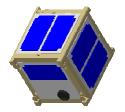
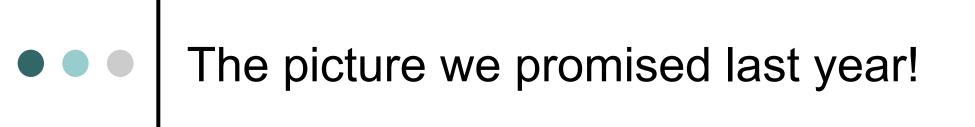
Design and Development of the i.T.Ü. pSAT I Engineering Prototype

Can Kurtulus, Taskin Baltaci, Tayfun Aydin Prof. Gokhan Inalhan, Prof. Rustem Aslan

Istanbul Technical University, Faculty of Aeronautics and Astronautics



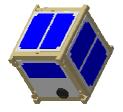






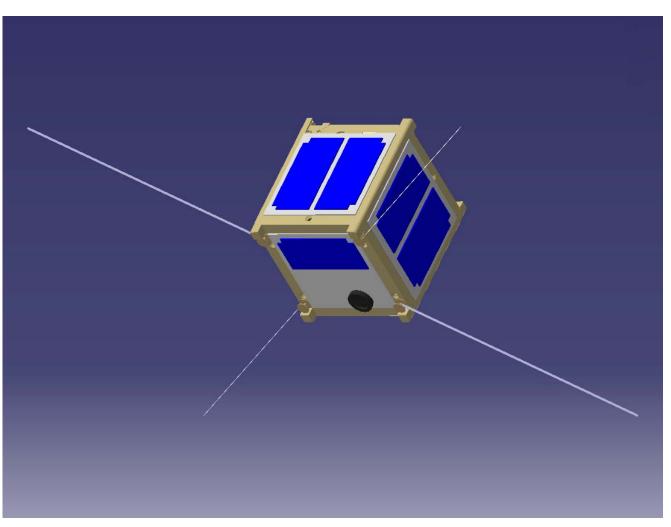


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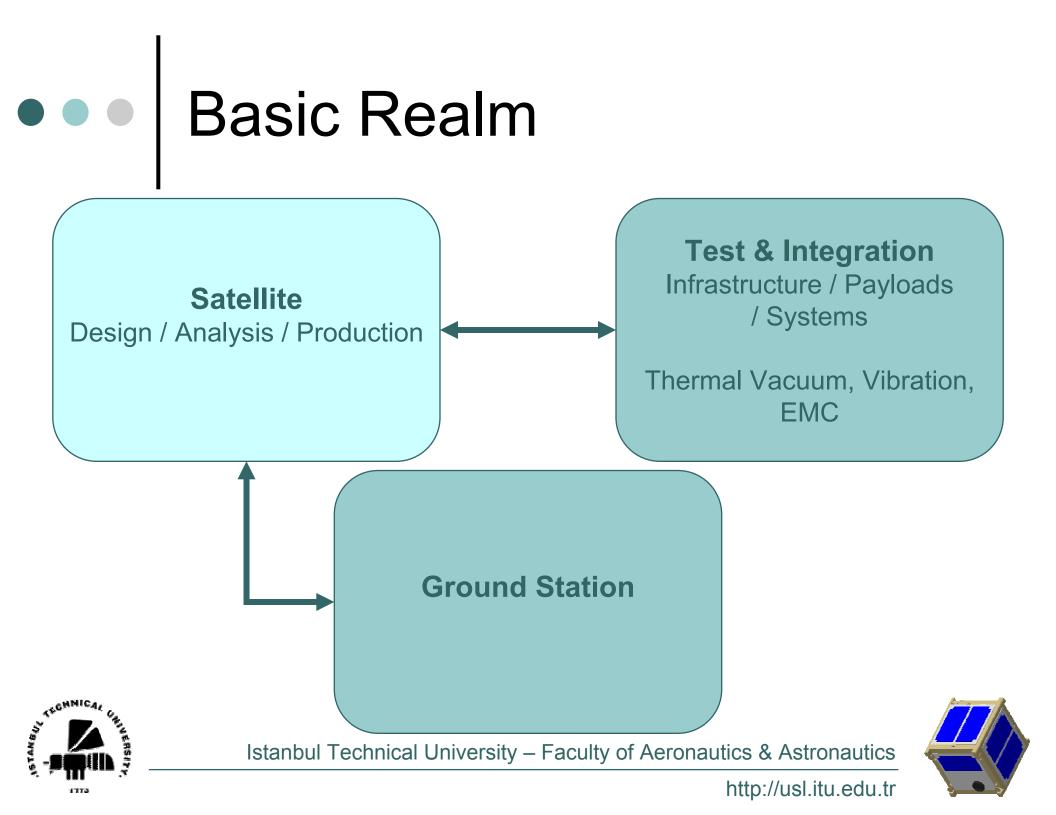
http://usl.itu.edu.tr

