



---

# **S-Band CubeSat Communications with the Open System of Agile Ground Stations (OSAGS)**

Kerri Cahoy, François Martel, Greg Huffman,  
Frank Hall Schmidt, Anne Marinan  
kcahoy@mit.edu  
fm@space.mit.edu

CubeSat Developer's Workshop  
August 12, 2012

---

- OSAGS S-Band Ground stations
  - Location
  - Control center
  - Heritage
  - Parameters
  - Current status
- CubeSat Payload Telemetry System
  - Parameters
  - Status
- Operations and licensing





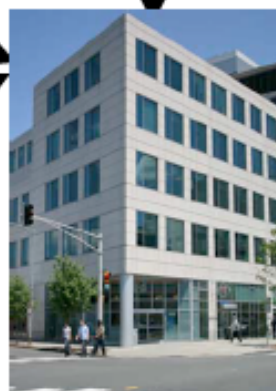
**Kwajalein**



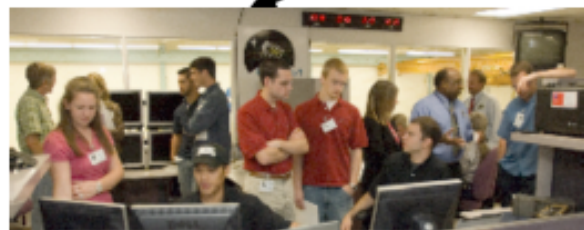
**Cayenne**



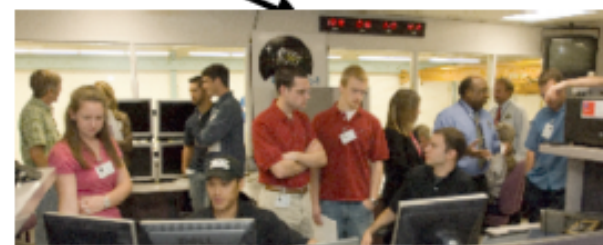
**Singapore**



**MIT NE 80**



**End-User #1  
Mission Operations Center**

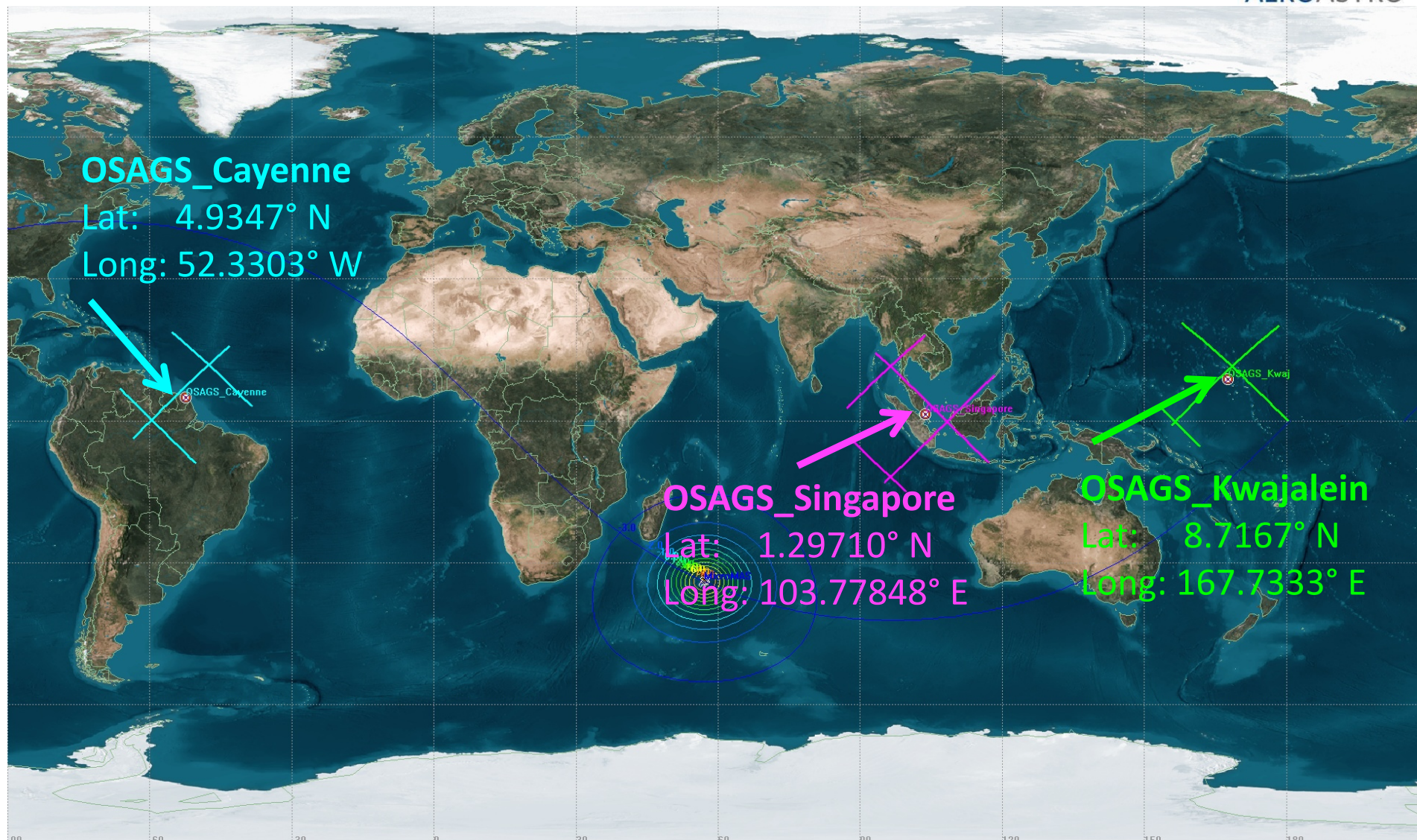


**End-User #2  
Mission Operations Center**

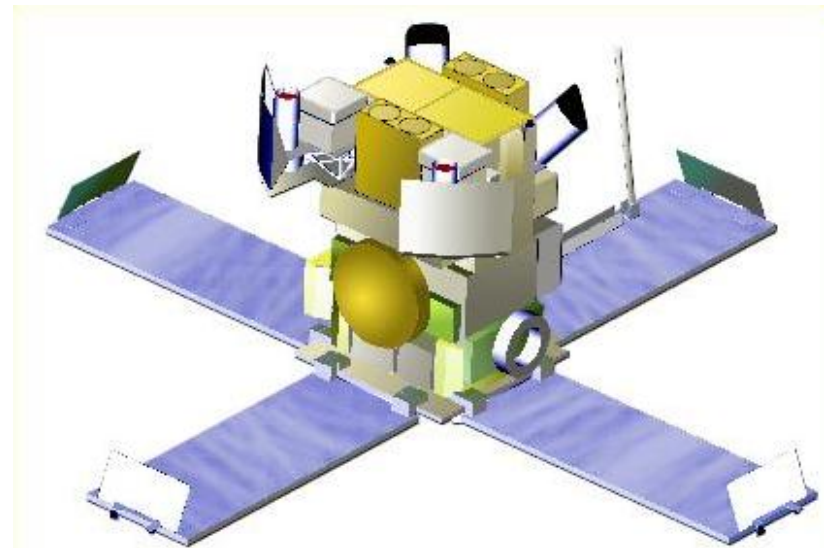
**Ground Station  
Maintenance &  
Management  
Center (GSMMC)**



# Location



- Ground stations originally successfully used to support the MIT HETE-2 mission
  - High Energy Transient Explorer (Oct. 2000)
- NASA SBIR with Espace to upgrade ground stations
  - Collaboration NASA ARC
