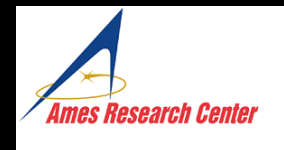


De-Orbit Mechanism for a Small Satellites

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Division focus:

NASA's ability to rapidly develop and launch ground-breaking technologies into space just got more efficient and less expensive.

Small and very small satellites (weighing between 2-440 pounds) can test innovative science and engineering technologies on a smaller scale in the space environment and better understand how hardware will survive the radiation, temperature and vacuum conditions encountered in space.

NASA nanosatellites are designed for a wide spectrum of space missions including biology experiments, testing advanced propulsion and communications technologies.



Presentation Overview



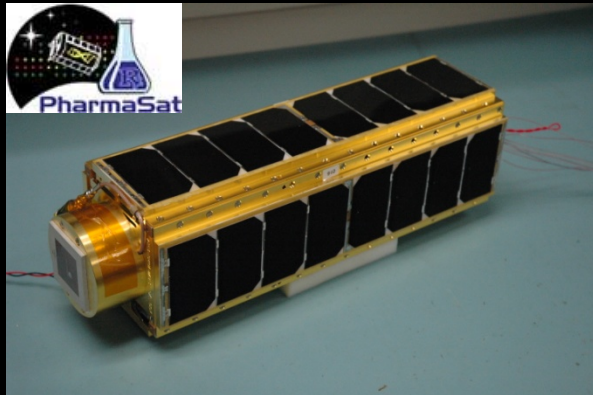
- 1) Heritage of 3 Cube Spacecraft
- 2) Why de-orbiting is important
- 3) Design
 - Mechanism, Spacecraft, Deployment
- 4) Future Development

Heritage of 3 Cube Spacecraft



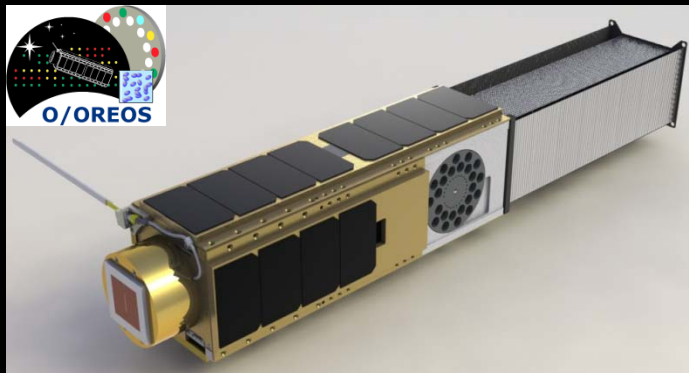
GeneSat-1

- Measured genetic changes E. coli bacteria in space environment
- Launched December 2006



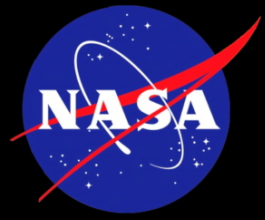
PharmaSat

- Examines yeast growth following exposure to antifungal in microgravity environment
- Launch May 2009