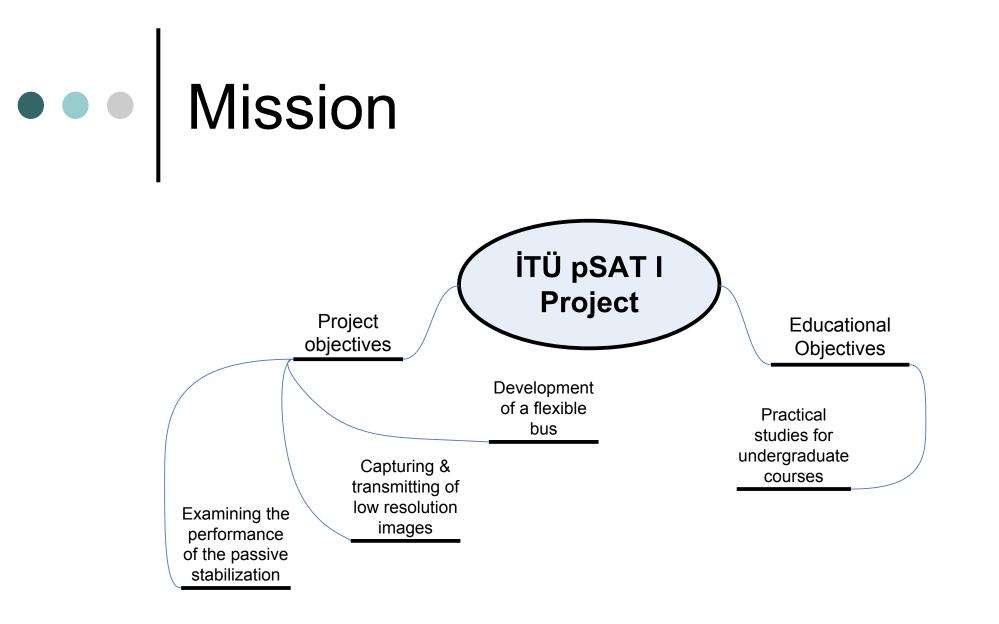
••• ITÜ-pSAT I



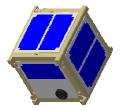


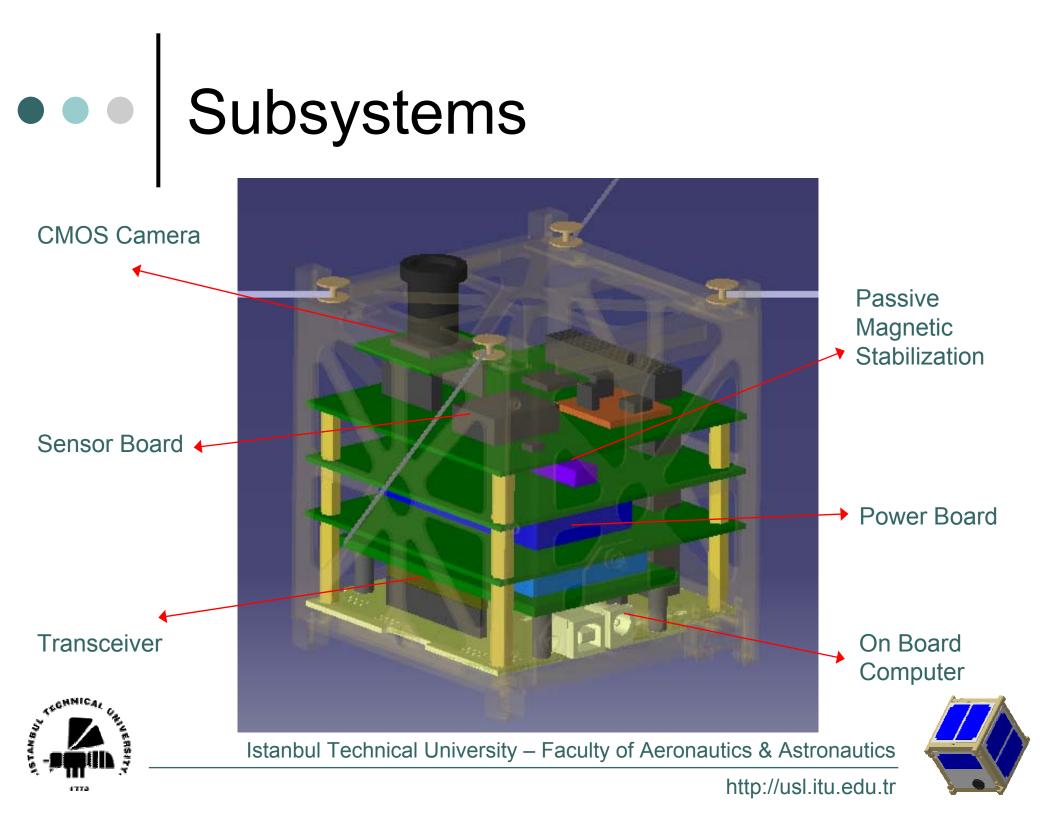
pSAT I is getting ready for its 2008 launch...







