



## Aeneas Mission Overview Lessons Learned

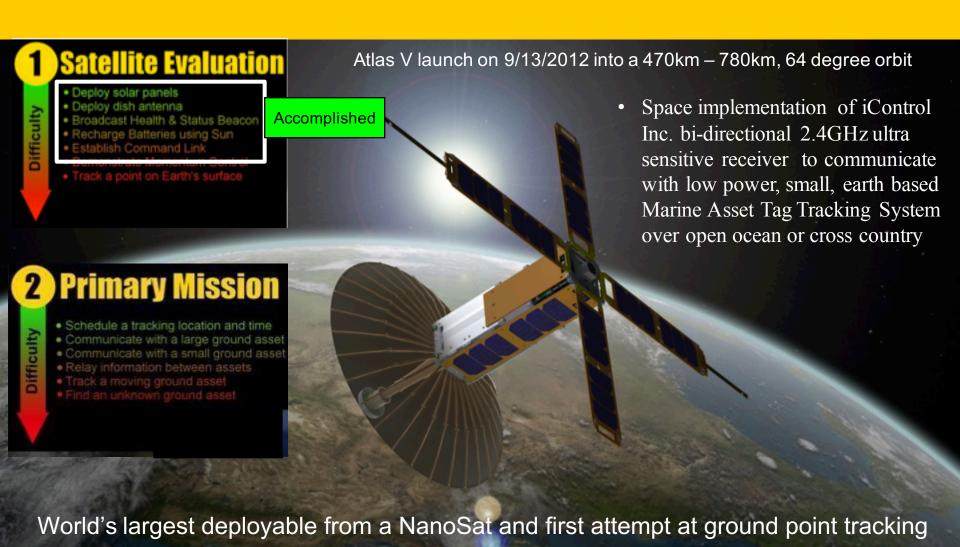
Dr. Mike Obal (571) 239-1855 mobal@isi.edu Tim Barrett (310) 448-8492 jtb@isi.edu





## **Aeneas NanoSat mission**





**USC** 



# L-36 P-POD locations and CubeSat manifest



P-POD locations and CubeSat manifest on launch vehicle

P-POD	Total P-POD Mass (kg)	CubeSat Name	Organization	Size	
1	6.750	SMDC-ONE B	SMDC	3U	
2	6.575	Aerocube	Aerospace Corporation	3 x 1U	
3	6.335	Aeneas	USC/ISI	3U	
4	6.200	CSSWE	University of Colorado	3U	
		CP5	Cal Poly	1U	
5	6.415	CXBN	Morehead State University	2U	
6	5.675	CINEMA	UC Berkeley	3U	
7	6.175	Re	LLNL	3U	
8	6.175	SMDC-ONE A	SMDC	3U	







## Mission Visibility



#### **Generated Interest in SERC**

- Largest Deployable from Cubesat
  - JPL evaluation for inter-plantery
     Ka band, Nanosat Yagi for Navy
- System/Bus/CubeSat knowledge and operational flight software and ground station experience
  - Northrop, Boeing
- 23 months on orbit operational
- First Government Agency CubeSat Rideshare
- Manifest slots into 2014
- Government support services training
- DARPA Phoenix program Maestro Flight experiment – ATK, Novawurks
- Next Generation Space Processor -Honeywell



Mission QSL card for amateur radio decodes

Significant mission brings interest





# Space Engineering Research Center Missions to orbit 2 in 3 years, 3 in 5, 4 in 6



- Effort vehicles in space in as many years
  - A few staff with a bunch of students worked very hard over 8 years
  - Funded by congressional money and some internal ISI funds
  - Designed and executed operational space-like and space vehicles

#### Culmination

- Aeneas on orbit 23 months, still operational
- Caerus Mayflower mission, Advanced Concepts Northrop-Grumman
- Ground infrastructure for satellite tracking and operations
- ½ meter deployable
- Generating a lot of interest both government and private industry

#### Lessons learned

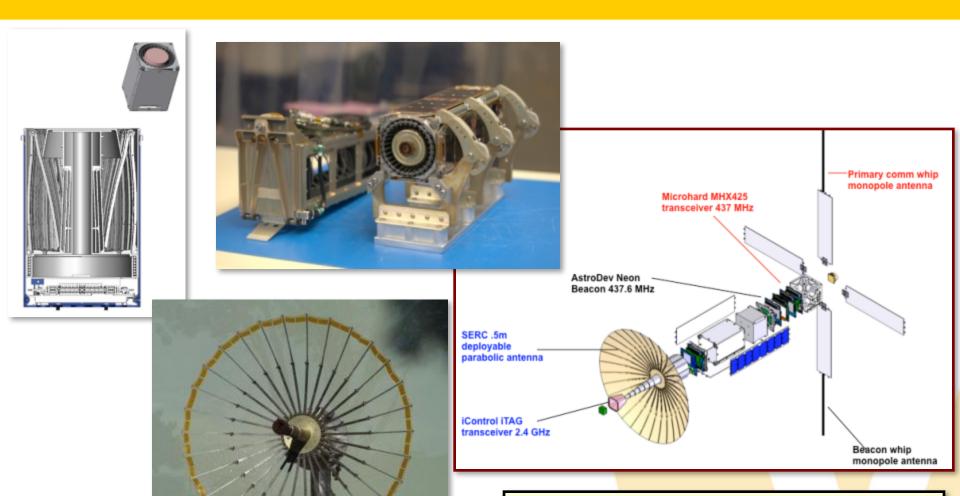
- Design, Testing, Verification and Instrumentation
- Mission Operations
- Champion buys Time and Funding





