

## **Delfi-C<sup>3</sup>**

#### Integration, & Status Update Wouter Jan Ubbels MSc., PE4WJ Project Manager Delft University of Technology, The Netherlands

CubeSat Developers' Workshop 2007



#### The Delfi-C3 satellite

- 3 unit CubeSat structure, 3kg
- 2 Sun Sensor payload units
- 4 deployable panels at 35 degrees (max/min power):

-Carbon Fiber Reinforced Plastic

-Thin Film Solar Cell payload suspension frame

–5 TEC1 GaAs TJ solar cells

- 2.5 W min. power available
- 8 antennas:
  - -4 VHF 50 cm downlink
  - -4 UHF 18 cm uplink
- No battery
- No active attitude control





#### The Delfi-C3 satellite

#### • EPS

- 1 DC DC converter per solar panel
- Current measurement
- CDHS
  - TI MSP430 OBC
  - Microchip PIC18LF4680 microcontrollers per subsystem
- Attitude Control
  - Magnetic hysteresis rods & permanent magnet
- COMMS
  - Two Redundant Transceivers, VHF 1200Baud BPSK AX.25 downlink, UHF uplink + Linear Transponder (after 3 months)
- Standard board interface
  - I<sup>2</sup>C bus
  - 12 V DC power bus
- Passive thermal subsystem (thermal tapes)

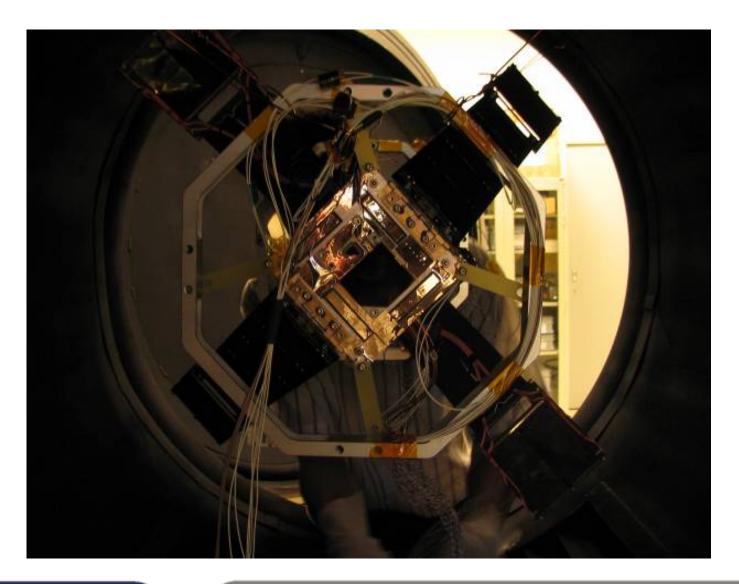


#### Status

- Flight model successfully integrated
- Vibe test  $\rightarrow$  ok
- TVac test  $\rightarrow$  ok
- Currently in Toronto, awaiting shipment to India
- Launch on NLS-4 PSLV Sept-Nov 2007



#### **TVAC** Test



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#### Vibe Test



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#### **Delfi-C3 flight stack**