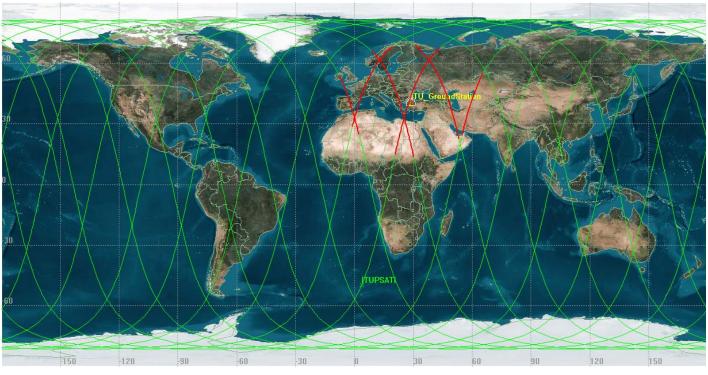




Launch & Orbit

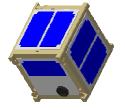
• DNEPR 2008

- 600-700 km circular polar orbit
- Allows a comm window of ~50 min/day

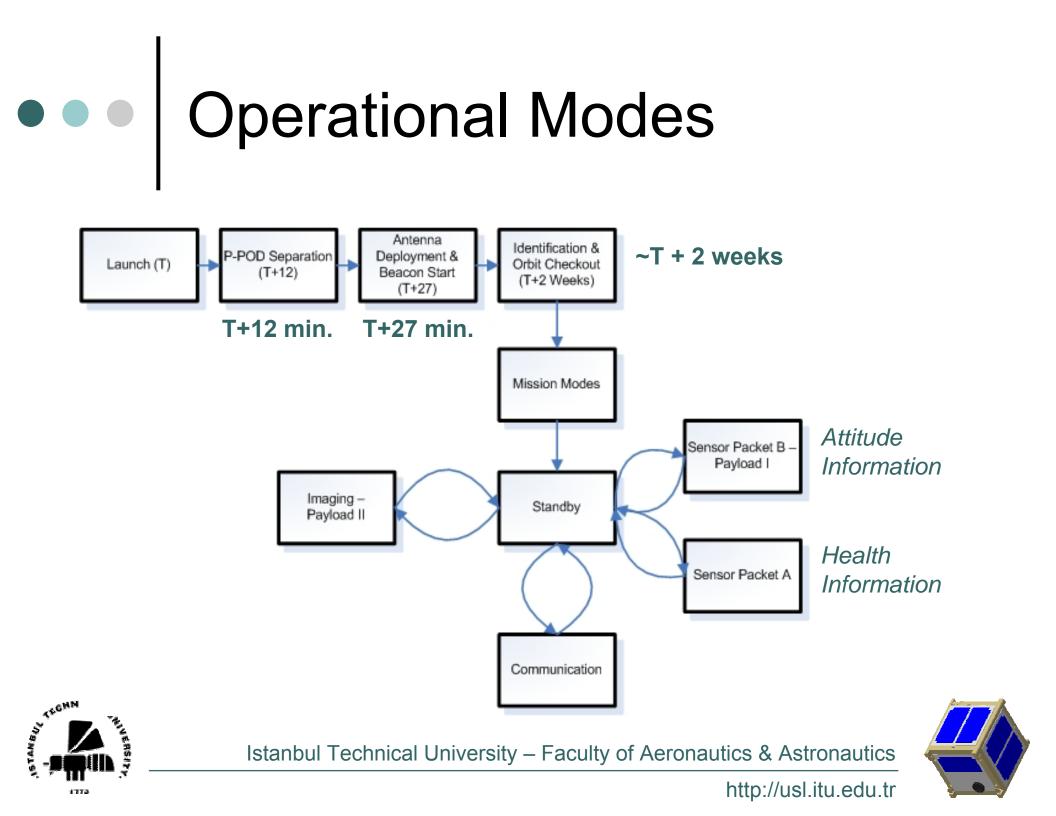




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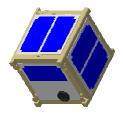


