

# SwissCube Project

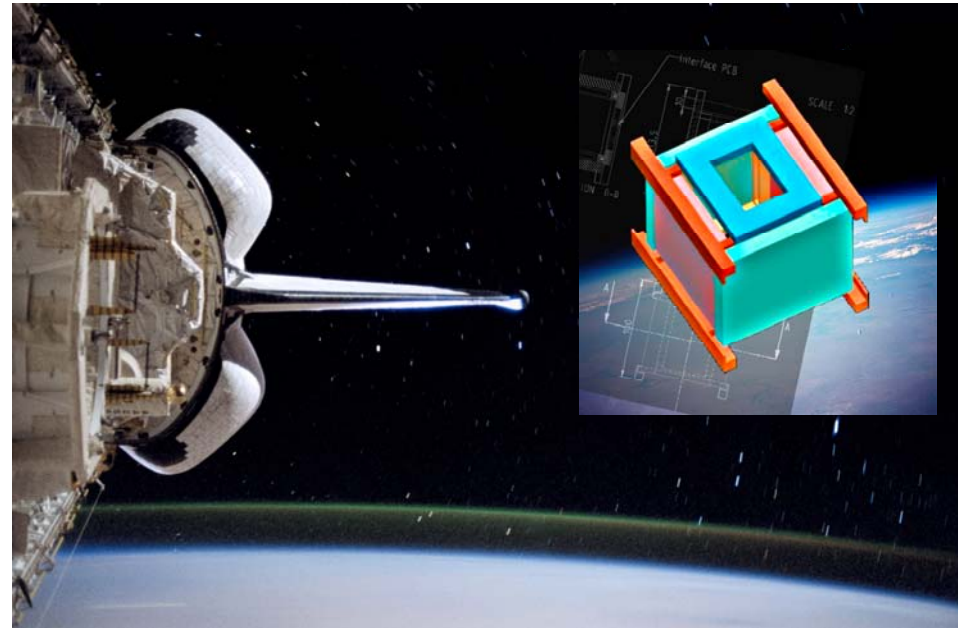
## 3rd Annual Cubesat Workshop April 27, 2006

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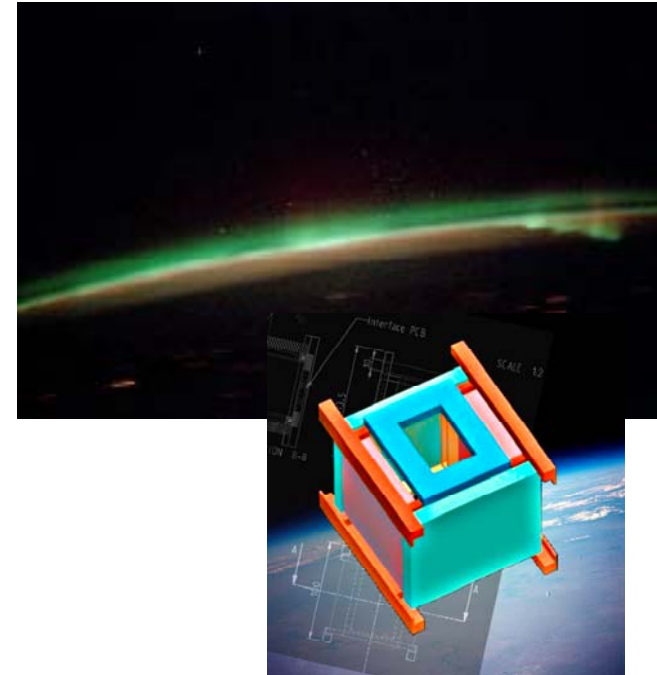


- Project and science objectives
- Preliminary mission assumptions
- Project organization (participants + schedule)
- Test and model philosophy
- Conclusion

# Project Objectives

- **Goal is to have a Swiss Cubesat in orbit by 2008**

- Satellite and ground segment defined, designed, built, tested and operated by students
  - Multi-disciplinary, multi-lab, multi-university collaboration
  - Strong educational aspect: student/industry ties, ESA/NASA development phases and standards
- Budget:
  - ~ \$ 200 k (including launch) over 2.5 years
  - already gathered 2/3 of budget
- High visibility project for Swiss universities and industry partners since it would be the first Swiss satellite launched