# 3U CubeSat design: 0.5 meter parabolic antenna; solar arrays; communication antennas deployed





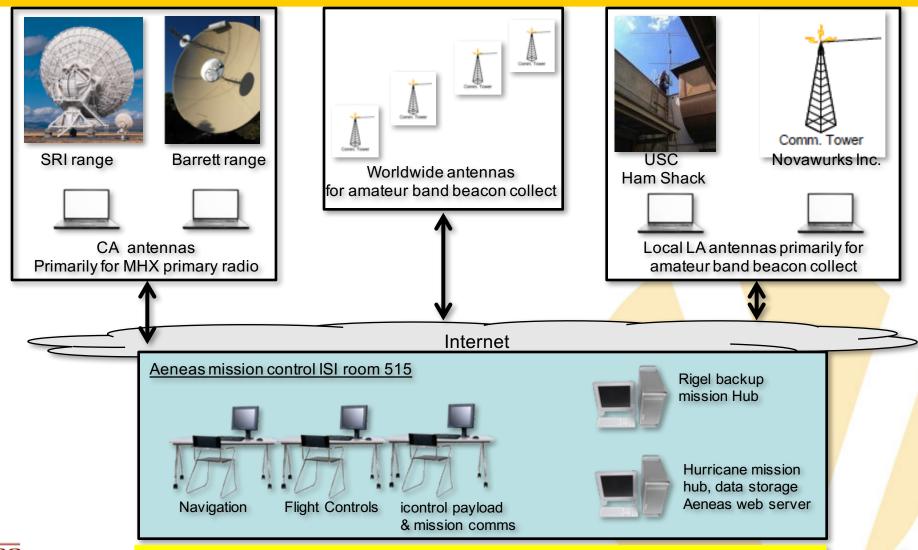
World's largest deployable antenna from a CubeSat, invented in SERC