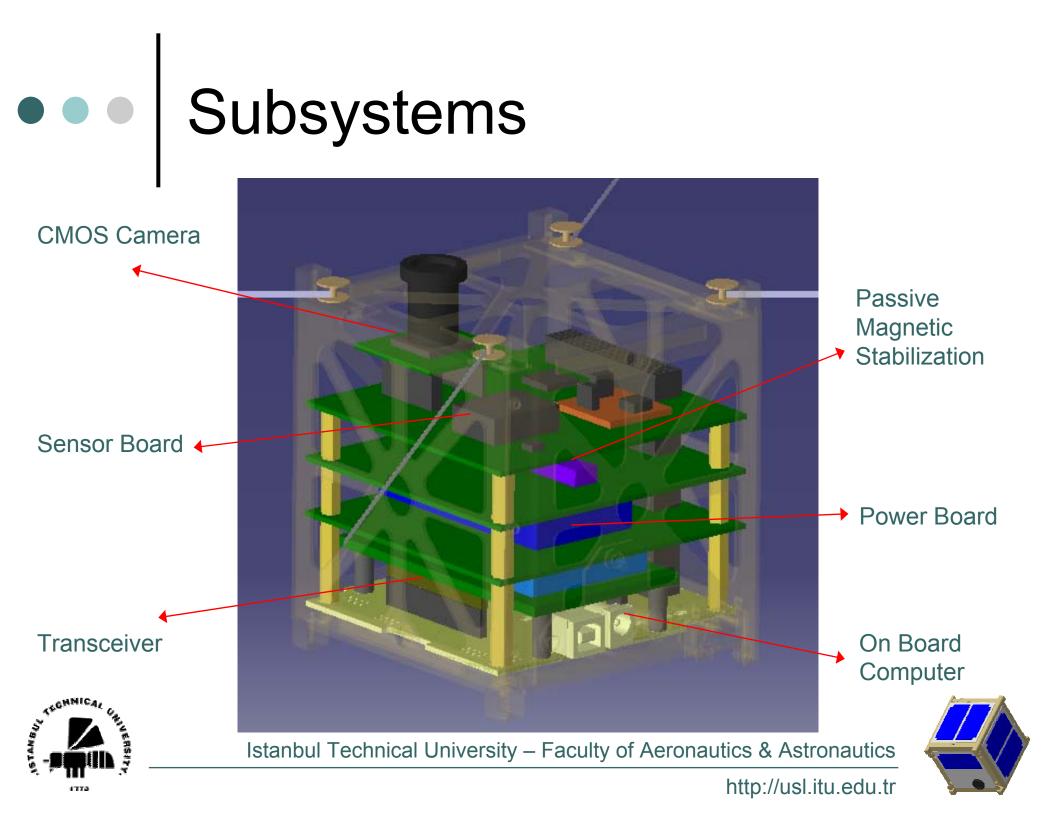
Mission Modes

		Beacon	On-Board Computer	Power Board	Radio- Modem	Payload 1	Payload 2
	Standby	On	On	On	Off	Off	Off
	Communic ation	On	On	On	On	Off	Off
	Sensor Packet SPA	On	On	On	Off	On	Off
	Sensor Packet SPB	On	On	On	Off	On	Off
	Taking a photograph	On	On	On	Off	Off	On