- **Two mission objectives:**

- Science: characterize variability of Nightglow phenomena in intensity and altitude
- Technology: test and qualify a new Single Photon Avalanche Detector

- **Primary success criteria:**                 **Deliver a fully tested cubesat to launch site**

- **Secondary success criteria:**           **1. Launch, close RF link and download telemetry**  
   **2. Receive Science data and characterize operations**

# Science Objectives

- Science Objectives:  
 Take comprehensive measurements of the NightGlow Phenomena over all latitudes and longitudes and over a period of 3 months (primary mission) to 12 months (extended mission)
- Measurements in the 75-110 km altitude with [5] km spatial resolution
- Measure two to three bands of emissions in the spectral range of [550 - 880] nm with spectral resolution less than [1] nm
  - Preliminary bands: 558, 762 and 840 nm
- Detector
  - A. Rochas, M. Gani, B. Furrer, P. A. Besse, and R. S. Popovic, "Single photon detector fabricated in a complementary metal-oxide-semiconductor high-voltage technology", *REVIEW OF SCIENTIFIC INSTRUMENTS*, vol. 74(7), July 2003

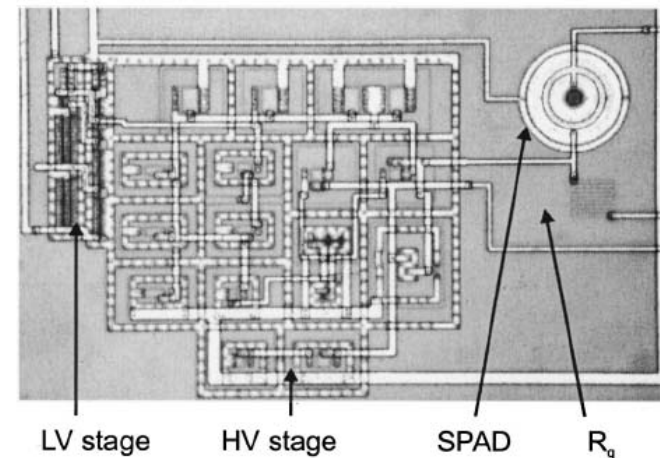
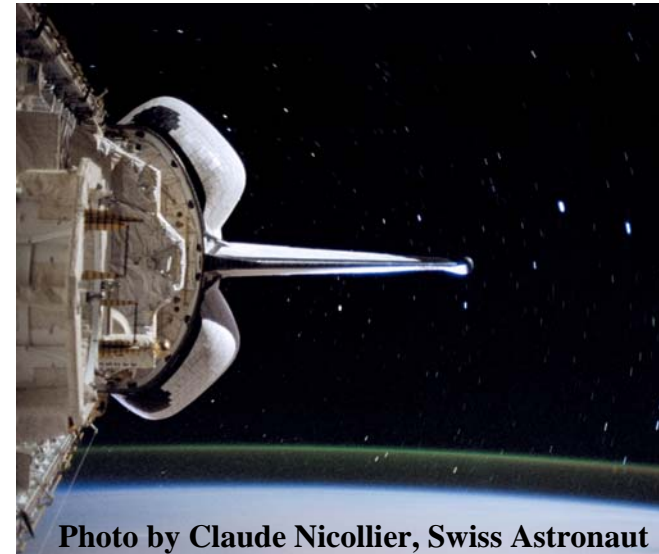
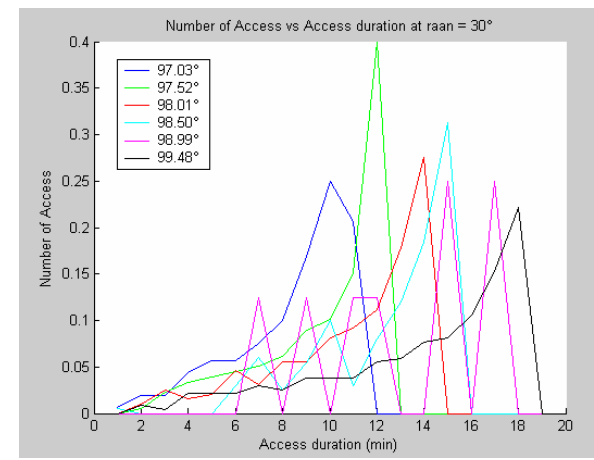
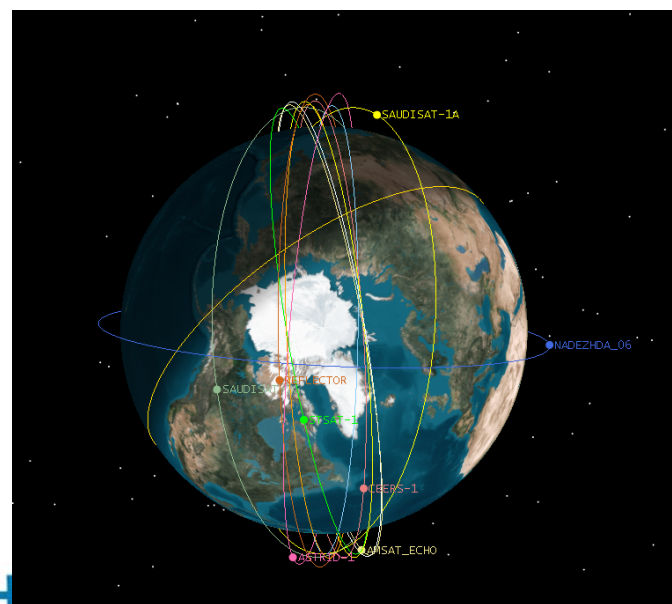
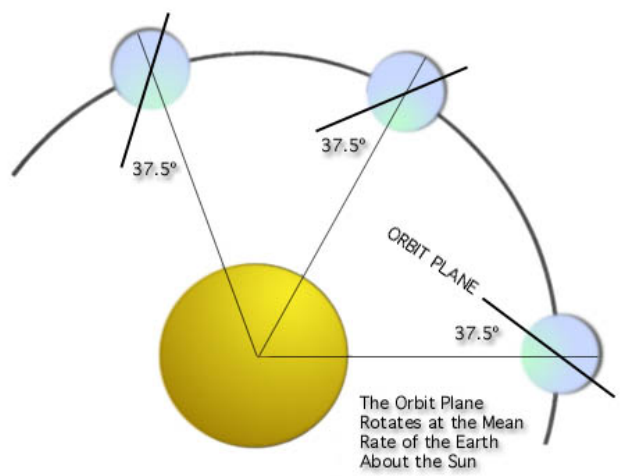
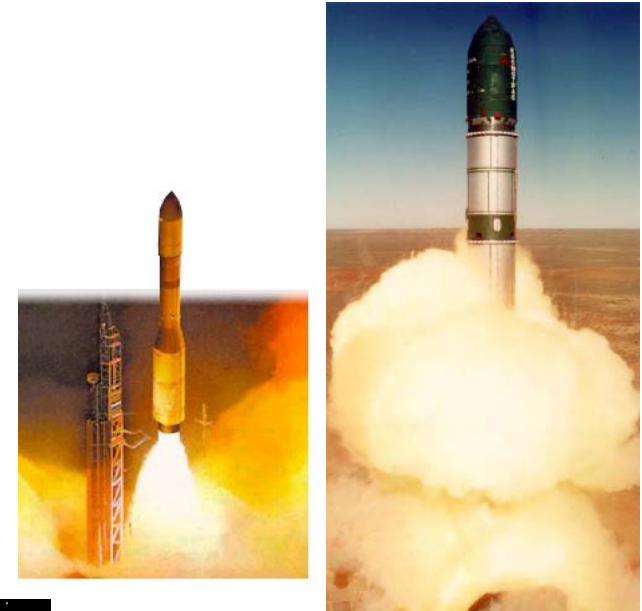


FIG. 3. Picture of the CMOS-high-voltage SPAD. The circular SPAD and the quenching resistor can be seen on the right-hand side. The low-voltage and the high-voltage parts of the comparator can be seen as well.

