

Applications of NanoSats at Small Body Objects



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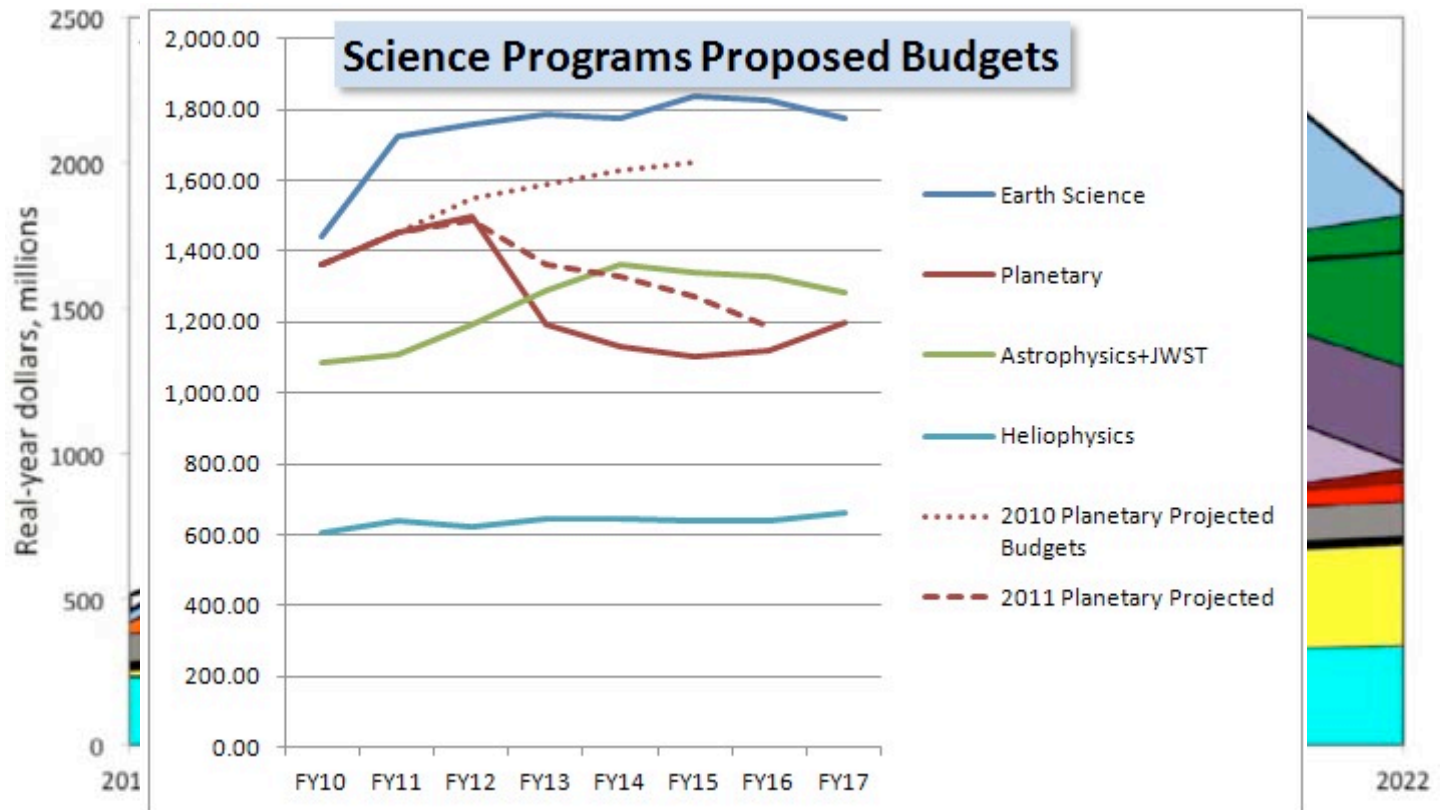
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Currently Flying Overhead...

- MSL - \$2.5 Billion
 - Cassini - \$3.27 Billion (with some contributions)
 - Dawn - \$446 Million
- + more than 25 other missions JPL is involved in (Voyager is still performing science!)



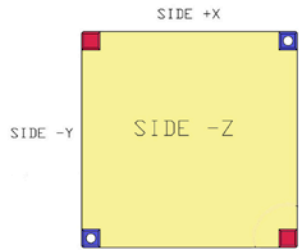
What is a CubeSat?