Active Payload



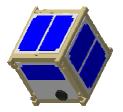


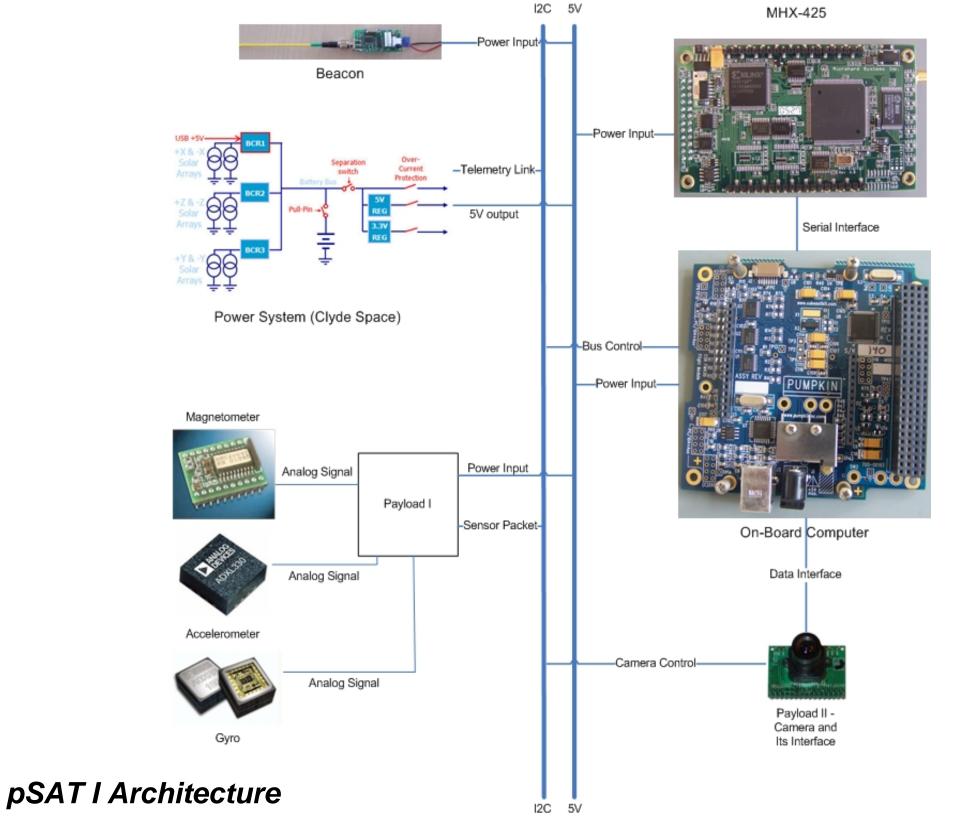
Bus – approach philosophy

 Simple design structured around COTS equipment

- o Low risk
- Concurrent in-house development within SSDTL
 - Ever-growing involvement of undergraduate and graduate students







Systems Engineering

- Mass Budget
 - Left some headroom for surprises
- Power Budget
- Link Budget

	Current (A)	Power (W)
Standby	0,055	0,275
Communication	1,155	5,775
Sensor Packet SPA	0,105	0,525
Sensor Packet SPB	0,105	0,525
Imaging	0,175	0,875
Peak	1,155	5,775

Downlink Budge

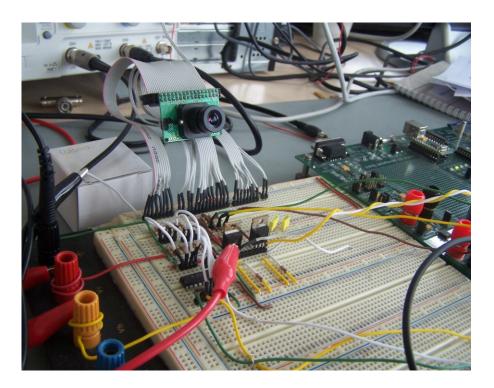
Downlink Budget	
Spacecraft Transmitter Power Output:	1.0 Watt / 30 dBm
Spacecraft Total Transmission Line Losses:	1.9 dB
Spacecraft Antenna Gain:	2.2 dBi
Spacecraft Antenna Pointing Loss:	0.3 dB
S/C-to-Ground Antenna Polarization Loss:	0.2 dB
Path Loss:	143.3 dB
Atmospheric Loss:	1.1 dB
Ionospheric Loss:	0.4 dB
Ground Station Antenna Pointing Loss:	0.7 dB
Ground Station Antenna Gain:	18.95 dBi
Ground Station Total Transmission Line Losses:	5.1 dB
Link Margin	13.15 dB