# Preliminary Mission Assumptions

- Launch DNEPR or VEGA launch vehicles
- Orbit Most likely Sun-synchronous
- Inclination  $\sim 97 - 99^\circ$
- Orbital Altitude  $\sim 400 - 1000$  km
- Orbital period  $\sim 90 - 105$  min
- Eclipses  $\sim 30\%$  of orbital period
- Avg. power  $\sim 1.5$  W
- Mean pass duration  $\sim 10$  min
- Data downlink rate  $\sim 1$  kbps



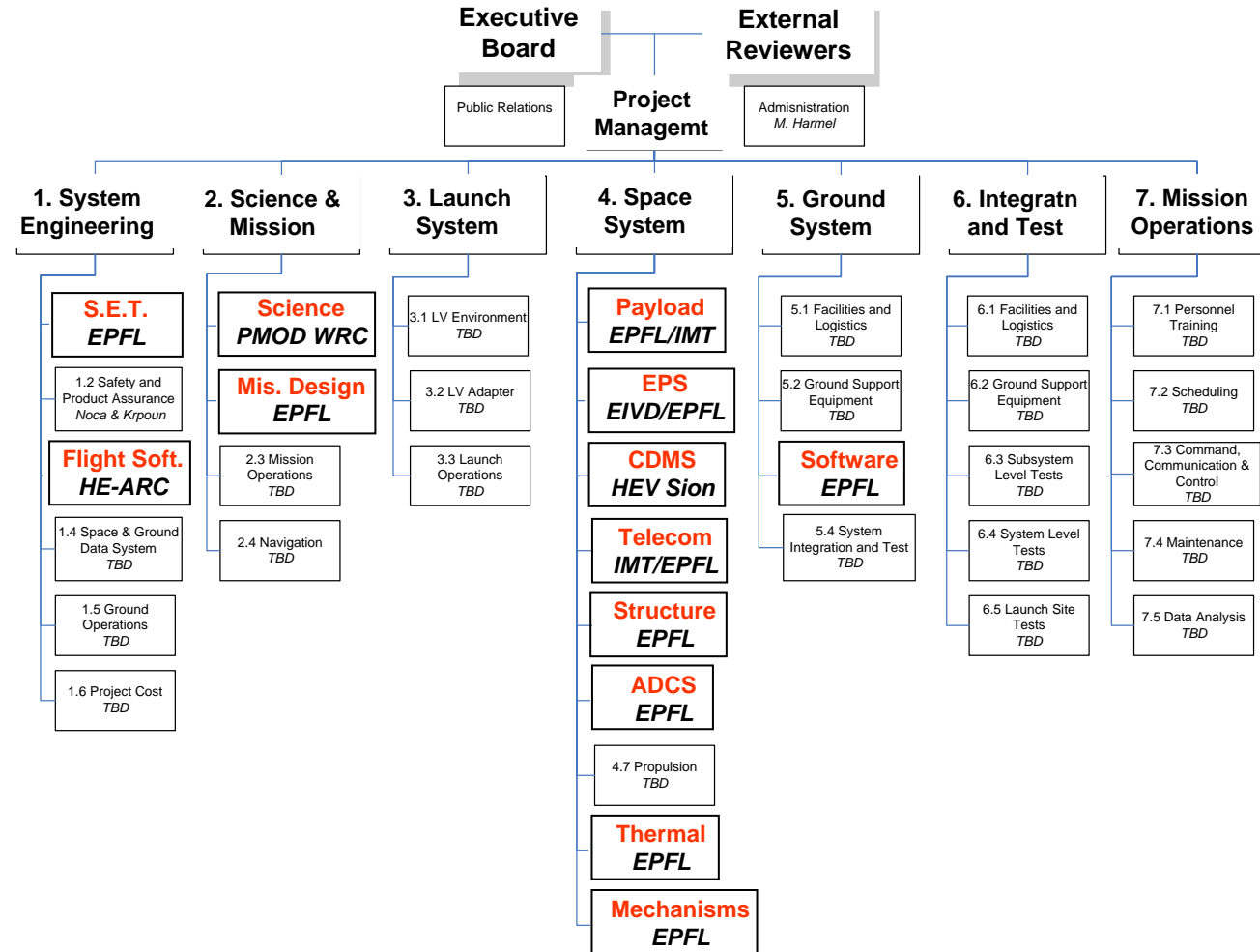
# Project Organization (1/2)

