



The CanX-4 and CanX-5 Formation Flying Mission

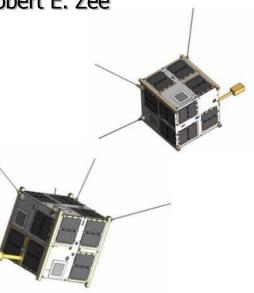
A Technology Pathfinder for Microsatellite and Small Satellite Constellations

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Presented by Ben Risi

August 10th, 2013





Mission Overview

- Two identical satellites
 - Less than 7 kg, cubic form factor 20 cm per side
- Launch on Polar Satellite Launch Vehicle (PSLV), late 2013
- Ejected from separate deployment systems on launch vehicle
- Demonstration of autonomous formation flight

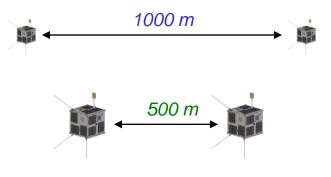
Performance	Minimum Requirement
Position Control	1 m
Relative Position Determination	10 cm
Minimum Relative Distance	50 m
Maximum Relative Distance	1000 m
Attitude Control	5°
Intersatellite Link Range	5 km
Intersatellite Link Data Rate	10 kbps



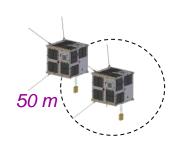
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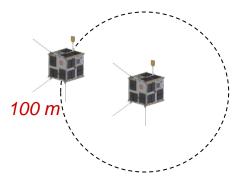
Formation Flying

- Two Formation Types:
 - 1. Along Track Orbit



2. Projected Circular Orbit

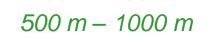






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Along Track Orbit



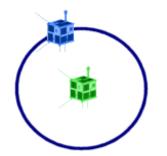
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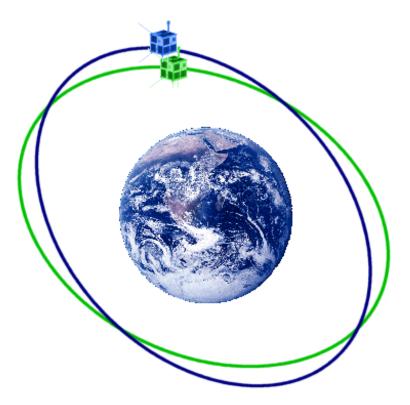