PLACEMENT OPTIONS FOR DEPLOYMENT SWITCHES AND SEPARATION SPRINGS

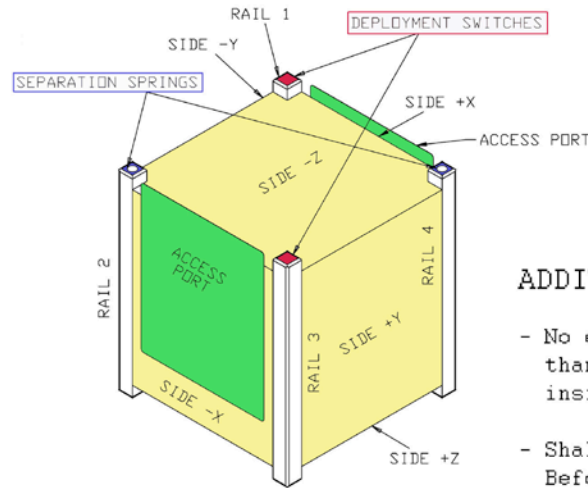
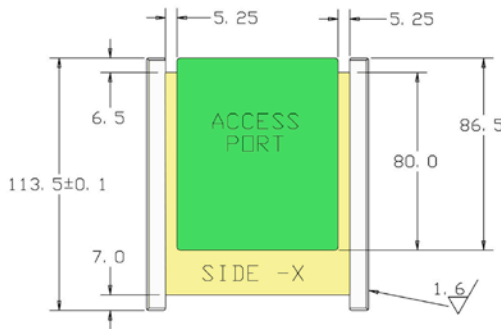
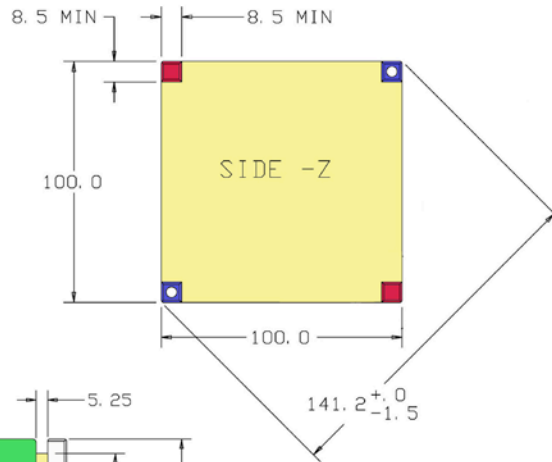
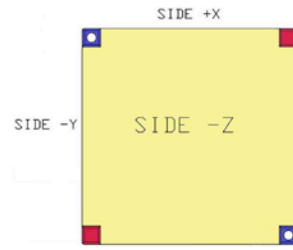
■ = DEPLOYMENT SWITCH

■ = SEPARATION SPRING

OPTION A



OPTION B

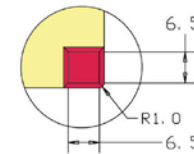


NOTE: Deployment switch and separation spring placement schemes shown in Option A and B. Deployment switch(es) should be compatible with +Z contact points.

ADDITIONAL NOTES:

- No external components other than the rails shall touch the inside of the P-POD.
- Shall incorporate a Remove Before Flight pin OR launch with batteries fully discharged.
- Rails shall be aluminum hard anodized.
- At least one (1) deployment switch shall be incorporated on all CubeSats.
- Center of gravity shall be located within a sphere of 2 cm from its geometric center.
- Separation springs can be found at McMaster Carr (P/N 84985A76).

CONTACT DETAIL FOR SIDE +Z



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	CALIFORNIA POLYTECHNIC STATE UNIVERSITY AEROSPACE ENGINEERING DEPARTMENT SAN LUIS OBISPO, CA 93407 (805) 756 - 5087	
ROUND ALL EDGES AND CORNERS.	DATE: August 1, 2009	NOT TO SCALE
+0.1 mm OR BETTER		



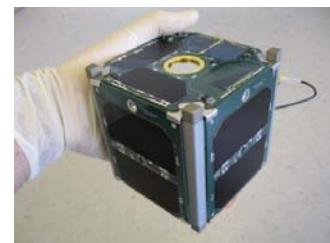
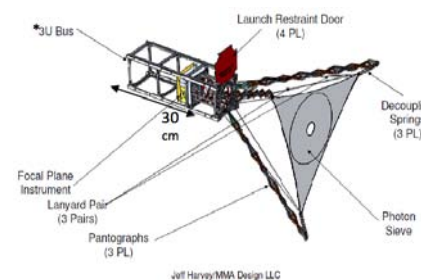
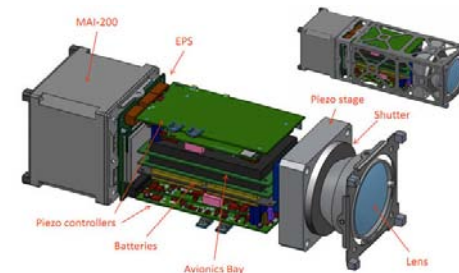
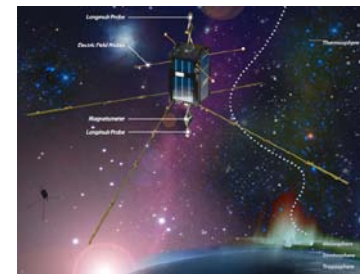
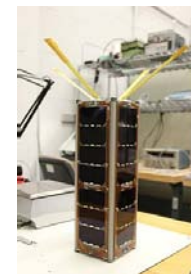
What is a CubeSat?

- A CubeSat is an accepted standard to enable low-cost access at the price of higher risk – but not to the primary **(encapsulation of risk)**
- **A CubeSat is a flexible platform** without defined “innards”
- **A CubeSat is an instrument** – with a few spacecraft parts tacked on.