- Key SwissCube subsystem responsibility is spread across several labs and universities.  
Current partners:

- EPFL: 10 labs
- Université de Neuchatel: 3 labs
- HES Sion: 1 lab
- HES Yverdon: 2 labs
- HE-ARC: 1 lab

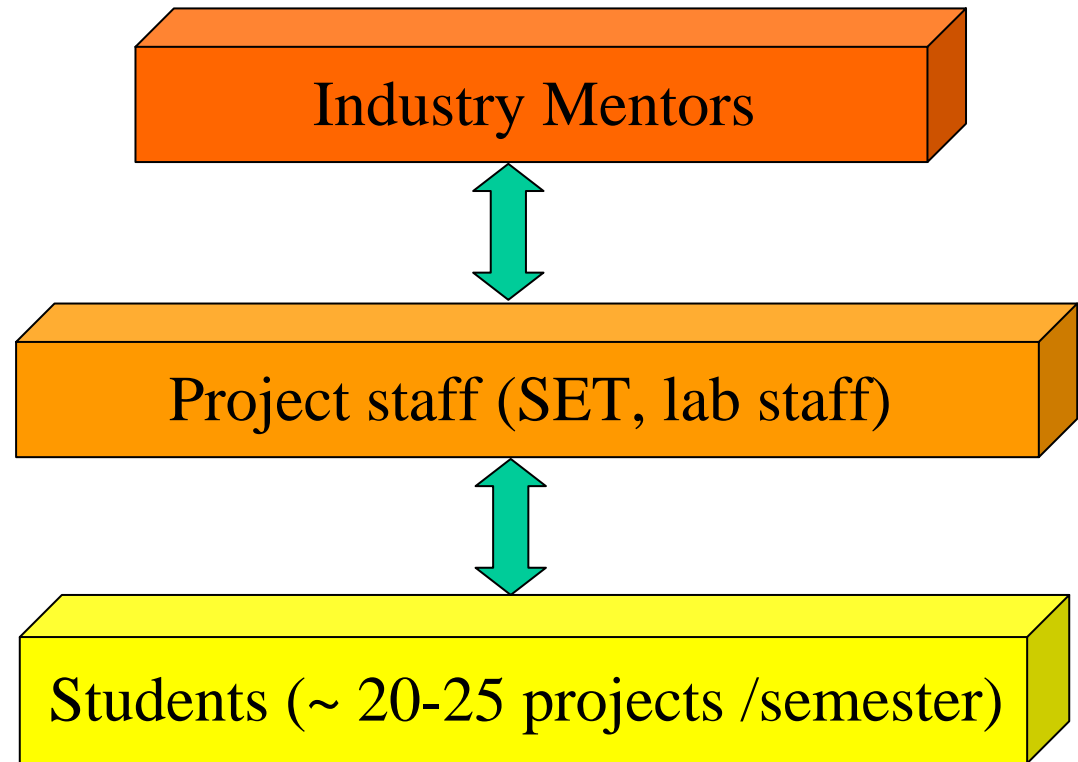
- Executive Board includes industry sponsors and university representatives

