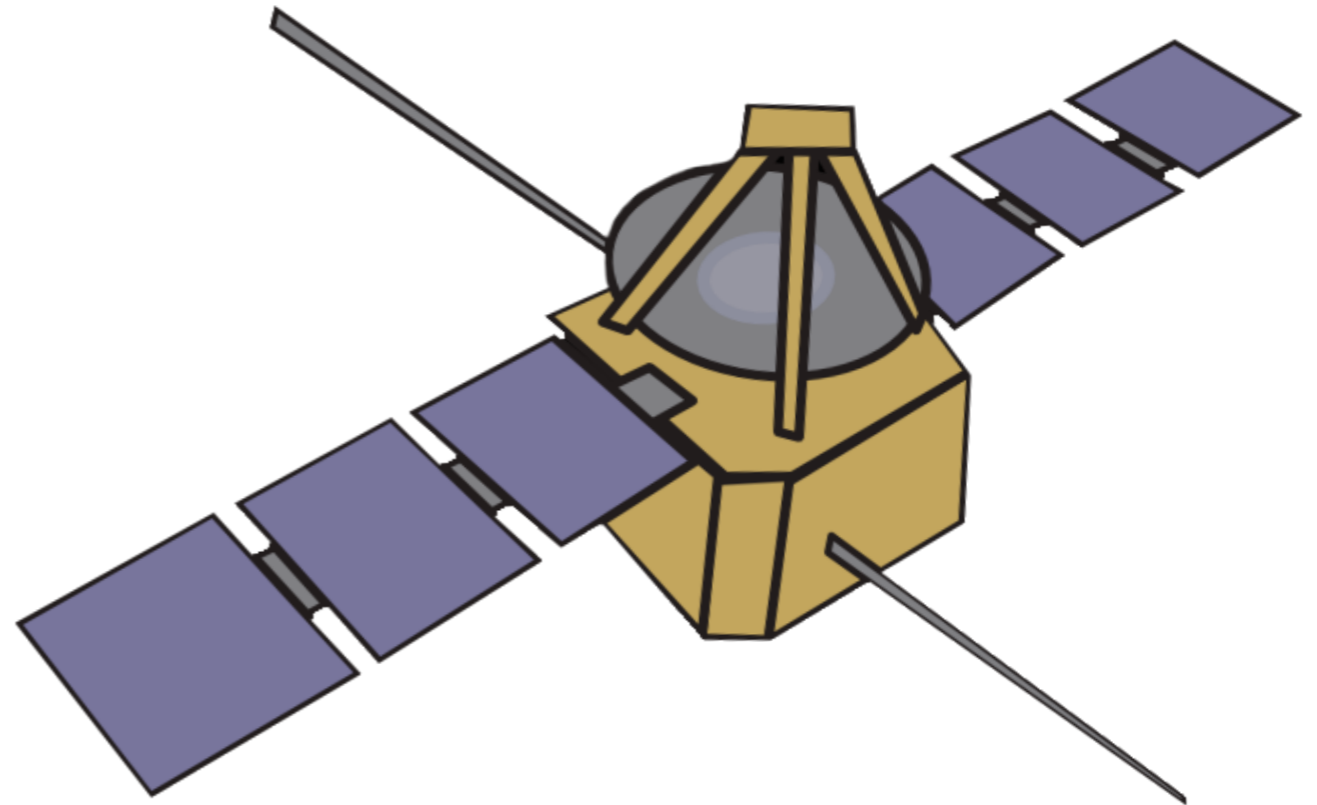


Application of 3D-Printing and COTS Components in Micro-Propulsion Systems

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Micro Propulsion

Cold gas propulsion

Compressed gas/Liquefied gas, Traditional/MEMS/3D-Printed

Chemical propulsion

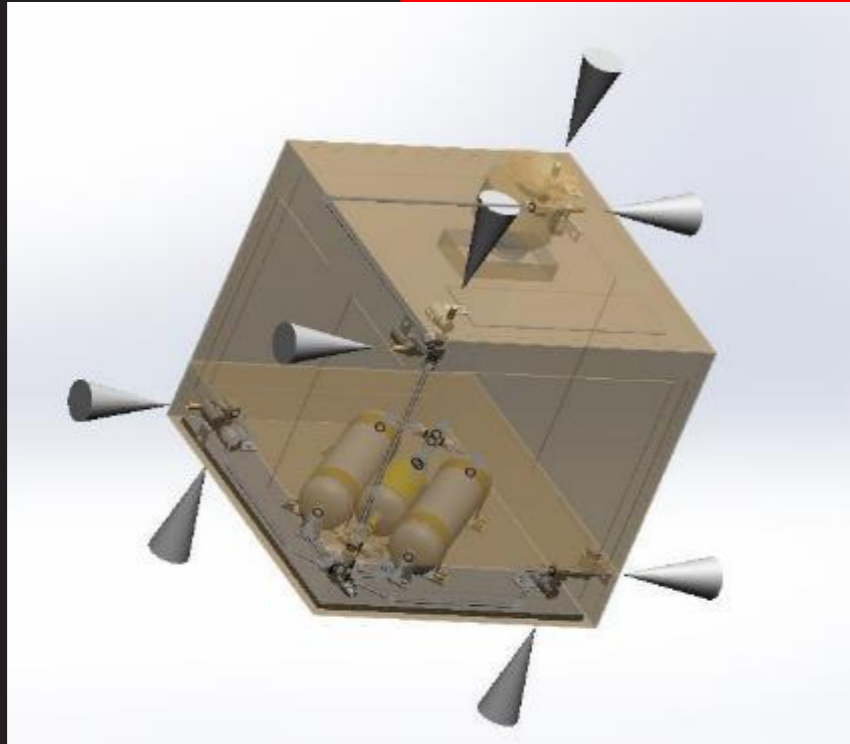
Monopropellant/Bipropellant

Electric propulsion

Arcjet/Hall Thruster/PPT/FEEP/Colloid thruster

Target Mission

Thrust Directions



25kg satellite

Power $\leq 15\text{W}$

System mass $\leq 6\text{kg}$

$\Delta V \geq 35\text{m/s}$

50mN (altitude control)

150mN (orbit control)

Thrust accuracy $\leq 5\%$

Target Mission

Cold gas system
(Liquefied gas)

Power $\leq 15\text{W}$

System mass $\leq 6\text{kg}$

$\Delta V \geq 35\text{m/s}$

Pressure/thrust control

50mN (altitude control)

150mN (orbit control)