  - Software defined radio
  - Available for nanosatellites and CubeSats





# Ground Station Parameters



Parameter	Value	Units
Uplink Frequency	2.025 – 2.120	GHz
Downlink Frequency	2.200 – 2.300	GHz
Antenna Diameter	2.3	m
Antenna Gain	31	dBi
Polarization	RHCP	-
Transmit Power	15.44	dBW
Data Rate	< 3.5	Mbits/sec
G/T	6.9	dB/K



# GS Link Budget Values



Parameter	Value	Units
Beamwidth	3.5	degrees
Efficiency	50	%
Back-lobe Gain	-30	dB
System Noise Temp	290	K
LNA Gain	0	dB
Antenna to LNA Loss	1	dB
LNA to Receiver Loss	0	dB
Pointing Loss*	1	dB

\*pointing 30% of main lobe beamwidth

- New:
  - 2.3 m antennas
  - Counterweights
  - Feed and feed arms
  - Diplexers (BPF)
  - LNA
  - 4 Ettus USRP2 **SDR** transceivers (redundancy)
- Reuses HETE-2 power amplifiers
- Can support several missions
  - 5 MHz NTIA S-band BW limit
  - Handles up to 3.5 Mbits/sec
- Remotely configurable







# Interface



The screenshot displays a web-based satellite tracking interface and a USRP2 FFT plot. The web interface includes a camera view of a satellite dish, control buttons for pan/tilt, and various tracking parameters.