- External reviewers include Swiss industry representative and ESA partners

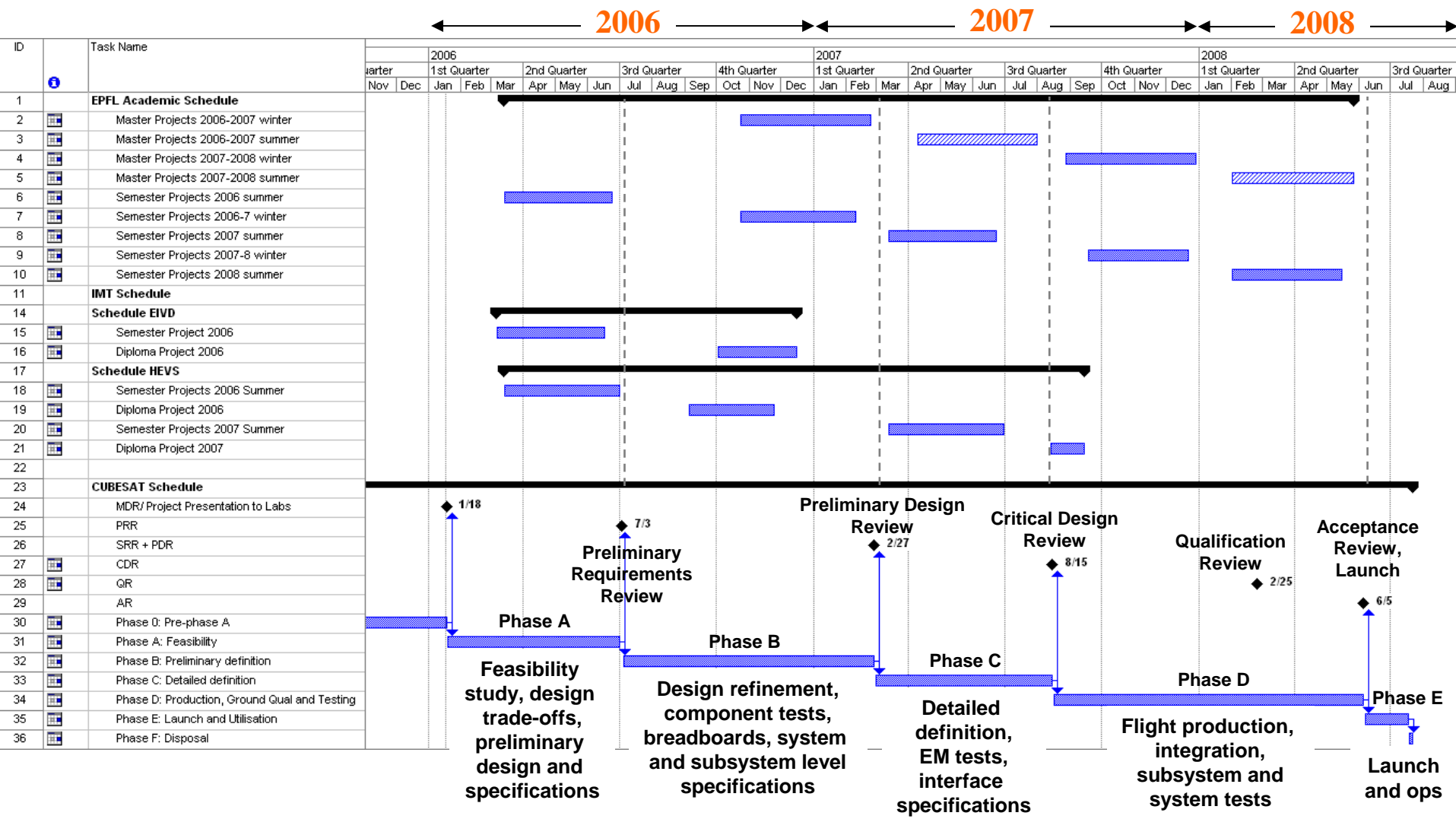


# Project Organization (2/2)

- Most of the work will be done by the students
  - Concurrent engineering environment
  - Multi-center communication via video conferences
- In each lab **one scientific staff** is leader of a given subsystem, supervises the student projects on that topic, coordinates with engineering team and provides continuity over time.
- **System Engineering Team (SET)** provides oversight and coordination.
- Each subsystem will have an expert mentor in industry.

