



NOAA EON-IR CubeSat Study for Operational Infrared Soundings

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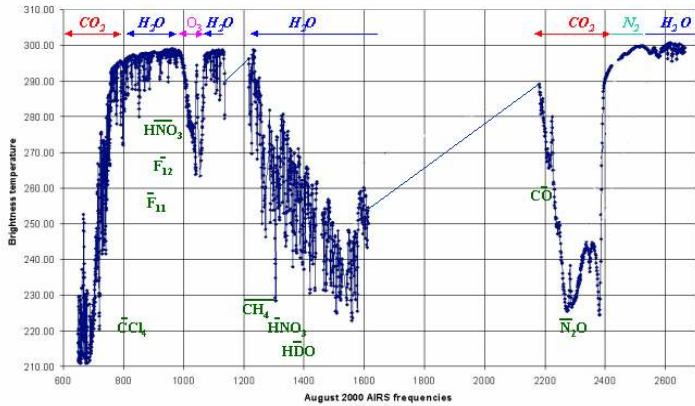
Riverside Technology



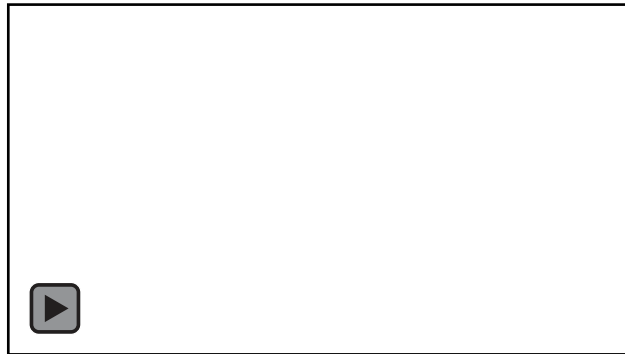
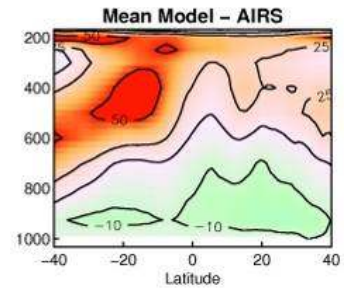
IR Sounders Support Weather Forecasting and Climate Science



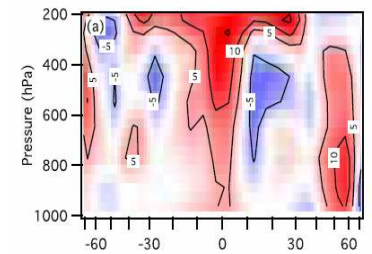
AIRS Channels for Topical Atmosphere with $T_{surf} = 301K$
Full Spectrum



Water Vapor Climatology
(Pierce, Scripps, 2006)



Water Vapor Feedback
(Dessler, Texas A&M, 2008)