**Tracking Parameters:**

Ground Sta : Kwajalein (KUS)	Date: Sun 31 Oct 10	Downlink Freq : 145.0500 MHz
Satellite : AIT	KST : 304 13:12:38	Uplink FREQ: 8.8800 MHz
Inclination: 97.847 deg		Downlink Loss: 132.8200 dB
Orbit : 19151 46.9 %	Object : 31304	Uplink Loss: 8.8800 dB
Run Az/El: 203.6 65.3 deg	Model : SEP+	Sun Angle : 158.1700 deg
Scan Az/El: 203.2 1.8 deg	Tracking: 01	Phase (256): 128.1800

**State Vector:**

X: -9811.392 km	Y: -5882.781 km	Z: 1006.339 km
Vx: -1.246 km/s	Vy: +8.193 km/s	Vz: -7.449 km/s

**Tracking History:**

Last AOS : 304/11:00:11 KST	AOS Azimuth : 17.488 deg
Duration : 8/88:18:18	REL Azimuth : 184.788 deg
Text LOS : 304/11:18:11 KST	LOS Azimuth : 187.888 deg
Count down : 8/88:05:11	Max Elevation: 72.188 deg
	REL Range : 628.188 km

**USRP2 FFT Plot:**

The plot shows a signal spectrum with a prominent peak at approximately 2.2825 GHz. The y-axis is labeled 'dB' and ranges from -50 to 10. The x-axis is labeled 'MHz' and ranges from 2.2818 to 2.2822. The center frequency is 2.2825 GHz and the gain is 35 dB.

**Options:**

- Average
- Persistence
- Peak Hold
- Set dB/div: 5 dB/div
- Adj Ref Lvl: +

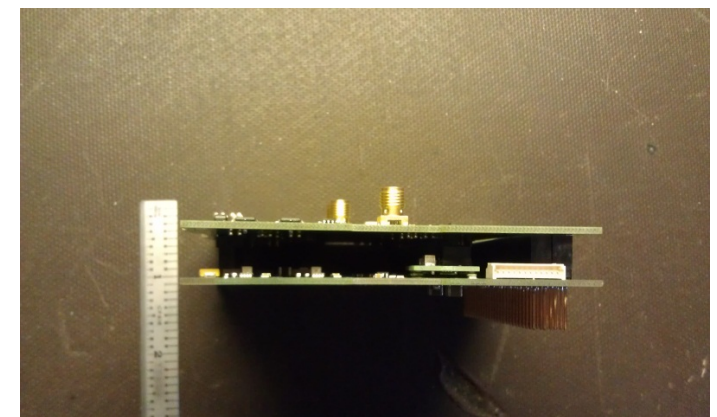
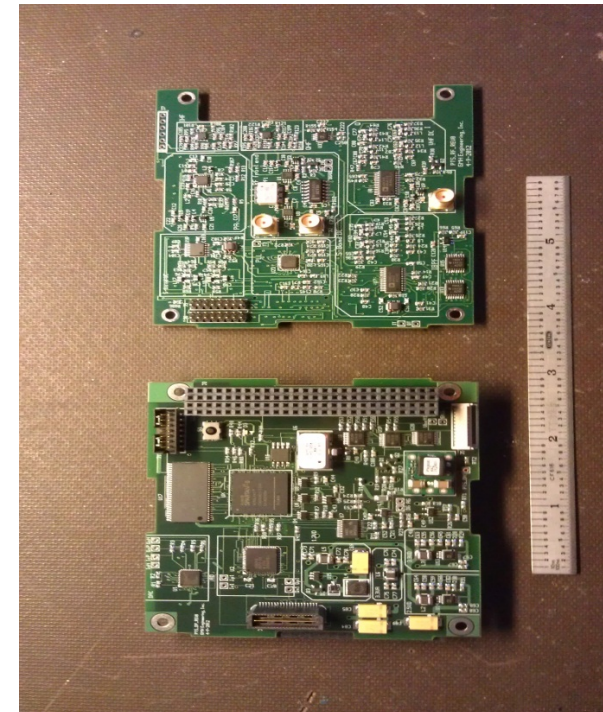


# S-Band Payload Telemetry System



- PTS
- RF Board + Digital Processing Board
  - 2 inputs, 2 outputs
  - Half-duplex

	Uplink	Downlink
Frequency	2.025-2.120 GHz	2.20-2.30 GHz
Data Rate	0.01 – 0.1 Mbps	0.01-1.0 Mbps
Power	2.0 W	3.6 W
Output Power	N/A	1.0 W
Modulation	BPSK, QPSK, OQPSK, CPFSK	
Standby Power: 0.75W		
Dimensions (LxWxH): 90.17mm x 95.89mm x 35mm		
Mass: 0.094kg		
Cost: \$6000 per unit		