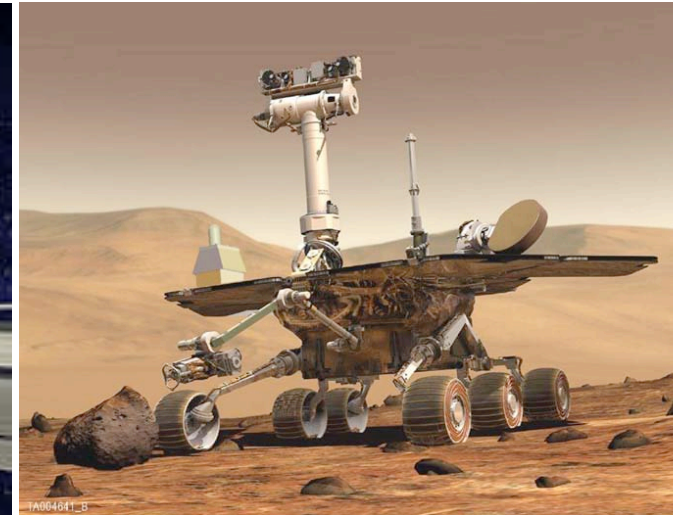
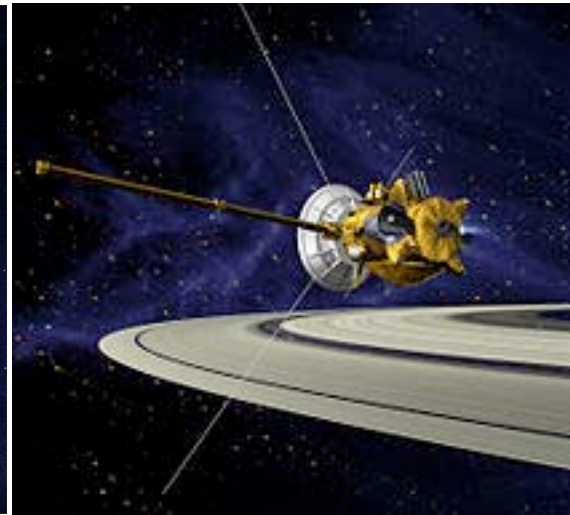


CubeSats – Example Missions

- Space Weather Science (NSF missions)
- Biological Science (NASA ARC missions)
- Astrophysics (Moorehead States's CXBN)
- Planetary Science (MIT's ExoplanetSat)
- Technology Demonstrations
 - Propulsion / Attitude (Nanosail-D)
 - Imagers (FalconSat-7)
 - Solar Arrays (NPS-SCAT)
 - FPGAs (M-Cubed/COVE)
 - Fractionated Space (DARPA F6)
 - Plug-n-Play architectures (AFRL)

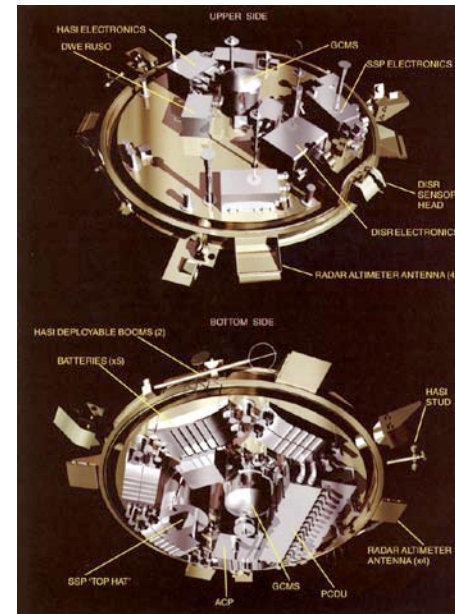


But JPL does BIG things!



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.... Except for these:

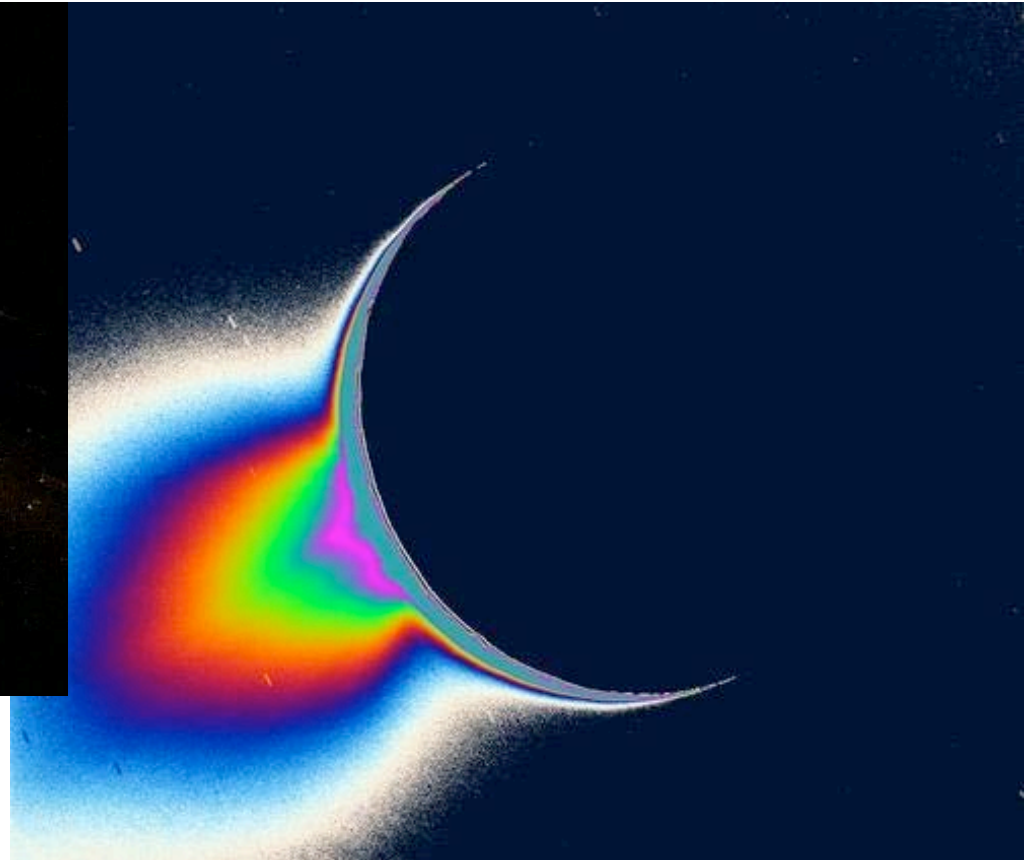
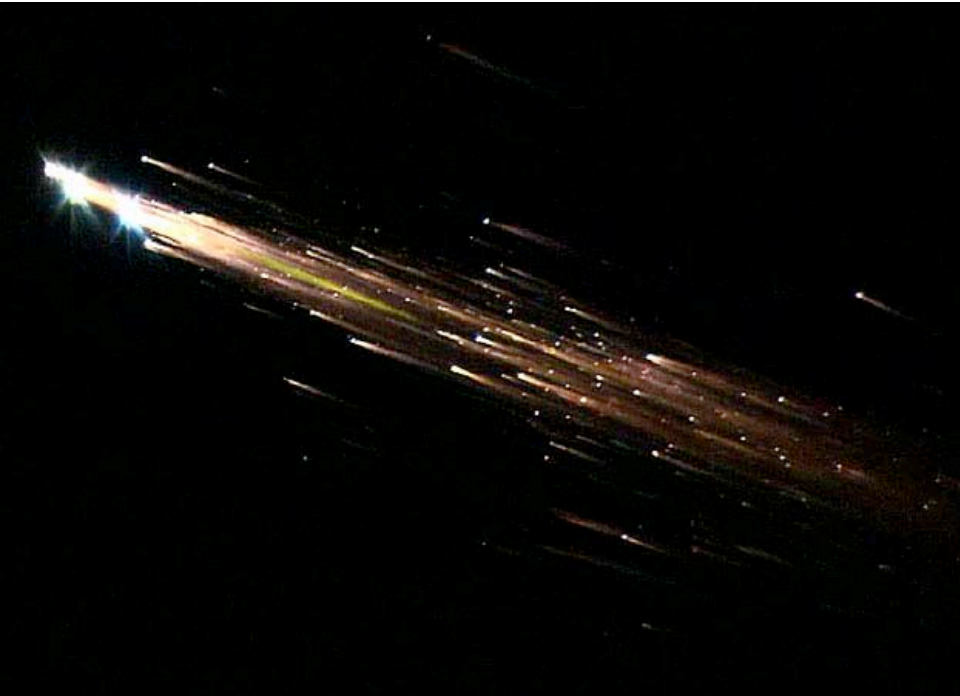


Motivation to Think Small



It's a Scary Solar System

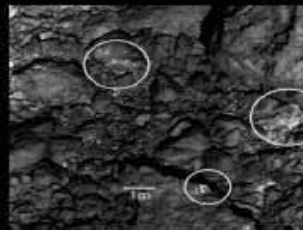
- NanoSatellites may be useful at other places in the solar system
- Is there compelling science enabled by this platform?