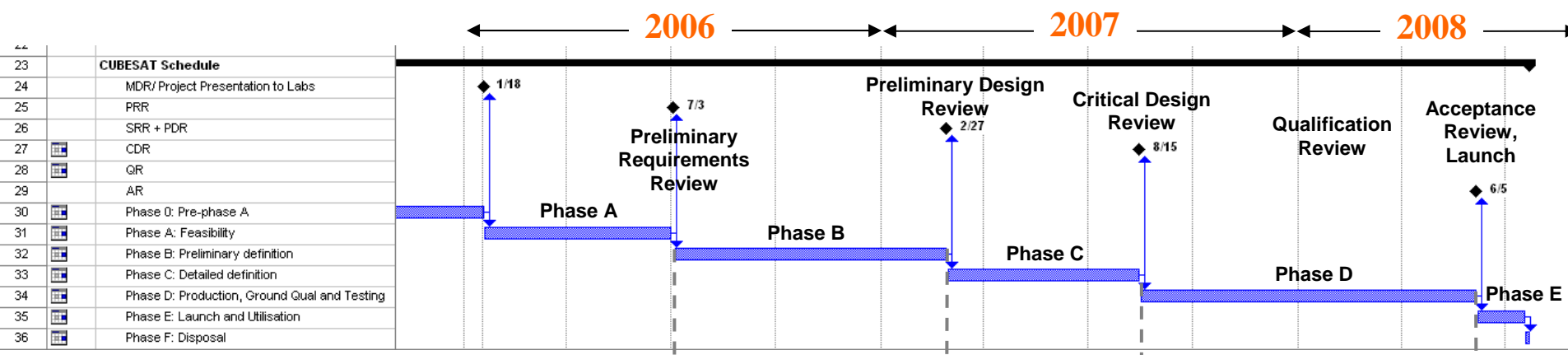


# Project Schedule

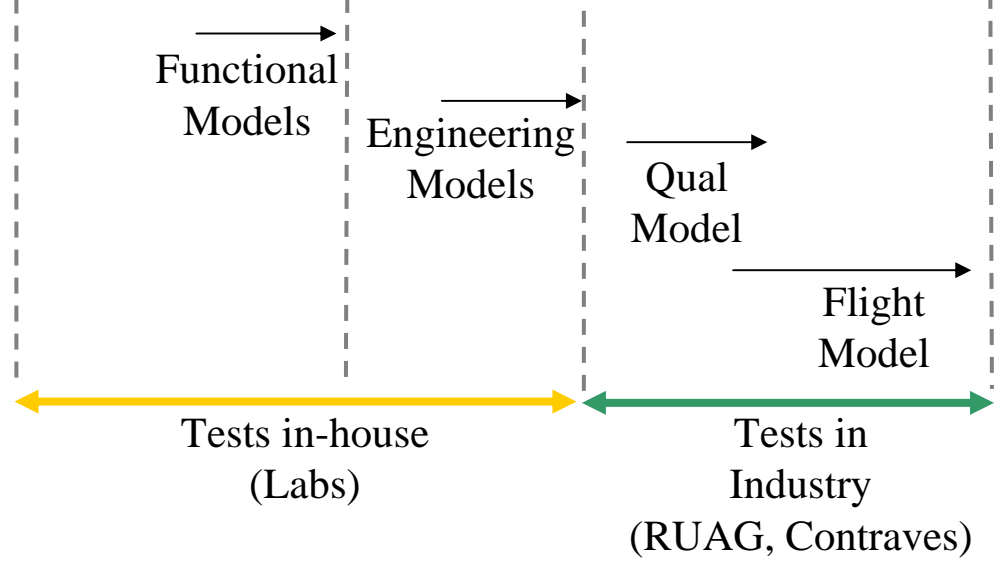




# Preliminary Test Schedule and Philosophy



- The SwissCube is a relatively high-risk low-cost development, but testing will parallel as much as possible a high-reliability mission to ensure success
- Model Philosophy:
  - Set of breadboards/mockups in Phase B (Functional Models)
  - Engineering Models in Phase C
  - Qual Model and Flight Model in Phase D (prototype approach)
- Compliant with ECSS as much as possible



- Today's programmatic challenge
  - Efficient transfer of information between students over the different phases of development
- Today's technical challenge
  - Current payload asks for relatively tight pointing and stabilization requirements, and volume requirements
  - System studies and trades are on-going to find solution
- Need for communication with YOU
  - Assume that a great deal of information already exists
    - Parts list, what worked, what didn't?
  - Experience of Cubesat developments within a university environment
    - What worked, what didn't?
- Conclusion
  - Starting project in a multi-university environment
  - All advices, sharing of experience are welcome!