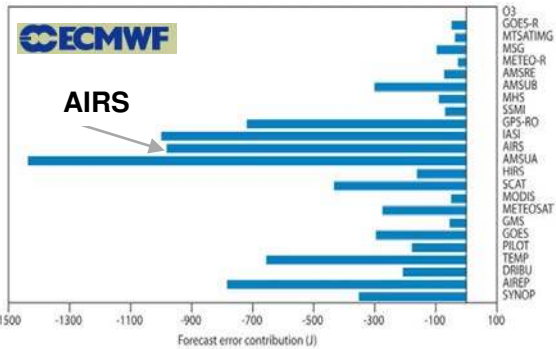
JPL/GSFC



NOAA
NESDIS/NCEP

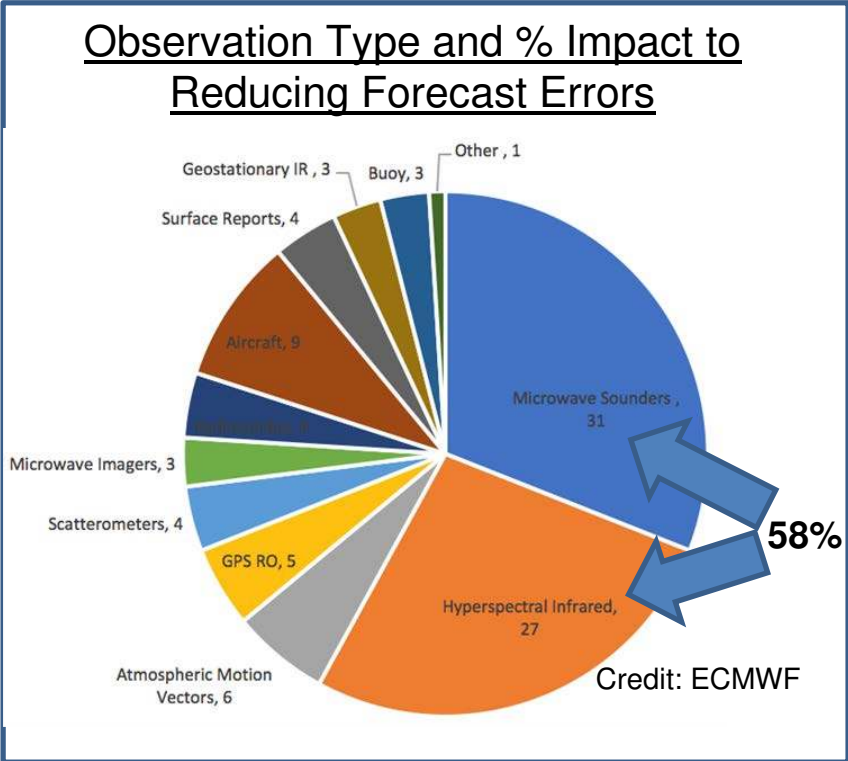
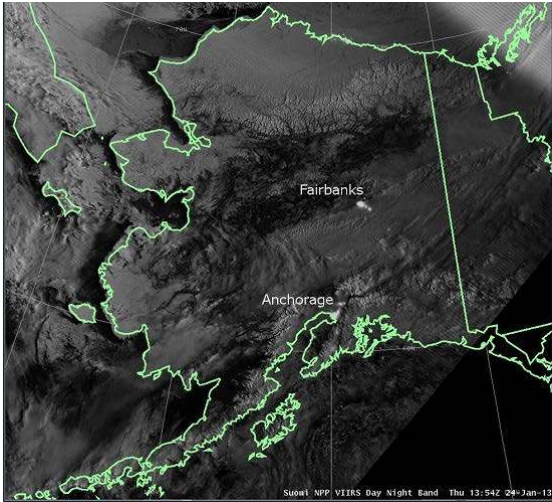


JCS



Key Polar Satellite Data

Microwave and infrared atmospheric sounders on polar orbiting satellites have a large positive impact on reducing numerical weather prediction forecast error



Imagery from polar orbiting satellites provides enhanced coverage in high-latitudes where geosynchronous satellite coverage is diminished