Thrust accuracy $\leq 5\%$

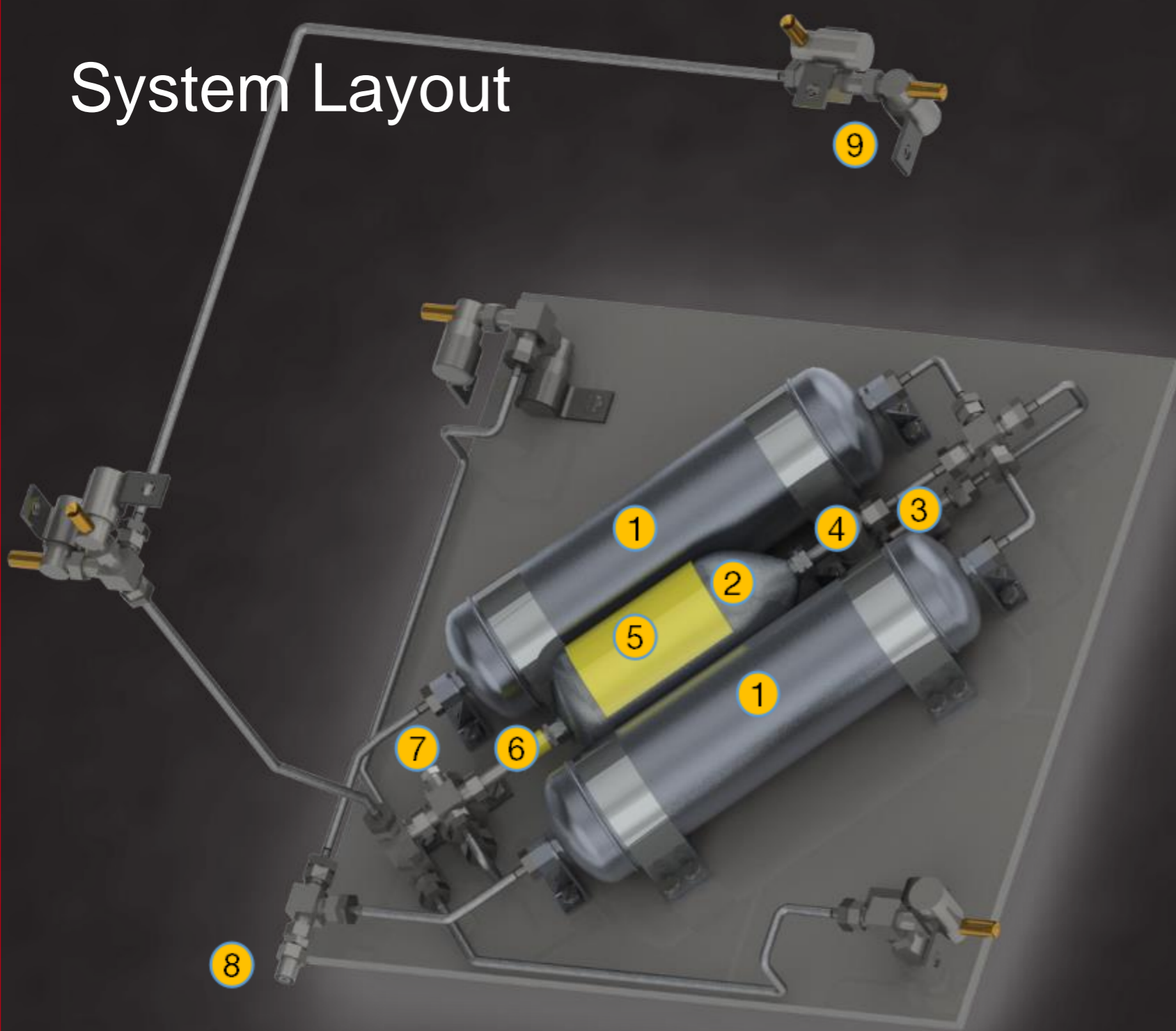
Target Mission

Quick and Cheap

Additive Manufacturing

Commercial Off-The-Shelf
Components

System Layout



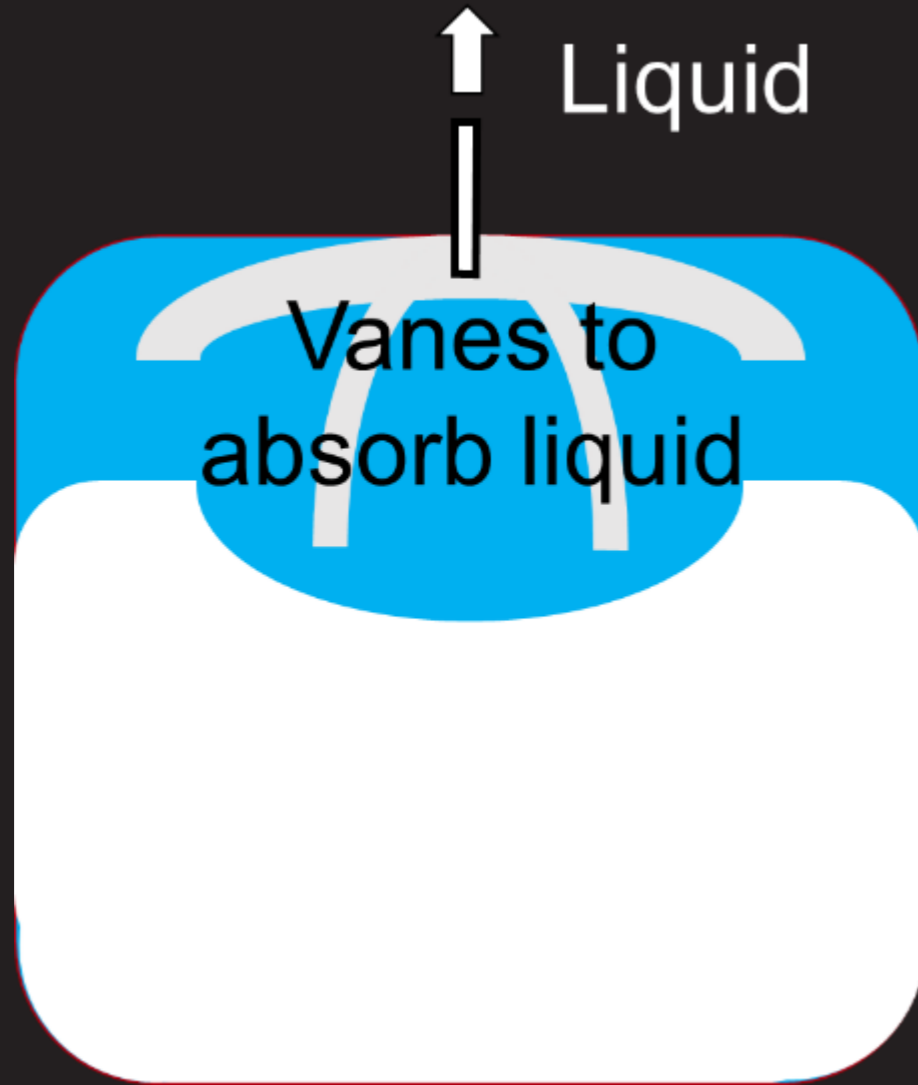
- 1 Tank×2
- 2 3D-Printed Plenum
- 3 Pressure Sensor ×2
- 4 Isolation Valve
- 5 Heater
- 6 Filter
- 7 Temperature Sensor
- 8 Fill/Drain Valve
- 9 Solenoid Valve&Nozzle ×8

- Simplified Tank Design
- Thrust Control
- Additive Manufactured Components
- Thrust Measurement

OUTLINE

Simplified Tank Design

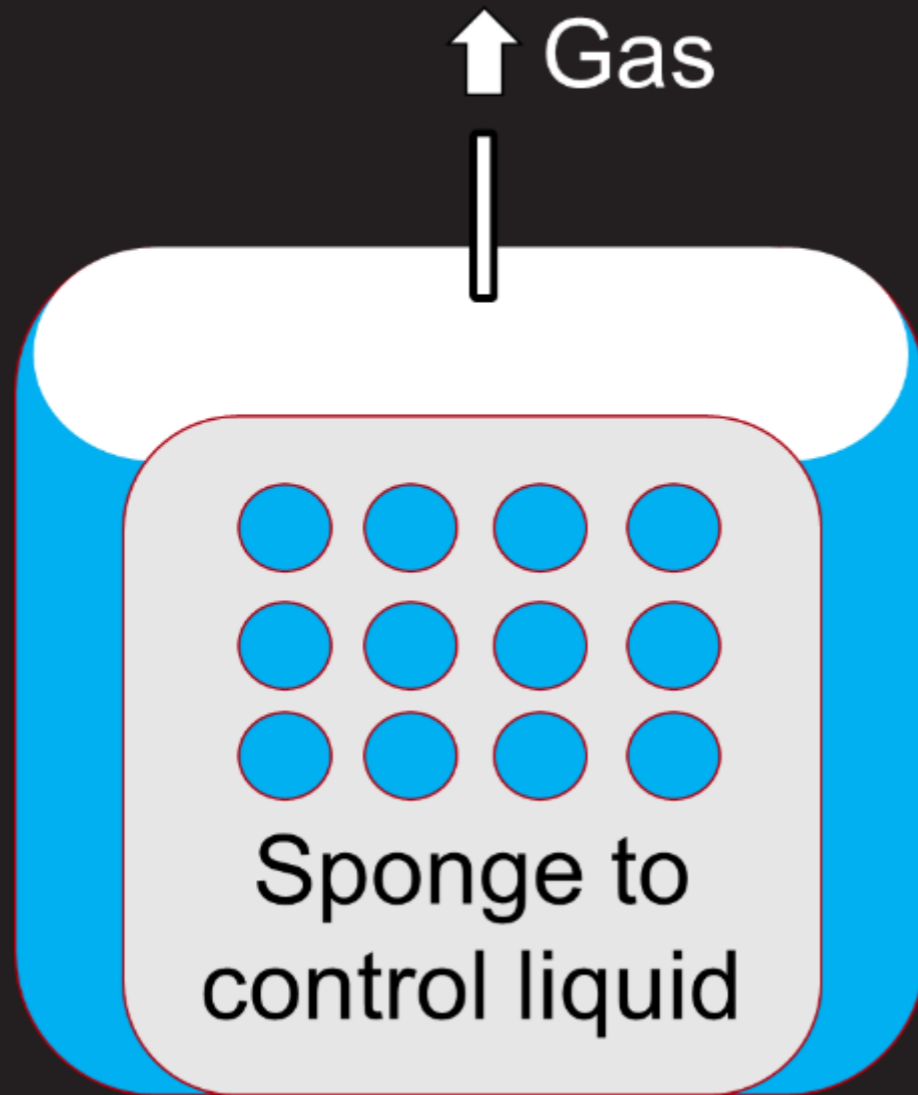
Traditional Tank Design with PMD



For chemical propulsion:

Light weight
Cheap
Reliable

Traditional Tank Design with PMD



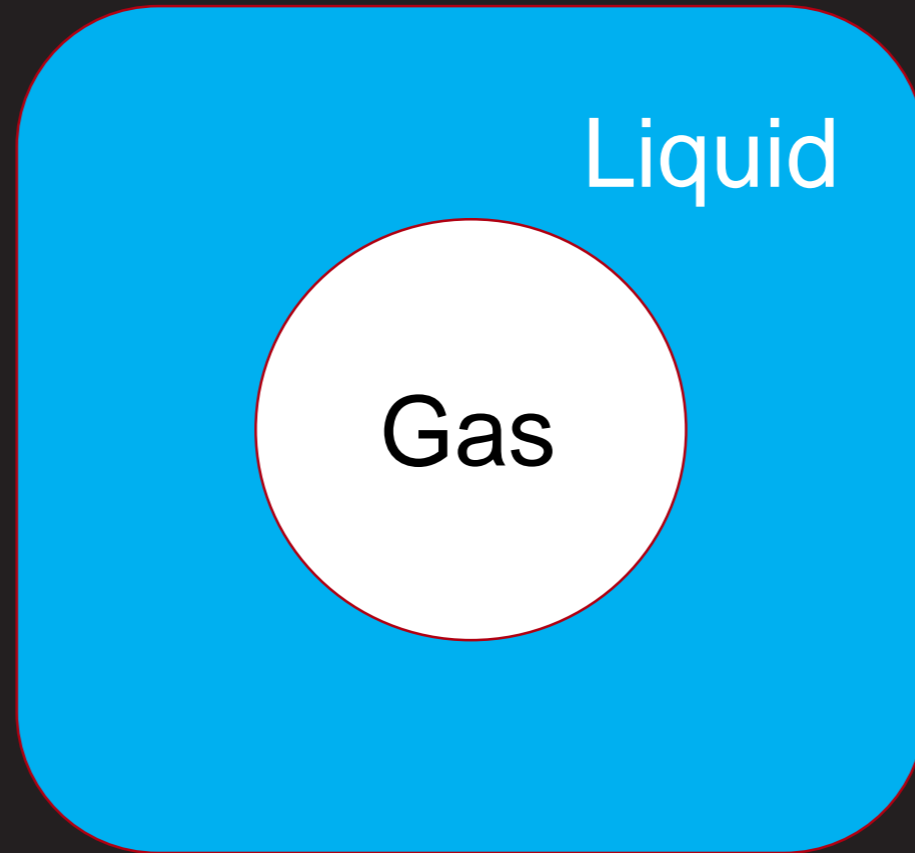
For liquefied gas propulsion:

Heavy

&

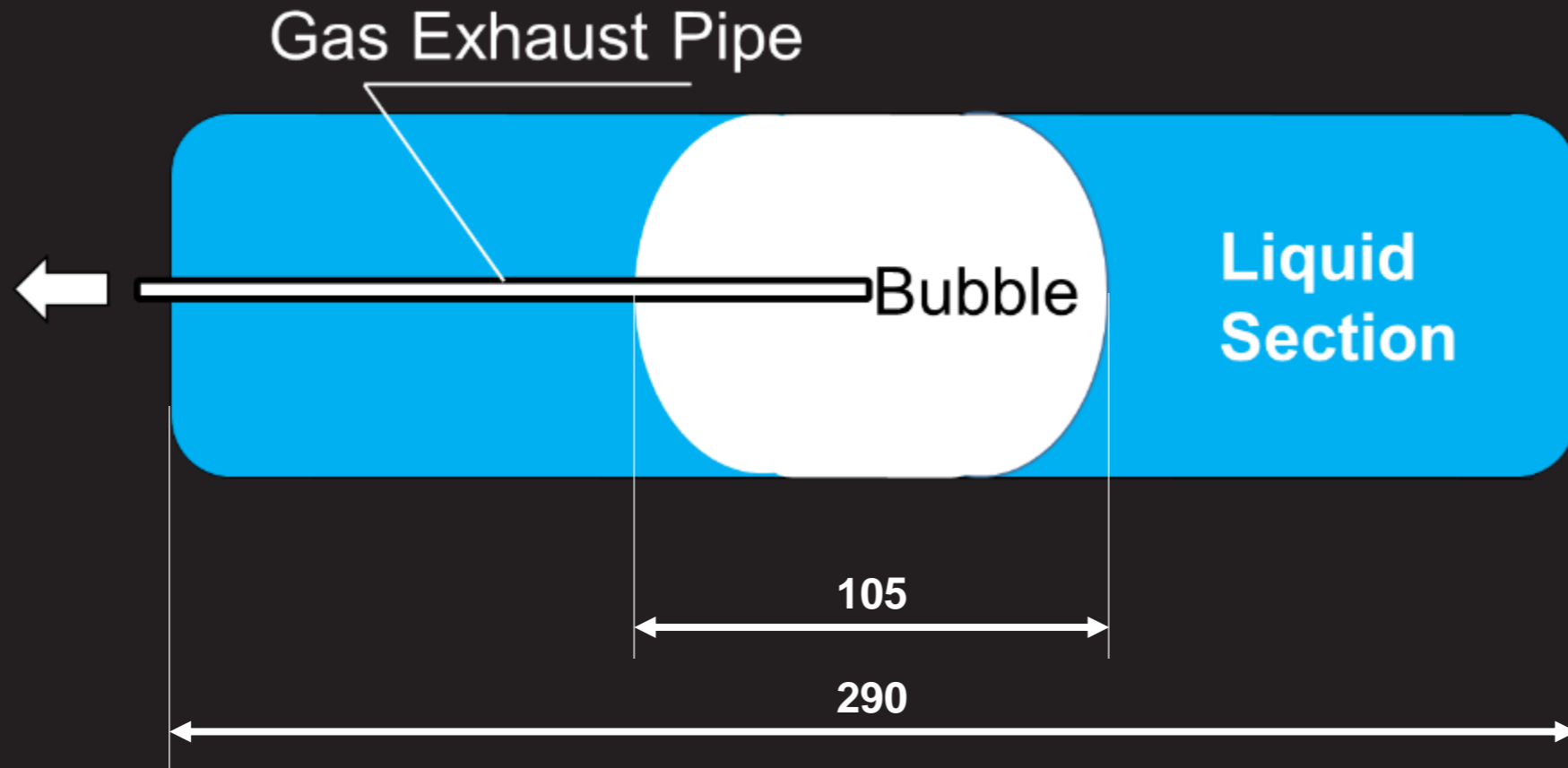
Size of existing tanks
don't match the satellite

Simplified Tank Design



Two-phase interface in microgravity
(when infiltrating)

Simplified Tank Design



Simplified Tank Design

Lightweight

Cheap

Only add a pipe to the tank

No sloshing control

Thrust Control

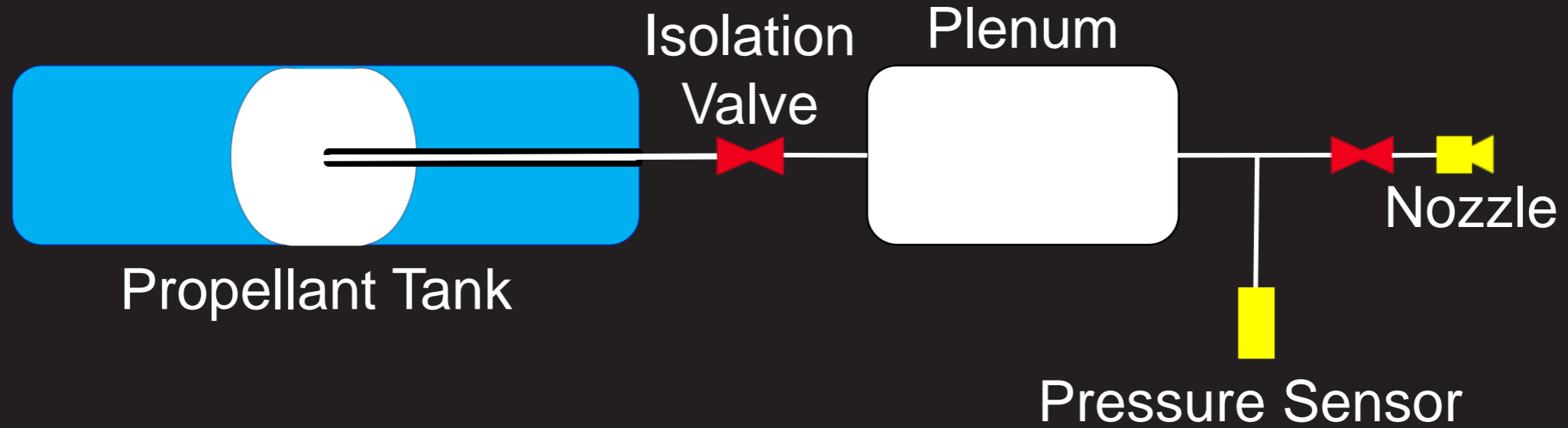
How to regulate pressure/thrust?