## Spacecraft Control/Data Processing Ground Setup

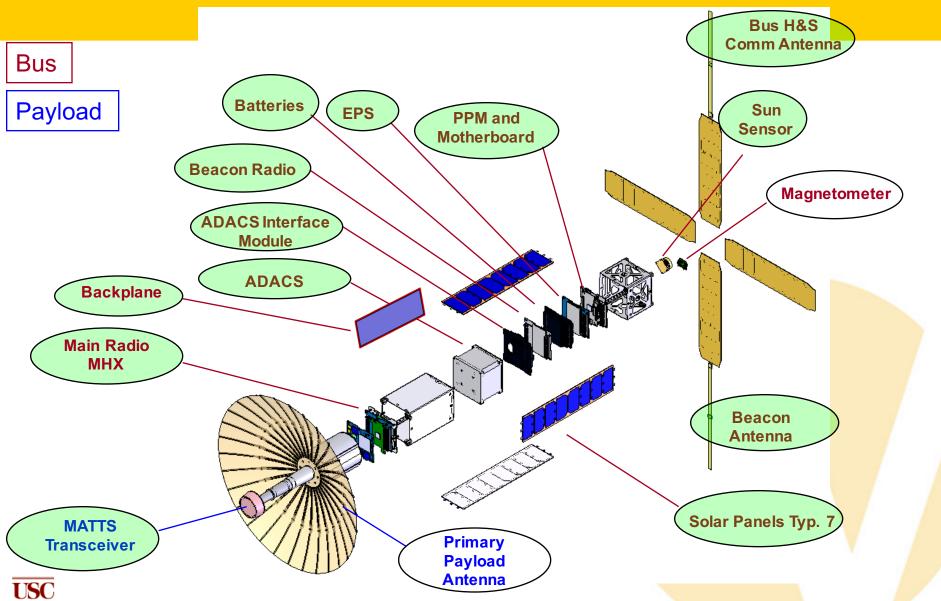






## Initial Operational Status - Expanded View of AENEAS

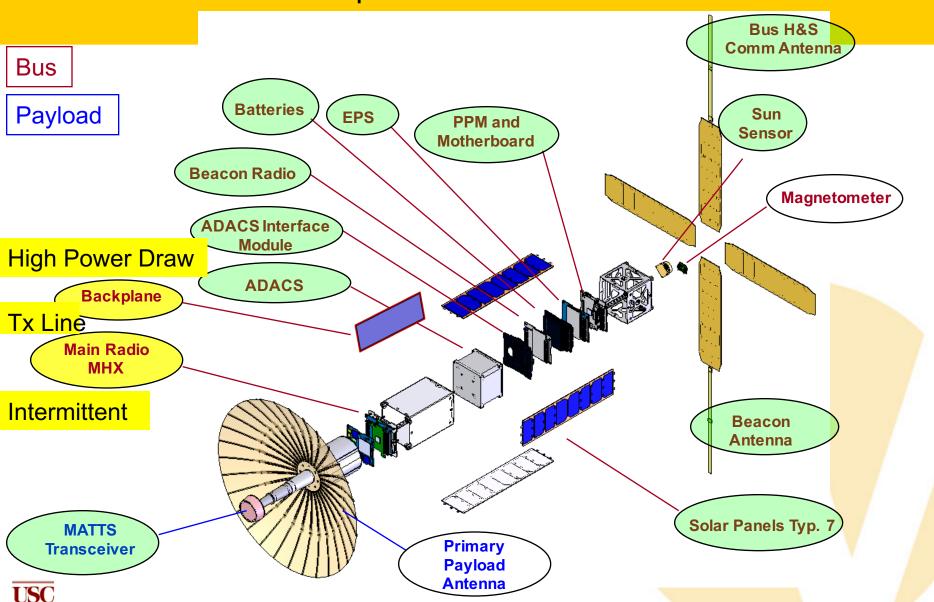






## Current Operational Status - Expanded View of AENEAS

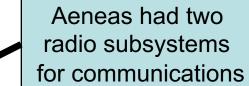






## Satellite Communication Subsystem Degradation





HF Beacons (worked as planned)



Provided state of health data



Received worldwide by many HF stations and locally at the HamShack & Novawurks ranges Data stored in server Thermal

heating



Could not duplicate in lab with spare radio and power supply regulator after raising temperatures 80+ deg. C

MHX Radio
Intermittent failures



Space radiation effects



Radio and power supply regulator not flown in space before. Suspect components TBD SEU possible LE would have not come back on ESD due to space charging (unlikely) Other? Launch environment



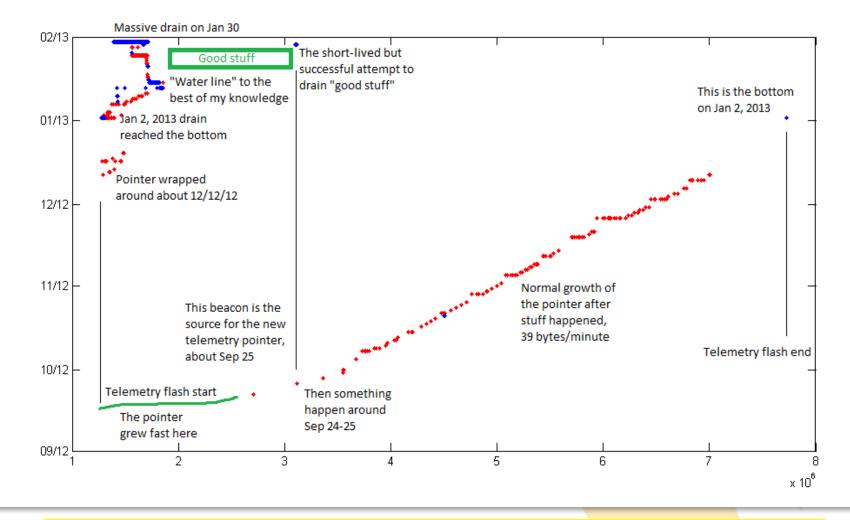
Mechanical weakening due to severe acoustic/vibration levels leading to to early thermal fatigue failure