O/OREOS

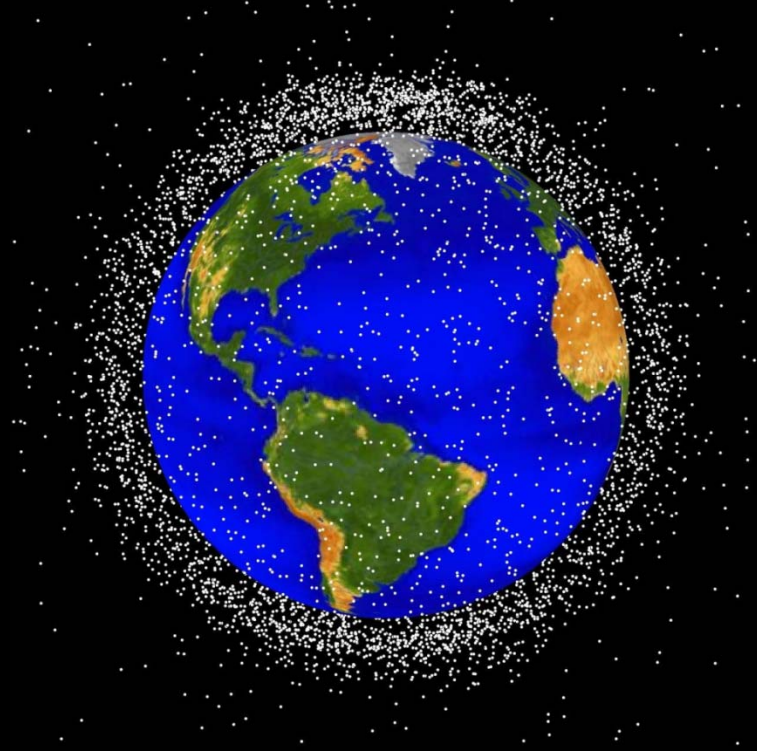
- Dual Payload
 - Biological growth experiment
 - Organics experiment using spectrometer
 - First de-orbit mechanism deployment
- Launch December 2009

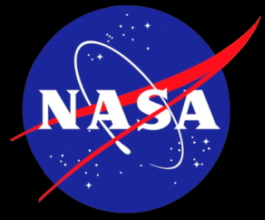


Why Satellite De-orbiting is Important:



- Space is getting cluttered
 - Increasing interest in small spacecraft usage
 - Unused satellites become “space debris”
 - Increasing challenge to track all debris
 - Recent satellite collision
- NASA-STD-8719.14:
Spacecraft in Earth orbit must re-enter