Mechanical regulating valve

Electronic closed-loop valve

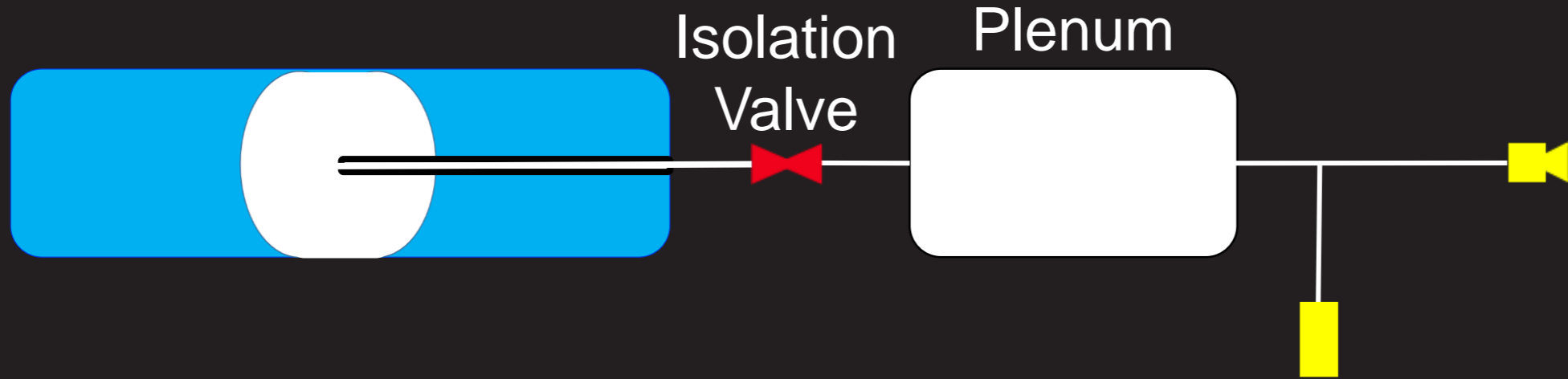
Bang-Bang control in a plenum

Bang-Bang Control in Plenum



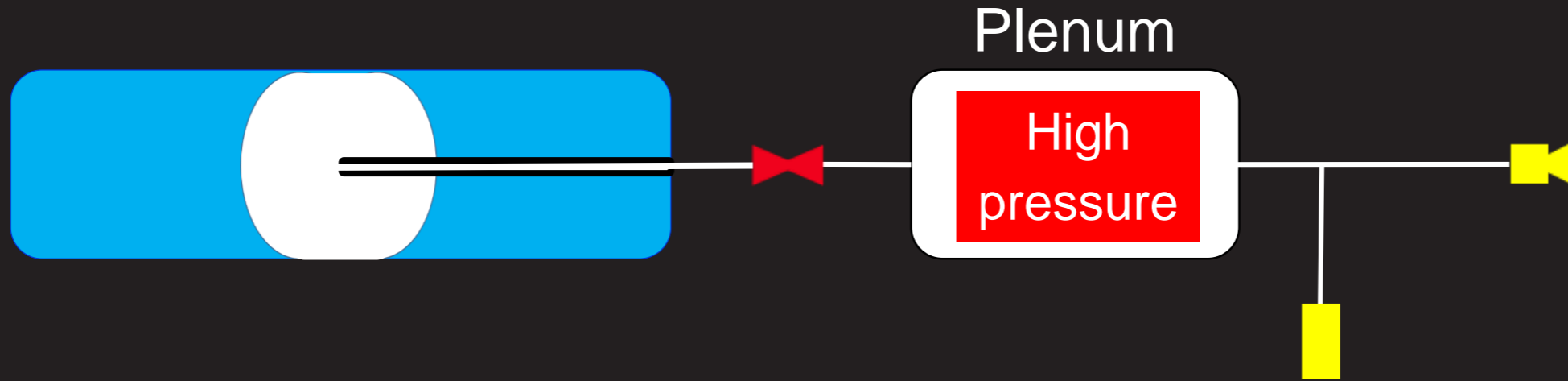
Pressure in Plenum → Thrust

Bang-Bang Control in Plenum



Nozzle valves are omitted here

Bang-Bang Control in Plenum

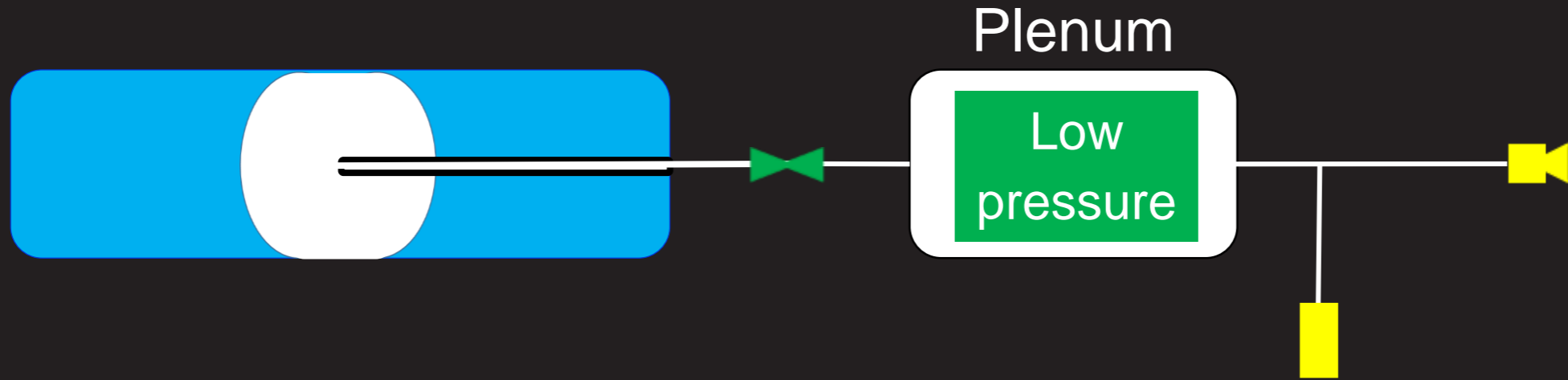


Closing time

Opening time

Nozzle valves are omitted here

Bang-Bang Control in Plenum



Closing time

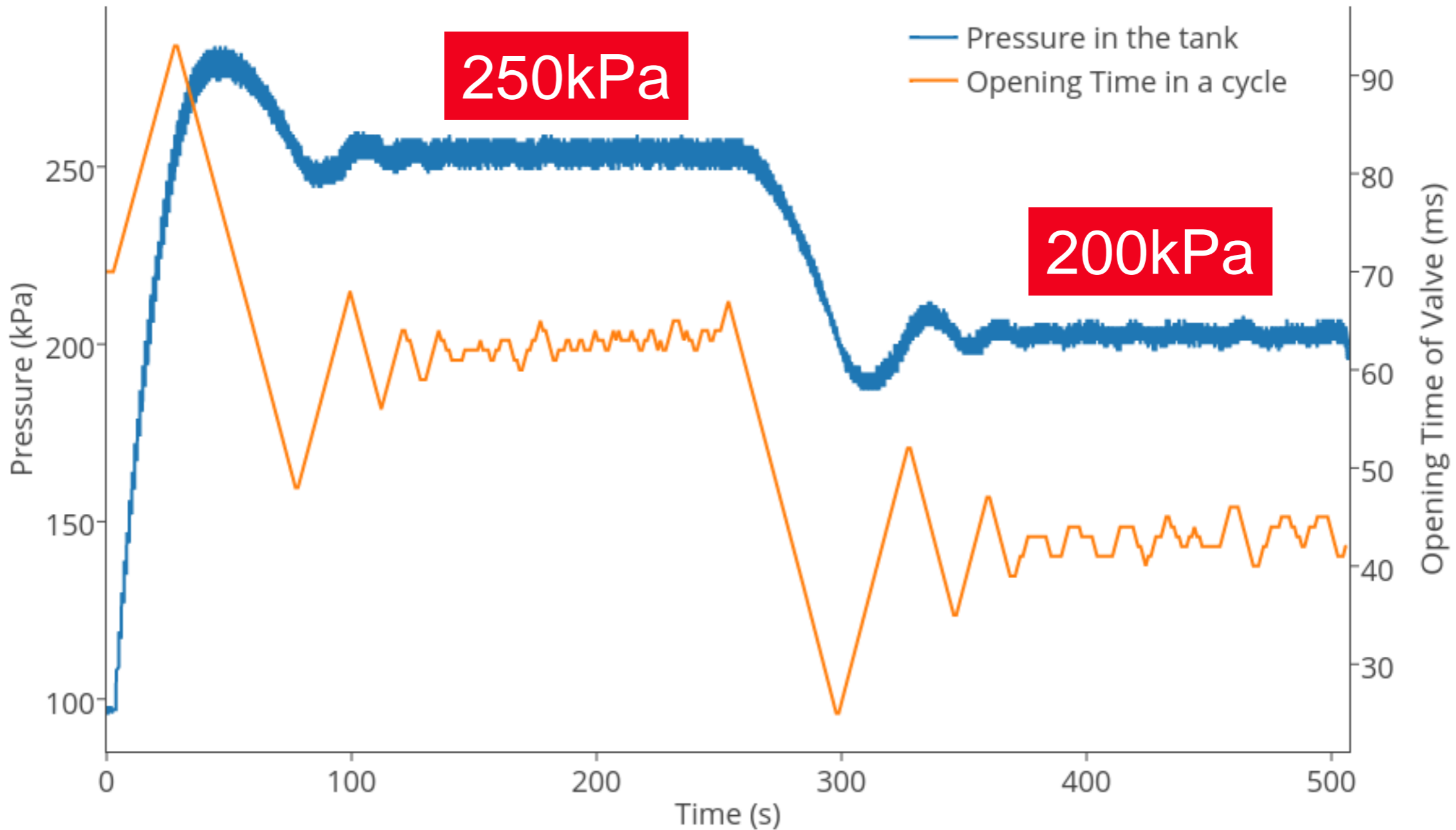
Opening time