## Telemetry Data in Flash









### **Data Reduction**



#### 4 Main Telemetry types

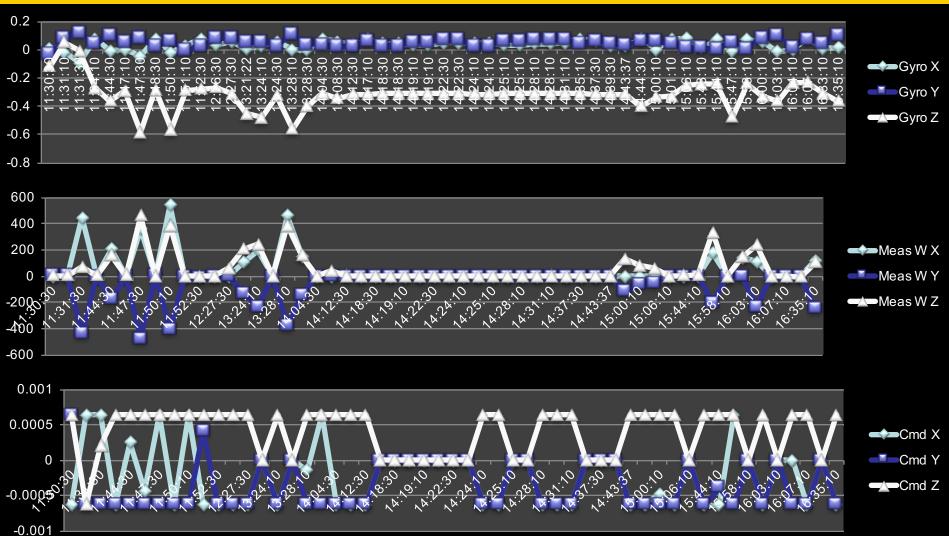
- 61& 62 flight controls, time and time, ephemeris
- 63 power, time temperature, lower reboots and reboot type
- 64 upper processor, time, upper reboots, reboot type
- To ground via main radio and beacon
  - beacon path stuck at type 64, minimal actionable data
- Flash filling ~600 bytes/min between wheels off and Sept 25
  - Sept is the cache of full on telemetry to mine
  - Past that slow type 64 growth until Feb 7<sup>th</sup>? 2013 shut off beacon to flash
  - Wrapped around flash during first main radio quiet time
- Tatiana's story pins down the rough dates of collect
  - Find it again for the timeline that I'm constructing





## Flight Software, Controls

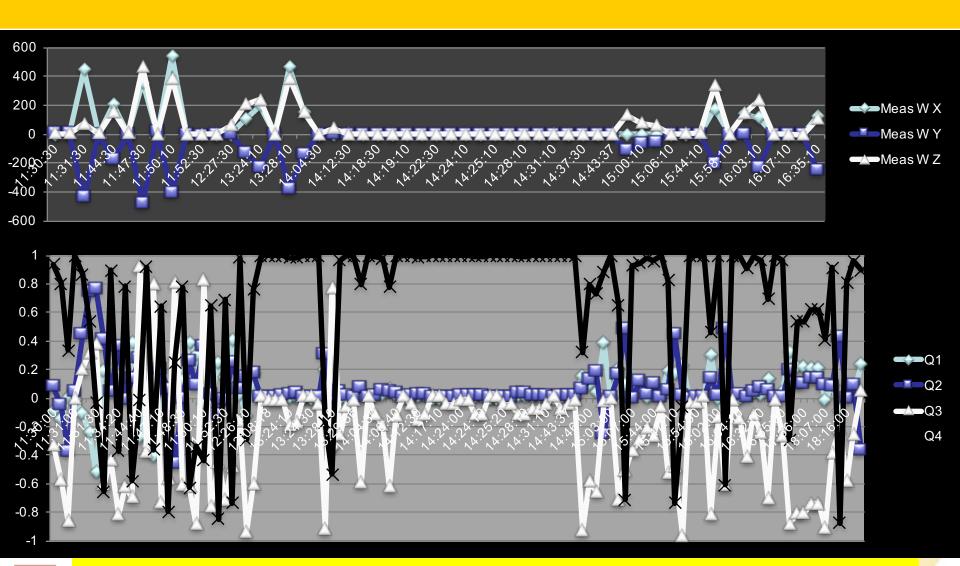






## Flight Software Error Quaternions









# Power, flight software modes, temperature, sun sensor



#### Power

- BCRs 1-4 (high power) all operational
  - at least 4 of 7 panels good
- BCR 5 (low power) not utilized
  - Z panels removed and USB not connected

#### Sun Sensor

Detected sun on two time stamps

#### Mode controller

Moved from idle to sun search mode

### Temperature

- 28-35 deg C, and two inverted sensor mechanisms, but all track
- Reportedly not reliable per Clydespace
- ISI mounted RTDs were not instrumented in fsw





## Communications Results 107 connects of 290 attempts



#### **Satellite Radios**

- Beacons Stensat Radio Beacon, 1 watt, 20 second intervals, omni whip
  - >47, 848 frames world wide
  - Continuing SOH and signal strength monitoring
- Main Radio Microhard MHX, 1 Watt, Spread Spectrum, freq Hop, omni whip
  - 1/2 Megabyte of telemetry data, much was duplicating Beacon data
  - Dozen Command types up
    - wheels off, beacon frequency, move telemetry pointer
    - iTag configuration, execute operational macros
- Itag Payload Radio 1 watt 2.4 GHz in the 0.5 meter dish

#### **Ground Stations**

- Barrett Range, Playa Del Rey, 10 Foot dish 4 turn helix and Auxiliary Beacon
- USC Ham shack, ACB building, 18 foot cross polarized yagi
- Novawurks, Los Alamitos, 18 foot cross polarized yagi
- Stanford Research Inc, (SRI) San Jose, 60 foot dish with cross polarized elements and 100 foot cable.
   >47848 Beacon Frames, ½ Megabyte of TT&C data