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Projected Circular Orbit

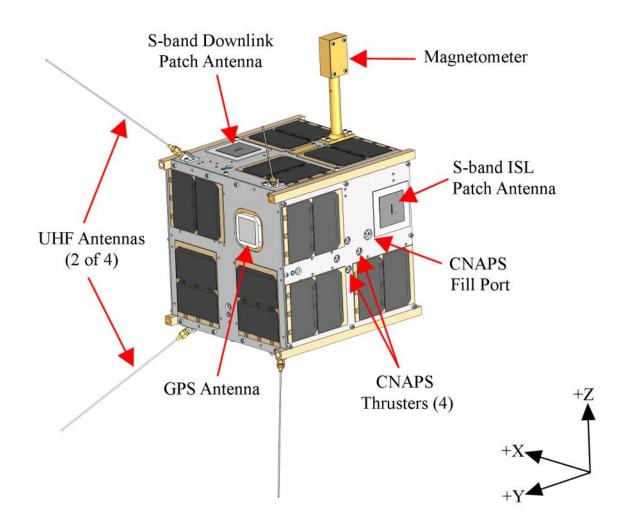
Viewed from Earth







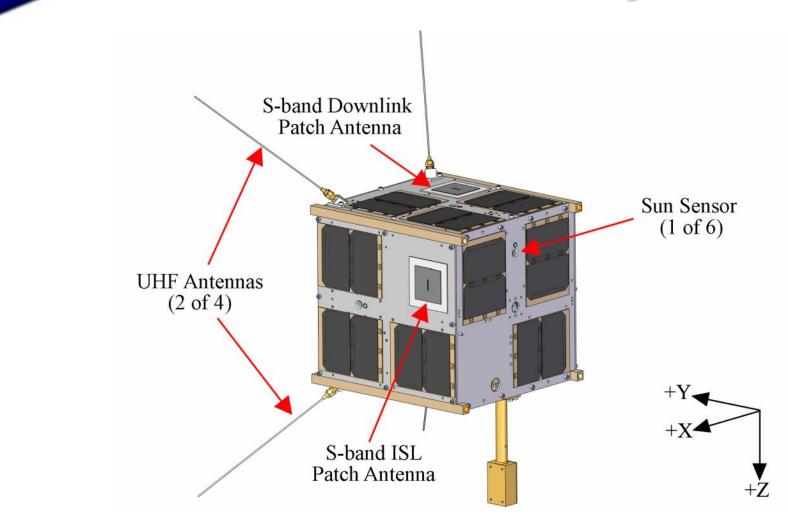
Satellite Layout



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Satellite Layout

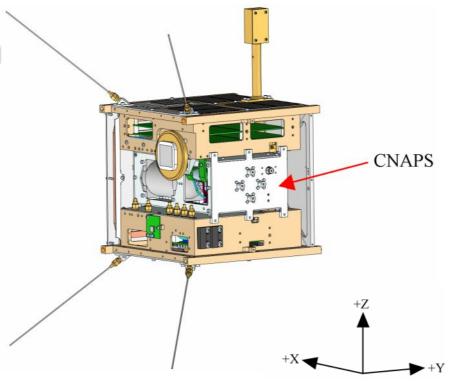




CanX-4/-5 Payloads

1. Propulsion System

- 2. GPS Receiver
- 3. Intersatellite Link



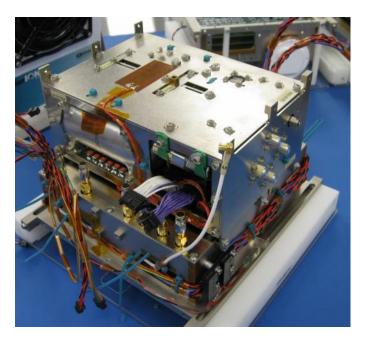




CNAPS

Canadian Nanosatellite Advanced Propulsion System

- Four-nozzle liquefied cold-gas thruster
- Sulfur hexafluoride (SF₆)
- I_{sp} > 35 seconds
- Thrust range of 10 to 40 mN
- Total ΔV ≈14 m/s



CNAPS-01 integration with CanX-5



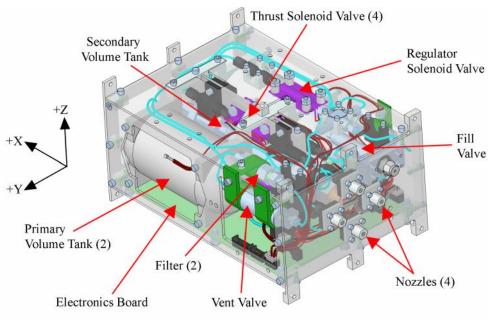


CNAPS

Canadian Nanosatellite Advanced Propulsion System



CNAPS-02 flight unit



CNAPS model



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GPS antenna

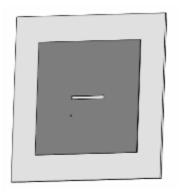


GPS receiver



CanX-4/-5 Payloads

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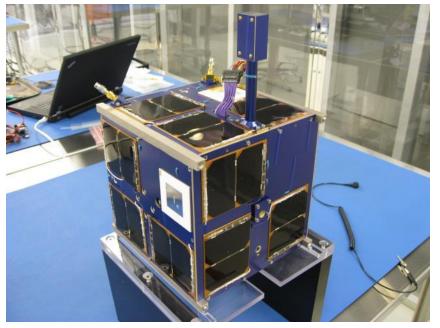


Intersatellite S-band radio transceiver

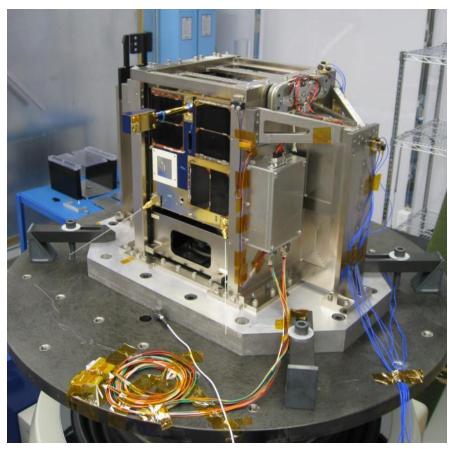


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Environmental Testing



Completed assembly of CanX-5



Vibration testing of CanX-5





Future Work

- Assembly of CanX-4 spacecraft
- Thermal vacuum testing of both spacecraft
- Launch planned for late 2013



UTIAS/SFL Thermal vacuum chamber