Nozzle valves are omitted here

Why choose plenum?

If residual liquid enter the pipeline,
the plenum will help it vaporize



5% accuracy achieved by a 0.5L plenum

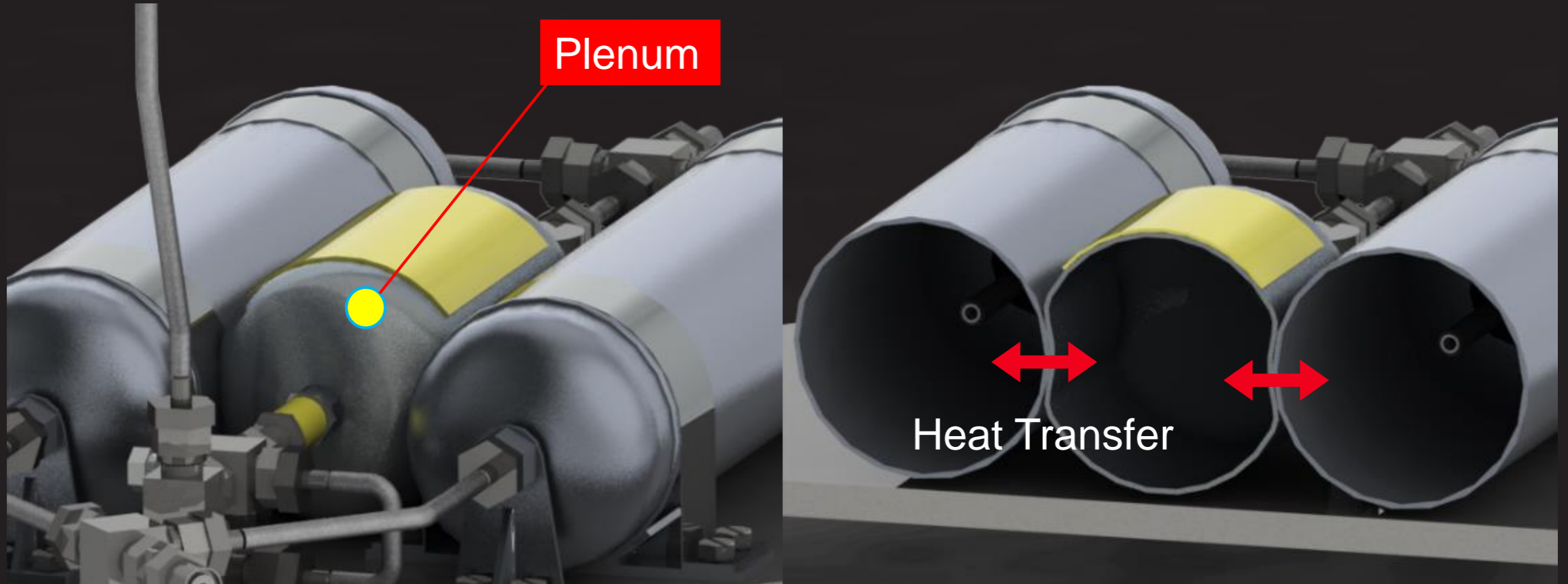
Additive Manufactured Micro Propulsion

Why use additive manufacturing?

The same electronics and valves can be integrated in different missions **quickly**

But tank production needs more time for **welding and examination**

Why use additive manufacturing?



The plenum is a special-shape pressure vessel

Which process?