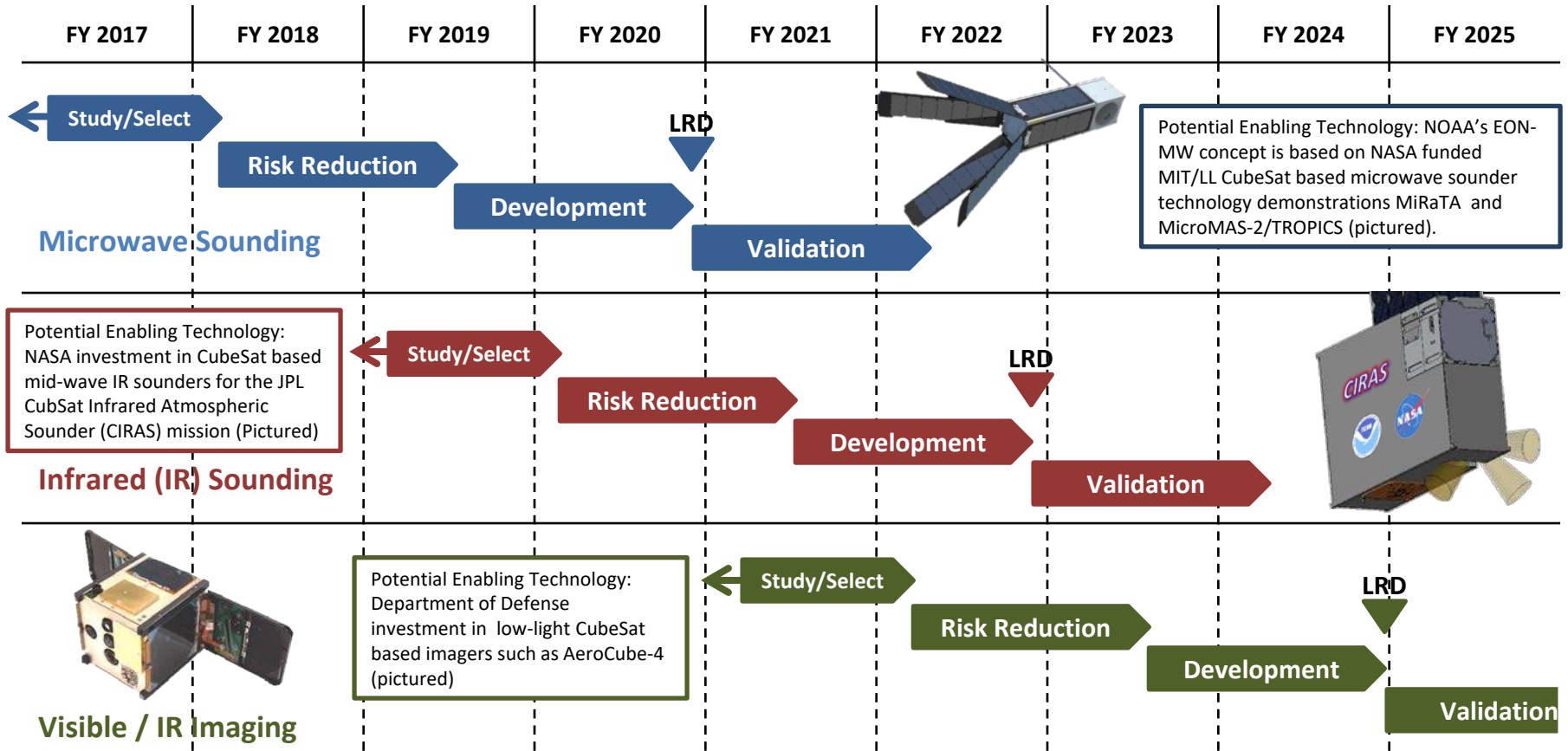
Why SmallSats?



- Current environmental satellites are expensive
 - No ability for spares in LEO orbit
 - Long development cycle
 - Failure means lack of data availability
- SmallSats could be the future for some observations
 - Lower cost alternatives
 - Use commercially available parts
 - Less weight means low launch costs
 - Can afford to have a spare for Gap Mitigation
 - Much shorter development time
 - Commercial launch availability
 - Loss of a single spacecraft does not result in the loss of all instruments
- Better capability for partnering opportunities ~ DoD and NASA

Incorporating SmallSats into future architecture plans

Strategy for SmallSat Integration



Challenges for CubeSat IR sounder

- Temperature control is among the highest challenges, along with a larger aperture size required for longer wavelengths.
- IR sensors are extremely sensitive to noise due to thermal emission of the optics and Johnson noise in the detectors, especially in the LWIR, and require a significant amount of cooling.
- Other technology risk areas include Focal Plane Array (FPA) technologies, miniature reliable cryo-coolers, compact optics, and IR Immersion grating spectrometers.