## **Ground Station Upgrades**



#### Main Helix

- 1, 2, 4 turns Z rot. gain within 3 dB
- Main radio on feed, 1.5 dB 100km
- Copper shield around radio
- Tuned with Smith chart Analyzer
- Cal at 100 feet & Malibu 21 dB

#### Aux 4 turn

- Continuous beacon monitoring
- Main radio measures main feed

#### WiFi

- Mounted to Az/El dish
- Operated for a time at feed point

#### Az-El mount

- Align with sun, north star
- Pinned for slippage









## Pass Log Spreadsheet



Aeneas Mission Summary

Rarrett Range & SRI & Anecdotes from LISC International

Ва	rrett Range & SRI	& Anecdotes f	rom USC, In	ternation	ıal														
	Date	Time	Begin Az	Max El	Alt km	Range km	TLE.txt	Main Radio MHX	MHX ATS	MHX Connects	Data	MHX Lights	RSSI dbm No	oise Floor dBm	Beacon dbm	Beacon hole	iGate, pwr		Notes
	1-Feb	13:53	200	75	753	794	13032?	Cu on Feed	19	Yes	15 Misc	1.5	-98	-110		yes	no	39	No Beacons under 1300km, CAERUSAM\$, CMD set TLM Pointer, 15 iTag packets  MHX 66 to 75, rising, nothing ondown side, iTag
	31-Jan	23:45	322	89	?	?	13031?	Cu on Feed	19	Yes	100K	1	-100	-110		yes	no	38	miscs 1 min, 11 hrs, 1 hr
	30-Jan	23:26	47	41	688	991	13029	Cu on Feed	19	Yes	3,500	1.5	-99	-109		yes	no	37	1 min connect, 447 tlm 11 misc packtes, no iTag 5 min beacon stop, tlm 23445-23800102958 (355
	30-Jan	14:52	296	63	756	861	13029	Cu on Feed	19	Yes	10,295	0.5	-102	-111		yes	no	36	pkts) +7 itags, cmd Q reset pointer 3 min connect time, tlm 16068-23359->7291 beacons + 60 iTag packets - Record! All
	30-Jan	0:46	231	43	680?	900?	13029	Cu on Feed	19	Yes	371,000	1.5	-98	-110		yes	no	35	commands up? Kill a beacon?  Copper bag MHX on Feedpiece, bkgnd tests,
	29-Jan	14:30	90	60	750	850?	13029	Cu on Feed	19	Yes	3859	1.5	-97	-111	-99	yes	no	34	electric saws running & not iGate & not Bookend 1-29 0812 gate, 1-18tag old, cmd Q set telem ptr upper globals, Wheels start, stop after
	29-Jan	0:26	90	62	650?	750	13028	Cu Screen Bx	19	Yes	60KB	1.5	-99	-109	-99	yes	yes	33	hard reset., 61,62,63
	27-Jan	15:30	270	75	750?	790	?	Cu Screen Bx	19	Yes	0	1.5	-97	-109		yes	yes	32	Bookend 1-27 2008 gate, 1-18 tag old Bookend 1-26 0805 gate, 1-18 tag old, Realterm and audacity fight for timeon laptop, kill sound cap, 1 min connect time, tim #15368-15585 (15591-15586is dupbeacons (nc doesn't
	26-Jan	1:06	90	65	680	800	?	Cu Screen Bx	19	Yes	8711	0.5	-99	-110		yes	yes	31	terminate cleanly) Bookend 1-24 2007, pass, Bkgnd on Realterm
	24-Jan	16:12	270	85	750?	760?	13024	Cu Screen Bx	19	Yes	16K	1.5	-97	-110		yes	yes	30	Argon & 1-24-13 pass No bookend, No data or commands,1 min in range, Rainy Fade, spec A and recorded beacons
	24-Jan	2:06	270	65	686	843	13023	Screen Box	19	Yes	0	1.5	-99	-109	S=2	yes	?	29	Audacity  No bookend, 10501-10530, 54-83 degrees in
	21-Jan	16:52	270	85	750?	760?	13021	Screen Box	19	Yes	850	1.5	-99	-110		yes		28	beam, then Hopelessly lost in Pierouette No bookend, but WIFI xmitting, No d Base connect, Binary backup is 2012-1-20-4:33 on Dell
	21-Jan	2:46	270 270	72 26	750?	800? 1400	13019	Screen Box	19 19	Yes No	0	?	-99	-110 -110	-109 -109	yes		27	(yesterdays date)
	20-Jan	18:17			757		13019	Screen Box			0					no			No bookend
	19-Jan	17:52	270	57	757	883	?	Screen Box	19	No	0			-108	-108	no	yes		No bookend, 2 decoded beacons on alt helix
	18-Jan	17:34	90	71	740?	?	13018	Screen Box	19	No	0			-109	-108	no	yes		No bookend, no blinks, B team TLE?
																			Bookend 2013-1-18 11:2:233 49, pass, 12:16:46, Q has save upper globals, wheels on , no restart on hard kill, tlm 61,62,63 <b>No Hurricane Cornet &amp;</b>
	18-Jan	3:26	90	85	750?	760?	13014	Screen Box	19	Yes	0	1.5	-99	-109		yes	yes	26	BLUE SCREEN, HAND OP DISH  Q has wheels on , no restart on hard kill, connect
	16-Jan	18:36	270	68	754	807	13014	Screen Box	19	Yes	2.5KB	1.5	-97		-102	yes	no	25	over 60 seconds, telem #1.0258-1.0348 Bookend 2013-1-15 11:51-50 dtm, pass, 12:27- 48 dbm, Two commands up, debug lower, ask telem 60, connect over 60 seconds, telem
	15-Jan	4:06	109	63	682	703	13014	Screen Box	19	Yes	72KB	1.5	-97		-103	yes, 5/min	yes	24	#7177-9758  Bookend 2013-1-14 3:00 -49dbm, pass, 03:47 -50
	13-Jan	19:13	270	67	746	790	13012	Screen Box	19	Yes	35KB	1.5	-97		-102	yes	yes	23	dbm, Two commands up (kill a be acon, ask tlm), connect over 13 seconds , telem #5 989-7175