Metal AM

Powder bed fusion **SLS/SLM/EBM**

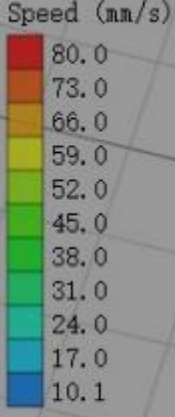
Small size, mature

Powder-based deposition **LENS**

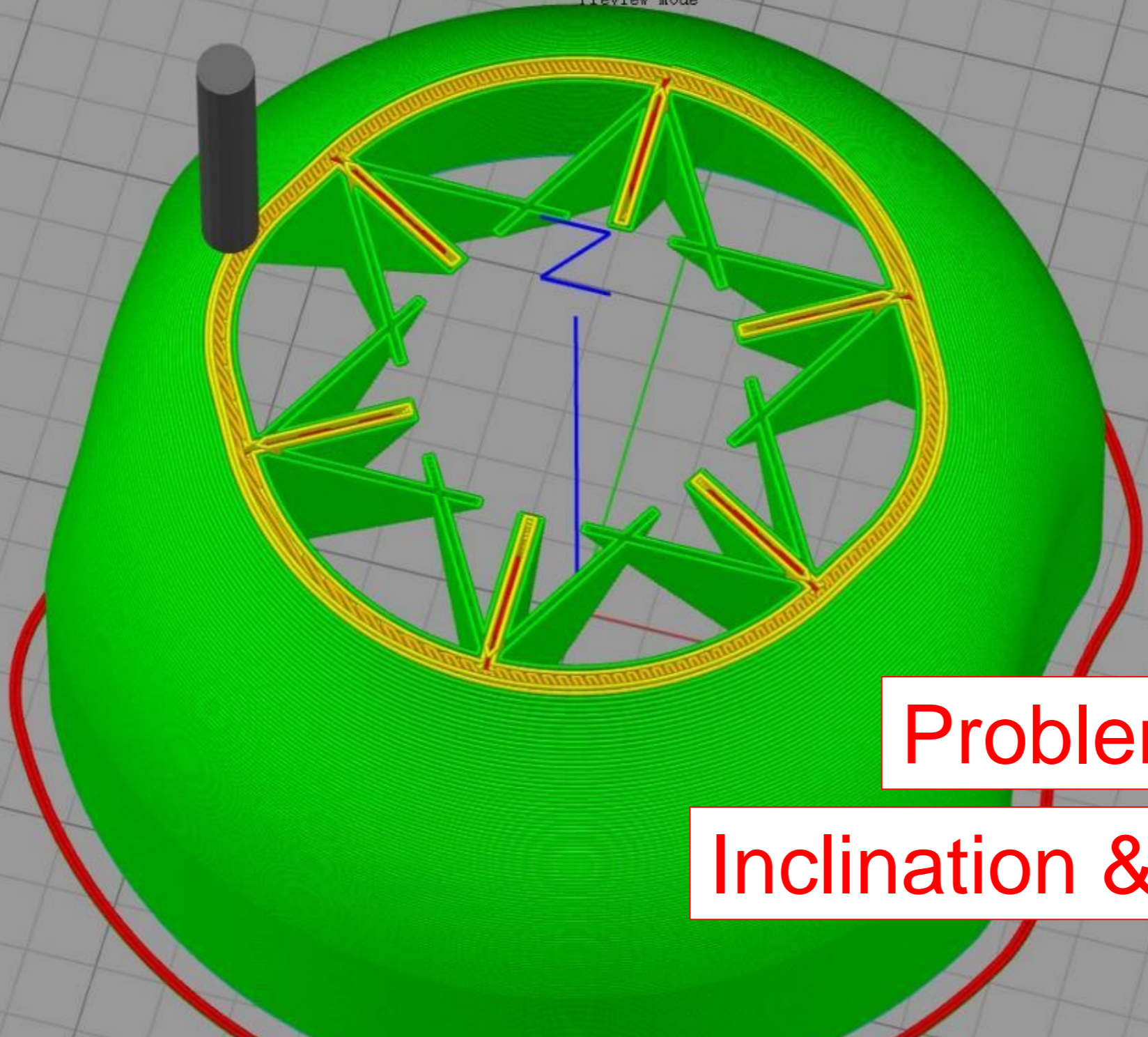
Large size

Wire-based deposition **EBF3**

Larger size, quicker



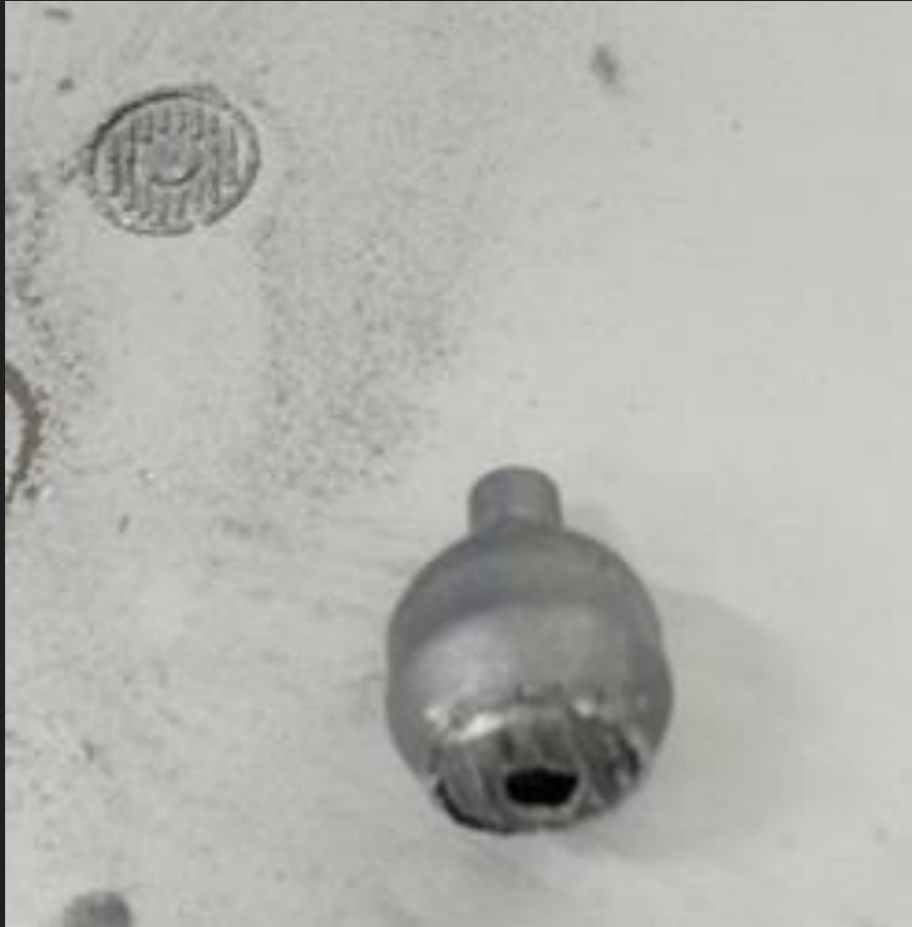
Preview Mode



Problem 1

Inclination & Support

3D-Printed Components