 - 25 years after end of mission (or)
 - 30 years after launch





De-Orbit Mechanism Design Requirements:



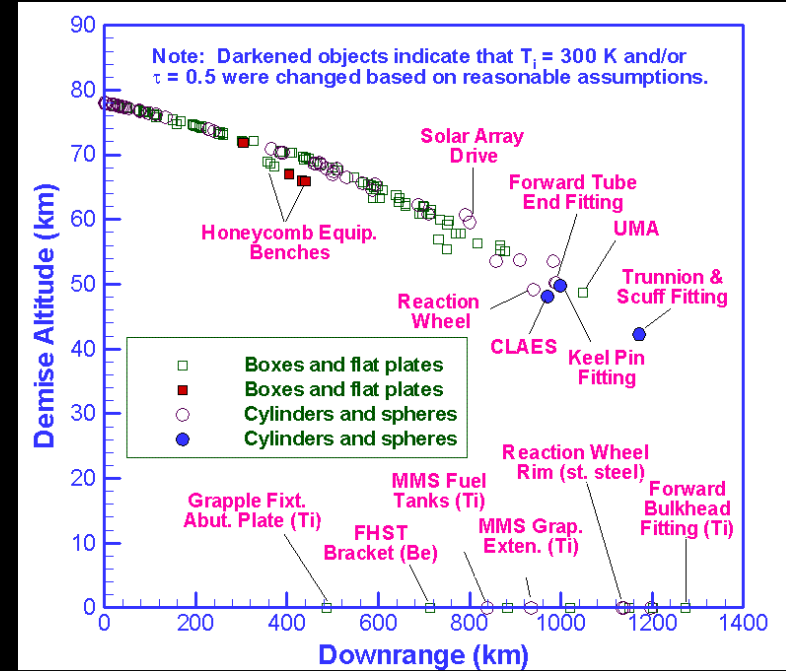
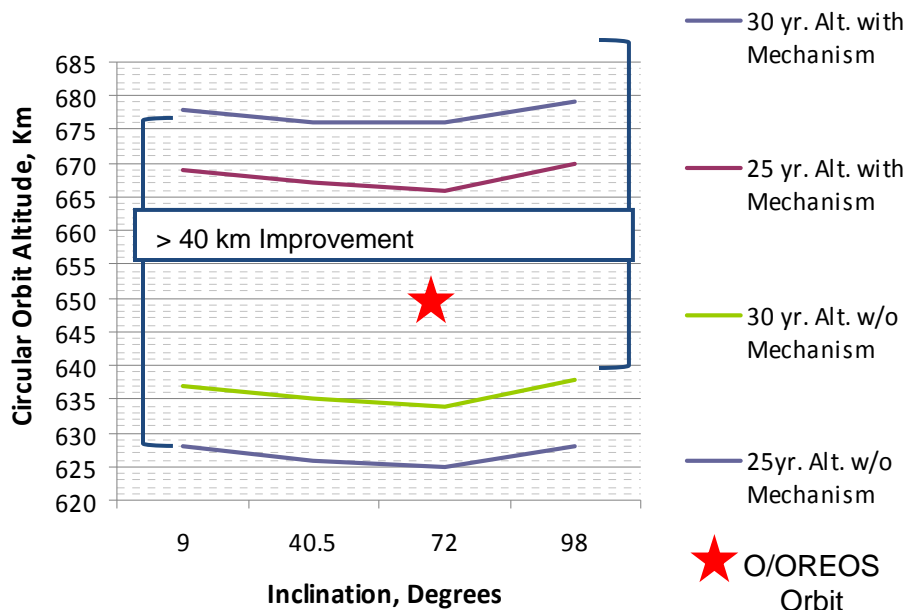
- Allows satellite to re-enter on reduced time table
- Simple, purely mechanical actuation
- Low profile package ≤ 0.30 " thick when collapsed
- System does not interfere with satellite operation
 - Solar panel exposure
 - Thermal properties
 - Tumble/Pointing is unaffected (magnetic passive control)
 - Deployment
- Deploys automatically during satellite release