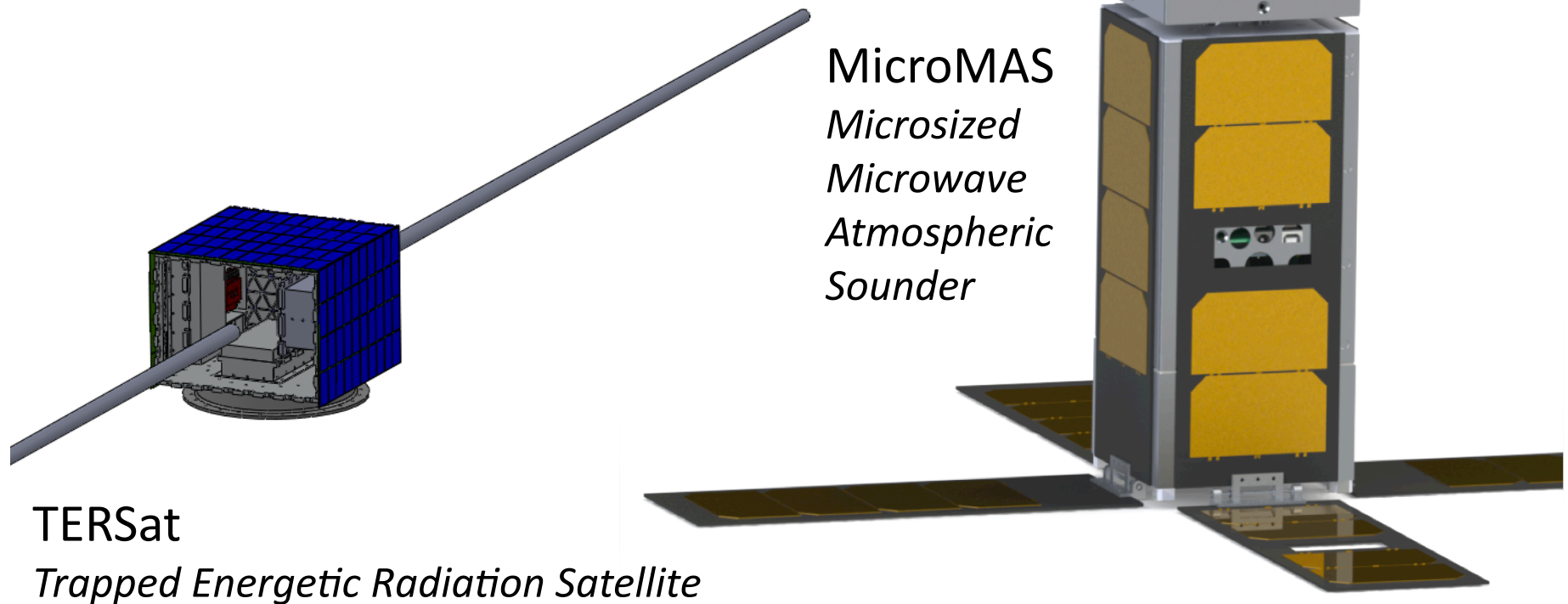
# Additional HF/VHF Rx Capability



- Takes GPS clock input

	HF Rx	VHF Rx
Frequency	100 kHz – 10 MHz selectable	60 MHz – 1 GHz selectable
Bandwidth	100 kHz – 10 MHz selectable	100 kHz – 10 MHz selectable
Sampling	14 bits I / Q up to 20 Ms/s	14 bits I / Q up to 20 Ms/s
Adjustable RF Gain	0 – 40 dB	0 – 44 dB
RF inputs	2	1
Power	1 W from 6 V supply	2.2 W from 6 V supply

- MicroMAS (MIT / MIT LL)
- TERSat (MIT / UNP-7)
- Mothercube (MIT / AFS)

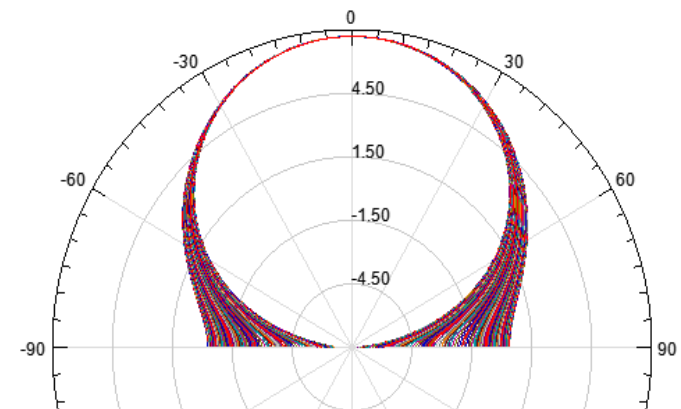
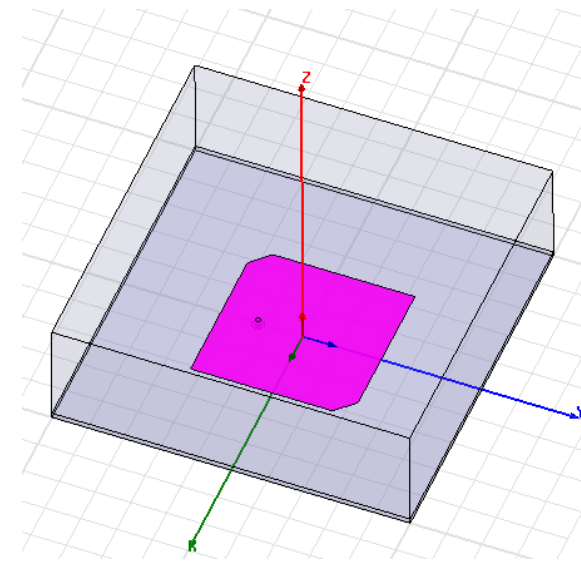