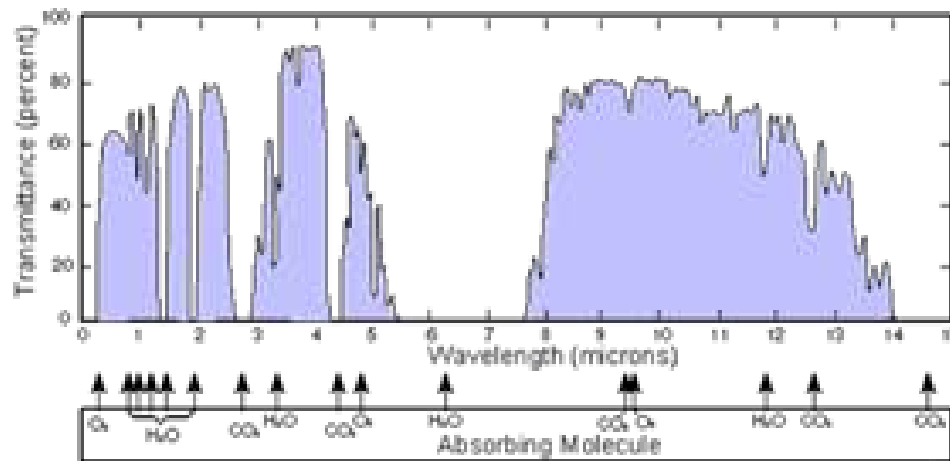


Figure 1: Electromagnetic Spectrum Transmission in the Infrared Wavelength



2015 Study



- NESDIS and JPL began studying the optimal performance of a CubeSat based infrared sounder in comparison to CrIS performance.
- TRL assessment of all mission components and subsystems
- Recognizing the difficulty with the thermal and power requirements of LWIR sounding, the study focused on design of the MWIR only in a 6U CubeSat.
- The study addressed the optical, mechanical, thermal, detector and electronic requirements from such a system.
- The EON-IR (MWIR-Only) instrument that resulted employed passive cooling for the spectrometer and a micro pulse tube cryocooler for cooling the detector.
- After completion of this first study, the ESTO funded CIRAS project began the design phase.
- Immediately it was found that the fully functional design arrived at during the NOAA study was not affordable for the CIRAS and a few changes were made including adding a second cryocooler for the spectrometer since passive cooling was more complex, and replacing the pulse tube cryocooler with a commercial less expensive and less reliable cooler.

- **HOT-BIRD Detectors (TRL 6)**

- The new High Operating Temperature Barrier Infrared Detector (HOT-BIRD) detector materials developed at JPL provide superior uniformity and operability, higher operating temperature, and low 1/f noise.
- Detector/ROIC (Sensor Chip Assembly, SCA) complete. SCA's under test.

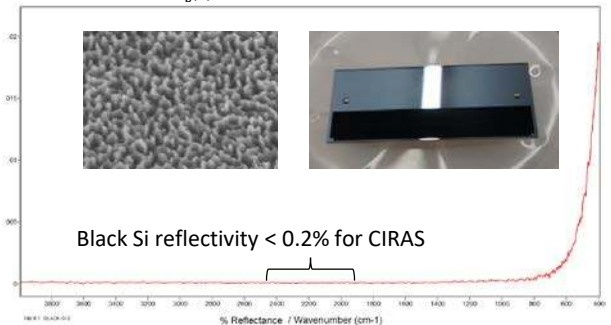
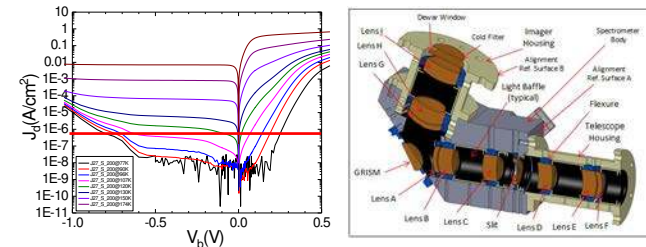
- **MWIR Grating Spectrometer (MGS) (TRL 5)**

- All refractive grating spectrometer with a 16 degree Field of View. Covers 4.08-5.13 μm and 625 channels. MGS design complete. Build by Ball Aerospace with immersion grating and slit by JPL.
- MGS in final design and parts procurement phase at Ball. Slit in design, procuring immersion grating substrate.