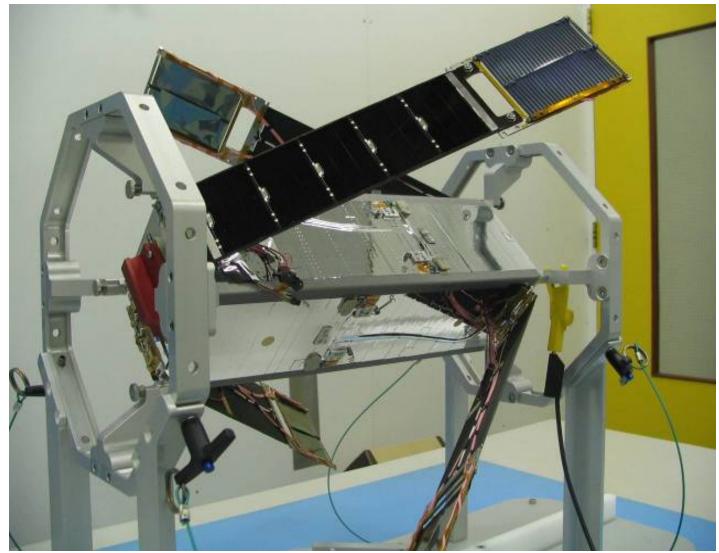
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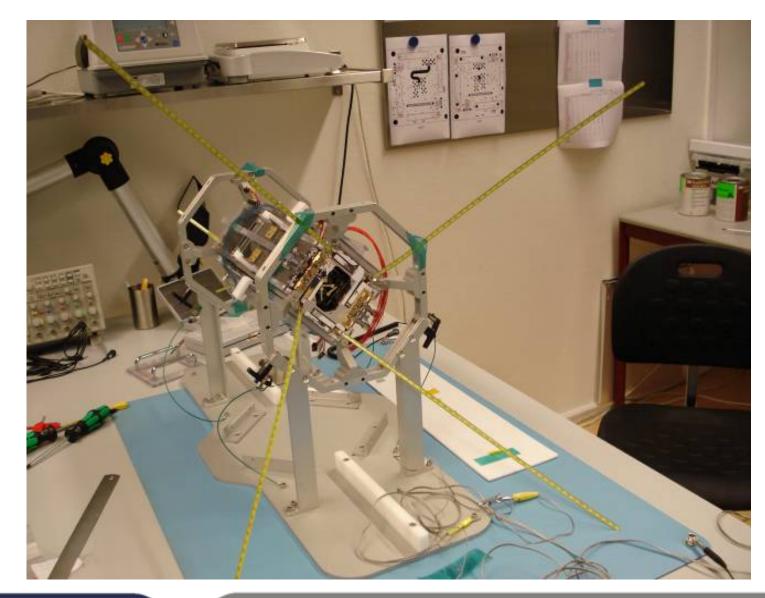
# Flight configuration (antennas not deployed)



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#### **Deployed antennas**



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#### **Space Art**



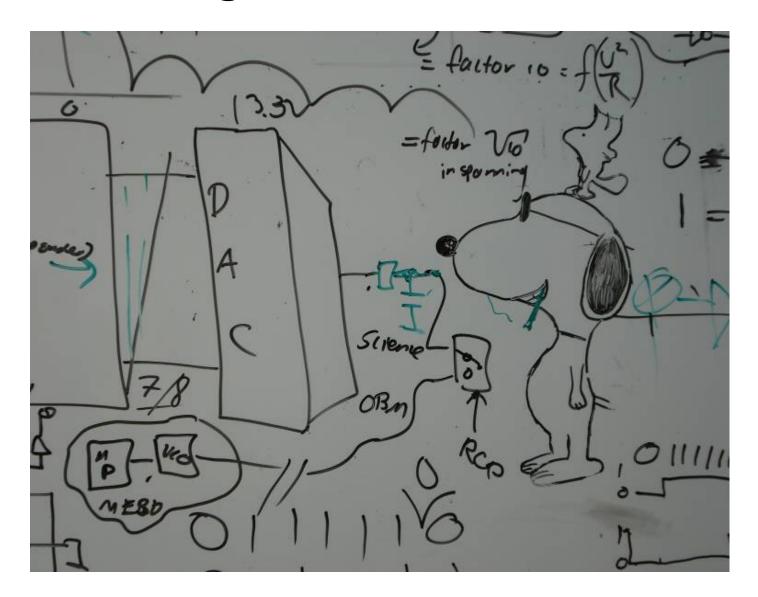
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#### How to design a satellite...



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#### Lessons Learnt

- Start building prototypes early!
- In RF circuits, power matching is not always necessary
- Opamps make nice RF/IF amplifiers
- Do not choose 0000 as bus reset command
- Parasitic oscillations can popup just about anywhere
- I2C repeaters / pull down by PICs
- Use ground lines between I2C data lines (Doh!)
- Reserve spare pins on connectors, you will need them!
- Use 0-ohm resistors to connect subcircuits
- And...