Object Re-entry Analysis



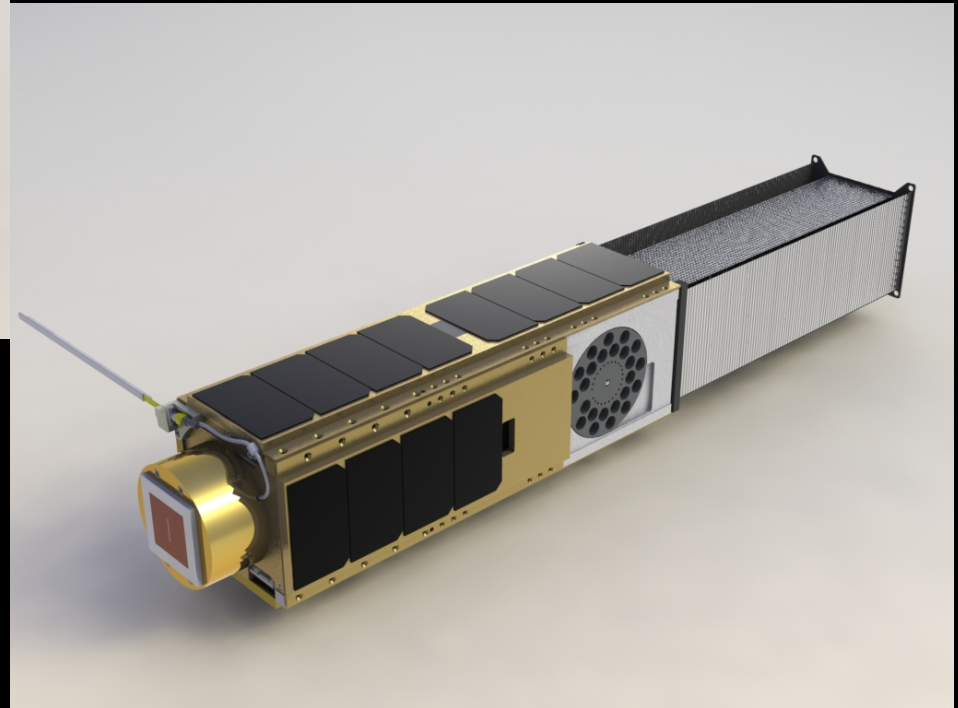
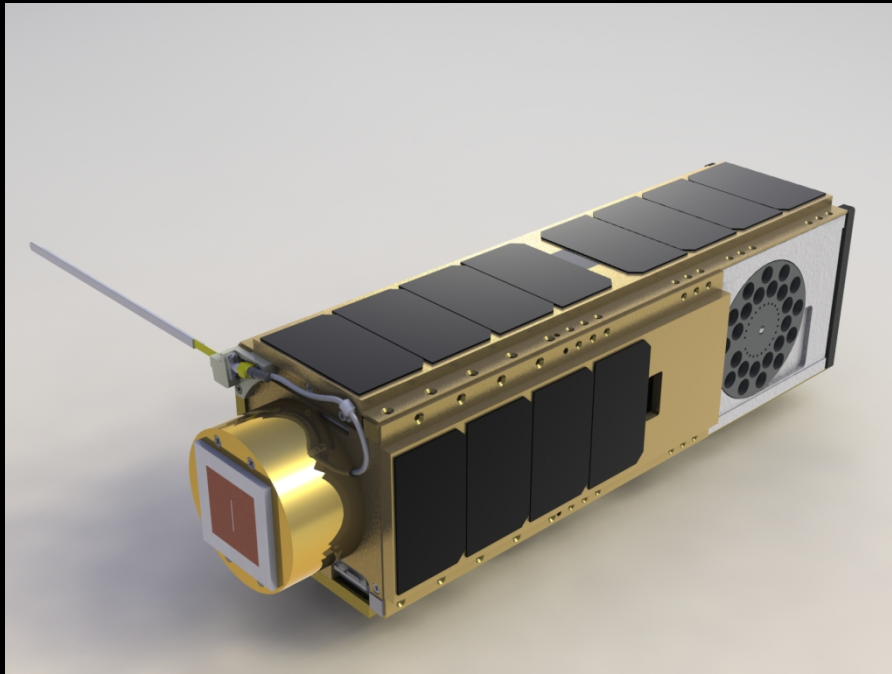
Deorbit Altitudes for a 5 kg, 3U CubeSat, Dec. 1 2009 Launch
(with and without ARC Deorbit Mechanism)



For O/OREOS characteristics, surface area must be increased by 60% to induce re-entry within 30 years