- **Black Silicon IR Blackbody (TRL 5)**

- A cryo-etched silicon surface that exhibits less than 0.2% reflectance across a broad spectral band. Developed at JPL
- CIRAS Black Si Slit and Blackbody currently in the design phase.

- **All technologies will be advanced to TRL 7 at the end of the spaceflight mission**



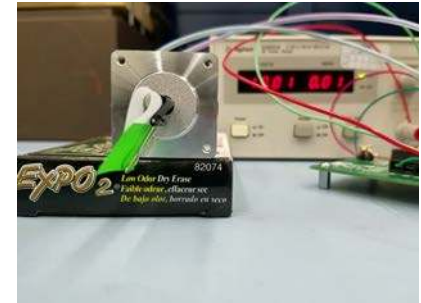


EON-IR 2016/2017 Statement of Work

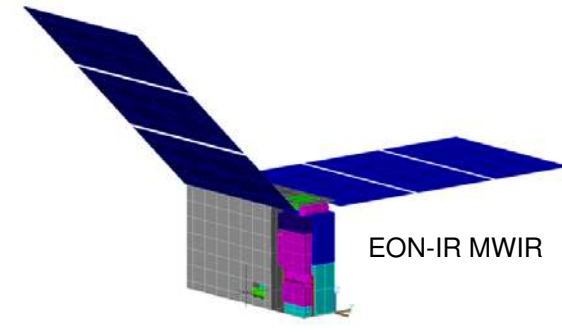


1. Improve the design of EON-IR to increase reliability commensurate with a mission of two years in length or longer. This task will examine the reliability and mission assurance of the EON-IR and its subsystems, primarily electronic, including the spacecraft.
2. Examine the ability to provide full swath scanning. This task will explore the ability to scan the EON-IR to achieve full swath as currently obtained from the operational sounders. Model the scanning mechanism and impacts on the sensor collection as well as the dwell times.
3. Improve the EON-IR thermal/mechanical design. Model designs for an FPA mount and cold shield/filter for EON-IR with sufficient fidelity to estimate total photon flux at the detector for accurate predictions of noise performance, and estimate total thermal load at the detector cryocooler cold finger. Provide a better estimate of the amount of heat needed to dissipate by the cryocoolers and radiators.
4. Identify drivers and limitations to expand the EON-IR pathfinder channel capability to CrIS sensor capabilities. The objective is to determine if there are viable options to expand EON-IR beyond the Mid-wavelength Infrared (MWIR) to include Long-wavelength Infrared (LWIR). This task should look at thermal impacts as well and also possible increase in CubeSat size to accommodate additional capability

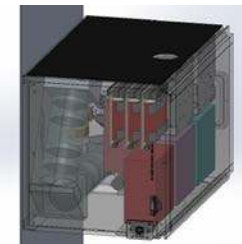
- The study was to benefit from the InVEST CIRAS program and identify the additional risks associated with the requirements of an EON-IR system.
- Task 1: Mission Reliability Improvement
 - Major portions of the EON-IR concept have low risk due to the commonality with CIRAS, however, further definition of EON-IR revealed several life limiting components which needed further reliability study and would possibly change the TRL of EON-IR.
 - Electronics: Parts identified with SEL sensitivity.
 - Scanning: Commercial scan motor has not undergone life testing.
 - Cryocoolers: Ricor K508N used on CIRAS not best choice for EON-IR. Alternate long-life microcoolers identified
- Task 2. Full Swath Scanning Study
 - Task demonstrated that full swath scanning is achievable with desired scan rates
- Task 3. Improve the MWIR portion of EON-IR Thermal/Mechanical Design
 - Results demonstrated that the heat generated by electronics and active cooling of the optics and detectors can be passively radiated by the 6U CubeSat structure
- Task 4. Expand the EON-IR Channel Capability
 - A Team-X study demonstrated that an LWIR Sounder can be designed to comfortably fit into a 12U CubeSat using a combination of active and passive cooling



