



# Plug and Play Attitude Control

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# ADCS Design Objectives

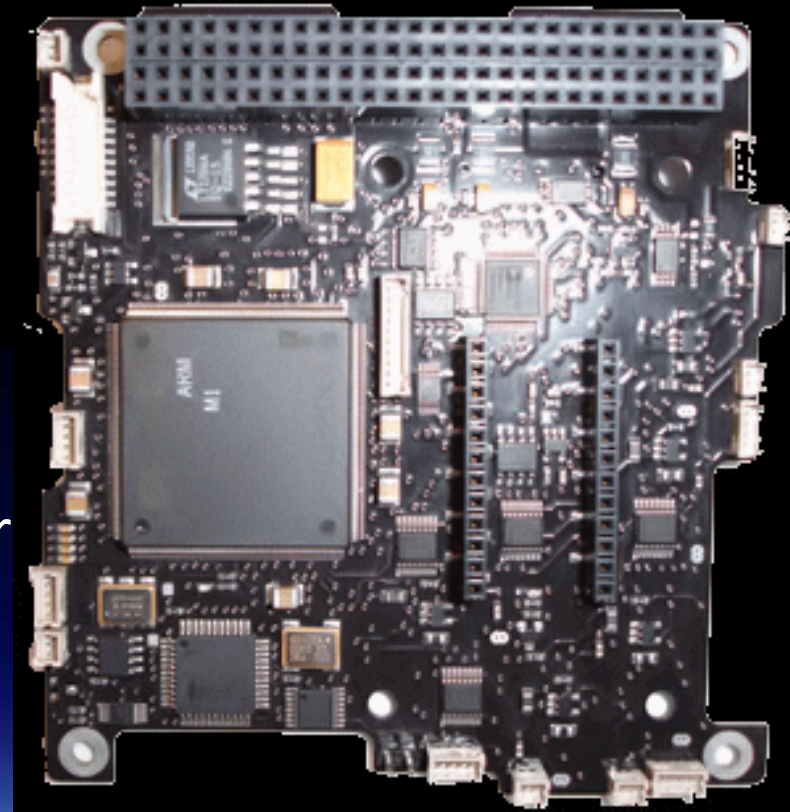


- Single board ADCS solution
- Handles complete ADCS task
  - Limited interaction from OBC required
    - Ability to set the autonomous mode transitions
  - Telemetry from the ADCS unit will keep you updated
- Include low power/safe mode ADCS on board
- Easily adaptable to different sensor and actuator technology.
- Inherently low power, low mass, minimal volume, low cost, low system overheads and high performance.