Uplink Budget	
Ground Station Transmitter Power Output:	75 Watt / 49 dBm
Ground Stn. Total Transmission Line Losses:	5.1 dB
Antenna Gain:	18.95 dBi
Ground Station Antenna Pointing Loss:	1.2 dB
Gnd-to-S/C Antenna Polarization Losses:	0.2 dB
Path Loss:	143.3 dB
Atmospheric Losses:	1.1 dB
Ionospheric Losses:	0.4 dB
Spacecraft Antenna Pointing Loss:	0.3 dB
Spacecraft Antenna Gain:	2.2 dBi
Spacecraft Total Transmission Line Losses:	2.0 dB
Link Margin	31.55 dB

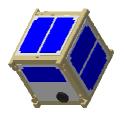
Subsystem	Component	Brief Explanation	Mass(g)
Structure			
	Base structure		165
	Cables		5
	Fasteners		15
Payload			
	Payload #1	Sensor Board	80
	Payload #2	Camera	40
Attitude Control			
	Magnet		50
On-board Computer			
	Board		55
	SD - Card		10
Communication			
	Transceiver		80
	Antenna		20
Power			
	Solar Panels	25g x 6	150
	Power System Board	Battery + circuit board	150
Total Mass			820

Payload II – Imaging Experiment

- CMOS camera and interface 0 board continental scale image capturing
- Camera C3188A (OV7620 chip) 0
 - Low resolution (VGA)
 - 5V
- Needs an interface circuit for 0 **MSP430!**
- Might switch to a MCU at the 0 same voltage level

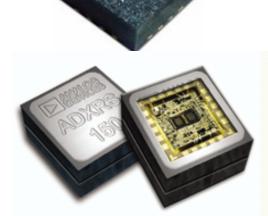






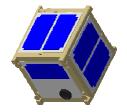
Payload I – Passive Attitude Control Experiment

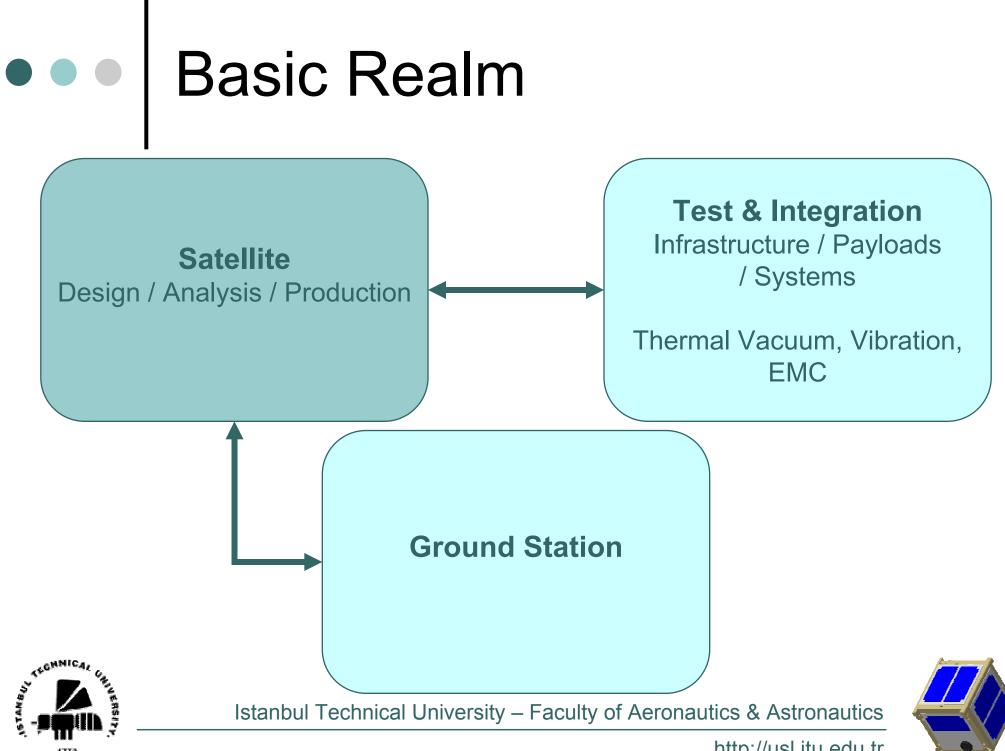
- Passive magnet and sensing board
 - Reports on I²C after ADC (Sensor Packet B)
- Sensors
 - Accelerometer ADXL330
 - Gyros ADXRS300
 - Magnetometer HMC2003
- Passive AlNiCo magnet