EON-IR Scan Control Demo



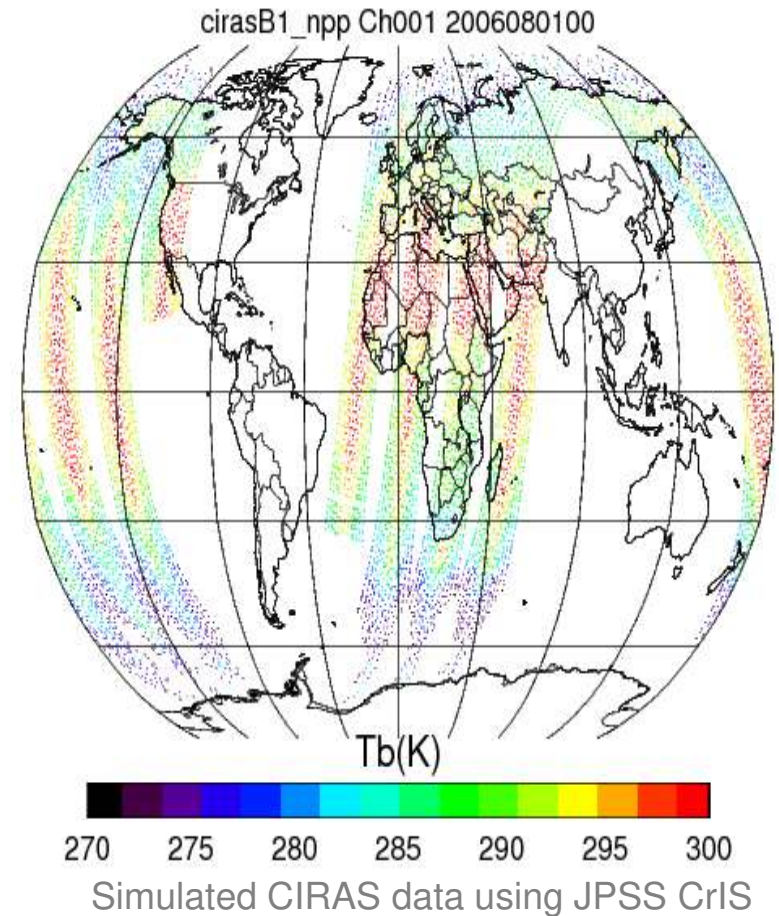
EON-IR MWIR



EON-IR LWIR 12U
Concept Layout

EON – Microwave and Infrared Data Impact Studies

- Scope:
 - Determine the quantitative value of MicroMAS-2 and CIRAS in the reduction of forecast error in global and regional numerical weather prediction (NWP) models:
 - Impact of MicroMAS-2 in the absence of ATMS
 - Impact of CIRAS in the absence of CrIS
- Recent Work:
 - Created simulated MicroMAS-2 and CIRAS data CubeSat Sounders for studying impact
 - Created orbit simulator for MicroMAS-2 and CIRAS
- Next Steps:
 - Complete work to quantify and summarize impacts on simulated global NWP models





Conclusion



- NOAA, NASA, and JPL are all working together to provide IR sounding technology in a CubeSat
- IR soundings have major impacts on weather forecast models
- MWIR is viable and being demonstrated on a CubeSat format in CIRAS
- LWIR concepts have been developed to fit onto a 12U CubeSat form
- EON-IR expands beyond the technology demonstration to a longer operational mission life