Per pass logging aided analysis,
Dish pointing, Feed positioning, Cable lengths, Data collects,
Beacon and TT&C radio signal strength, antenna deployment





## Payload Results



## iTag Wifi Transceiver

- Telemetry shows it has been alive for months straight, without resets
- Commanded it to go CW mode for 8 minutes
  - Digital control responses indicated it was on for that time
- Commanded it to stream data to iGate (1/2link)
  - No reception on ground
- Commanded to send back its configuration and parameters
  - Digital control responses indicated it is still setup appropriately
- Main Dish Deployment
  - Working the issue







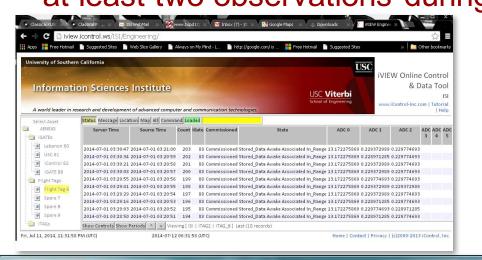
mysql>

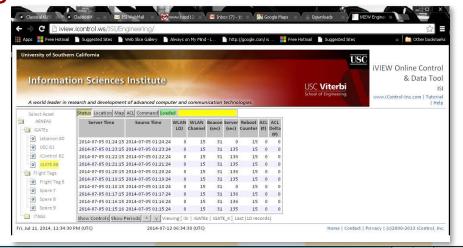
Celebrating 10<sup>2</sup> Years

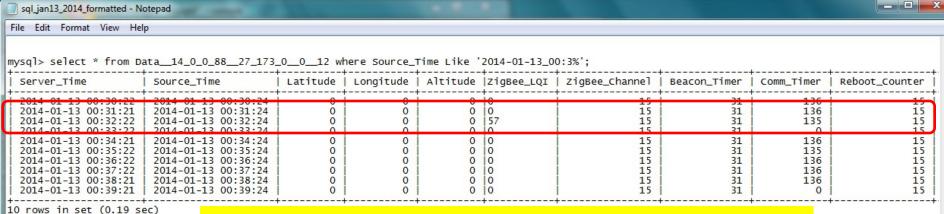


## Payload Results

 Mysql search of 28652 records shows handful of Non-Zero LQI, at least two observations during AOS



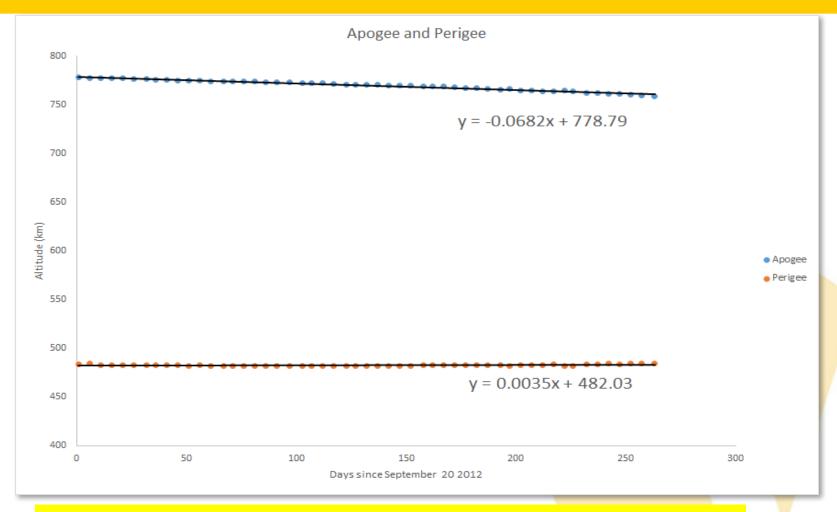






## **Aeneas Orbit Stability**





Orbit degradation is not changing our actual location vs. TLE, Adjusted timing of sat tracker computer/software - likely not issue