# S-Band CubeSat Antenna Design



- Two custom patch antennas
  - Uplink, downlink
  - Truncated corners – RHC
  - Probe feed
    - SMA – coax – PTS board
  - Dielectric RT Duroid 5880
    - Thickness 1.57 mm
    - $\epsilon_r = 2.2$
  - Mount on nadir facing body panels





# S-Band CubeSat Antenna Design



Parameter	Uplink Antenna	Downlink Antenna
Length	47 mm	43 mm
Corner Truncation	5.4 mm	4.9 mm
Center Frequency	2.088 GHz	2.27 GHz
Return Loss	-21.2 dB	-16.02 dB
Gain	7.20 dBi	7.45 dBi
Half-Power Angle	85 deg	84 deg
Bandwidth	38 MHz	35 MHz
Mass	11.2 g	11.08 g
Price	\$350	\$350

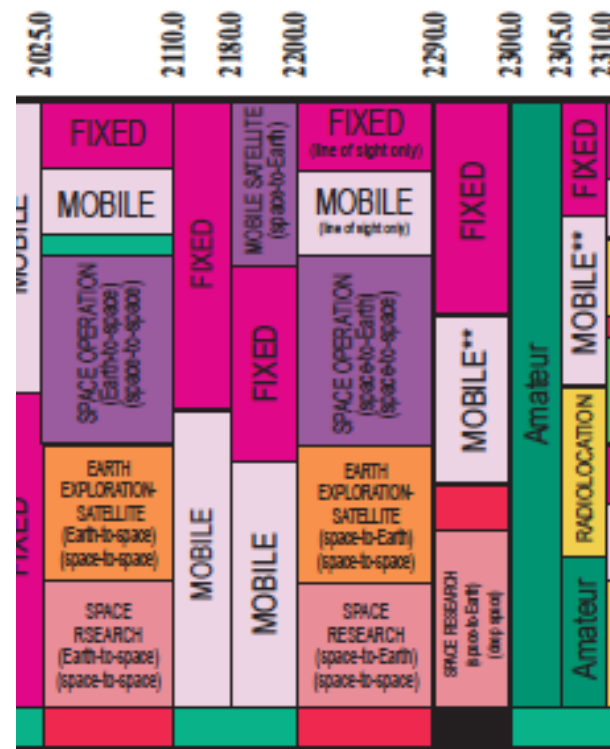


# MicroMAS Link Budget S/C



- Frequency = 2.2 GHz
- Power = 0 dBW
- Antenna – Nadir Facing (Gaussian Gain Pattern)
- 3dB Beamwidth =  $70^\circ$
- Main Lobe Gain = 7.4 dBi
- Efficiency = 50 %
- Polarization: RHCP
- Coded Data Rate: 694 kbps
- Information Rate: 347 kbps
- Modulation: OQPSK

- 2.2—2.3 GHz S-band
- Gov't. rights to spectrum
- Two approaches
  - DD-1494 with gov't. sponsor
  - FCC Commercial Experimental License
- Foreign ground stations
  - OSAGS has established representatives at Singapore and Cayenne (France)