Scaling Model

Problem 2

Thin wall



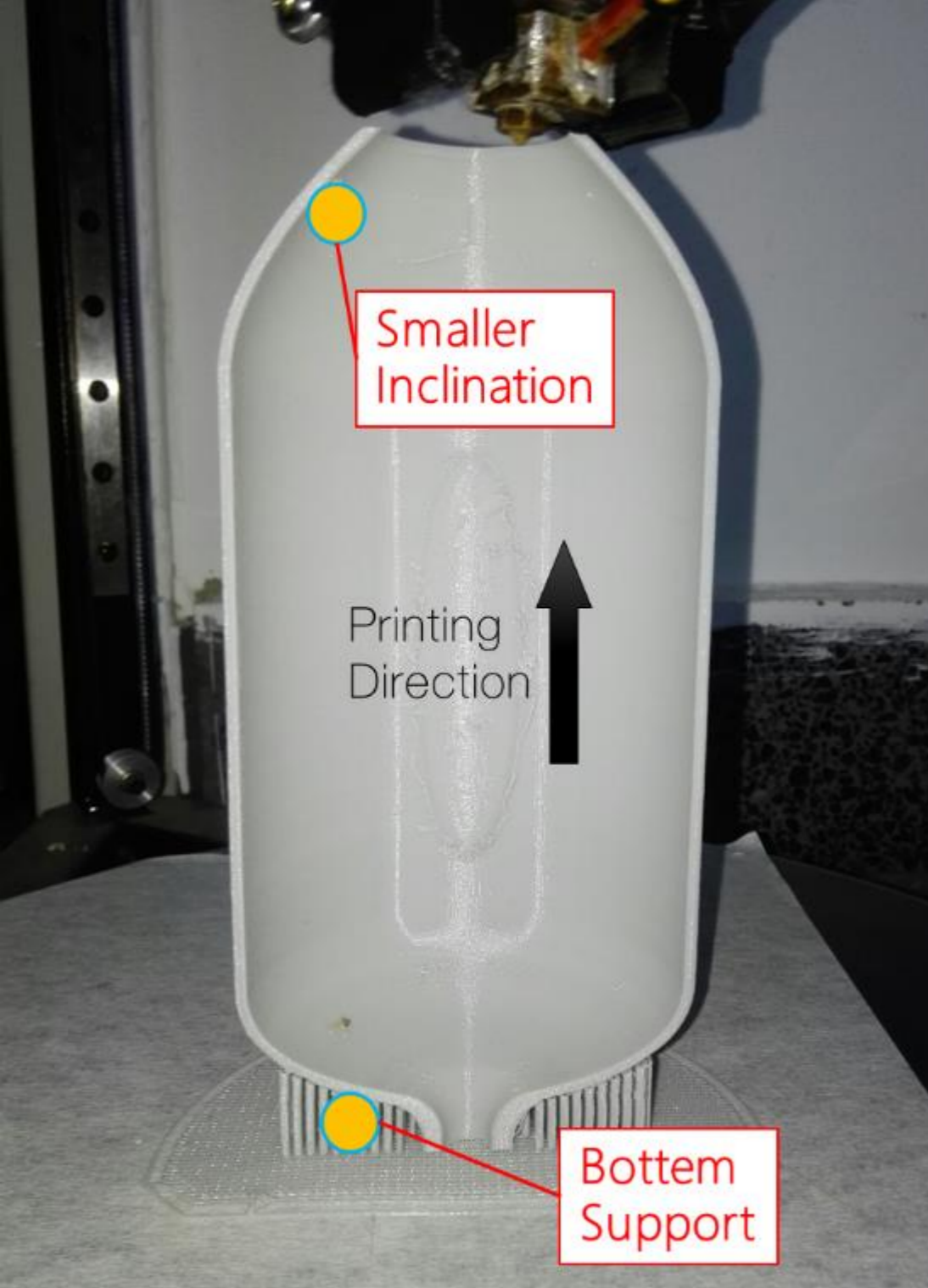
Damage during removing

1:1 Model of the Plenum

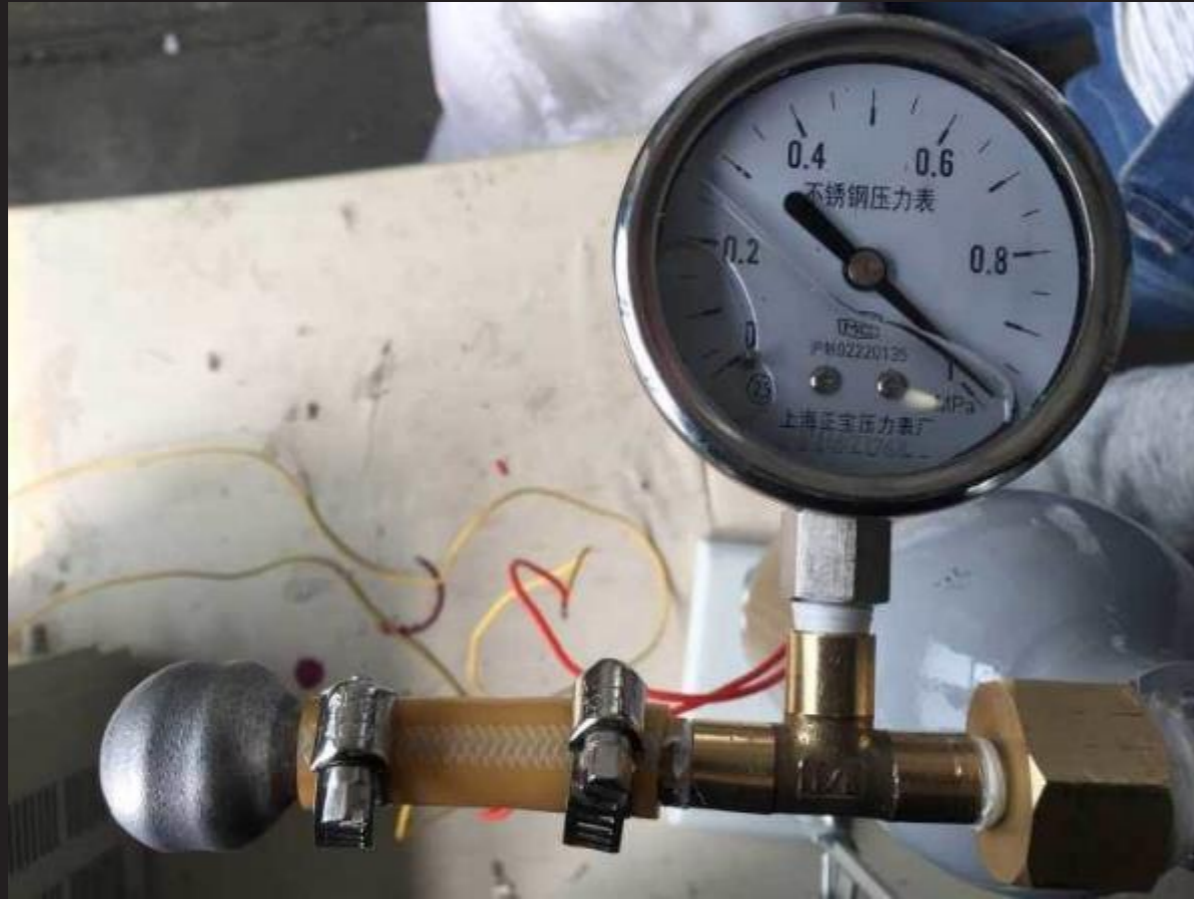
Solutions:

Different Inclinations

Minimum wall thickness
 $\geq 1.5\text{mm}$



3D-Printed Components

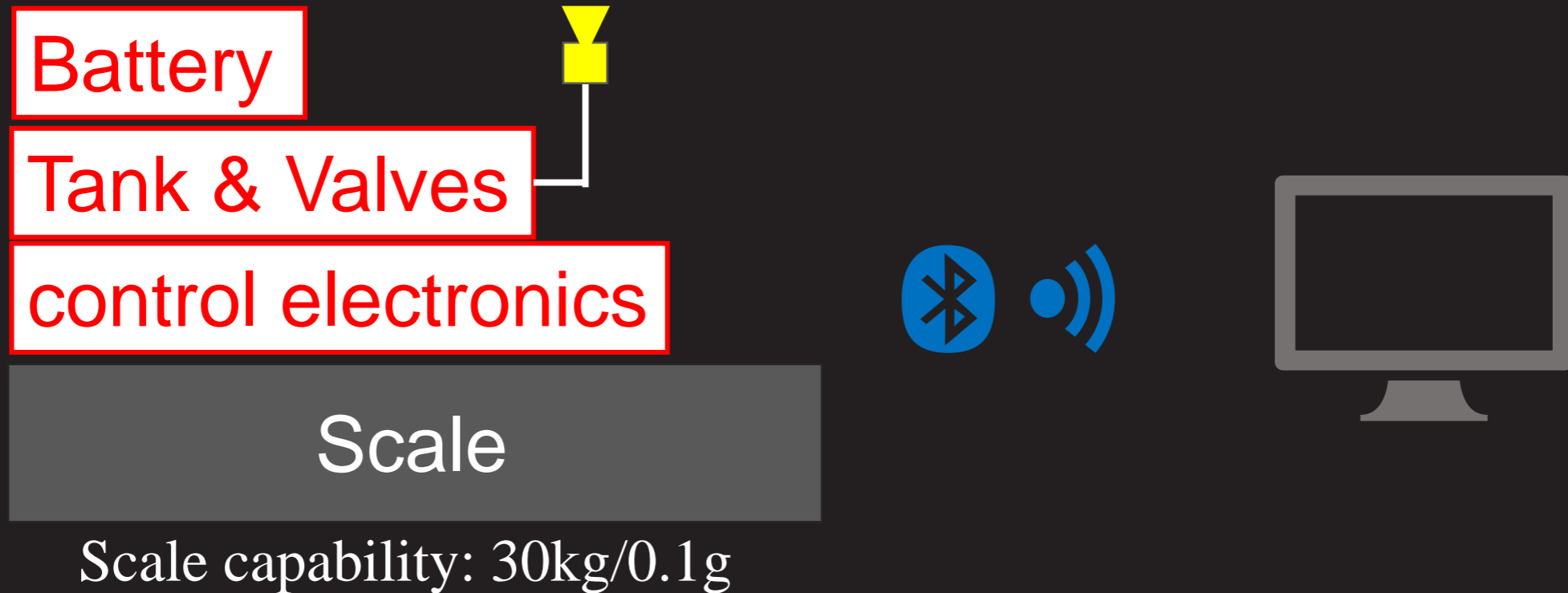


Scaling model withstood a 1MPa(145Psi) test.