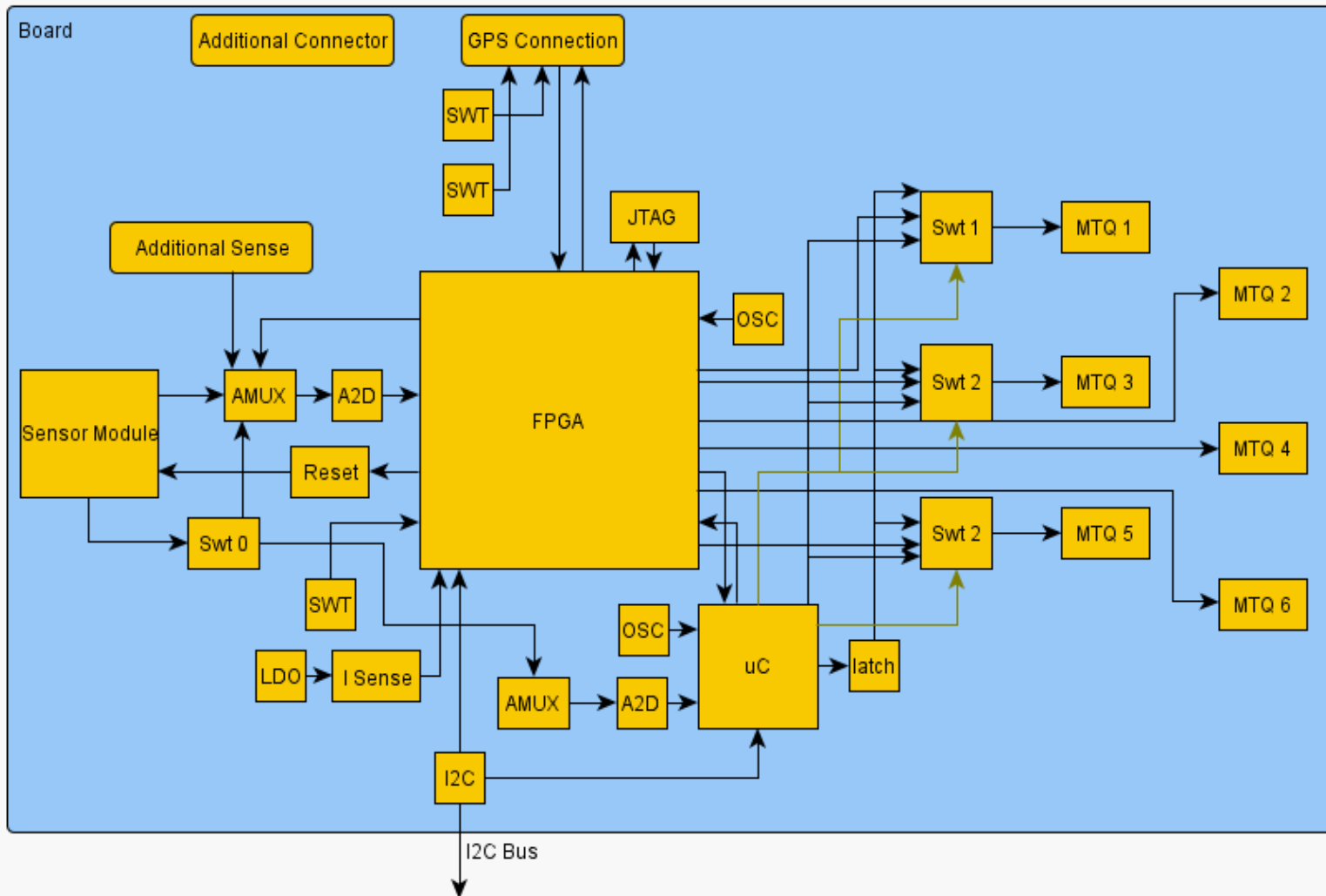
# Just add Magnetorquers...