# MicroMAS Coverage Study



Max P/L Data Rate to Bus & Required Storage Onboard Bus

ASSUME Constant 19.2 kbps

7/20/2012

CBC

	0°		30°		42°		60°		90°						
300km	h = 300km, i = 0°	Gap	Access	h = 300km, i = 30°	Gap	Access	h = 300km, i = 42°	Gap	Access	h = 300km, i = 60°	Gap	Access	h = 300km, i = 90°	Gap	Access
	Mean (s)	2256.4	332.1	Mean (s)	6442.2	282.3	Mean (s)	8648.7	280.1	Mean (s)	10953.0	273.7	Mean (s)	12289.7	267.3
	Max (s)	2904.1	384.3	Max (s)	30314.9	412.0	Max (s)	30457.6	408.9	Max (s)	30476.2	402.7	Max (s)	30421.8	390.3
	Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200	
	Max P/L Rate (bps)	44000		Max P/L Rate (bps)	14500		Max P/L Rate (bps)	10750		Max P/L Rate (bps)	8250		Max P/L Rate (bps)	7750	
	Req. Storage (MB)	18.50		Req. Storage (MB)	63.00		Req. Storage (MB)	58.00		Req. Storage (MB)	49.25		Req. Storage (MB)	41.13	
400km	h = 400km, i = 0°	Gap	Access	h = 400km, i = 30°	Gap	Access	h = 400km, i = 42°	Gap	Access	h = 400km, i = 60°	Gap	Access	h = 400km, i = 90°	Gap	Access
	Mean (s)	2396.4	425.9	Mean (s)	5579.3	346.5	Mean (s)	7486.1	342.4	Mean (s)	9589.5	337.3	Mean (s)	10858.1	326.6
	Max (s)	3057.9	479.7	Max (s)	25229.3	501.7	Max (s)	31255.9	497.8	Max (s)	31241.6	490.1	Max (s)	41864.1	474.6
	Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200	
	Max P/L Rate (bps)	52000		Max P/L Rate (bps)	20250		Max P/L Rate (bps)	15100		Max P/L Rate (bps)	11750		Max P/L Rate (bps)	10000	
	Req. Storage (MB)	22.25		Req. Storage (MB)	79.88		Req. Storage (MB)	72.75		Req. Storage (MB)	60.25		Req. Storage (MB)	46.38	
500km	h = 500km, i = 0°	Gap	Access	h = 500km, i = 30°	Gap	Access	h = 500km, i = 42°	Gap	Access	h = 500km, i = 60°	Gap	Access	h = 500km, i = 90°	Gap	Access
	Mean (s)	2528.6	510.3	Mean (s)	4921.2	398.7	Mean (s)	6823.4	403.4	Mean (s)	8374.6	365.1	Mean (s)	9701.7	377.5
	Max (s)	3204.6	566.6	Max (s)	25842.9	584.8	Max (s)	26050.6	580.0	Max (s)	31984.8	570.8	Max (s)	37043.6	552.3
	Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200	
	Max P/L Rate (bps)	58000		Max P/L Rate (bps)	26000		Max P/L Rate (bps)	19400		Max P/L Rate (bps)	14600		Max P/L Rate (bps)	13100	
	Req. Storage (MB)	25.63		Req. Storage (MB)	104.50		Req. Storage (MB)	105.88		Req. Storage (MB)	161.00		Req. Storage (MB)	91.00	
600km	h = 600km, i = 0°	Gap	Access	h = 600km, i = 30°	Gap	Access	h = 600km, i = 42°	Gap	Access	h = 600km, i = 60°	Gap	Access	h = 600km, i = 90°	Gap	Access
	Mean (s)	2653.0	588.1	Mean (s)	4170.2	414.4	Mean (s)	5765.4	414.8	Mean (s)	7290.7	404.6	Mean (s)	8435.5	401.9
	Max (s)	3347.1	648.1	Max (s)	23049.7	663.1	Max (s)	26736.2	657.5	Max (s)	26835.6	646.7	Max (s)	29547.7	625.3
	Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200	
	Max P/L Rate (bps)	62000		Max P/L Rate (bps)	31100		Max P/L Rate (bps)	23000		Max P/L Rate (bps)	18100		Max P/L Rate (bps)	15800	
	Req. Storage (MB)	28.13		Req. Storage (MB)	97.38		Req. Storage (MB)	92.63		Req. Storage (MB)	83.38		Req. Storage (MB)	96.75	
700km	h = 700km, i = 0°	Gap	Access	h = 700km, i = 30°	Gap	Access	h = 700km, i = 42°	Gap	Access	h = 700km, i = 60°	Gap	Access	h = 700km, i = 90°	Gap	Access
	Mean (s)	2755.8	642.7	Mean (s)	4452.9	497.6	Mean (s)	6205.4	499.4	Mean (s)	7719.5	478.9	Mean (s)	8967.5	478.2
	Max (s)	3463.1	702.1	Max (s)	20865.7	715.5	Max (s)	27392.3	709.2	Max (s)	30099.6	697.5	Max (s)	27540.7	673.9
	Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200		Req P/L Rate (bps)	19200	
	Max P/L Rate (bps)	65000		Max P/L Rate (bps)	34800		Max P/L Rate (bps)	25800		Max P/L Rate (bps)	20200		Max P/L Rate (bps)	17700	
	Req. Storage (MB)	30.63		Req. Storage (MB)	104.50		Req. Storage (MB)	102.50		Req. Storage (MB)	93.88		Req. Storage (MB)	82.75	

\* Req. P/L rate is the information rate output by the P/L assuming a constant 19.2kbps output rate from the P/L.

\*\*Max P/L (bps) is the **maximum** amount of payload data (information) per second that can be **continuously** sent to the bus and downlinked without interruption.

