



2014 CubeSat Developers Workshop



Integrity ★ Service ★ Excellence

A Tale of Two CubeSats: GEARRS & VPM

25/04/2014

**James Meub
ATA-Aerospace
AFRL/RVEP**



Introduction

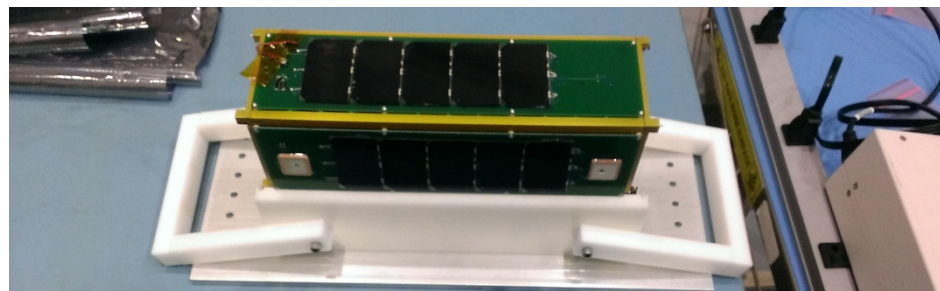
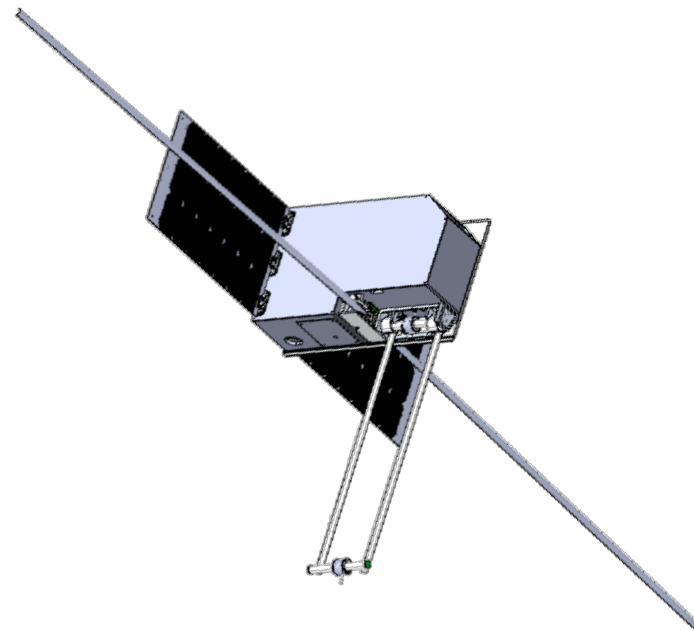


•Two different CubeSat Programs

- VPM
 - Highly capable
 - Long term program
- GEARRS
 - Quick turn
 - Specifically scoped mission

•How do you manage each end of the spectrum?

- Tailoring mission assurance to mission scope, goals, and constraints





VPM



- **Very Low Frequency and Particle Mapper**
 - 6U CubeSat (2Ux3U)
 - 4 year development
 - **Payload: VLF dipole receiver, High energy particle detectors**
 - Correlate VLF events to electron precipitation
 - Large program effort
 - Low risk posture
- **Approaching CDR stage in design**
 - Currently building Flat-Sat/EM
 - Developing payload maturity



GEARRS



- **GlobalStar Experiment and Risk Reduction Satellite (GEARRS)**
 - **Two radios**
 - GlobalStar Duplex
 - GlobalStar Simplex
 - **Characterize and map GlobalStar duplex and simplex coverage for LEO assets**
 - Provide 98% orbital TT&C capability for spacecraft @ 9600 bps
 - Demonstrate NSL commercial product for future AFRL missions
- **90 Day CubeSat**
 - December 2013 to March 2014
- **Collaboration between AFRL, Near Space Launch, Space Test Program, and NanoRacks**
 - NSL is the sole space GlobalStar Value Added Reseller (VAR) and built the satellite leveraging a previous bus design
 - AFRL performed mission planning, launch interface, and environmental stress screening
 - Space Test Program and NanoRacks



Early December



- **AFRL learned of a quick launch opportunity through the Space Test Program**
 - **March 2014 Delivery (~3 months from kickoff)**
- **AFRL was aware of the GlobalStar payload from a University NanoSat Program mission**
- **Hesitant to fly the radio on higher value programs without better understanding the on-orbit performance**
- **Perfect quick turn mission with well scoped payload and a documented user need (AFRL flight programs, UNP programs)**
 - **Can we fly a risk reduction in the required time frame?**
 - **How do we scope the mission assurance to accept risk, while still setting us up for mission success?**