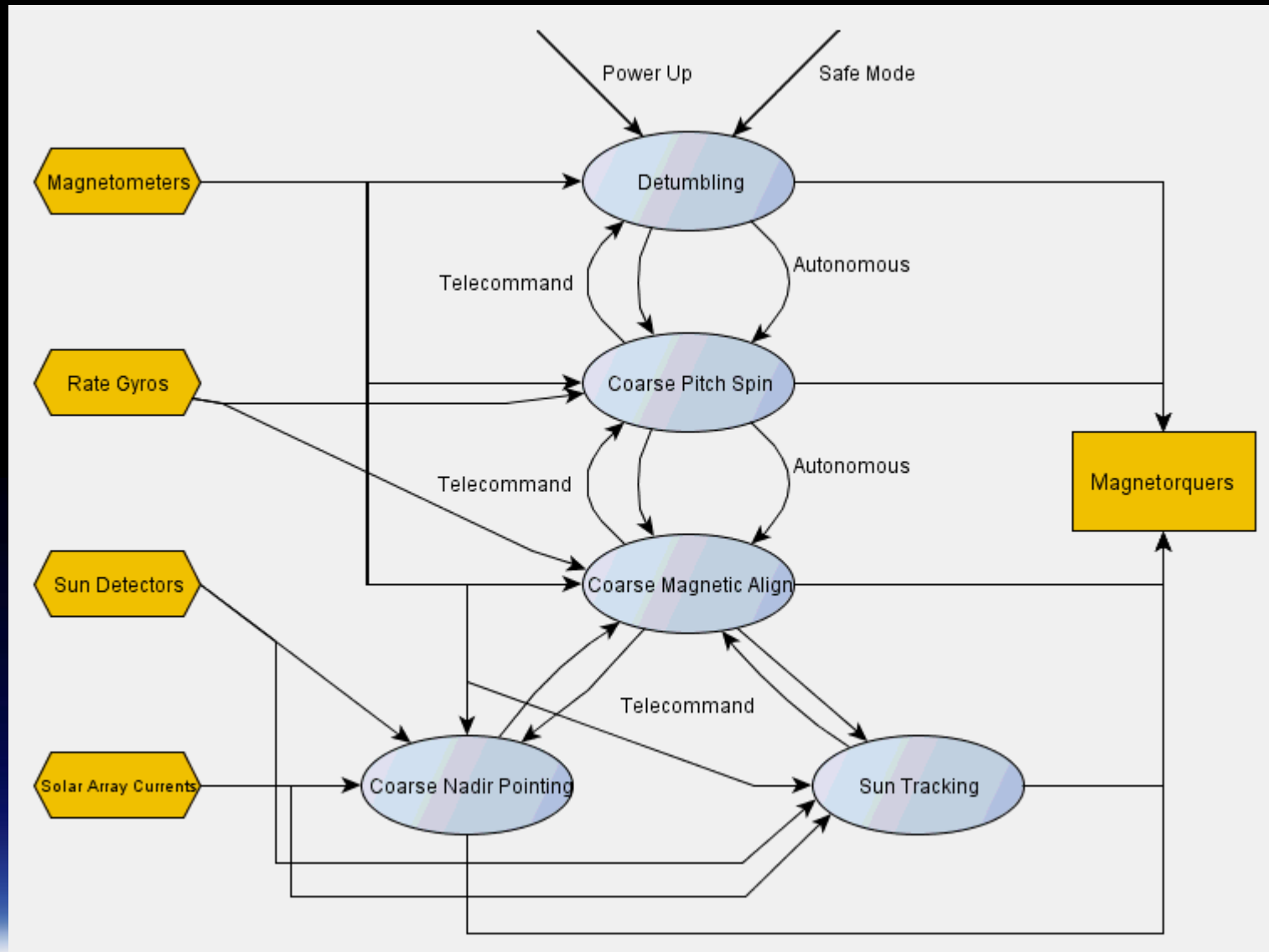
- Complete ADCS task runs on board
  - I2C telemetry and telecommand
- On board sensors include:
  - Magnetometers
  - Rate gyro's
  - Optional GPS
  - Interfaces for available for SIX 2-axis Analog Sun Sensors
- SIX current controlled Magnetorquer Drivers On Board
- Internal Parameters can be optimised easily for different mission requirements.