\*\*\*Req. Storage (MB) is the **maximum** amount of payload mission data that needs to be held for downlink at any given time assuming continuous payload operation @ the Max P/L Rate.

## Assumptions:

MicroMAS Information Downlink Rate: 347kbps // Collection occurs during downlink time // **NO COMPRESSION**

Analysis shown is for 30 day period // **Link Constraint: Eb/No > 7.5 dB, Elevation > 5°**

- Effort to refurbish S-band dish on MIT building 54
  - Led by EAPS (Sara Seager), AeroAstro (Kerri Cahoy) and MIT Radio Club
  - 42.359758, -71.093556



MIT Hack the day Hitchhikers Guide to the Galaxy came out: “Don’t Panic”



- Open System of Agile Ground Stations
  - S-band
  - Increase data downlink from CubeSats
- Espace Payload Telemetry System
  - Fall 2012
- MIT Control Center; networked
- MIT Campus ground station refurbishment
- Contact:
  - [fm@space.mit.edu](mailto:fm@space.mit.edu) (François Martel)
  - [kcahoy@mit.edu](mailto:kcahoy@mit.edu)



# Acknowledgements



- MIT and MIT LL Microsized Microwave Atmospheric Satellite(MicroMAS) team
  - Especially IO, CC, and RK
- MIT Trapped Energetic Radiation Satellite (TERSat) team
- MIT Kavli Institute
- MIT and Aurora Flight Sciences Mothercube team

- *Anne Marinan*
  - *Ad-hoc CubeSat Constellations*
    - 11:30 am Saturday, 8/11
- Emily Clements
  - Trapped Energetic Radiation Satellite PQR (TERSat)
    - 9:25 am Monday, 8/13
- Bill Blackwell
  - Nanosatellites for Earth Environmental Monitoring:  
The MicroMAS Project
    - 3:15 pm Monday, 8/13
- Emily Clements
  - Trapped Energetic Radiation Satellite
    - 8:30 am Wednesday, 8/15
- Come say hi and visit our TERSat Booth!



# Backup Slides

---



- The Open System of Agile Ground Stations (OSAGS) consists of 3 equatorial S-band ground stations located in Kwajalein, Cayenne, and Singapore. The new OSAGS is the result of updates in 2010—2011 to the previously existing HETE-2 (High Energy Transient Explorer) ground stations by Espace, Inc., and was supported by the NASA Small Business Innovation Research (SBIR) program. OSAGS operates in the space exploration and operation bands of 2.025—2.0120 GHz uplink and 2.20—2.30 GHz downlink. The 2.3 m OSAGS ground stations can support numerous missions, handling communications requirements up to 3.5 Mbits per second. The system is based on Software Defined Radio (SDR) and the three ground stations are remotely configurable via the internet from a secure portal located on campus at MIT. The Espace, Inc. Payload Telemetry System (PTS) currently consists of a CubeSat-sized S-band transceiver, with additional HF/VHF/UHF receive capability. The prototype PTS units have been developed, and the first flight PTS units are scheduled for delivery in August 2012. We discuss applications of both the PTS and OSAGS for CubeSat missions, as well as operational considerations.



