3.5	8

QuickSAT Satellite Designer
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Spacecraft Data
Subsystem Mass Summary Chart
Satellite Mass, Dry (kg): 18.00
Mass Propellants (kg): 18.00
Satellite Mass (Dry + Propellants, kg): 36.00
TOTAL Satellite Mass w/Margin (Mass + Mass Margin, kg): 36.00

Network Diagram
Description: Carrier Frequency (MHz): 2.2
Transmitter Power (W): 100
Transmitter Line Loss (dB): 10
Transmit Antenna Diameter (cm): 70
Transmit Antenna Beam Width (deg): 2
Antenna Efficiency: 0
Peak Transmitted Antenna Gain (dB): 18.758
Transmit Antenna Pointing Offset (deg): 0
Transmit Antenna Pointing Loss (dB): 0
Peak Received Antenna Gain (dB): 18.758
Rflxtn: Isotropic Radiated Power (W): 22,220
Propagation Path Length (km): 1000.0
Space Loss (dB): 170.184
Propagation & Polarization Loss (dB): 0
Receive Antenna Type: Receive Antenna Diameter (cm): 0.28
Peak Receive Antenna Gain (dB): 2.714
Receive Antenna Beam Width (deg): 120



QS/VMS Vehicle Management System
Mode: SIMULATION
Mission: AP7-0 SAT
Status: Unfinished
Percent Unfinished: 0
Percent Progress: Full

Parameter Lists
Select Parameters
Data Display
Plot
Group

Test Units C
Test Units B
Test Units A
Test Units 1
Test Units 2

Parameter List:
Parameter Name
FC_BAT_TEMP_FC_BAT_1
FC_BAT_TEMP_FC_BAT_2
FC_BAT_TEMP_FC_BAT_3
FC_BAT_TEMP_FC_BAT_4
FC_BAT_TEMP_FC_BAT_5
FC_BAT_TEMP_FC_BAT_6
FC_BAT_TEMP_FC_BAT_7
FC_BAT_TEMP_FC_BAT_8
FC_BAT_TEMP_FC_BAT_9
FC_BAT_TEMP_FC_BAT_10
FC_BAT_TEMP_FC_BAT_11
FC_BAT_TEMP_FC_BAT_12
FC_BAT_TEMP_FC_BAT_13
FC_BAT_TEMP_FC_BAT_14
FC_BAT_TEMP_FC_BAT_15
FC_BAT_TEMP_FC_BAT_16
FC_BAT_TEMP_FC_BAT_17
FC_BAT_TEMP_FC_BAT_18
FC_BAT_TEMP_FC_BAT_19
FC_BAT_TEMP_FC_BAT_20
FC_BAT_TEMP_FC_BAT_21
FC_BAT_TEMP_FC_BAT_22
FC_BAT_TEMP_FC_BAT_23
FC_BAT_TEMP_FC_BAT_24
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FC_BAT_TEMP_FC_BAT_44
FC_BAT_TEMP_FC_BAT_45
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FC_BAT_TEMP_FC_BAT_47
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FC_BAT_TEMP_FC_BAT_51
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FC_BAT_TEMP_FC_BAT_53
FC_BAT_TEMP_FC_BAT_54
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FC_BAT_TEMP_FC_BAT_84
FC_BAT_TEMP_FC_BAT_85
FC_BAT_TEMP_FC_BAT_86
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FC_BAT_TEMP_FC_BAT_88
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FC_BAT_TEMP_FC_BAT_96
FC_BAT_TEMP_FC_BAT_97
FC_BAT_TEMP_FC_BAT_98
FC_BAT_TEMP_FC_BAT_99
FC_BAT_TEMP_FC_BAT_100

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