# ADCS Architecture



# Attitude modes



# ADCS Simulation



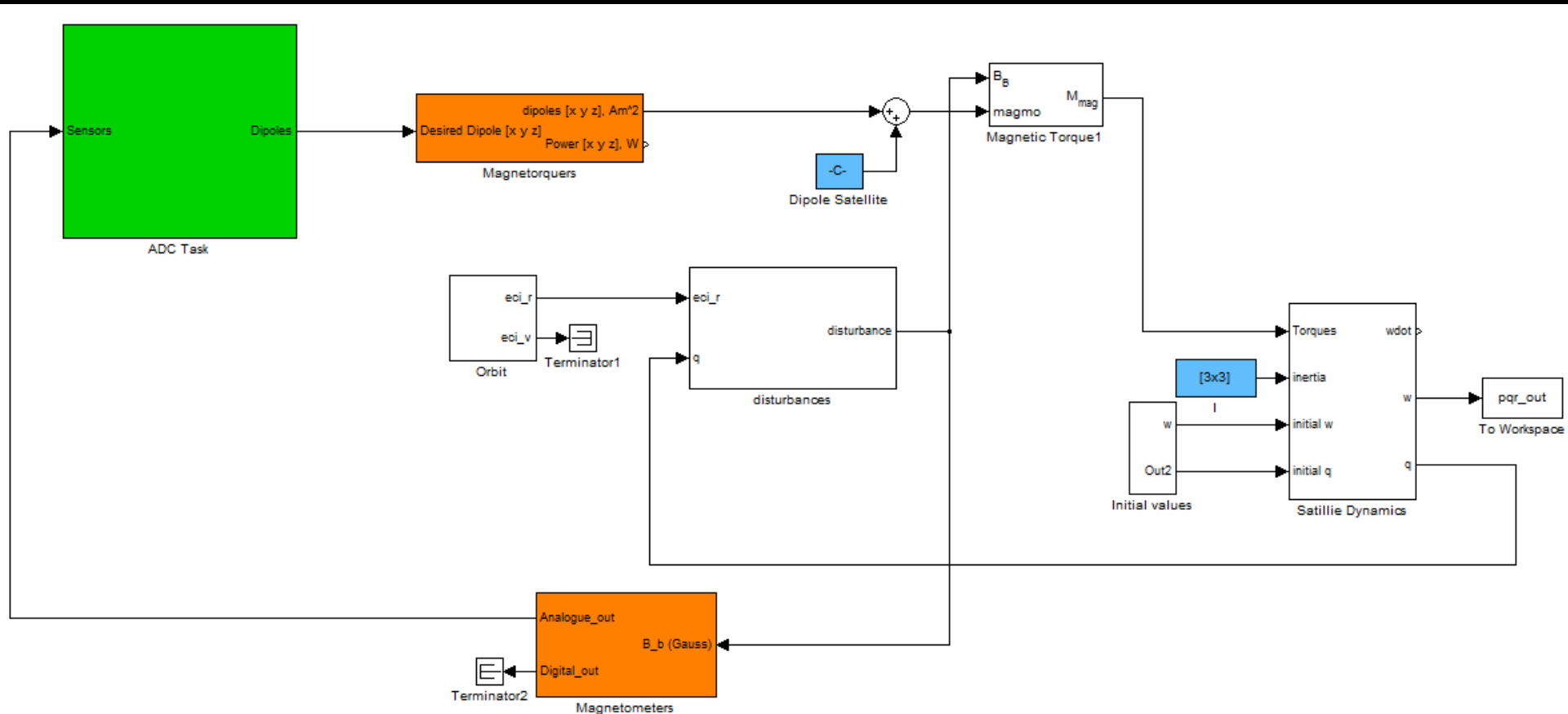
- To provide confidence in the performance for specific missions, bespoke orbits and parameters can be simulated within our Matlab model.
- From this, changes can be made to the control parameters to obtain the optimally performance
- The basic controller design suitable for most scenarios unmodified (plug and play).

# Matlab Simulation Model



- All sensors outputs based on measured data from real sensors.
- All actuators models based on measured data
- Disturbances modelled using high fidelity models

# Simulation Model





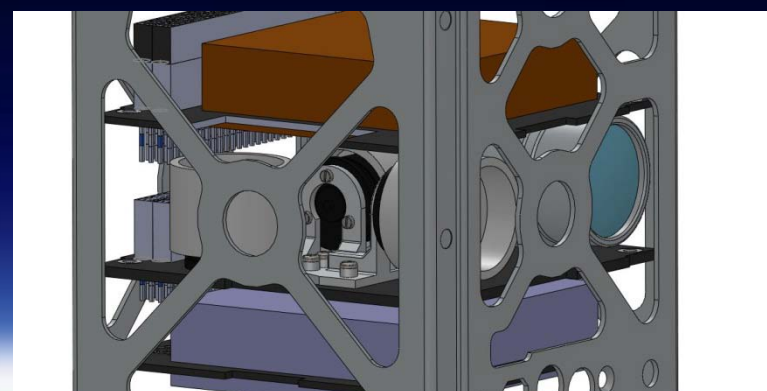
# ADCS Customisation



- Additional modes can be included
  - If additional modes or experimental modes are required, they can be included
- Additional sensors can be included
  - New sensors? Improved sensors?
  - Interfaces exist for these to be added in and they can be easily combined with the other sensor data
- Additional actuators can be included
- FPGA can be reprogrammed in flight
  - Upload new versions of HDL code to improve performance.
- Development roadmap includes the addition of mass memory for log data

# 3 Axis Reaction Wheels

- Inertia disk  $5 \times 10^{-6} \text{kgm}^2$
- Angular momentum  $2.6 \times 5 \times 10^{-3} \text{kgm}^2/\text{s}$  @ 5000RPM
- Nominal torque 2-3mNm
- Estimated total mass (3 wheels plus electronics) <200g.
- Ready for Ukube-1 end 2011.



A small satellite is shown in space, oriented vertically. The satellite has a rectangular body with a series of solar panels or sensors on its side. The Earth's blue and white horizon is visible in the bottom right corner of the frame. The background is black with some faint stars.

questions?

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