

EDDE ElectroDynamic: Delivery Express <u>and</u> Debris Eliminator

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Propellantless Propulsion

Demonstrated by PMG (Plasma Motor Generator)





STAR, INC.

100's of km/day altitude change at ~400-1000 km
>1°/day orbital plane change (inclination & node)



The EDDE Vehicle

STAR, INC.



Each vehicle is <100 kg, and packs into 28"x24"x12"
Each ESPA secondary payload slot can carry 2 EDDEs
EDDE can distribute payloads, then collect orbital debris

STAR, INC.

Key EDDE Components

Conductor/collector

Reinforced Al tape, 30 mm x 38 μm

Winding:

Stack:





Dynamics/control

Orbit transfers optimized Computer controls current



Packaging
 28"x24"x12", 2 fit in 1 ESPA slot

Electronics

Emitters, folding solar arrays



STAR, INC. Secondary Payload Launch



star, INC. Small Payload Ride Options

- 1. Conventional secondary payload accommodations
- 2. Conventional + on-board propulsion
- 3. Conventional + upper stage with payload carrier
- 4. Dedicated nanosat launch, whenever you're ready

Each step down provides a better service, but raises costs and/or constraints, and (at least now) decreases credibility. What is the potential market size for each of these options?

STAR, INC. "EDDE, Inc." Business Vision

- Deliver small payloads to custom LEO orbits, at costs closer to secondary than dedicated launch.
- After distributing payloads, EDDE can go on to capture and relocate large debris in LEO
- Key questions:

 How much extra will secondary payloads be willing to pay for custom orbits?

• Who will pay to clean up LEO—and how much?





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- Cosmos-Iridium collision of February 2009
- After collision, debris clouds from each object are clearly distinct
- Each object contributes its own mass in debris fragments
- With time, the clouds spread, polluting their altitude ranges



STAR, INC. Lethal Impactors in LEO

Туре	Characteristics	Hazard
"Cars"	Tracked, >2 kg, ~1% of all lethal objects	Primary source of new shrapnel; ~99% of collision area and mass
"Hubcaps"	Tracked, >10 cm, <2 kg, ~4% of all lethal objects	Cause most known conjunctions and drive avoidance maneuvers
"Shrapnel"	Untracked, 1-2 to ~10 cm, ~95% of all lethal objects	Primary direct threat to satellites: too small to track and dodge, but too massive to shield against

Most new shrapnel comes from "car/car" collisions

Must remove cars to prevent LEO pollution by shrapnel



Current EDDE Status

- Now 2 months into a 2 year NASA OCT contract
- We are maturing key components, stowage and deployment concepts, solar array steering, active collision avoidance, and failure-tolerant controls.
- Contract goal is a successful PDR for a subscale Mini-EDDE flight test using full-scale components
- We are looking for good nanosat-class sensors & actuators relevant to multi-year LEO missions
- NRL's TEPCE will test parts of EDDE next year



4-for-4 Flight Record





5:33 AM APR . 6.1994



SEDS-1, NASA Marshall

• Deployed 20 km braided Spectra tether; sent 26 kg end-mass into controlled reentry

PMG (Plasma Motor Generator), NASA JSC

 Demonstrated motor/generator operation, hollow cathode as part of large current loop

SEDS-2, NASA Marshall

• Stabilized 20 km Spectra tether near vertical; cut after 4 days, but 7 km lasted till reentry

TiPS, Naval Research Laboratory

Libration damped out over several months, 2 mm x 4 km tether survived >10 yrs

All tethers & deployers by Tether Applications

If you're interested in:

• EDDE for payload delivery/distribution in LEO

Star, Inc.

- Or an agile sensing platform (anywhere in LEO)
- Or wholesale LEO debris capture and relocation
- Or providing or collaborating on nanosat-class sensors & actuators for multi-year LEO missions

Then please contact me at 619-980-1248 (cell) or tether@cox.net

star, INC. LEO Congestion & Cleanup

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