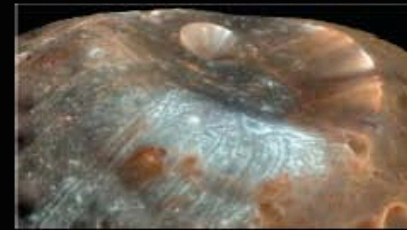
THEME

HUMAN
EXPLORATION

Potential Landing Sites



Phobos



WORKINGS
OF SOLAR
SYSTEMS

*Cometary
Vents*



*Cryoflow
on comet*



*Craters
on Vesta*



BUILDING
NEW
WORLDS

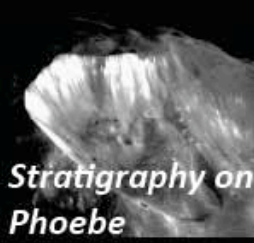
*Comet
Grain*



*Volatile
Mineralogy*



*Stratigraphy on
Phoebe*

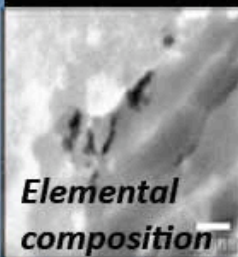


Ice on Tempel 1

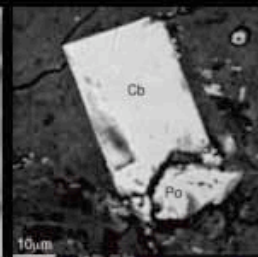


PLANETARY
HABITATS

*Elemental
composition*



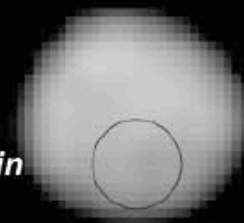
10µm



*Cubanite --
Stardust*



*Large basin
on Pallas*



nm

µm

mm

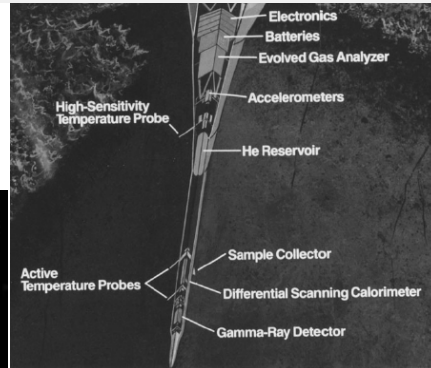
cm

m

km

SCALE

Interplanetary NanoSats



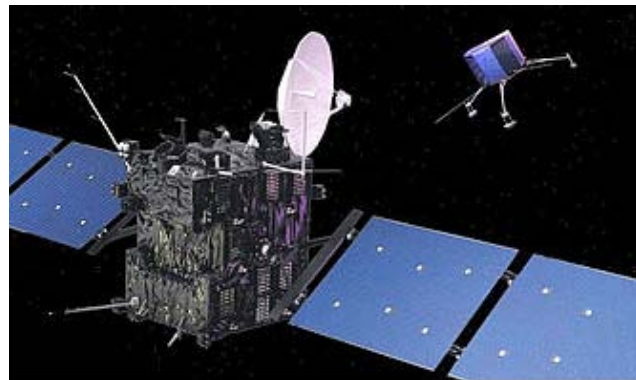
Penetrator(s) and Orbiter (e.g., CRAF)



Mothership and tailored hopper (e.g., Hayabusa/Minerva)

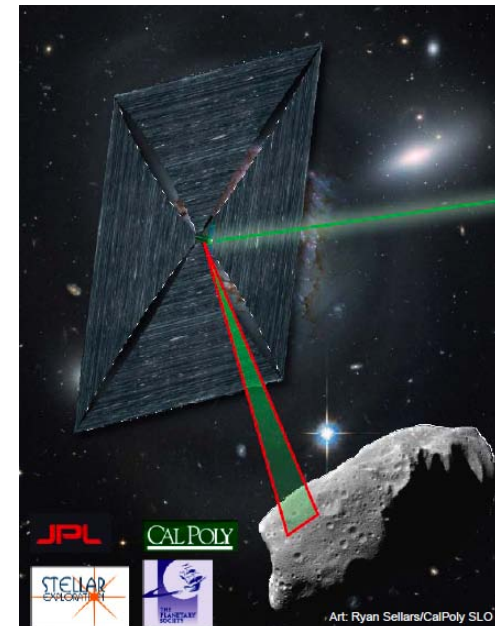
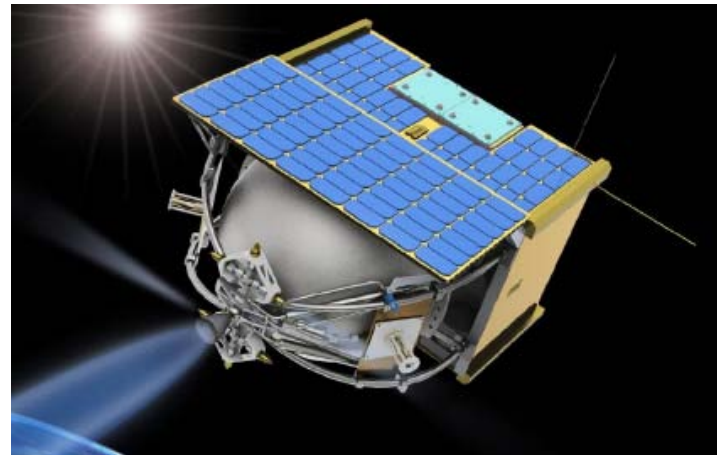


Single Cubesats (e.g., NIAC)



Mothership and tailored static lander (e.g., Rosetta/Philae)

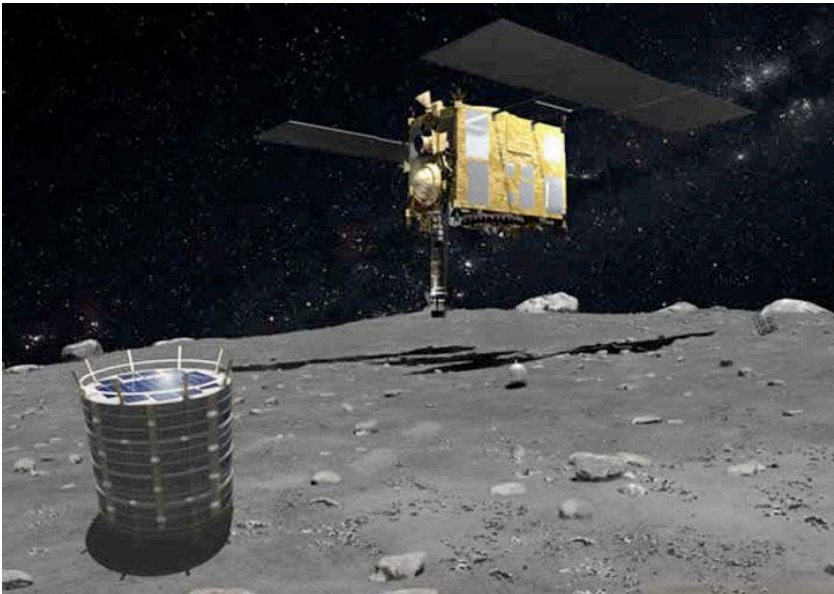
Mothership and multiple Cubesats (e.g., Planetary Hitchhiker)