GEARRS



•Kick-off

- December 17th 2013, D - 94
- Decided that the only way to converge on a March delivery:
 - Perform very targeted mission assurance
 - Focused (i.e. reduced) documentation and analysis
 - Increase hardware testing
 - Nano-team format
 - Everyone-knows-everything mentality
 - Everyone present at daily meetings
 - Only possible with a well scoped effort
 - Leverage prior hardware designs and COTS components
 - KISS



Kickoff Program Decisions



- 1. We don't have the time or budget to do all of this.**
- 2. How do we pick and choose?**

- Goddard Gold Rules
- MIL-STD-1540C
 - Random Vibration
 - Shock
 - Thermal Cycling
 - Thermal Vacuum
- Cleanliness
- ESD safe-guarding
- Requirement definition
- Requirement V&V
- Configuration control
 - Documentation
 - Software
 - Revision Control
- Signatures
- Design Reviews
- Configuration Control Boards (CCB)
- Management structure
- Engineering Change Requests (ECR)
- Functional Testing
 - Go/No-Go Tests
 - Day-in-the-life (DITL)
- Mate/De-mate logs
- Conformal Coating
- Etc.



Focus on the Essentials



GEARRS: Think Sputnik

- Target for 72 hour operations
 - Required coverage map fidelity can be generated in the first 72 hours
- Simple Payload
 - Leverage simplicity of mission
 - What is the bare minimum to achieve the required data products?
 - Don't overcomplicate it
- Soft-stow vibe loads

VPM: Not Sputnik

- Target for 1 year operations
 - Required measurement conjunctions with larger space systems
- Complex Payload
 - 3 deployables
 - Specific FoV requirements
 - Strict attitude pointing during science mission phases
 - Low data latency requirements

A simple payload allows for quick assessment of mission assurance trades



Documentation



GEARRS: Focus on essential documents

- Small, 4 man tiger team
- Everyone CC'd on every email and at every teleconference
 - Contributed to low-overhead and more face-to-face discussions
- Critical items still produced
 - SWaP Budgets
 - CONOPS
 - Mission Success Criteria
 - Requirement Verification Matrix
 - Integration and test procedures

VPM: Very document heavy

- Configuration control policies
 - PM/CE Signatures
 - Subsystem lead reviews
 - Configuration control boards after CDR -2 Months
- In-depth document production schedule
 - Design review oriented

A small team decreases documents and configuration control overhead



Analysis & Testing



GEARRS: Trade for testing

- One well scoped test is worth a 1000 simulations
 - For GEARRS, it was faster and cheaper to make an EM and ESS
- Exceed mission levels
 - Vibe to 9 grms
 - Thermal +/- 15
- PV Array I/V curve with controlled light source
- Leveraged known analysis from previous missions

VPM: Analyze everything, then test it

- Structural
- Thermal
- ADCS
- GNC
 - GPS antenna pointing analysis and gain patterns
- Science Data
 - Compare to MLTB and mission success criteria

Analysis *and* testing is preferred if resources allow, however testing alone *can* be a valid alternative



CONOPS



GEARRS: Simple software design

- **Two modes**
 - **Fast Beacon**
 - **Slow Beacon**
- **Always power positive**
- **Limited Commanding**
 - **Commands are restricted to changing a parameter**
 - **Change reported via simplex beacon**
- **Run DITL many, many times**

VPM: Robust software scheme

- **Safe mode accounts for most error sources**
- **Autonomous ‘Lights-out’ operation paradigm**
- **Various mission stages**
 - **LEO checkout**
 - **Payload checkout**
 - **Operations**

Simple payloads mean simple CONOPS



Targeted Mission Assurance



- **One size does not fit all**
 - Cannot apply one level of mission assurance to all CubeSats
 - Use the scope of the mission to determine
 - Risks that can be accepted
 - Risks that need to be mitigated or avoided
- **Constantly re-evaluate**
 - Need to be flexible
 - Keep the mission goals in mind and don't be afraid to cut pieces to meet them
- **GEARRS paradigm doesn't work for every system**
 - It can be done with the right payload, but not all of them
 - Many satellites require a lower risk posture
- **VPM paradigm doesn't work for every system**
 - It's applicable to some systems, but not most
 - Sometimes it's more than what the mission requires