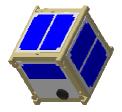


Space Systems Development and **Test Laboratory**

- Thermal/Vac Chamber
 - 350 litre, 10⁻⁶ torr
 - -60° C 90° C at 1°C / min steps
 - Class 1000 Clean Room,
 - 25 m²
- Access to university resources
 - Shake Table
 - EMI/EMC
- Ground Station





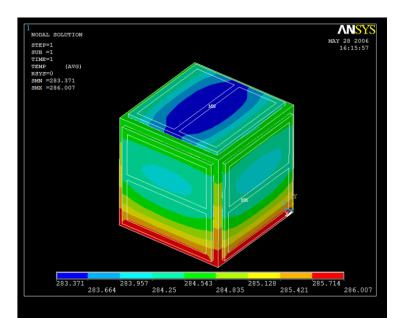


Facilities

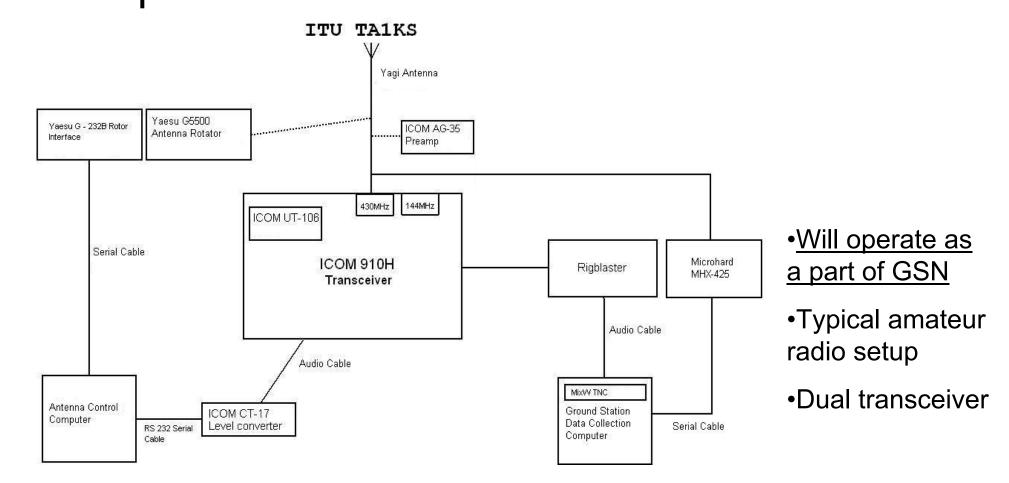




- Access to researchers, students and small business initiatives from all around the world
 - <u>Assemble</u> and <u>Test</u> science payloads, and nano-to-micro satellites.

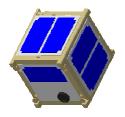


Ground Station – ITU TA1KS [©]