NanoSats and Planetary Exploration

- **Motivation**

- 2013 Planetary Science Decadal Survey shows high-interest in small-body objects: both close proximity and in situ observations
- The next 5-10 years may provide several mission opportunities
- Several nano-spacecraft (Minerva and Muses-CN) have been designed for small-body exploration
- The CubeSat community has experience with nano-spacecraft – Can we leverage this to explore small-bodies (or the solar system) in a whole new way?



Key Science Drivers for Small, Flexible Spacecrafts

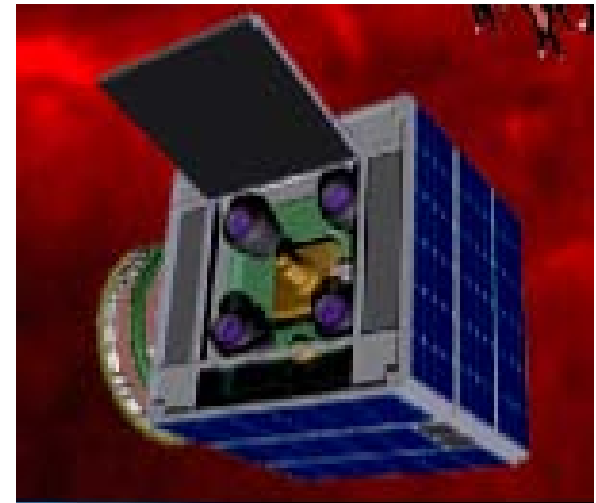
- Capture the diversity of small bodies
 - Regionally (e.g., satellite systems)
 - Globally
- Need for diversified and coordinated exploration
- Characterize surface properties at all scales
 - Heterogeneity (physical and chemical)
 - Exploration of outstanding, discrete features vs. global context
- Characterize z-dimension
 - Local environment (dust, exosphere, radiations)
 - Subsurface (gravity structure, water)
- Visit high-risk/high-science return areas

Diversified and Coordinated Exploration

- Systematic, self-consistent approach for chemistry measurements
 - Elemental composition (e.g., neutron detection), isotopic ratios (e.g., mass spectrometry)
 - Correlation between remote observations, theory, and reality
- Changing the way geophysics is done from space
 - Local or regional gravity/magnetic field characterization
 - Assess risk by using expendable assets, e.g., dust dynamics, electrostatic charging, etc.

Interplanetary NanoSat Definitions

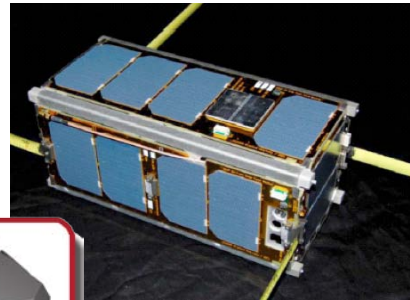
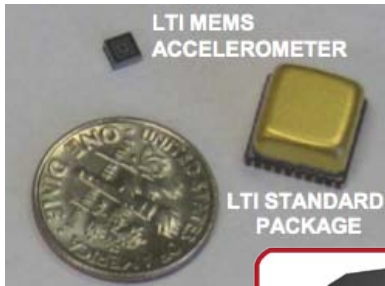
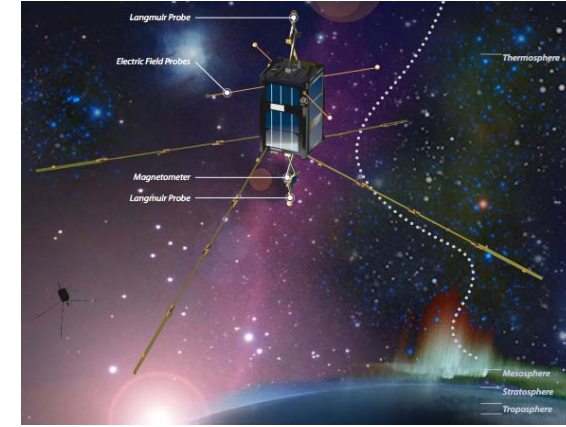
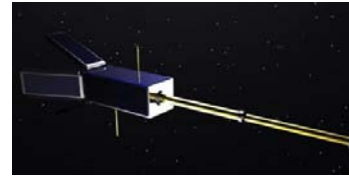
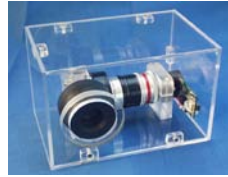
- On the order of 10 kg and 6U in volume
- Propulsion
 - Primary Propulsion (Direct from LEO) – See Rob Staehle’s NIAC research
 - Secondary “Propulsion” (Launched from Mothership)
- Communications
 - Direct-to-Earth
 - Relay through Mothership
- Power
 - Solar Power (Inner Solar System)
 - Primary Batteries