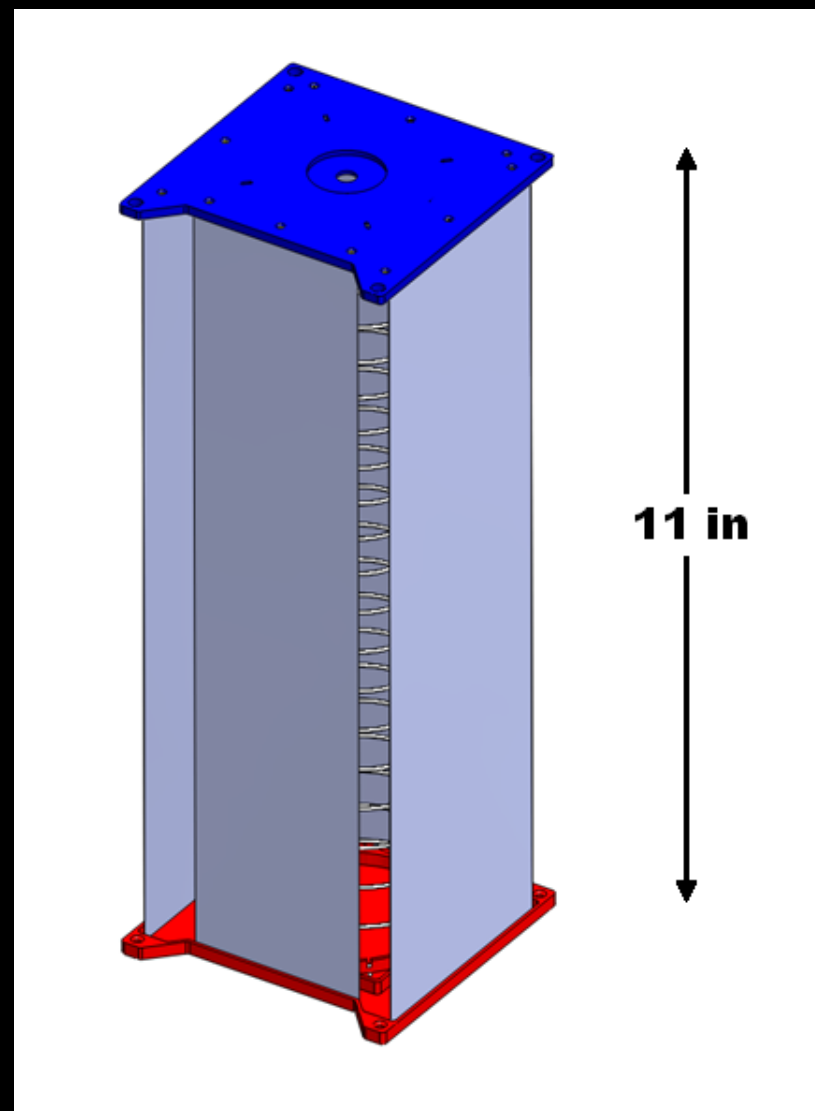
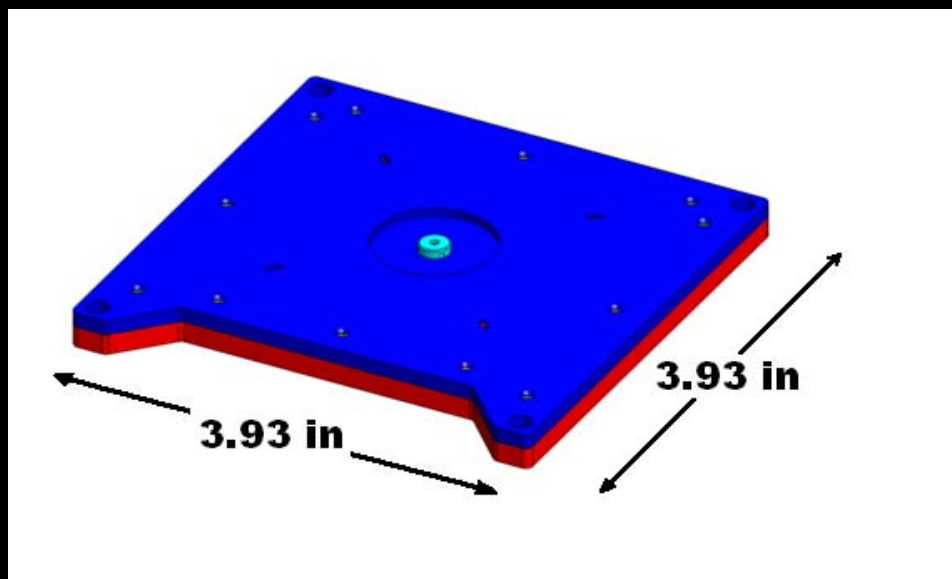
O/OREOS Spacecraft With De-Orbit Mechanism