## Radiation Belt Crossings



3-4 passes twice daily, 10-15 minutes per day, Flux O >0.5MeV electrons, 23-94 MeV protons 1-10K /cm² sec str, 1990-1996, 2000-2006, CEASE/TSX-5, LET vs Cross-Section curve starts at 2MeV.cm²/mg, probabilistic, must collide





## Lessons – Design, Test, Verification



#### Design

- Duplicate interconnect wiring
  - Inter-processor interconnect, one way, went bad at connector, partially cured arathane
- B dot controller
  - Torque coil controller to kill tip off rates, momentum control too great for wheelbox

#### Testing

- Beacon checkout
- Inter-processor communication interconnect
- Long range and Doppler Radio tests

#### Verification

Power consumption for 'functionally identical' components





## Lessons – Agencies, Mission Ops, Champion



#### FCC and IARU

- Start early and keep checking and circling back until license in hand
- Amateur IARU coordination first before FCC, big fan of beacons
- Mission Operations
  - Tracking
    - TLE versus momentarily operational vehicle
    - HRD sat track great but can easily stick to old TLE
    - Keep a dish near you
  - Don't wait too long for a test on orbit
    - Comms may not last forever
- Champion to buy Time and Funding
  - Will buy you more test time
  - Can keep a team together for Mission Operations





## **Going Forward**



- Monitor Aeneas from campus/Millennium Space Systems while beacon, batteries and power continue
  - 1 year inside 2 years outside
- Delivered 3 element Crossed-Dipole Yagi deployed from 10x10x8 cm
- Deliver Maestro Flight Experiment for DARPA Phoenix Program
- Build and Fly PTecSat
  - From lessons learned
  - Rev II of post shock tested antenna
- SERC is a successful concept for:
  - Developing low cost platforms for space science research
  - Evolving from designing and flying short life (< 1 year) CubeSats to long life (~ 36 months) NanoSats to meet customer mission needs</li>
  - Educating undergraduate and graduate students in space systems engineering





## **Questions?**



- Thanks to the Cubesat Program office for Colony I bus(es) launch opportunity and support
- Mike Obal for championing the effort to the end
- David Barnhart for providing the mission, initial inspiration and first champion
- Talbot Jaeger for Mayflower/Caerus risk reduction opportunity
- iControl for flight software, Matrix X flight simulation, gyro PPMs, itags and igates and untold extra hours
- Team Aeneas Mike Aherne, Tatiana Kichkaylo, Jeff Sachs, John Smolik, Will Bezouska, Lucy Hoag, Siamak Hesar, Alex Shim, Maria Guzman, Melissa Jawaharlal, and a dozen others that made this happen









# Improvements to the baseline Pumpkin Colony I CubeSat design for the AENEAS mission



#### • Colony I BASELINE BUS

- Attitude determination
  - 3-axis magnetometer on ADACS interface module
  - Coarse sun sensing requires panels on all 6 faces
- Power
  - · Deployable solar panels
  - Fixed cells on Aft, Fwd, and 3 Sides
  - EPS with Li-lon-Polymer batteries
- Communications
  - None
- Structures & Mechanisms
  - Payload structure plus <1U structure
- CD&H & FSW
  - 8051 Processor
  - Driver software only
- Attitude control
  - Miniature reaction wheels
  - Magnetic torque rods

#### USC/ISI MODIFICATIONS

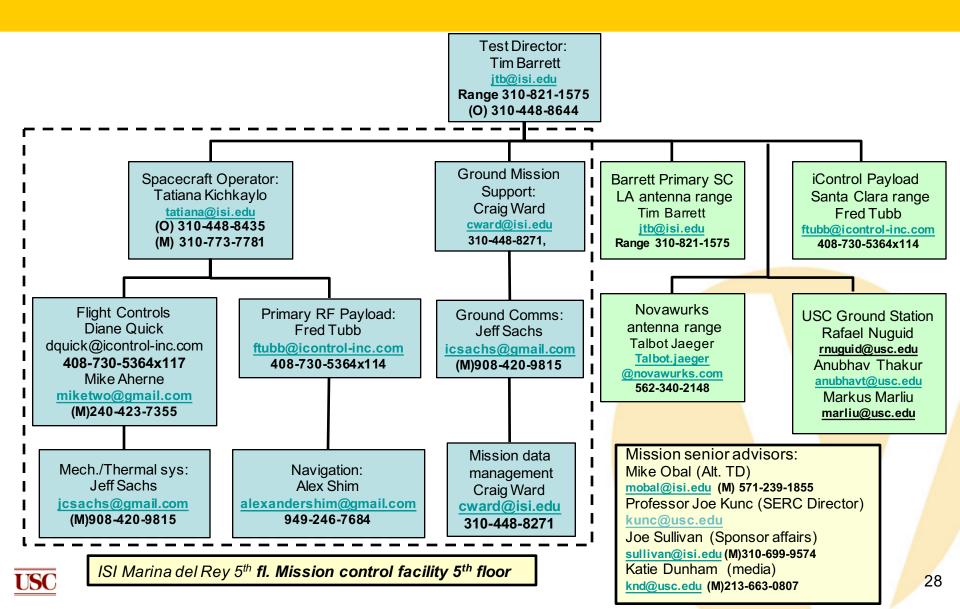
- Attitude determination
  - Relocated magnetometer reduce field input
  - Added sun sensor increase pointing accuracy
  - Added MEMS Gyros increase fidelity of bus for pointing
- Power
  - Changed panel mounting ("flower petal")
  - Removed aft cells for sun sensor
- Communications
  - 115 kbps Transceiver plus Antenna for Bus TT&C
  - 9.6 kbps Beacon plus Antenna for TLE
- Structures & Mechanisms
  - Added Deployment Mechanisms for Solar Panels/Antenna
  - Replaced Payload top plate
- CD&H & FSW
  - Replaced 8051 with PIC-24 Microprocessor
  - Full compliment of Flight, Mission and TT&C Software provided