CubeSat Instruments/Sensors

Instrument Examples

- Imagers (Vis, IR, Hyperspectral)
- Spectrometers
- Bolometers
- Field/particle sensors
- Radio science
- Magnetometers



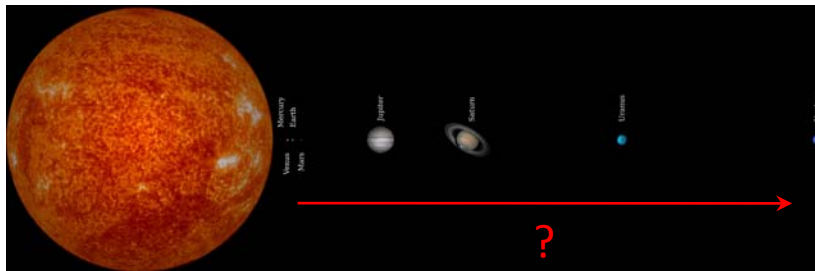
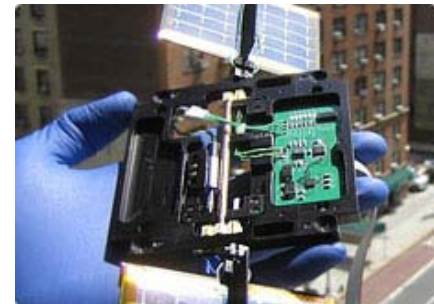
Navigation Sensor Examples

- Star trackers
- Sun sensors
- Horizon sensors
- MEMS gyros and accelerometers (<math><1 \mu\text{g}</math>)
- X-band transponders

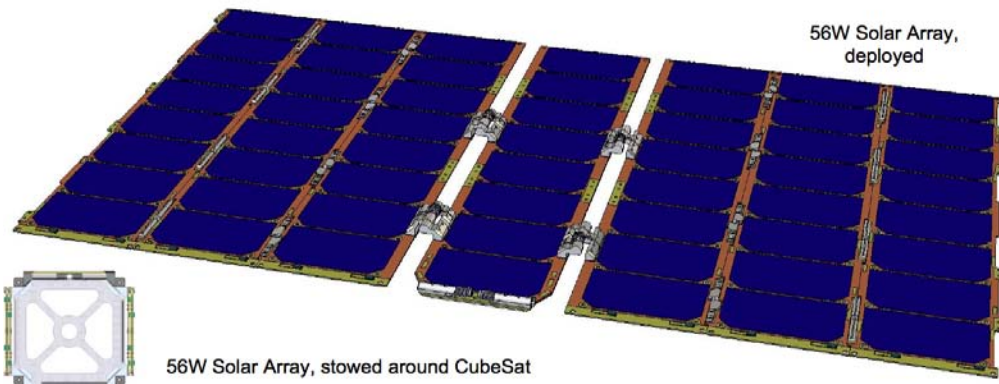
Some instruments/sensors can be dual-purposed!

CubeSat Capabilities – Power

- Deployable solar arrays pushing the power envelope
 - Current COTS-available maximum is 56 Watts instantaneous
 - Up to 80-Watt deployable arrays on the horizon
 - Solar array drive systems
- COTS Power Systems
 - 30 Watt nominal power regulation, up to 70 Watts instantaneous
 - Battery packs with up to 33 W-hr have been flown in LEO
- What about power beyond Earth/Lunar orbits?



	Distance (AU)	Solar Power Reduction Factor	80-Watt Earth Array Power at Target (W)
Mars	1.52	2	34.63
Ceres	2.77	8	10.43
Jupiter	5.2	27	2.96
Saturn	9.5	90	0.89
Uranus	19.6	384	0.21
Neptune	30	900	0.09