Surface area
increased by
60%



De-orbit Mechanism Design Overview

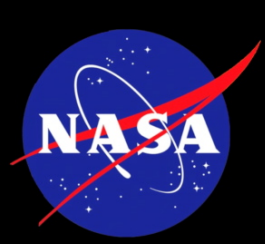




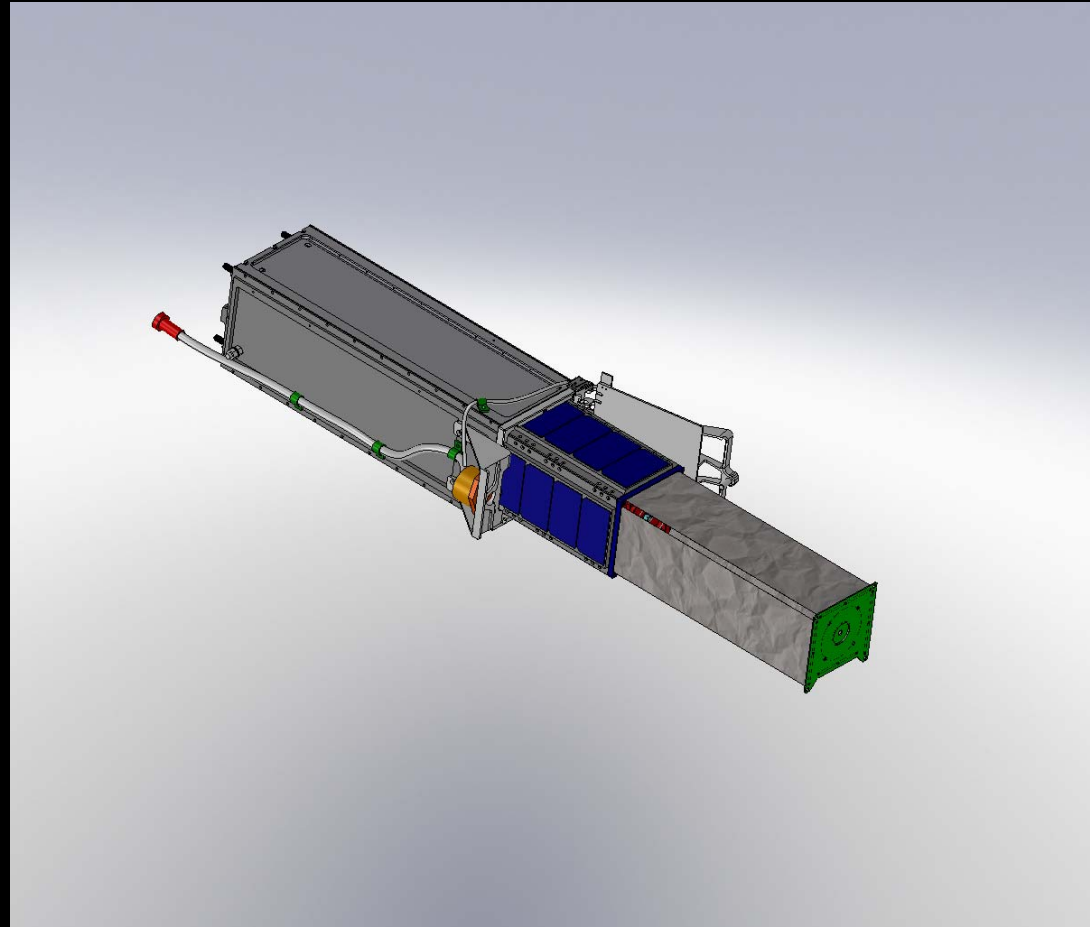
De-Orbit Mechanism Design Features:



- Mechanism completely contained during launch:
 - Protects fragile wall material
 - Reduced to small volume before deployment
 - Independent structural stability
- Ease of assembly and system reset ability
- Design Flexibility
 - Adjustable extended length (up to 15 inches = 184 sq in surface area)
 - Can be modified to meet specific requirements
- Passive deployment using P-POD door



P-POD Deployment

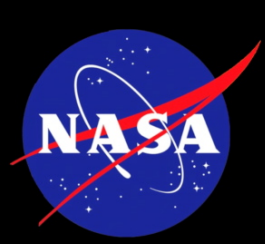




Future Work



- Additional testing
 - Vibration and shock testing
 - Thermal Studies
- Design refinement
 - Simplify components
- Implementation with other spacecraft
 - 1U CubeSat
 - 10+ kg spacecraft
- Additional deployment methods



Questions