### **Delfi-C**<sup>3</sup>

# ....Satellites are indeed, entirely constructed out of pizzas



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#### Latest photos

http://www.delfic3.nl/photoblog/

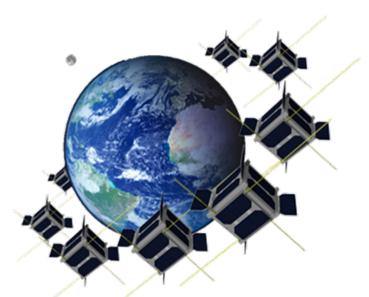
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## ISIS Innovative Solutions In Space



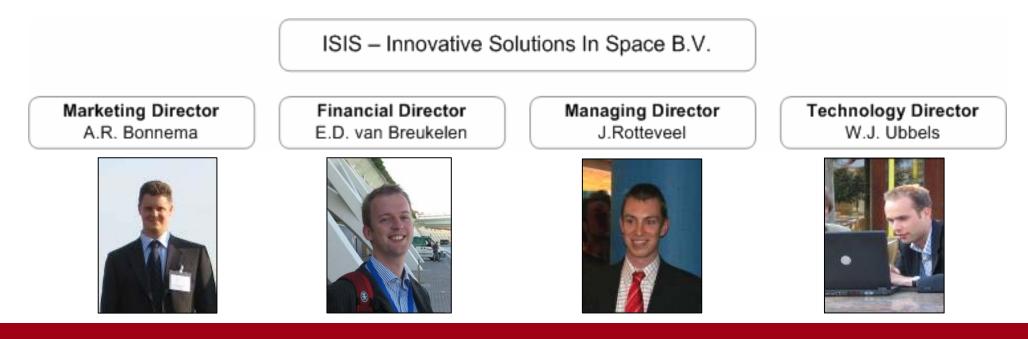
Small Satellite Subsystems & Piggyback opportunities for (Cube)Sats

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## ISIS - Innovative Solutions In Space B.V. Satellites for fun, science and profit

- Founded in January 2006, indepently held company
- Board of 4 aerospace engineers
- Space systems engineering background
- Offices on campus of Delft University of Technology
- Spin-off from Delfi-C3 satellite project. This is not OUR project, but we all participate(d) in it and find it great fun.



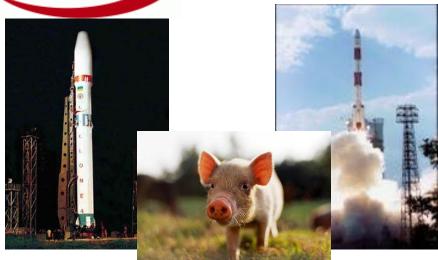


## Vision

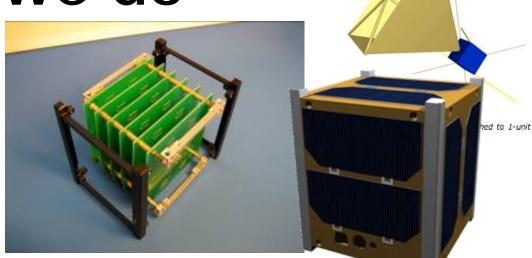
- Advances in electronics miniaturization
  - Commercial (micro-)systems technology applicable in space,
  - Smaller satellite systems as a result of improved functionality per unit mass
- Improved access to space
  - New launchers: Vega, Falcon, Stihl-II, Cyclone-4
  - Secondary payload rings: (ASAP, KAP, ESPA, etc)
- Temporal resolution will grow in importance
  - Growing need for small satellite constellations



## What we do



Piggyback launches 2008, 2009, ...