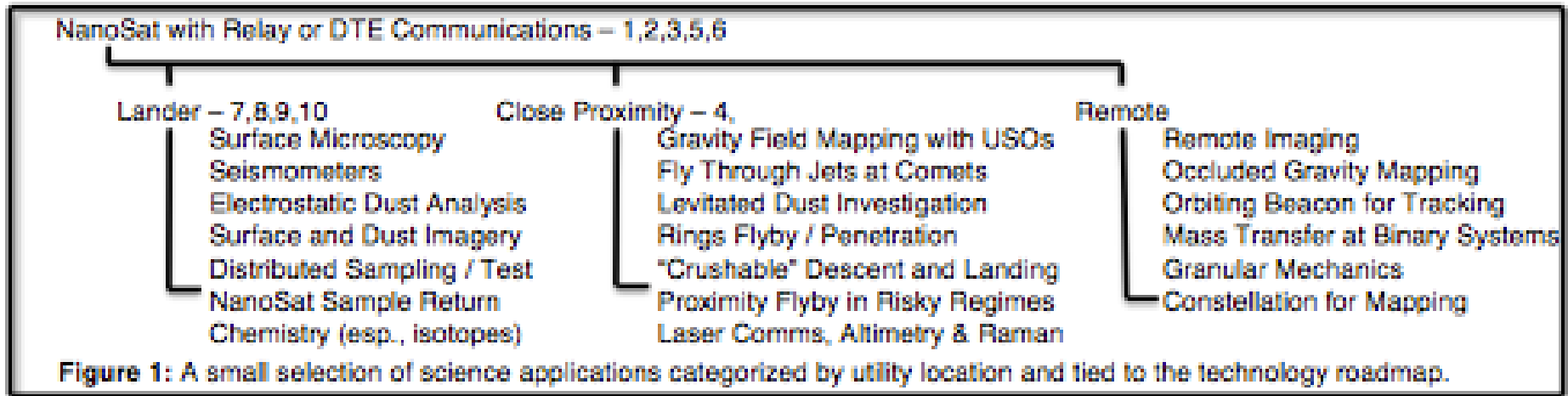
CubeSat Capabilities Demonstrated

- Structures: 1U, 3U structures available (6U this year)
- Communications: Flexible radios in multiple bands (VHF, UHF, S-Band)
- C&DH: Multiple levels of complexity and radiation-hardening
- Power: Deployable Panels (and other pieces)
- Propulsion: Cold-Gas and EP on the market, sail decelerator demonstrated – further developments upcoming through Edison
- Payloads: 2U and larger payloads available on 3U systems

- **CubeSats are ready to go interplanetary – but what will we do?**

Applications of NanoSats at Small Body Objects

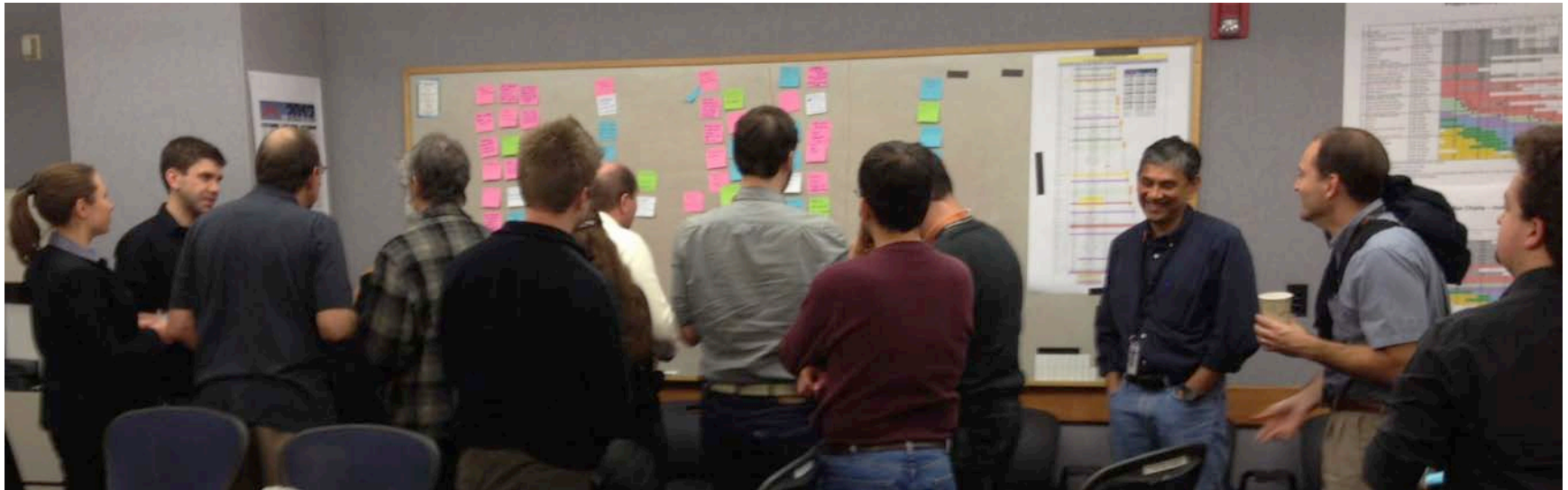
- Workshop in January held at JPL on the use of NanoSpacecraft at Small-Body Objects



- U. Michigan, U. Colorado, Naval Postgraduate School, CalPoly – SLO, JPL
- A paper summary will be released at GLEX 2012
- Interest in exploring secondary support options for large missions
- Interest in using a “standard” bus with multiple deployments throughout the solar system.

Applications of NanoSats at Small Body Objects

- Identified 100 science applications enabled by NanoSats
- Developed a technology roadmap towards enabling interplanetary NanoSats
- High-TRL, miniaturized instruments are already available and cover broad range of measurements
- Technologies for the development of small mass spectrometers are in PIDDP/ASTID
- 3 kg payload can combine reconnaissance (Pancam) + geophysical (accelerometers, thermoprobe) + chemical (XRD/XRF, mass specs) observational capability



Finding the Partnership

- There is a strong desire to explore small-body objects, yet **funding is slim**.
- **CubeSats are not providing solutions to traditional planetary problems** (getting lots of data back from far away places) – big spacecraft are just better at that.
- **Partnerships between the CubeSat community and planetary scientists** may result in novel approaches to traditional missions – enabling a new type of exploration.
- **Lessons learned from CubeSats can inform appropriate decisions on risk** – sacrificial elements and cheap vehicles can still be used to perform spectacular science.
- **Co-developing platforms (scientists with engineers)** for interplanetary space is critical – better integration between heritage hardware and novel investigations.

Where will NanoSats go next?