# Aeneas flight mission operations team organization and locations



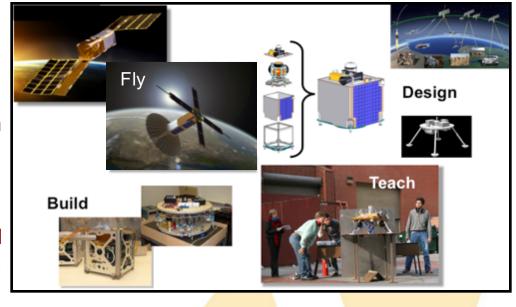




## USC/ISI SERC history of educational programs and rapid growth



- Dr. Max Nikias, Aerospace Corp. and the USAF initiated the Microsatellite systems center at ISI in 2003 later merged with the Astro dept. into the SERC
- Since 2006 over \$3.8M in sponsorship from USC, government and industry
- ~100 undergraduate, graduate and PhD. students have been involved in all aspects
- Caerus CubeSat payload flown in 2010 and Aeneas CubeSat flown September of 2012 MaVEx manifested for Oct. 2014 launch
- Improved key subsystems to enhance the capabilities of an industry designed CubeSat
- Designed the LOC-Comm Encapsulated NanoSat (Life >24 months) that will conduct a number of advanced "Radio Science" experiments



USC/ISI is evolving from design and flying short life (< 1 year) CubeSats to long life (~ 36 months) NanoSats to meet customer mission needs





# Timeline Microsoft project to collect the arc



ID	0	Task Name	Duration	Start	Finish	Predecessors	Resource Names	Sep '12 2 9	16	23	Oct '12 30 7	14	21	Nov 28		18
1	000	Aeneas Mission Timeline	230 days?	Thu 9/13/12	Wed 7/31/13				10	6.07	39.1.1.	esherolistical	-800	AV.	the state of	and the same of th
2	TE	Beacons-Weak	5 days	Thu 9/13/12	Wed 9/19/12				=							
3		Beacons-Strong	225 days	Thu 9/20/12	Wed 7/31/13	2		77								
4	THE .	Main Radio-PreLocation	11 days?	Thu 9/13/12	Thu 9/27/12											
5	THE STATE OF THE S	Main Radio - First Run	20 days	Fri 9/28/12	Thu 10/25/12	4							_			
6	HE.	Main Radio - Darkness	47 days	Fri 10/26/12	Mon 12/31/12	5										
7	THE .	Main Radio - Second Run	76 days	Tue 1/1/13	Tue 4/16/13	6										
8	1000	Main Radio - Darkness	69 days	Wed 4/17/13	Mon 7/22/13	7										
9	HH.	Full Sun-One	14 days	Mon 11/26/12	Thu 12/13/12											
10	THE STATE OF	Full Sun-Two	11 days	Fri 2/22/13	Fri 3/8/13											
11	HE	Full Sun-Three	12 days	Mon 4/8/13	Tue 4/23/13											
12	THE STATE OF	Full Sun-Four	11 days	Fri 7/5/13	Fri 7/19/13											
13	110	Tumbling Z17, Y2.8, X1.7 deg/s	78 days	Thu 9/13/12	Mon 12/31/12			- 1								
14	THE R	Tumbling minimal	151 days	Wed 1/2/13	Wed 7/31/13	13										
15	<b>EE</b>	Lower Tx Breaks	0 days	Tue 9/25/12	Tue 9/25/12					9/2	25					
16		Command - Wheels off	0 days	Sun 9/30/12	Sun 9/30/12					0	9/30					
17	110	Command - Beacons 20 sec	0 days	Thu 2/7/13	Thu 2/7/13			The second								