(Cube)Sat products: Structure, VHF/UHF Radio, Ground Station, microPropulsion, inflatables

Expertise,
track record,
Technology
development

Small Satellite Engineering consultancy and subsidy projects for R&D

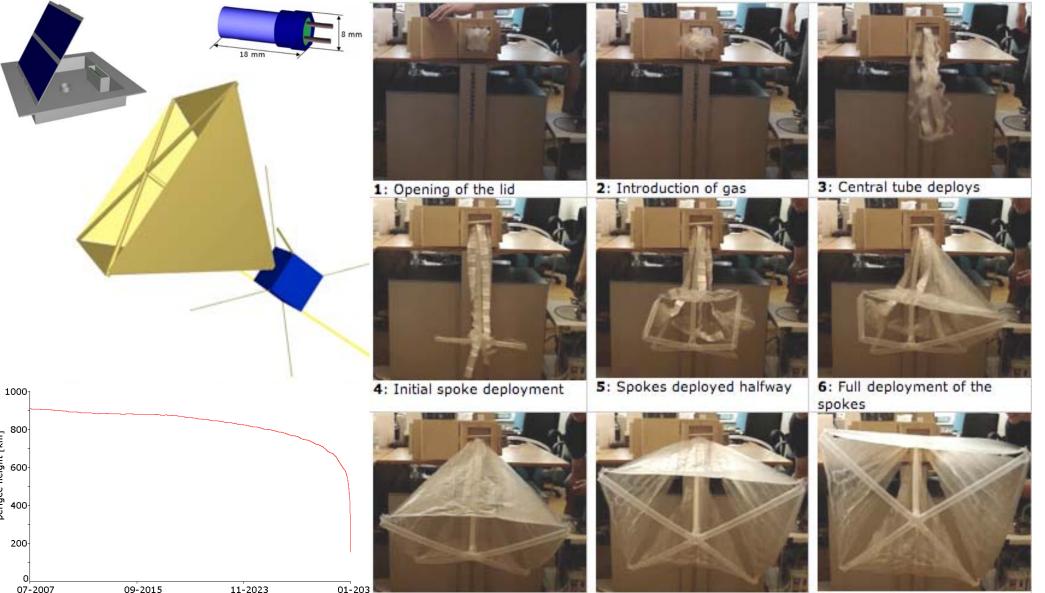
Educational space projects: CanSat, ..



## Launch Services

- Access to space for payloads up to 10 kg
  - CubeSats
  - Nanosats
  - Fixed payloads
- LEO, piggy back
- Multiple configurations up to 20 kg
- Flight opportunities:
  - PSLV 2008Q2, one 1U CubeSat slot still available! (come and talk to me in the break if you're interested)
  - PSLV 2008Q4, ...
  - Cyclone-4 maiden flight 2009Q4

### What about space junk? Long term R&D: inflatable De-orbit device



8: Internal pressure

increases

7: Deployment achieved

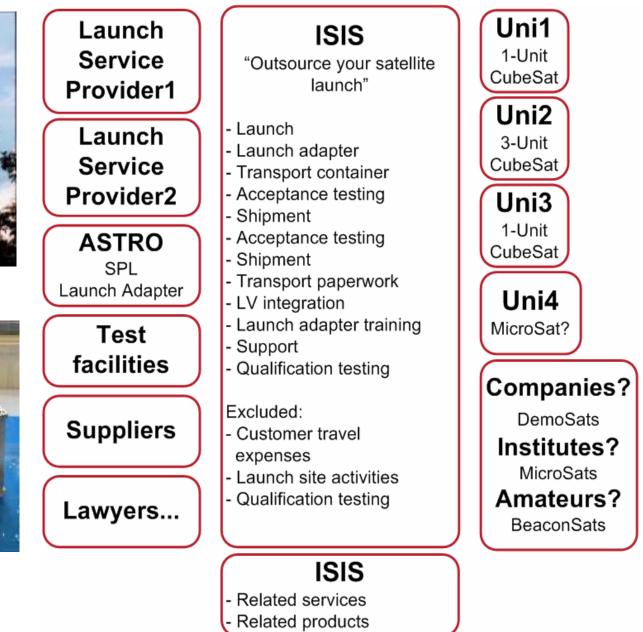
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Date [mo-yr]

9: Internal pressure maximal

## **Launching CubeSats**







## Questions?

w.j.ubbels@isispace.nl / info@isispace.nl www.isispace.nl www.cansat.nl