Processing cost of the complete plenum is **\$1860.**

Thrusts Measurement

Thrusts Measurement

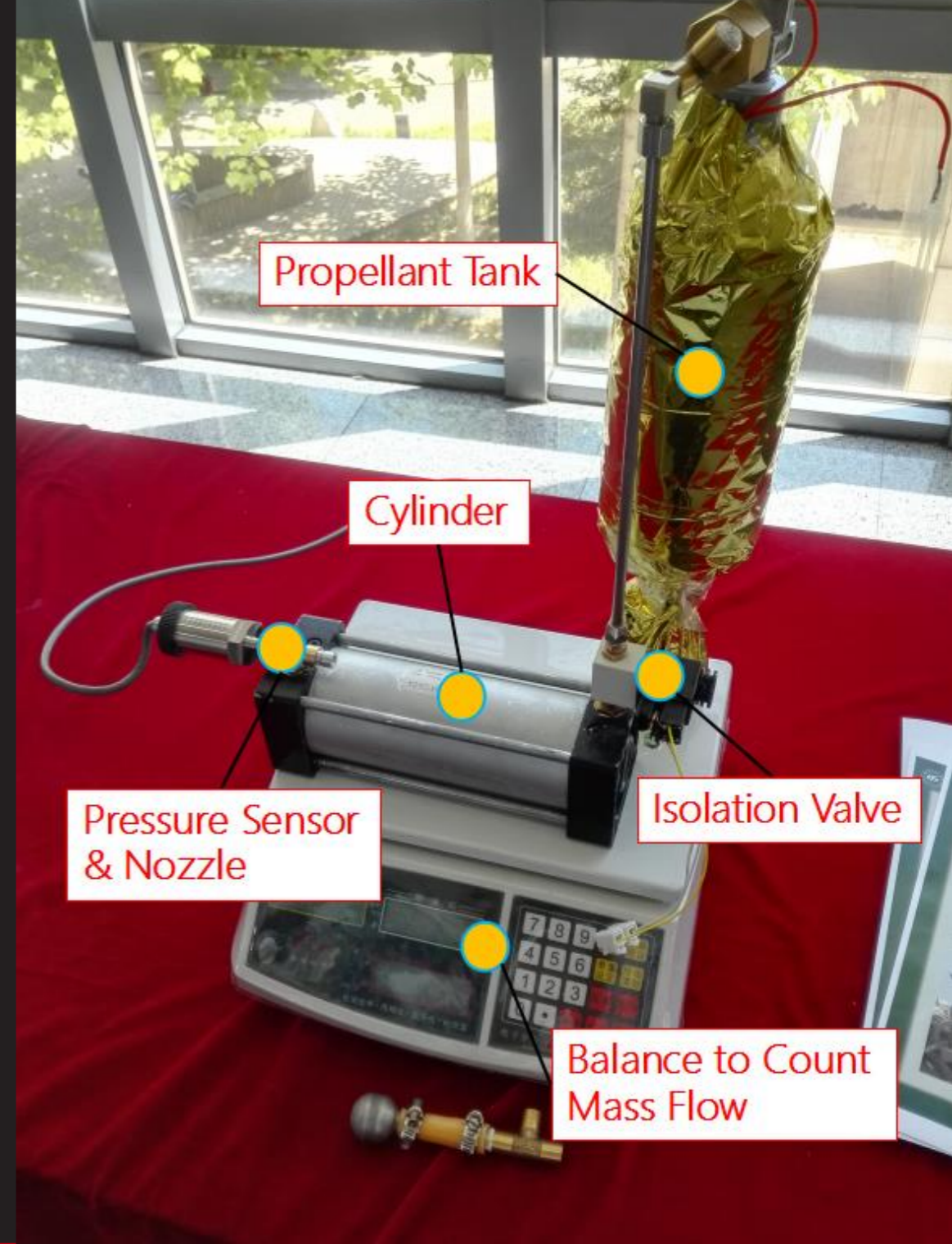


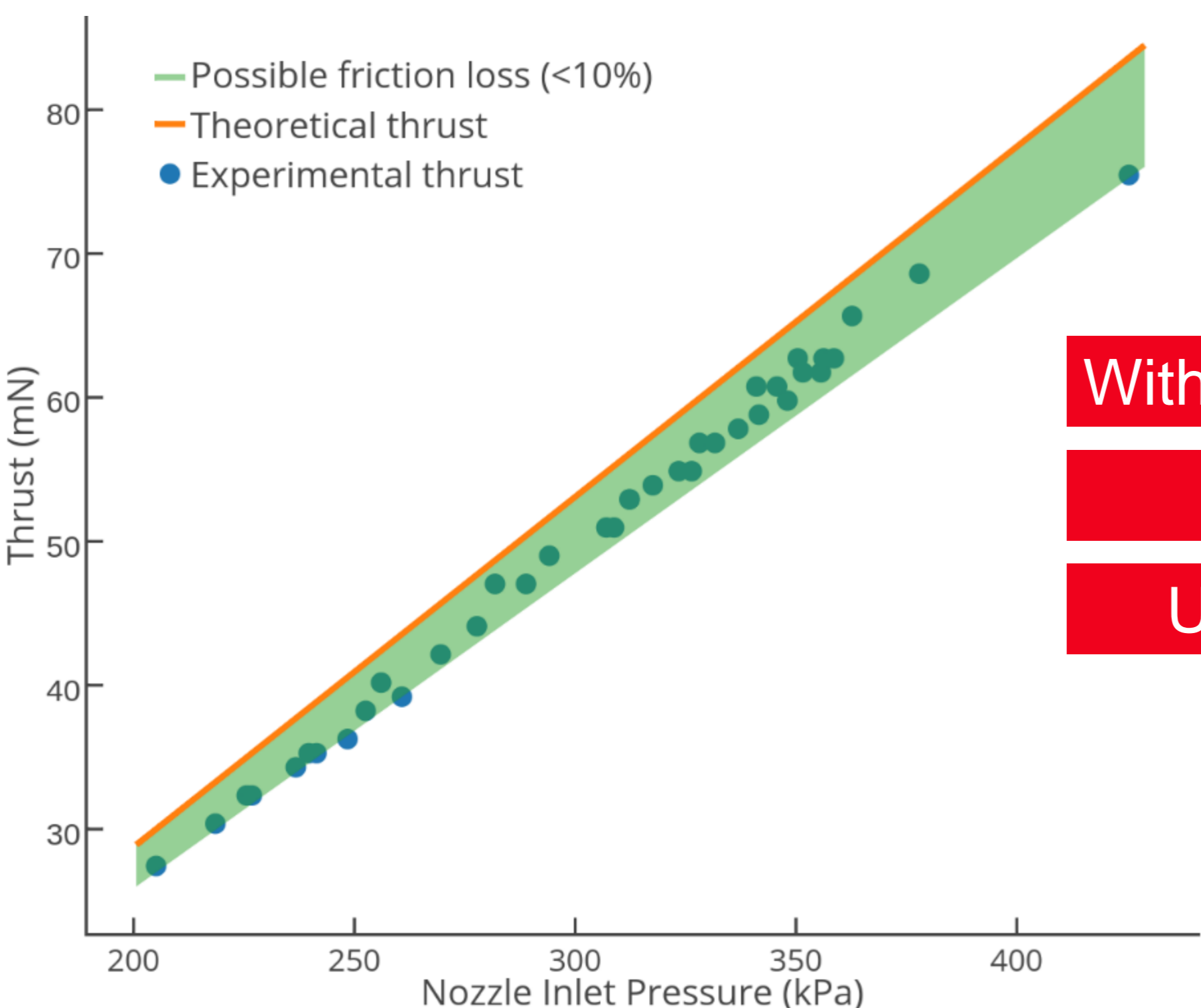
Whole propulsion system is placed on the scale

No wiring/pipeline interference

Thrusts Measurement

Thrust measurement of prototype

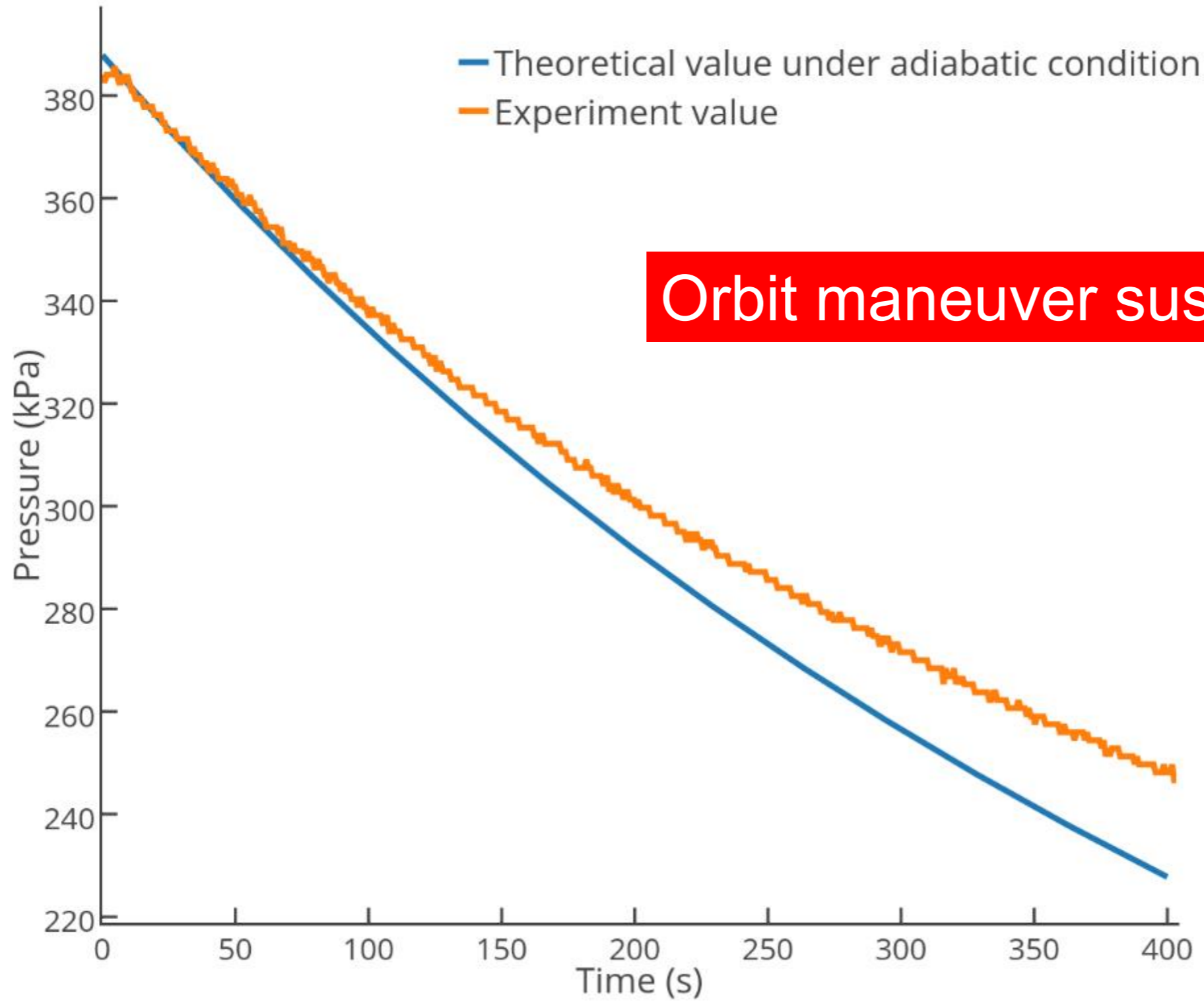




Within friction loss range

Calibrated

Uncertainty $\leq 1\text{mN}$



Orbit maneuver sustainable time ≥ 200 s

Conclusion

Conclusion

1. Prototype test have been completed, and the design meets the requirement of the target mission.
2. Commercial SLM method can meet the requirement of small-satellite-size components now.
3. Novel gas extract method has been proposed.

A sophomore research program in BUAA

Thank you for listening!
And questions?



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