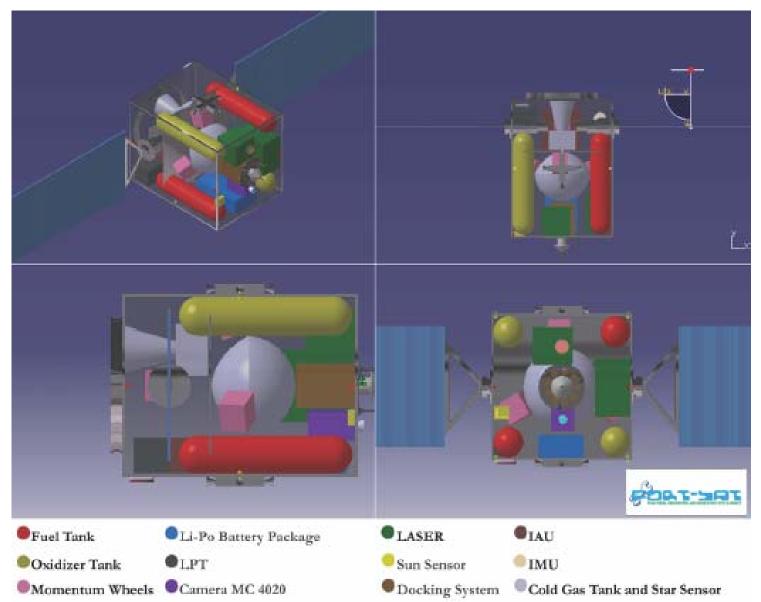


Will be up and running by the summer of 2007

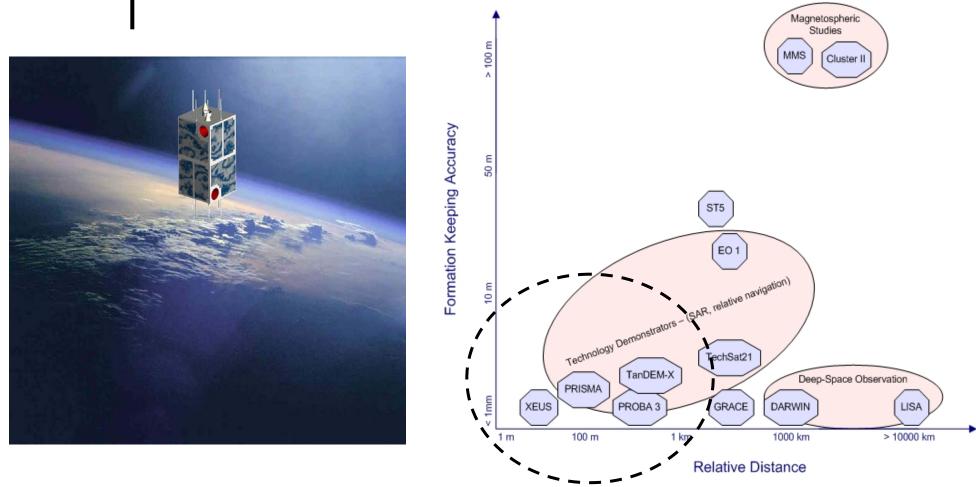




Spacecraft Systems Design Course – The winner is ...



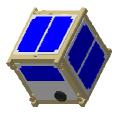
ITU-nSAT Concept



Need for cheap on-orbit demonstration against key technologies.



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http://usl.itu.edu.tr

Nano-satellite Formation Experiment

• A true research platform for key control, micro-propulsion and imaging technologies :





20x20x20cm per unit; Total 12-14kg



Concepts developing...

- Space Qualifying New Processors
 - LPC2294 Arm
 - MPC 555 Motorola PPC



LPC2294



MPC555

- Transition to CAN Bus
 - PIC based sensor interface boards
 - Other sensors

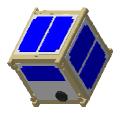




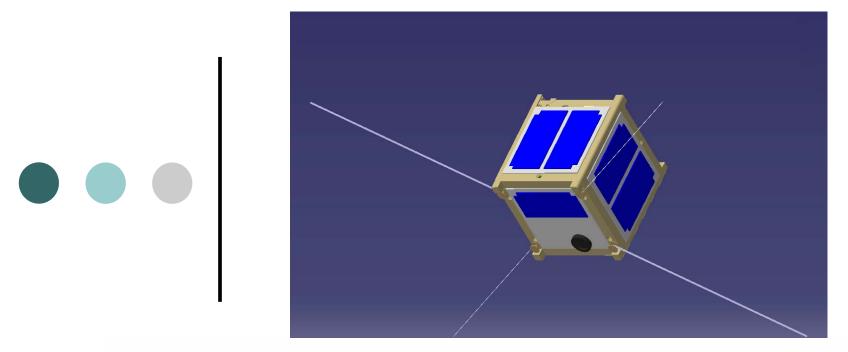
Smart Sensor Nodes

CAN Interface





İTÜ pSAT I is on schedule for its 2008 launch...



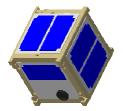


See "it" at Recent Advances in Space Technologies 2007 June 14th -17th at Istanbul.

Thank You!

Any Questions?





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http://usl.itu.edu.tr