# Link Budget



## • Detailed link budget

MicroMAS Link Budget							
h=500 km      i = 42°							
Item	Symbol	Units	Downlink		Uplink		Comments
			Worst Case	Best Case	Worst Case	Best Case	
<b>EIRP:</b>							
Transmitter Power	P	dBW	0.00	0.00	15.44	15.44	0 dB is power output for Espace PTS Radio.
Transmitter Line Loss	L <sub>t</sub>	dB	0.00	0.00	0.00	0.00	TBR: Attenuation and length of cable needed for accurate calculation (Assume negligible for now)
Transmit Antenna Gain (net)	G <sub>t</sub>	dBi	1.00	7.40	31.68	31.68	TBR: Downlink transmit antenna gain is the requirement for onboard patch antennas. Antennas design is underway.
Equiv. Isotropic Radiated Power	EIRP	dBW	1.00	7.40	47.12	47.12	See SMAD eq. (13-5)
<b>Receive Antenna Gain:</b>							
Frequency	f	Ghz	2.25	2.25	2.08	2.08	Downlink Frequency: 2.25 GHz, mid point of 2.2 - 2.3 GHz range of OSAGS ground stations. Uplink Frequency is 2.075 GHz. Mid point of 2.025 - 2.120 GHz for OSAGS ground stations.
Receive Antenna Diameter	D <sub>r</sub>	m	2.30	2.30	0.05	0.05	2.3 m diameter for OSAGS from Espace-OSAGS-PTS-Slides-2-21-2012.pdf from Francois Martel.
Receive Antenna efficiency	η	n/a	0.50	0.50	0.50	0.50	General Assumption
Receive Antenna Gain	G <sub>r</sub>	dBi	31.68	31.68	6.00	-10.00	See SMAD eq. (13-18b)
<b>Free Space Loss:</b>							
Propagation Path Length	S	km	2,076.00	500.00	2,076.00	500.00	Based on Max Range from STK Analysis (Assuming a cutoff of Eb/No > 7.5, or Elevation > 5°)
Free Space Loss	L <sub>s</sub>	dB	-165.84	-153.47	-165.13	-152.77	See SMAD eq. (13-23b)
<b>Transmission Path and Pointing Losses:</b>							
Transmit Antenna Pointing Loss	L <sub>pt</sub>	dB	0.00	0.00	-1.00	0.00	Assume no antenna pointing loss on S/C since antenna is always nadir facing and does not point to the ground station.
Receive Antenna Pointing Loss	L <sub>pr</sub>	dB	-0.50	-0.50	-0.50	0.00	Assumed based on conversations with OSAGS engineer.
Ionospheric Loss	L <sub>ion</sub>	dB	-1.00	0.00	-1.00	0.00	Best: Assumed 0 to match STK simulation. Worst: Based on 3 years of atmospheric attenuation research prior to CASTOR program (CASTOR link budget)
Atmospheric Loss (H2O and O2 losses)	L <sub>atmo</sub>	dB	-0.34	-0.34	-0.34	-0.34	Based on 3 years of atmospheric attenuation research prior to CASTOR program (CASTOR link budget)
Loss due to Rain	L <sub>rain</sub>	dB	-2.00	-0.01	-2.00	-0.01	Based on 3 years of atmospheric attenuation research prior to CASTOR program (CASTOR link budget)
Demodulator Loss	L <sub>dmd</sub>	dB	-0.15	0.00	-0.15	0.00	"derived from generic communication knowledge" (CASTOR link budget). TBR - Plan to Close loop with PTS Designer
Splitter Loss	L <sub>spl</sub>	dB	0.00	0.00	0.00	0.00	Current MicroMAS design uses two patch antennas, one for uplink and one for downlink.
Implementation Loss		dB	-0.50	0.00	-2.00	-0.50	Best: Assumed 0, Worst: Assumed -2dB for OSAGS, -0.5dB for MicroMAS
Total Additional Losses		dB	-4.49	-0.85	-6.99	-0.85	
<b>Data Rate:</b>							
Data Rate	R	bps	694,444.00	694,444.00	25,600.00	25,600.00	Espace PTS max downlink data rate is 1 Mbps. Max uplink frequency is 0.1 Mbps.
Data Rate	10 log( R )	dBbps	58.42	58.42	44.08	44.08	Original data downlink calculations were done with 115,200 bps, but MicroMAS will need as high a data rate as possible
<b>Boltzman's Constant:</b>							
Boltzman's Constant	10 log(k)	dBW/(Hz*K)	-228.60	-228.60	-228.60	-228.60	
<b>System Noise Temperature:</b>							
Antenna Noise Temperature	T <sub>ant</sub>	K	290.00	290.00	340.00	340.00	Conservative Estimate Uplink: per Alessandra's suggestions (290K from Earth + 50K from Cosmic Background = 340K)
Receiver Noise Temperature	T <sub>r</sub>	K	0.00	0.00	0.00	0.00	
System Noise Temperature	T <sub>s</sub>	K	290.00	290.00	340.00	340.00	Conservative Estimate
System Noise Temperature	10 log(T <sub>s</sub> )	dBK	24.62	24.62	25.31	25.31	Conservative Estimate
E <sub>b</sub> /N <sub>0</sub>		dB	7.91	30.31	40.20	42.70	See SMAD eq. (13-14)
E <sub>b</sub> /N <sub>0</sub> required		dB	7.50	7.50	7.50	7.50	Required for BER = 10 <sup>-5</sup> . An Eb/No of 7.5 will yield a BER = 10 <sup>-5</sup> . This assumes no coding gain. Once we take credit for the coding gain, the numbers should get better.
Margin		dB	0.41	22.81	32.70	35.20	Looking for a value greater than or equal to 3dB, which we achieve once we take credit for the coding gain of approx 5.5dB (not shown here).





# STK Link Budget Parameters



- Baseline Scenario Assumptions
  - $h = 500\text{km}$ ,  $T = 94.8$  minutes
  - $i = 42^\circ$
  - Scenario Time Period: 30 Days
- Rain Model: ITU-R P618-9
  - Rain Outage: 0.1%
  - Surface Temp:  $24^\circ\text{C}$
- Atmospheric Absorption Model
  - Water Vapor:  $7.5\text{g}/\text{m}^3$
  - Surface Temp:  $24^\circ\text{C}$



# 30 Day Ground Trace

